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# SEMI-ENGINEERED EARTHQUAKE-RESISTANT STRUCTURES: ONE STORY BUILDINGS MADE WITH BHATAR CONSTRUCTION TECHNIQUE

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This thesis is focused on an engineering field; at the same time, it has been developed through multidisciplinary knowledge and eyes. The word Design involves all the fields I have ever appreciated. The word Design is the link-word, which connects all my academic path.

Ten years ago I started studying Industrial Design in Architecture atUniversità di Genova, then through the years I crossed civil and environmental studies, finishing focusing on Structural engineering curriculum in Bologna. Often the people thinks this is a strange path but I have never considered myself strictly pure. I have always considered varied and mix knowledge as a wealth.

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## **SOMMARIO**

Questa tesi studia il comportamento statico e sismico di strutture semplici realizzate con un sistema costruttivo utilizzato da vari secoli, in zone piuttosto remote dei paesi che oggi vengono definiti come terzo mondo. A secondo delle zone il nome cambia. Per quanto riguarda le regioni Himalayane tra Nepal e Pakistán il nome comune è Bhatar. Questo sistema costruttivo vede come materiali utilizzati legno e pietra locali. Il Bhatar è costituito da pareti portanti composte da strati di pietra non perfettamente uniforme, i comuni muretti a secco, intervallati orizzontalmente da travi composte da elementi lignei i quali incastrati tra di loro risultano paragonabili a cordoli.Il sistema Bhatar è conosciuto come intrínsecamente antisísmico poichè esistono costruzioni di alcuni secoli che hanno resistito a fenomeni sismici importanti. Le analisi sono condotte con riferimento ad un edificio ad un piano, di dimensioni (pianta 3.60m x 3.6m) e con tetto in legno e terra. Questa tecnologia costruttiva, di carattere semi-ingegneristico, è già ampiamente utilizzata nelle regioni Himalayane, in Pakistan e India, ma è anche indirizzata alle popolazioni di nazioni in via di sviluppo poiché offre un vantaggio sia di tipo economico che di tipo tecnico rispetto ai materiali convenzionali (muratura in mattoni e cemento). Le informazioni ad oggi disponibili su questo genere di strutture sono molto limitate a causa della scarsa e poco approfondita ricerca eseguita sul tema. Di grande utilità è stato il materiale elaborato dall'architetto Tom Schacher technical advisor per la Swiss Agency for Development and Cooperation. Tom Schacher col suo lavoro ha stilato delle linee guida, tramite immagini per popolazioni semi-analfabete, che consigliano particolari dimensionamenti e rapporti tra dimensioni nella costruzione di sistemi a Bhatar.

L'obiettivo principale di questa ricerca è di definire gli aspetti principali del comportamento sismico di un edificio ad un piano composto secondo le linee guida dettate da Tom Schacher, con scopo di prevenire crolli causati da azioni sismiche e quindi ridurre il rischio sismico in quelle regioni del mondo dove questi disastri hanno intensità significative. Non esistono attualmente in letteratura ricerche specifiche su pareti costruite con il sistema Bhatar.

Per quanto riguarda le pareti , sono stati effettuati calcoli e analisi allo scopo di capire il comportamento statico e sismico. In analisi statica, è stata condotta una verifica a sforzo normale calcolando lo sforzo normale agente alla base del muro e la corrispondente capacità resistente. Per quanto riguarda l'analisi sismica del muro, si è studiato sia il comportamento nel piano sia quello fuori dal piano. Per l'analisi in piano ci si è concentrati sul materiale roccioso ed è stato utilizzato il modello di Barton che definisce la relazione non lineare che tra le tensioni normali e tangenziali nelle discontinuità degli ammassi rocciosi in presenza di pietre non uniformi. Per quanto riguarda l'analisi fuori dal piano l'attezione è stata rivolta alle connessioni degli elementi lignei che diventano fondamentali nelle reazioni a sollecitazioni di tipo orizzontale e prevengono ribaltamento e gli altri meccanismi di collasso, questo scopo le connessioni e le strutture in legno suggerite da Tom Schacher sono state esaminate alla luce delle norme tecniche Eurocodice 5 : Design of timber structures.

Grazie alle analisi effettuate è possibile avere una prima idea di quanto questo tipo di costruzioni siano effettivamente antisismiche. Importante è sottolineare che questa tesi è l'inizio di un lungo lavoro che per essere affrontato al meglio necessita di prove di laboratorio su materiali e prove di laboratorio su modelli in scala reale.

## **ABSTRACT**

After the 2005 M7.6 Kashmir earthquake (Pakistan), field observations reported that several buildings manufactured with traditional techniques well resisted to this strong seismic event. Nonetheless, these techniques have never been deeply studied from a structural engineering point of view yet. The high number of people living in such structures highlights the importance of focusing on this subject.

This paper reports a full analytical study on the static and seismic behavior of simple one-storey buildings made with a typical construction technique commonly named as "Bhatar" system, used for several centuries and widely diffused in rather remote areas of the Himalayan regions like India, Nepal and Pakistan.

The Bhatar system consists of load-bearing walls made of common dry-stacked rubble stone masonry held together by horizontal wooden bands disposed at several levels (spaced at intervals of about 60 cm). It is widely adopted in developing countries due to its advantages from both economical and constructive point of view with respect to the conventional constructions techniques (i.e. brick masonry and concrete structures).

Despite its wide diffusion, the information currently available on the actual static and seismic behavior of such construction technique are very limited due to little attention paid on such topic.

In the present work, analytical analyses are conducted with reference to a one-storey building modulus characterized by a 3.6 m x 3.6 m square plan covered by an heavy wooden roof with 20 cm thick earth coverage, in order to investigate its response under both gravity and seismic inertial loadings. In detail, in-plane and out-of-plane response of a single wall under horizontal actions is discussed and particular attention is focused on the connections between the timber elements, which are fundamental for the transmission of the horizontal actions and for preventing overturning and other failure mechanisms.

The main aim is twofold: (i) to provide a first insight into the actual seismic response of such construction technique, as a basis for the specific design of ad-hoc laboratory tests on full-scale models, and (ii) to give some rules of thumb for a proper dimensioning and construction of this kind of structures.

# **INDEX**

ACKNOWL	EDGEMENTS	I
SOMMARIO	)	II
ABSTRACT		III
INDEX		IV
LIST OF FIG	GURES	XI
LIST OF TA	BLES	XVIII
SIMBOLOG	Y	XX
1 INTRODU	CTION	1
1.1 Bac	kground	1
1.2 Just	ification of the document and objectives	1
1.2.1	General objectives	2
1.2.2	Specific objectives	2
1.3 Org	anization of the thesis	2
2 BHATA	AR	2
2.1 Trac	ditional definition of Bhatar	2
2.2 Ton	n Schacher Manual : Bhatar construction - An illustrated guide for craftsmen	4
2.2.1	Gross shape and dimensions	4
2.2.2	Foundation and plinth band	5
2.2.3	The Walls	5
2.2.4	Wall - joints	6
2.2.5	Kashmiri joint or Keyed scarf joint	7
2.2.6	Connections - Corners	7
2.2.7	Connections – Cross Pieces	8
2.2.8	Connections – Internal wall	8
2.2.9	Openings	9
2.2.10	Doors	9
2.2.11	Windows	10
2.2.12	The Roof	10
3 STUDY	CASE	11
3.1 Sing	gle modular unit	11
3.1.1	Orthogonal projection	12
3.2 Sing	gle Wall	13

	3.2	2.1	Orthogonal projection	.13
	3.3	One	e room box	.14
	3.3	3.1	Orthogonal projection	.15
	3.3	3.2	The Roof	.16
4	M	ATEI	RIALS	.18
	4.1	Tin	nber : Shorea Robusta	.18
	4.1	.1	Botanic Characteristics	.18
	4.1	.2	Mechanical properties of Shorea Robusta	.18
	4.1	3	Characteristic Values from EN 338	.19
	4.1 CN		Design Values from EC 5 and en.1995.1.1.2004 and NICOLE – Istruzioni 0T206_2007	.20
	4.2	Sto	nes : Main construction material since the Stone Age	.20
	4.2	2.1	Limestone mechanical properties.	.22
5	BA	ARTO	ON MODEL AND SHEAR STRENGTH OF ROCKFILL	.26
	5.1	Inte	erfaces between material : Timber-Stone and Stone-Stone	.26
	5.2	She	ear strength of rock discontinuities	.28
	5.3	Pla	ne smooth joint	.28
	5.4	Idea	alised rough joint (Patton , 1966)	.29
	5.5	Rea	ıl rough joint (Barton, 1973)	.31
	5.5	5.1	Barton's failure criterion	.32
	5.5	5.2	Barton's empirical model:	.32
	5.6	She	ear Strength of Rockfill	.37
	5.6	5.1	The shear strength of rockfill as measured	.38
	5.6	5.2	Estimating the shear strength of rockfill	.41
	5.6	5.3	Interface shear strength	.42
	5.6	5.4	R-controlled or JRC-controlled behavior.	.44
	5.7	Bar	ton model applied on Bhatar system	.44
	5.7	7.1	Rockjoint	.45
	5.7	7.2	Rockfill	.45
	5.7	7.3	Voids ratio and Porosity	.46
	5.7	7.4	Limestone Mechanical Properties for application of Barton model	.47
	5.7	7.5	Rockjoint results	.48
	5.7	7.6	Rockfill results	.50
	5.8	Cor	nclusions	.52

6 TIMBE	R ELEMENTS AND CARPENTRY CONNECTIONS	54
6.1 Ge	ometry of Timber elements	54
6.1.1	Rafter	55
6.1.2	Roof rafter	56
6.1.3	Cross piece	57
6.2 Ass	sembling	58
6.2.1	Timber Band	58
6.2.2	Roof Timber Band	60
6.3 Por	tions of Rafter and Roof rafter	62
6.3.1	Rafter Head + Rafter Body + Rafter Head	62
6.3.2	Roof Rafter Head + Rafter Body + Roof Rafter Head	62
6.3.3	Subdivisions of the timber elements	63
6.4 Are	ea under stresses	64
6.4.1	Cross Piece	64
6.4.1	Rafter	64
6.4.2	Roof Rafter	65
6.4.3	Measures for area under stresses	66
6.5 Sai	nt Venant for Timber elements	67
6.5.1	Rafter	67
6.5.2	Roof rafter	70
6.5.3	Cross piece	73
6.6 Eu	rocode 5 : EN 1995-1-1 :2004+A 1	76
6.6.1	Tension parallel to the grain	76
6.6.2	Tension parallel to the grain with keyed scarf joint	76
6.6.3	Compression parallel to the grain	76
6.6.4	Compression perpendicular to the grain	77
6.6.5	Tension perpendicular to the grain	77
6.6.6	Bending	77
6.6.7	Shear	78
6.6.8	Torsion	79
6.6.9	Combined bending and axial tension	80
6.6.10	Combined bending and axial compression	81
6.6.11	Combined Torsion and Shear - CNR-DT 206/2007	82
67 Res	sistances - Rafter Body	83

	6.7.1	Longitudinal to the grain	83
	6.8 Res	sistances - Cross piece Notch	85
	6.8.1	Longitudinal to the grain	85
	6.9 Act	tivation of the chains	87
	6.9.1	Overturning Mechanism	88
	6.9.2	Activation of the chain along Roof Rafter Head	90
	6.9.3	Activation of the chain along Rafter Head	111
	6.9.1	Possible actions along cross pieces	131
	6.10 I	nternal developed bending moments	141
	6.10.1	Mytf bending moment due to tension	141
	6.10.2	Mycf bending moment due to compression	142
	6.10.3	Mz bending moment	143
	6.10.4	Torsional Mx	144
	6.11 <b>k</b>	Keyed scarf joint	146
	6.11.1	Geometry and resistance	146
	6.11.2	Influence of keyed scarf joint on element subjected to tension	147
7	STATIO	C ANALYSIS	148
	7.1 Air	n of static analysis	148
	7.2 Sin	gle modular unit	148
	7.2.1	Material properties	148
	7.2.2	Volumes	148
	7.2.3	Weights and stresses	150
	7.1 Ro	of	150
	7.1.1	Material properties	150
	7.1.2	Volumes	151
	7.1.3	Weights and linear load	151
	7.2 No	rmal Stresses	152
	7.2.1	Normal Stress inside stones layer	153
	7.2.2	Normal Stress below timber beam	154
8	SEISM	IC ANALYSIS IN PLANE	156
	8.1 She	ear strength for rockfill with Barton empirical model	156
	8.1.1	Normal Stress and Coefficients of friction inside stones layer	156
	8.1.2	Normal Stress and Coefficients of friction below timber beam	157
	8.2 Sei	smic load multiplier.	157

	8	3.2.1	Critical multiplier for inside stones layer	157
	8	3.2.2	Critical Multiplier below the timber band	165
	8	3.2.1	Conclusions on seismic analysis in-plane	170
9	5	SEISM	IC ANALYSIS OUT OF PLANE – OVERTURNING RIGID BEHAVIOR	174
	9.1	Нур	pothesis of rigid body behavior	174
	9.2	Rig	id body over rigid soil by Equilibrium – Tmin as function of $\alpha$ load multiplier - Har	ıd
	cal	culation	1	175
	9	9.2.1	Horizontal equilibrium	176
	9	9.2.2	Rotational equilibrium	177
	9.3	Rig	id body over rigid soil by PVW - $\alpha$ load multiplier - Hand calculation	179
	9	9.3.1	Unique seismic force on the top	180
	9	9.3.2	Roof force + Wall force	183
		9.3.3 on α	Unique seismic force on the top with timber tie-beams - Minimum Tension dependent 185	lent
	9	9.3.4	Roof force + Wall force with timber tie-beams - Minimum Traction dependent on 188	α
	9.4	Cor	nclusions about the highest required tension strength Tmin	191
	9	9.4.1	Horizontal equilibrium and Rotational equilibrium – Tmin	191
	9	9.4.2	Unique seismic force on the top - $\alpha$ critical	192
	9	9.4.3	Roof force + Wall force - α critical	192
	9	9.4.4	Unique seismic force on the top with timber tie-beams - Tmin	193
	9	9.4.5	Roof force + Wall force with timber tie-beams - Tmin	193
	9.5	Ver	ifications for Overturning Rigidbehavior	194
		9.5.1 Γmin	Analyzing the worst case : Unique seismic force on the top with timber tie-beams 194	-
	9	9.5.2	Equal distribution of the reactions T1=T2 and R1=R2	194
	9	9.5.3	Verifications T1=T2	196
	9	9.5.4	Verifications R1=R2	199
	9	9.5.5	Verifications on corner joint, seismic event parallel to Roof Rafter	201
	9	9.5.6	Verifications on corner joint, seismic event parallel to Rafter	206
	9.6	Cor	nclusions on seismic analysis out of plane – Overturning	208
	9	9.6.1	Safetybehavior under seismic multiplier α=0,15	208
10		SEISM 210	IC ANALYSIS OUT OF PLANE - FLEXIBLE RESPONSE BENDING BEHAVIO	ЭR
	10	1 1	Ivnothesis of Flexible response – Rending behavior	210

10.1.1	Hypothesis of Flexible behavior	211
10.1.2	Static scheme of the timber tie- beam	211
10.1.3	Hyperstatic scheme of the corner joint and actions from static scheme of the time	
	m	
10.1.4	Hyperstatic rigid-jointed frame	213
10.2 F	Force method with Müller-Breslau equations	
10.2.1	Force method	
10.2.2	Degree of indeterminacy Rigid-Jointed Frame	215
10.2.3	Solved released systems	217
10.2.4	Functions of the diagrams	222
10.2.5	Müller-Breslau equations	222
10.2.6	Solutions of the complete isostatic structure	229
10.3	Friangular distribution of seismic load $qlpha$	232
10.3.1	Scheme of wall Flexible response – Bendingbehavior	232
10.3.2	Masses involved and heights of each timber beam	233
10.3.3	Seismic load $q\alpha$ and Distribution factor $\beta j$	233
10.4 F	Reactions for each beam	235
10.4.1	Rafter body reactions for each beam in the corner joint	235
10.4.2	Rigid- jointed frame reactions for each beam	236
10.4.3	T1 in compression & T2 in tension	238
10.5 V	Verifications for Flexible response – Bendingbehavior	240
10.5.1	Analyzing the worst case : Roof level with maximum Seismic load $q\alpha$ ( $\alpha$ =1)	240
10.5.2	Distribution of the reactions T1≠T2 and R1=R2	240
10.5.3	Verifications T1 - compression	241
10.5.4	Verifications T2 - tension	243
10.5.5	Verifications R1=R2	244
10.5.6	Verifications on corner joint, seismic event parallel to Roof Rafter	248
10.5.7	Verifications on corner joint, seismic event parallel to Rafter	264
10.6	Conclusions on seismic analysis out of plane – Flexible	267
10.6.1	Safetybehavior under seismic multiplier α=0,125	267
11 PRACT	TICAL RULES OF THUMB FOR CONSTRUCTION OF BHATAR SYSTEM	268
11.1 A	Arch Tom Schacher's rule of thumb an new specifications	268
11.1.1	Specifications on wall joints	268
11.2 N	New Rules of thumb	270

11.2.1	Consideration about vertical component of the seismic event,	270
11.2.2	Steel wire connectors	270
11.2.1	Vertical rafters	279
11.2.2	Roof timber band	282
12 CONC	LUSIONS	284
12.1	Analysis performed	284
12.2	Results	285
12.2.1	Results on seismic analysis in-plane	285
12.2.2	Results on seismic analysis out of plane	287
12.3	Possible research developemnts	289
BIBLIOGR	APHY	290
SITOGRAF	PHY	291

# LIST OF FIGURES

Figure 2-1 Project entry 2008 Asia Pacific - "Advocacy of traditional earthquake-resistant	
construction, North-West Frontier Province, Pakistan": "Bhatar" at Besham Fort	2
Figure 2-2 Regions of the world where Bhatar is still used	3
Figure 2-3 Nepal peak ground acceleration	3
Figure 2-4 Bhatar construction-An illustrated guide for craftsmen	4
Figure 2-5 Divided rectangular structures	4
Figure 2-6 Gross dimension - ratio length/width	5
Figure 2-7 Foundations	5
Figure 2-8 The plint	5
Figure 2-9 Wall dimensions	6
Figure 2-10 Spread the connection points	6
Figure 2-11 Raise all walls together to avoid vertical joints	7
Figure 2-12 Kashmiri joint or Keyed Scarf Joint	7
Figure 2-13 minimum size of the beams/rafters	7
Figure 2-14 Lap joint – dimension	8
Figure 2-15 Cross Pieces	8
Figure 2-16 Internal wall joint	9
Figure 2-17 Openings	9
Figure 2-18 Openings 2	9
Figure 2-19 Lintel reinforcement	10
Figure 2-20 the flat heavy roof with earth cover	10
Figure 3-1 Modular unit-perspective	11
Figure 3-2 Modular unit - Orthogonal projection in cm	12
Figure 3-3Largest wall possible , length of 3.6m	13
Figure 3-4 Wallt - Orthogonal projection in cm	13
Figure 3-5 One room box	14
Figure 3-6 One room box orthogonal projections in cm	15
Figure 3-7 Section AA - studied wall	16
Figure 3-8 Flat earth heavy roof – exploded	17
Figure 4-1 Shorea Robusta – SAL	18
Figure 4-2 Architect Martijn Schildkamp - bhatar stones	21
Figure 4-3 Sedimentary rocks	21
Figure 4-4 Limestone/Calcarea	22
Figure 5-1 Stone layer (black box above) - Timber beam (black box below)	26
Figure 5-2 Contact surfaces	27
Figure 5-3 Plane and smooth joint surface	28
Figure 5-4 Stress vs Strain diagram and Mohr-Coulob failure criterion	29

Figure 5-5 Rough joint surface	29
Figure 5-6 45-D0566/A Profilometer (Barton comb), 150 mm length. ControlsGroup	32
Figure 5-7 Roughness profiles and their corresponding JRC values (Barton and Choubey 1977)3	33
Figure 5-8 Tilt test (or self-weight gravity shear test) for characterizing rock joints. Note	
measurement	33
Figure 5-9 Tilt Test apparatus	34
Figure 5-10 Alternative method for estimating JRC from Measuremens of surface roughness	
amplitude from a straight edge (Barton 1982)	35
Figure 5-11 Estimate of joint wall compressive strength from Schmidt hardness	36
Figure 5-12 When peak shear strength is approached (joints and rockfill), the actual rock-to-rock	
contact stress levels are extremely high, due to small contact areas	37
Figure 5-13 Illustration of the tilt test principle for rockfill (Barton and Kjærnsli, 1981)	38
Figure 5-14 Leps ( 1970)	38
Figure 5-15 The peak shear strength envelopes for rockfill have remarkable similarity to those for	•
medium rough, medium strength rock joints. Large-scale test data from Marsal (1973)	39
Figure 5-16 Large rock dumps are a familiar feature of mines in the Chilean Andes. Large-scale	
triaxial shear tests performed in Chile, with important results (black dots and Mohr circles)	
showing non-linear stress- ependent friction angles (Linero and Palma 2006)	40
Figure 5-17 The same non-linearity with effective stress level is seen in large-scale triaxial tests	
performed at NGI (Strøm, 1974, 1975, 1978), with particle size-dependence, rock strength	
dependence, and porosity effects also indicated	40
Figure 5-18 Shear strength envelopes (and peak dilation angles) predicted for rock joints, using the	ıe
JRC-JCS non-linear model of Figure 5-10. Rockfill generally lies between curves #2 and #34	41
Figure 5-19 An empirical method for estimating the equivalent roughness R of rockfill as a	
function of porosity and particle origin, roundedness and smoothness. Barton and Kjærnsli (1981	)
	41
Figure 5-20 Particle size strongly effects the strength of contacts points in rockfill. Triaxial or plane	e
shear also influencesbehavior. Empirical S/UCS reduction factors for estimating S when evaluating	g
equation 3	42
Figure 5-21 Asperity contact across stressed rock joints, and rockfill inter-particle contact, and	
rockfill lying on a rock foundation.	43
Figure 5-22 A review of interface shear tests was performed in response to concern over	
insufficient roughness for the rockfill dam foundation, in the glaciated mountain terrain in	
Norway	44
Figure 5-23 Rockjoint function for Bhatar	49
Figure 5-24 Rockjoint function for Bhatar range of interest	49
Figure 5-25 Rockfill function for Bhatar	51
Figure 5-26 Rockfill function for Bhatar range of interest	52
Figure 6-1 Continuous Bhatar wall	54
Figure 6-2 Carpentery connections	
Figure 6-3 Rafter beam	55
Figure 6-4 Rafter beam Orthogonal projections in cm	55

Figure 6-5 Roof rafter beam	56
Figure 6-6 Roof rafter beam Orthogonal projections in cm	56
Figure 6-7Cross piece	57
Figure 6-8 Cross Piece Orthogonal projections in cm	57
Figure 6-9 Timber Band	58
Figure 6-10 Timber band Rafter exploded	58
Figure 6-11 Timber band Cross pieces exploded	59
Figure 6-12 Timber band All exploded	59
Figure 6-13 Roof Timber band	60
Figure 6-14Roof timber band Roof rafter exploded	60
Figure 6-15 Roof timber band Cross pieces exploded	61
Figure 6-16 Roof Timber band All exploded	61
Figure 6-17 6.3.1 Rafter Head + Rafter Body + Rafter Head	62
Figure 6-18 6.3.2 Roof Rafter Head + Rafter Body + Roof Rafter Head	62
Figure 6-19 Subdivisions of the timber elements	63
Figure 6-20 Area under stresses - Cross piece	64
Figure 6-21 Area under stresses - Rafter	64
Figure 6-22 Area under stresses - Roof Rafter	
Figure 6-23 All areas under stresses	66
Figure 6-24 Rafter -Compression along X axis	
Figure 6-25 Rafter -Tension along X axis	67
Figure 6-26 Rafter -Shear on Y axis	68
Figure 6-27 Rafter -Shear on Z axis	68
Figure 6-28 Rafter - Bending Moment My on Y axis	69
Figure 6-29 Rafter - Bending Moment Mz on Z axis	69
Figure 6-30 Rafter - Torsion: Mx on x axis	69
Figure 6-31 Roof Rafter -Compression along X axis	70
Figure 6-32 Roof Rafter -Tension along X axis	70
Figure 6-33 Roof Rafter -Shear on Y axis	71
Figure 6-34 Roof Rafter -Shear on Zaxis	71
Figure 6-35 Roof Rafter - Bending Moment My on Y axis	72
Figure 6-36 Roof Rafter - Bending Moment Mz on Z axis	72
Figure 6-37 Roof Rafter - Torsion: Mx on x axis	72
Figure 6-38 Cross Piece -Compression along X axis	73
Figure 6-39 Cross Piece -Tension along X axis	73
Figure 6-40 Cross Piece -Shear on Y axis	74
Figure 6-41 Cross Piece -Shear on Z axis	74
Figure 6-42 Cross Piece -Bending Moment My on Y axis	75
Figure 6-43 Cross Piece -Bending Moment Mz on Z axis	75
Figure 6-44 Cross Piece -Torsion: Mx on x axis	75
Figure 6-45 Jourawky stress distribution	78

Figure 6-46 (a) Member with a shear stress component parallel to the grain (b) Member with b	oth
stress components perpendicular to the grain (rolling shear)	79
Figure 6-47 Torsional stress distribution	80
Figure 6-48 Combined bending with axial compression/tension	
Figure 6-49 Combined biaxial bending with axial compression/tension:	81
Figure 6-50 Overview of the room box	
Figure 6-51Section of the studied wall	
Figure 6-52 Overturning mechanism	
Figure 6-53Overturning mechanism - Orthogonal projections	
Figure 6-54 - activation of the chains Overturning mechanism	90
Figure 6-55 Figure 6 53Overturning mechanism - Orthogonal projections activation of the chair	าร 90
Figure 6-56 Roof timber beam subjected to seismic actions	91
Figure 6-57 Repartitions of forces - Roof timber beam subjected to seismic actions	91
Figure 6-58 Descriptions of the rafters crossed at the roof timber beam	92
Figure 6-59 Description of the crossing rafters at roof level	92
Figure 6-60 Roof Rafter Head Axial stresses : Crossing rafters T2-R2	
Figure 6-61 Roof Rafter Head Axial stresses : Crossing rafters T1-R1	95
Figure 6-62 Roof Rafter Head Axial stresses : Crossing rafters T1-R2	96
Figure 6-63 Roof Rafter Head Axial stresses: Crossing rafters T2-R1	
Figure 6-64 Roof Rafter Head Tangential stresses: Crossing rafters T2-R2	99
Figure 6-65 Roof Rafter Head Tangential stresses: Crossing rafters T1-R1	.100
Figure 6-66 Roof Rafter Head Tangential stresses: Crossing rafters T1-R2	.101
Figure 6-67 Roof Rafter Head Tangential stresses: Crossing rafters T2-R1	.102
Figure 6-68 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T2-R2	.103
Figure 6-69 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T1-R1	.104
Figure 6-70 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T1-R2	.105
Figure 6-71 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T2-R1	.106
Figure 6-72 Roof Rafter Head Bending moments: Torsnion: Tangential stresses: Crossing rafter	rs
T2-R2	.107
Figure 6-73 Roof Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters	T1-
R1	.108
Figure 6-74 Roof Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters	T1-
R2	.109
Figure 6-75 Roof Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters	T2-
R1	.110
Figure 6-76 Roof timber beam subjected to seismic actions (normal rafter)	.111
Figure 6-77 Repartitions of forces - Roof timber beam subjected to seismic actions (normal raft	er)
	.111
Figure 6-78 Descriptions of the rafters crossed at the roof timber beam actions (normal rafter)	.112
Figure 6-79 Description of the crossing rafters at roof level actions (normal rafter)	.112
Figure 6-80 Rafter Head Axial stresses: Crossing rafters T2-R2	.114
Figure 6-81 Rafter Head Axial stresses: Crossing rafters T1-R1	.115

Figure 6-82 Rafter Head Axial stresses: Crossing rafters T1-R2	116
Figure 6-83 Rafter Head Axial stresses: Crossing rafters T2-R1	117
Figure 6-84 Rafter Head Tangential stresses: Crossing rafters T2-R2	119
Figure 6-85 Rafter Head Tangential stresses: Crossing rafters T1-R1	120
Figure 6-86 Rafter Head Tangential stresses: Crossing rafters T1-R2	121
Figure 6-87 Rafter Head Tangential stresses: Crossing rafters T2-R1	122
Figure 6-88 Rafter Head Bending moments: Axial stresses: Crossing	rafters T2-R2123
Figure 6-89 Rafter Head Bending moments: Axial stresses: Crossing	rafters T1-R1124
Figure 6-90 Rafter Head Bending moments: Axial stresses: Crossing	rafters T1-R2125
Figure 6-91 Rafter Head Bending moments: Axial stresses: Crossing	rafters T2-R1126
Figure 6-92 Rafter Head Bending moments: Torsnion: Tangential str	esses : Crossing rafters T2-R2
	127
Figure 6-93 Rafter Head Bending moments: Torsion: Tangential stres	ses: Crossing rafters T1-R1128
Figure 6-94 Rafter Head Bending moments: Torsion: Tangential stre	sses: Crossing rafters T1-R2
	129
Figure 6-95 Rafter Head Bending moments: Torsion: Tangential stre	sses: Crossing rafters T2-R1
	130
Figure 6-96 Cross Piece – Compression	131
Figure 6-97 Cross Piece - Compression - Axial stresses	132
Figure 6-98 Cross Piece - Compression - Tangential stresses	133
Figure 6-99 Cross Piece – Tension	134
Figure 6-100 Cross Piece - Tension - Axial stresses	136
Figure 6-101 Cross Piece - Tension - Tangential stresses	137
Figure 6-102 Cross Piece – Friction/Inertia	138
Figure 6-103 Cross Piece - Friction/Inertia - Axial stresses	139
Figure 6-104 Cross Piece - Friction/Inertia - Tangential stresses	140
Figure 6-105 Parasitic Bending moment along Y axis due to tension a	nd flexion141
Figure 6-106 Parasitic Bending moment along Y axis due to compress	sion and flexion142
Figure 6-107 Parasitic Bending moment along Z axis due to compress	sion and flexion143
Figure 6-108 Parasitic Torsional Bending moment along X axis due to	compression on the notch144
Figure 6-109 Parasitic Torsional Bending moment along X axis due to	compression on the body
section	145
Figure 6-110 Figure 6 109 Parasitic Torsional Bending moment along	X axis due to Friction/Inertia
case on the body section	146
Figure 6-111 Kashmir Joint or Keyed Scarf Joint	146
Figure 7-1 Single modular unit – Decomposed	149
Figure 7-2 Single modular unit – Large -Decomposed	149
Figure 7-3 Roof – Decomposed	151
Figure 7-4 Normal stresses - Inside stones layers - Studied surfaces	Figure 7-5 Normal stresses -
Inside stones layers - Sigma Stresses	153
Figure 7-6 Normal stresses - Below timber beam - Sigma Stresses Fig	ure 7-7 Normal stresses -
Below timber beam - Studied surfaces	154

Figure 8-1 Analyzed layers for inside stones layer case	157
Figure 8-2 Force applied at the top of the wall	158
Figure 8-3 Triangular lateral distribution over the height of the wall for inside stones layer	159
Figure 8-4 Triangular lateral distribution over the height of the wall for inside stones layer- Hei	ghts
(cm)	160
Figure 8-3 Uniform lateral distribution over the height of the wall for inside stones layer	162
Figure 8-5 Analyzed layers for the below timber bands case	165
Figure 8-6 Triangular lateral distribution over the height of the wall for below timber bands	166
Figure 8-7 Triangular lateral distribution over the height of the wall for below timber bands -	
Heights	166
Figure 8-6 Triangular lateral distribution over the height of the wall for below timber bands	168
Figure 8-8 Critical layers for the in-plane seismic analysis	171
Figure 9-1 Overturning mechanism – example scheme	174
Figure 9-2 Overturning mechanism - tie-timber beam chains activation	174
Figure 9-3 Heights of the rafters and distances between the timber beams bands	176
Figure 9-4 Horizontal equilibrium – equilibrium method	176
Figure 9-5 Rotational equilibrium – equilibrium method	178
Figure 9-6 Hinges posotions Figure 9-7 Hinges heights and Blocks Heights	180
Figure 9-8 Unique seismic force on the top - Overturning Wall - $\alpha$ critical	181
Figure 9-9 Unique seismic force on the top - Overturning Blocks - $\alpha$ critical	182
Figure 9-10 Roof force + Wall force - Overturning Wall - $\alpha$ critical	183
Figure 9-11 Roof force + Wall force - Overturning Blocks - $\alpha$ critical	184
Figure 9-12 Unique seismic force on the top - Overturning Wall - Tmin	186
Figure 9-13 Unique seismic force on the top - Overturning Blocks - Tmin	187
Figure 9-14 Roof force + Wall force - Overturning Wall - Tmin	189
Figure 9-15 Roof force + Wall force - Overturning Blocks — Tmin	190
Figure 9-16 Overturning - RH90Shear most critical section	208
Figure 10-1 Flexible mechanism – example scheme	210
Figure 10-2 Flexible mechanism - tie-timber beam chains activation	211
Figure 10-3 Flexible mechanism - deformed tie-timber beam chains and activation	211
Figure 10-4 Static scheme of the timber tie-beam (clamped ends)	212
Figure 10-5 Hyperstatic scheme of the corner joint and actions from static scheme of the timb	er
tie- beam	213
Figure 10-6 Hyperstatic rigid-jointed frame	214
Figure 10-7 Rigid-Jointed Frame - names of the corners	215
Figure 10-8 Primary structure - Static system	216
Figure 10-9 Decomposition of the redundant frame	216
Figure 10-10 System 0 Figure 10-11 External Equilibrium System "0"	217
Figure 10-12 Internal Equilibrium System "0"	
Figure 10-13 Internal reactions System "0"	
Figure 10-14 System 1 Figure 10-15 External Equilibrium System "1"	218
Figure 10-16 Internal Equilibrium System "1"	219

Figure 10-17 Internal Reactions System "1"	219
Figure 10-18 System 2 Figure 10-19 External Equilibrium System "2"	219
Figure 10-20 Internal Equilibrium System "2"	220
Figure 10-21 Internal Reactions System "2"	220
Figure 10-22 System "3" Figure 10-23 External Equilibrium System "3"	220
Figure 10-24 Internal Equilibrium System "3"	221
Figure 10-25 Internal reactions System "3"	.221
Figure 10-26 Flexible response – Bendingbehavior - Analyzed beams	232
Figure 10-27 Bending behavior Pertinent masses for each timber band Figure 10-28 Bending	
behavior Heights of each timber band	233
Figure 10-29 Bending behavior -Distribution factors	234
Figure 10-30 Bending behavior - Rigid- jointed frame reactions for each beam	236
Figure 10-31 Bending behavior - Rigid- jointed frame reactions for each beam - Resisting rafter	s R
	237
Figure 10-32 Bending behavior - Distribution of the forces on the rafters	238
Figure 10-33 Bending behavior - Distribution of the forces on the rafters- corner joint	238
Figure 10-34 Rigid-Jointed Frame - names of the corners- scheme	248
Figure 10-35 Flexible - RH90Shear most critical section	267
Figure 11-1 Spread the connection points	.268
Figure 11-2 Pattern of Keyed scarf joint (or Kashmir joint)	.269
Figure 11-3Pattern for internal and external surface of the same wall	269
Figure 11-4 Forces acting on the steel wire connectors	270
Figure 11-5 Pattern of vertical fasten connector	271
Figure 11-6 Example of single diagonal connector with positive orientation	272
Figure 11-7 Example of single diagonal connector with negative orientation	.272
Figure 11-8 Preliminary design of diagonal connectors	273
Figure 11-9 Connectors for foundation	.276
Figure 11-10 Vertical connectors total wall - external	277
Figure 11-11 Vertical connectors total wall - internal	278
Figure 11-12 Connectors on total wall - external	278
Figure 11-13 Vertical Rafters – gross measuraments in cm	279
Figure 11-14 Connectors for vertical rafters	279
Figure 11-15 Thrifty Solution orthogonal projections	280
Figure 11-16 Thrifty solution	280
Figure 11-17 Optimal sSolution orthogonal projections	.281
Figure 11-18 Optimal solution	281
Figure 11-19 Rule of thumb for the roof	.282
Figure 11-20 Rule of thumb for the roof - Timber band at roof level exploded	283
Figure 12-1 Critical layers for the in-plane seismic analysis. Sliding	286
Figure 12-1 Critical layers for the in-plane seismic analysis. Sliding	286
Figure 12-2 Critical sections on the bhatar construction	.288
Figure 12-3 RH90Shear most ctitical section	.288

# LIST OF TABLES

Table 1 Shorea Robusta mechanical properties 1	19
Table 2 Shorea Robusta mechanical properties 2	19
Table 3 Design Value EC5-Nicole -1	20
Table 4 Design Value EC5-Nicole -2	20
Table 5 Rock characterization results	23
Table 6 Miller's correlation 1972	24
Table 7Miller's correlation 1965	24
Table 8 Reduction factor due to the presence of the timber beam	27
Table 9 Rockjoint data	48
Table 10 Barton method for Rockjoint Bhatar results	49
Table 11 Rockfill data	51
Table 12 Barton method for Rockfill Bhatar results	51
Table 13 Measures for all areas under stresses	66
Table 14 Roof rafter Head and Rafter head	93
Table 15 Roof rafter Head and Rafter head	.113
Table 16 Material Properties for single modular unit	.148
Table 17 Elementary parts of single modular unit - Volumes	.149
Table 18 Elementary parts of single modular unit - Weights and stresses	.150
Table 19 Material Properties for roof	.150
Table 20 Elementary parts of Roof - Volumes	.151
Table 21 Elementary parts of Roof - Weights and linear load	.152
Table 22 Total weight of roof on wall and on module	.152
Table 23 Normal stresses - Inside stones layers - Sigma Stresses	.153
Table 24 Normal stresses - Below timber beam - Sigma Stresses	.154
Table 25 Normal Stress and Coefficients of friction inside stones layer	.156
Table 26 Normal Stress and Coefficients of friction below timber beam	.157
Table 27 Force applied at the top of the wall - Data	.158
Table 28 Safe limit multipliers - Force applied at the top of the wall -inside stones layer case $$	.159
Table 29 Triangular distribution of the forces - inside stones layer case	.161
Table 30 Safe limit multipliers- Triangular lateral distribution-inside stones layer case	.162
Table 29 Uniform Distribution of the forces - inside stones layer case	.163
Table 30 Safe limit multipliers- Triangular lateral distribution-inside stones layer case	.164
Table 31 Safe limit multipliers - Force applied at the top of the wall –below timber band case	.165
Table 29 Triangular distribution of the forces – below the timber bands case	.167
Table 32 Safe limit multipliers- Triangular lateral distribution- below timber bands case	.167
Table 29 Uniform distribution of the forces – below the thimber bands case	.168
Table 32 Safe limit multipliers- Triangular lateral distribution- below timber bands case	.169

Table 33 Summary of results for the in-plane seismic analysis	170
Table 39 Summary of results for the in-plane seismic analysis Reduced by Safety factor $\gamma_b$ = 1.5	.172
Table 34 Masses of each analyzed layer	175
Table 35 Total weight and mass of the wall composed by 3 single modular unit	175
Table 36 Heights of the considered rafters	175
Table 37 Horizontal equilibrium - minimum tensions	177
Table 38 Centroid of the section of the wall - data	178
Table 39 Rotational equilibrium - minimum tensions	179
Table 40 Weights and masses pertinent to studied blocks	180
Table 41 Heights and ratios for $\Delta$ proportional multiplier between 0 and 1	180
Table 42 Roof force + Wall force - Overturning Blocks - α critical multipliers	185
Table 43 Weights and masses pertinent to studied blocks - Tmin	185
Table 44 Heights and ratios for $\Delta$ proportional multiplier between 0 and 1 - Tmin	185
Table 45 Unique seismic force on the top - Overturning Wall - Tmin	187
Table 46 Unique seismic force on the top - Overturning Blocks - Tmin -data and results	188
Table 47 Unique seismic force on the top - Overturning Blocks - Tmin	188
Table 48 Roof force + Wall force - Overturning Wall - Tmin	189
Table 49 Roof force + Wall force - Overturning Blocks - Tmin -data and results	191
Table 50 Roof force + Wall force - Overturning Blocks – Tmin	191
Table 51 Geometric dimensions for Notch and Body Areas	195
Table 52 Pertinent masses foe each timber bands	233
Table 53 Bending behavior - Distribution of the weight over the height	234
Table 54 Bending behavior - Wall lenght	234
Table 55 Bending behavior - Distribution factors and seismic loads	235
Table 56 Flexible behavior - Rafter body reactions for each beam	235
Table 57 Bending behavior - Rigid- jointed frame reactions for each beam	236
Table 58 Bending behavior - Rigid- jointed frame reactions for each beam - Resisting rafters R	237
Table 59 Bending behavior - External rafter T1 - compression	239
Table 60 Bending behavior - Internal rafter T2 - tension	239
Table 61 Flexible behavior - Rafter body reactions for each beam - Verifications	248
Table 62 Summary of results for the in-plane seismic analysis	285

## **SIMBOLOGY**

#### Symbols and abbreviations for Shorea Robusta EN388

E<sub>0,mean</sub> mean characteristic value of modulus of elasticity parallel to grain (in kN/mm2)

 $E_{0.05}$  5-percentile characteristic value of modulus of elasticity parallel to grain (in kN/mm2)

 $E_{90,mean}$  mean characteristic value of modulus of elasticity perpendicular to grain (in kN/mm2)

 $f_{c,0,k}$  characteristic value of compressive strength parallel to grain (in N/mm2)

 $f_{c,90,k}$  characteristic value of compressive strength perpendicular to grain (in N/mm2)

 $f_{m,k}$  characteristic value of bending strength (in N/mm2)

 $f_{t,0,k}$  characteristic value of tensile strength parallel to grain (in N/mm2)

 $f_{t,90,k}$  characteristic value of tensile strength perpendicular to grain (in N/mm2)

 $f_{v,k}$  characteristic value of shear strength (in N/mm2)

G mean mean characteristic value of shear modulus (in kN/mm2)

 $\rho_k$  characteristic value of density (in kg/m3)

 $\rho_{mean}$  mean value of density (in kg/m)

#### **ANNEX A Determination of values**

Tensile strength parallel to grain  $f_{t,0,k} = 0.6 f_{t,m,k}$ 

Compression strength parallel to grain  $fc,0,k = 5*(fm,k)^0,45$ 

Shear strength

fv,k shall be taken from Table 1 Tensile strength perpendicular to grain

 $f_{t,90,k} = 0.4 \text{ N/(mm}^2)$  for softwoods

 $f_{t,90,k} = 0.6 \text{ N/(mm^2)}$  for hardwoods

Compressive strength perpendicular to grain

f (c,90,k)=0,007 \*  $\rho$  k for softwoods

f  $(c,90,k)=0,015*\rho k$  for hardwoods

Modulus of elasticity parallel to grain

 $E_0,05=0,67*E_0,mean$  for softwoods

 $E_0,05=0,84*E_0,mean$  for hardwoods

Mean modulus of elasticity perpendicular to grain

 $E_{(90,mean)}=E_{(0,mean)}/30$  for softwoods

 $E_{(90,mean)}=E_{(0,mean)}/15$  for hardwoods

Mean shear modulus G\_mean=E\_(0,mean)/16

Mean density  $\rho_{\text{mean}=1,2} \rho_{k}$ 

γ: Specific weight

λ: Slenderness

 $\sigma_N$ : Normal stress

 $\tau$ = shear stress

 $\mu$ = friction coefficient

As= Surface area

B= Base of the wall

c= cohesion

F<sub>s</sub>= seismic force

H= Height of the wall

L= Length of the wall

Mext: External moment

M<sub>sp</sub>: moment due to the sprigs

t: Thickness

W<sub>roof</sub>: Weight of the roof

Wt: Weight of the wall

 $M_{\text{ytf}}\,$  : Parasitic Bending moment along Y axis due to tension and flexion

 $M_{\text{ycf}}\,$  : Parasitic Bending moment along Y axis due to tension and flexion

M<sub>v1</sub> Parasitic Bending moment along Y axis due to tension and flexion on external notch

M<sub>y2</sub> Parasitic Bending moment along Y axis due to tension and flexion on internal notch

M<sub>z1</sub> Parasitic Bending moment along Z axis due to compression and flexion

M<sub>z2</sub> Parasitic Bending moment along Z axis due to compression and flexion

M<sub>x1</sub> Parasitic Bending moment along X axis due to compression and flexion

M<sub>x2</sub> Parasitic Bending moment along X axis due to compression and flexion

Wtb weight Timber Band

Wrs weight A - roof support

Wmb weight C - main block

Wof weight D - outer foundation

**6tb Stress under Timber Band** 

Ors Stress under A - roof support

бтb Stress under С - main block

боf Stress under С - main block

Wearth weight of Earth/clay

Wtwigs weight of Twigs

Wringstones weight of Ring of stones

Wplanks weight of Planks

Wrb weight of Roof beams

Wearth linear: linear load of Earth/clay

Wtwigs linear: linear load of Twigs

Wringstones linear: linear load of Ring of stones

Wplanks linear: linear load of Planks

Wrb linear: linear load of Roof beams

 $\alpha$  is the load multiplier

Wtot is the total weight of the box structure and of the roof

*PGA* is the peak ground acceleration

μi is the friction coefficient of the i<sup>th</sup> layer

Wi is the pertinent weight on the i<sup>th</sup> layer

 $\beta_i$ : is the distribution factor corresponding to the analyzed layer

 $W_i$ : is the weight corresponding to the analyzed layer

 $h_i$ : is the height corresponding to the analyzed layer

 $\sum_{i=1}^{N} W_i * h_i + W_{roof} * H$ : is the summation of all the of all the masses times the corresponding heights

 $\mu s_j$  is the friction coefficient obtained by the Barton models for rockfill corresponding to the analyzed layer

Nj is the pertinent normal force acting on the on the analyzed layer

Chapter 9

 $T_{min}$  is the minimum tension allowed for resisting to the seismic action

n is the total number of the rafters , for 3,6 m length wall = 12 (Each timber tie-beam is composed by 2 rafters)

 $M_{tot}$  is the total mass of the 3,6 m length wall

 $a_q$  is the seismic acceleration in g

g is the gravity acceleration constant =  $9.81 \text{ m/s}^2$ 

Htchain is the height from the ground of the centroid of the roof rafter beam

H is the height of the centroid of the section of the wall

B is the horizontal component of the centroid of the section of the wall

 $E_{ext}$  is the external energy

 $E_{int}$  is the internal energy

 $\beta$  is the rotation angle for the overturning mechanism

 $\delta 1$  is the displacement of the centroid

 $\delta 2$  is the displacement of the application point of the considered seismic force (in same case just the roof force)

 $\Delta$  is the proportional multiplier between 0 and 1

 $W_{roof}$  is the weight of the roof

 $W_{wall}$  is the weight of the wall

 $\delta 3$  is the displacement of the application point of the considered seismic force of the wall

 $T_{min}$  is the minimum tension due to the seismic event on the roof tie timber beam

 $\delta t_{chain}$  is the displacement of the application point of the roof timber beams acting as a chain

 $Ht_{chain}$  is the height of the roof timber beams acting as a chain

#### Chapter 10

n: number of rigid joints n = 4

m: number members m = 4

r: support reactions r = 3

i: degree of indeterminacy i = ?

 $\eta_i$ : is the effective displacement in the effective structure

 $\eta_{i0}$  : is the displacement due to the primary system on the i released

 $X_i$ : is the unitary force in the position of the i released

 $\eta_{ik}$ : is the displacement of the point of application of the released  $X_i$  due to the redoundant  $X_k = 1$ 

n: is the number of the released equal to the degree if indeterminacy i

 $\eta_{ik} = \eta_{ki}$  due to Maxwell Theorem

*Mass<sub>i</sub>*: is the mass involved for the specific tie-timber beam

g: is the gravity accelleration

 $\alpha$ : is the seismic load multiplier

L: is the length of the wall

L: is the length of the wall equal to 3.6 m

l: is the length of the wall where the load is distributed, equal to 2.78 m

d: is the distances between all the timber elements, equal to 0.36 m

 $\gamma_b = 1.5$  safety factor for the amplification of the seismic actions.

## 1 INTRODUCTION

# 1.1 Background

Bhatar system is a traditional method of construction which involves a vertical succession of dry stacked stones masonry and timber beam. Through the century and countries this kind of architecture has been used for many different purpose and different scale, temples for religions, forts for military camps and houses for civil use. Along the time some of these structures of the past are still standing after important earthquake, this suggest us that bhatar system has somehow a good seismic behavior. The different between the constructions that have survived and those who did not may be due to many factors. The knowledge of the know-how goes from an old generation to a new one, because of this there are many differences about materials, about the proper place where to build but most of all the differences about the techniques are the most important.

In the poor and lost areas where this kind of architecture is used is important to use local material and to avoid the use of material or component which need to be imported from somewhere else, this is not just because it is important to save money but most of all because there are no proper infrastructures and this means more obstacles and some time the impossibility to be done.

In order to give a reference point, international organizations such as ERRA, UN-HABITAT, SDC and FRC have published "Bhatar construction - An illustrated guide for craftsmen".

Guidebook prepared by the Swiss Agency for Development and Cooperation SDC (Tom Schacher, technical advisor). In collaboration with: French Red Cross and Belgian Red Cross (technical research and development) UN Habitat, NSET and NESPAK (revisions) French Red Cross (Translation into Urdu) Mansehra, NWFP, April 2007

This guide shows how to built-up a bhatar house and the gross dimensions that must be satisfied.

Thus, this research was performed to ensure that this alternative building technique can be built in a seismic region knowing that it will be a safe structure and that can be used for a post-disaster reconstruction in developing countries.

# 1.2 Justification of the document and objectives

The use of bhatar system is a traditional technique in the construction field and it is widely used all over the Himalayan area due to some factors such as durability of the structure, low environmental impact, cost-effective ratio.

Considering the advantages that this system carries, it can be an alternative building technique and post-disaster reconstruction for houses in developing countries where it can be used for individual housing or for community facilities. Thereby, this technique can be built in remote areas, locations

difficult to reach and poorly supplied areas with the advantage that gabion boxes are easily installed and that deployment can be performed without special equipment and there is no need of highly trained personnel.

On the other hand, from a seismic point of view, there will be "weight issues" because the bhatar are heavy due to the rocks (it's known that the seismic forces acting on the structure are proportional to the weight). Thus, the need of research has been identified in order to understand the static and seismic behavior of this kind of structures focusing on the limitations of the system and the structural safety under a certain seismic action.

#### 1.2.1 General objectives

Based on the justification of this document, this dissertation aims at understand the behavior inplane and out-of-plane under seismic actions of a modular box composed by walls built-up with bhatar method and to give practical suggestions and simple formulas for the dimensioning of the structure, satisfying structural safety conditions.

#### 1.2.2 Specific objectives

- To comprehend the compression behavior and strength of a single Wall, composed by elementar modules under vertical loads.
- To verify the structural safety under seismic actions in-plane and out-of-plane of a wall build-up with bhatar system.
- Conduct analytical considerations to examine the effect of lateral forces on the behavior of a bhatar system.
- Propose constructions details and limitations to acquire an assure good seismic behavior of the structure
- To develop rules of thumb for a proper dimensioning and construction of this kind of structures in order to be a seismic resistant structure.

# 1.3 Organization of the thesis

The work has been organized starting from the elementary elements used in the Bhatar system thus starting from the geometry following the guide line of Architect Tom Schacher.

The following points shows the steps of the logic path followed in the work:

- Studies of Tom Sacher manual
- Definition of a single module

#### Definition of the wall

- Definition of one room module (box)
- Definition of material properties: Timber SHOREA ROBUSTA
- Definition of material properties: Stones LIMESTONE
- Studies on Rock discontinuities: Barton model –
- Connections Eurocode 5 : EN 1995-1-1 :2004+A 1- DESIGN ULS
- Static Analysis
- Seismic analysis in plane application of Barton model
- Seismic analysis out of plane Overturning
- Seismic analysis out of plane Bending
- Practical rules of thumb.

# 2 BHATAR

#### 2.1 Traditional definition of Bhatar

*Bhatar* is a traditional construction system consisting of stone mortarless masonry walls reinforced with horizontal timber ladder-beams, which combine to resist and dissipate the energy and stresses induced during an earthquake.

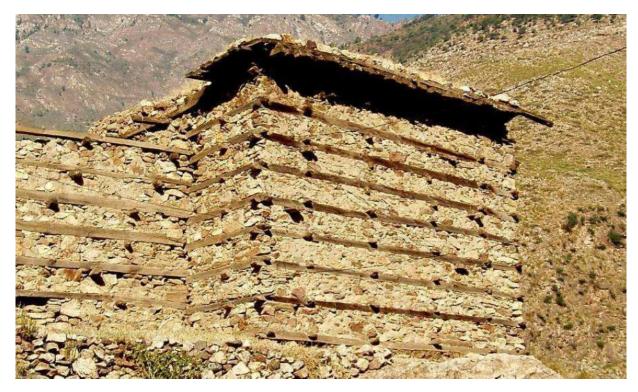


Figure 2-1 Project entry 2008 Asia Pacific - "Advocacy of traditional earthquake-resistant construction, North-West Frontier Province, Pakistan": "Bhatar" at Besham Fort.

Through the century and countries this kind of architecture has been used for many different purpose and different scale, temples for religions forts for military camp and houses for civil use.

Along the time some of these structures of the past are still standing after important earthquake, this suggest us that bhatar system has somehow a good seismic behavior. The different between the constructions that have survived and those who did not may be due to many factors. The knowledge of the know-how goes from an old generation to a new one, because of this there are many differences about materials, about the proper place where to build but most of all the differences about the techniques are the most important.

In the poor and lost areas where this kind of architecture is used is important to use local material and to avoid the use of material or component which need to be imported from somewhere else, this is not just because it is important to save money but most of all because there are no proper infrastructures and this means more obstacles and some time the impossibility to be done.

This type of construction has been extensively used in Turkey, Afghanistan, Pakistan, India and Nepal for many centuries, as shown in figure below. Nepal is the country taken as reference point for the local material.

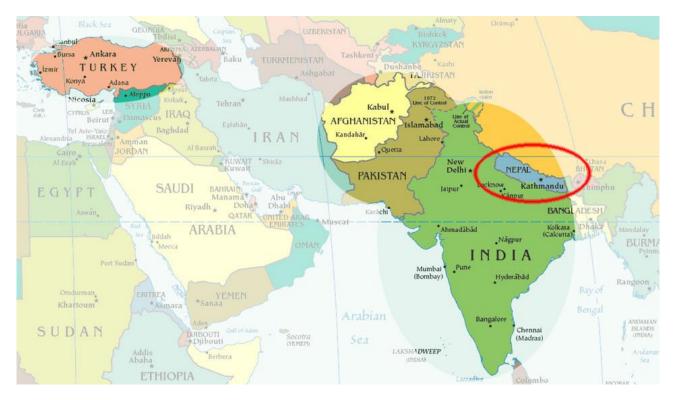


Figure 2-2 Regions of the world where Bhatar is still used

Nepal is subjected to very strong earthquake because its characteristic positionas shown in the picture 2-3.

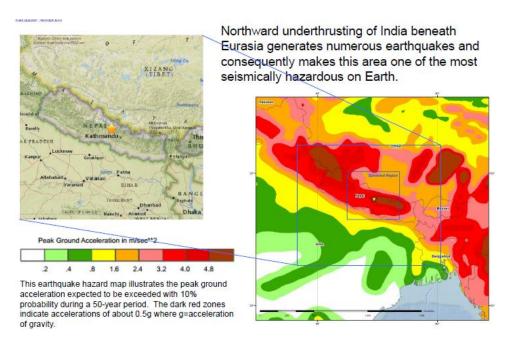


Figure 2-3 Nepal peak ground acceleration

# 2.2 Tom Schacher Manual : Bhatar construction - An illustrated guide for craftsmen

Arch Tom Schacher

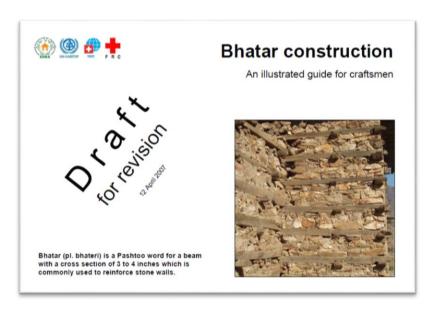


Figure 2-4 Bhatar construction-An illustrated guide for craftsmen

#### 2.2.1 Gross shape and dimensions

The fisrt thing described is the position of the structure and the gross shape. As it is shown in the figure 2-5 it is always better to choose a simple and regular structure, if necessary it is better to subdivide it into rectangular parts.

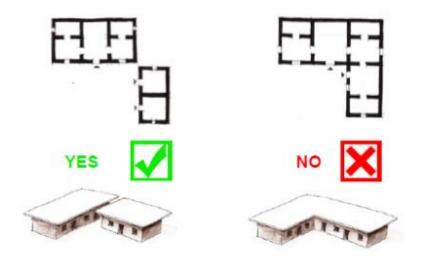


Figure 2-5 Divided rectangular structures

The first suggestion about the dimensions is the relation about the length and the width. The house must not be longer than three times the width, as it is shown in figure 2-6.

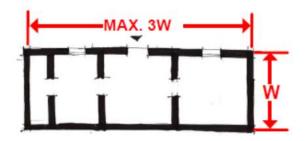


Figure 2-6 Gross dimension - ratio length/width

# 2.2.2 Foundation and plinth band

The foundation should be at least  $2\frac{1}{2}$  feet(0,762 m) wide and 3 feet (0,91 m) deep. The plinth band should be placed 1 foot (0,3 m) above the foundation (1 foot out of the ground) in order to avoid the contact with water, as it is shown in figure 2-7.

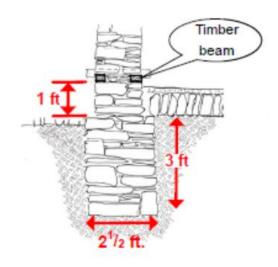


Figure 2-7 Foundations

The plinth band must pass under the door. It should be continuous along all the perimeter (better if it is made in RC, it will not rot),, as it is shown in figure 2-8.

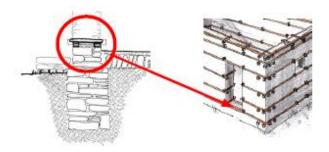


Figure 2-8 The plint

#### 2.2.3 The Walls

The walls must be smaller than the values reported in the figure 2-9.

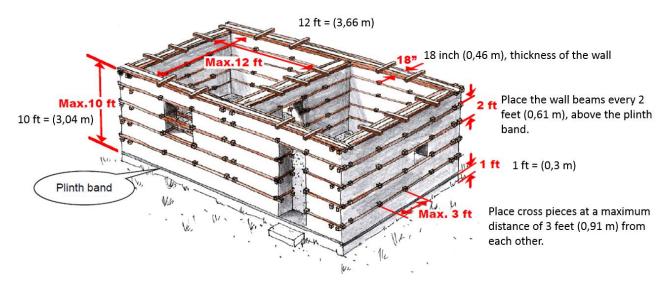


Figure 2-9 Wall dimensions

The drawing is not in scale, in a real scale it would be appreciated the fact that the spaces are quite small then the necessity to add the rooms.

# 2.2.4 Wall - joints

The timber elements may be not enough long to cover all the length of the wall so it is suggested to use scarf keyed joint along their length but taking into account that at each level they must be in different position and not along a vertical line as shown in figure 2-10., at the same time the position of the stones must be always laid down in order to have a dovetail as shown in figure 2-11

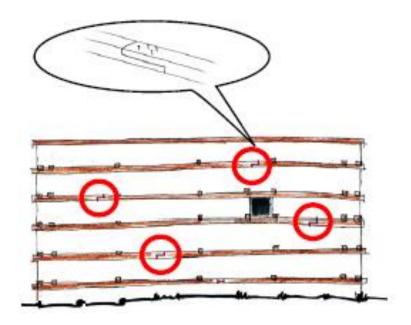


Figure 2-10 Spread the connection points.

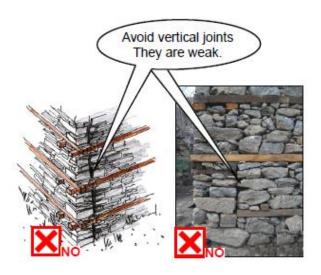


Figure 2-11 Raise all walls together to avoid vertical joints

#### 2.2.5 Kashmiri joint or Keyed scarf joint

The joints in the timber element must be done with Kashmiri joint or normally known as keyed scarf joint as shown in figure 2-12.

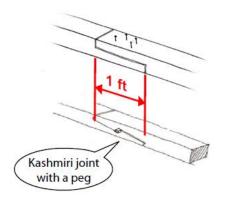
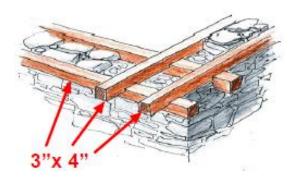


Figure 2-12 Kashmiri joint or Keyed Scarf Joint

#### 2.2.6 Connections - Corners

The connections on the corners stand due to lap joints and Minimum size of beam is 3" (7,62 cm) high by 4" (10,16 cm) wide, as shown in figure 2-13.



Figure~2-13~minimum~size~of~the~beams/rafters

Beams must be hooked together in the corners. Cut a notch of 1" (2,54 cm) into all four corner beams. Add 2 nails (3" =7,62 cm) for more security. Keep 4" (10,16 cm) of wood after all notches for strength. As shown in figure 2-14.

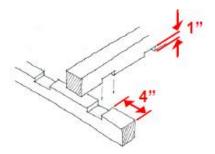


Figure 2-14 Lap joint – dimension

#### 2.2.7 Connections – Cross Pieces

Along the wall cross pieces must be insert in order to assure stability. Cross pieces help to hold the beams and walls together. You need notches only on the cross pieces, but not on the main beams. As shown in figure 2-15

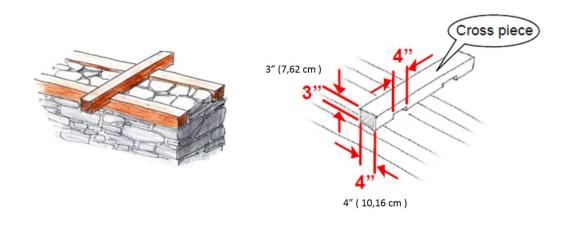


Figure 2-15 Cross Pieces

#### 2.2.8 Connections – Internal wall

In case of double room they are specified how the connections between the walls must be done. Minimum size of beam is 3" (7,62 cm) high by 4" (10,16 cm) wide. Where internal walls connect, only notch the internal wall beams, not the main beams, as shown in figure 2-16.

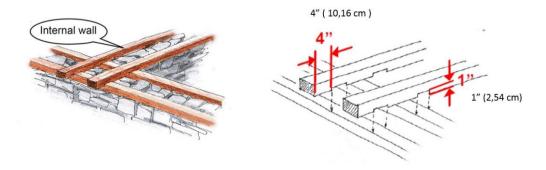


Figure 2-16 Internal wall joint

# 2.2.9 Openings

The distance between openings should be minimum 3 feet (0,91 m) ,windows and doors must not be wider than 3 feet (0,91 m) ,the windows must be between the beams. As shown in the figure 2-17

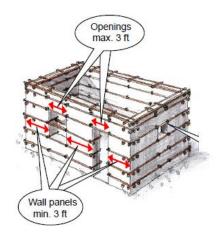


Figure 2-17 Openings

# **2.2.10 Doors**

The integrity of the structure must be assured thus it must be avoided any modification and all the openings must be bounded with cross pieces as shown in figure 2-18

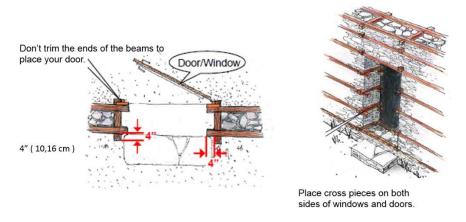


Figure 2-18 Openings 2

#### **2.2.11 Windows**

The windows must be reinforced with beams, for lintel must be added two pieces of wood in between the existing beams to support stones above. It must pass at least 1 foot(0,3 m) into masonry on each side of the opening, as shown in figure 2-19

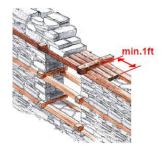


Figure 2-19 Lintel reinforcement

#### **2.2.12The Roof**

The roof considered for this research is the flat heavy roof with earth cover which is the worst case but it does not need metal sheet to cover which are difficult to be found in far regions.

Some suggestions are given referring the figure below. 1-Let the top beams (bhateri) stick out of the wall 1 foot on each side. Connect them with nailed cross pieces. 2-Add the 4"x6"roof beams and let them too stick out 1 ft on each side (also over the retaining back-wall if there is) to protect the wall against rain. 3- Nail the planks on the roof beams leaving a half inch gap between each. 4- Place flat stones along the edge of the roof to contain the earth. 5- Add twigs and small branches in a layer 4 to 6 inch thick. 6 Cover with earth 4 to 6 inch thick. 7-Avoid to make the earth cover thicker over the years.

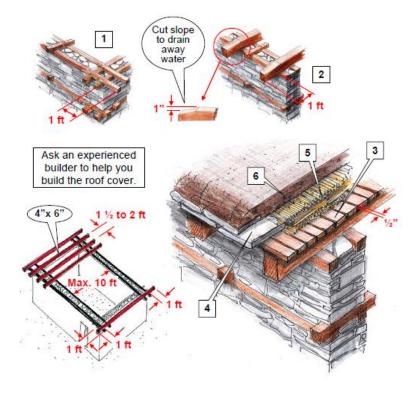


Figure 2-20 the flat heavy roof with earth cover

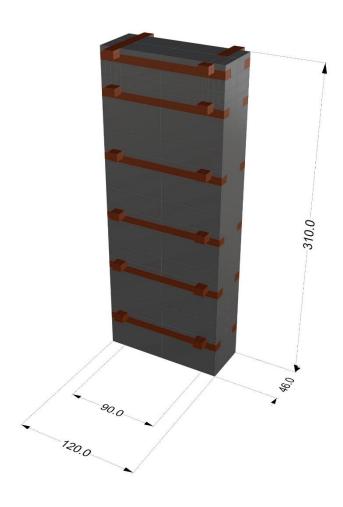
# 3 STUDY CASE

Following the Tom Schacher Manual

Following the guide lines given by Architect Tom Schacher it has been defined a basic module of the wall which can be used as modular unit in order to built square or rectangular housing unit.

# 3.1 Single modular unit

In accordance to the manual the single unit has been drawn starting from the ground layer until the roof support. The beams are placed every 60 cm exept the first beam from the bottom and the roof beam that are placed at 30 cm. The global measure are shown in the figure 3-1.



 $Figure \ 3-1 \ Modular \ unit-perspective$ 

# 3.1.1 Orthogonal projection

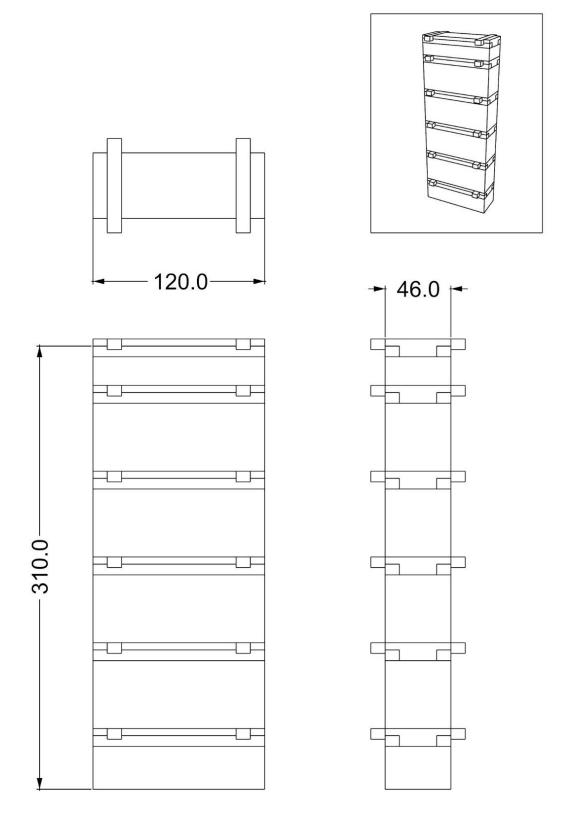
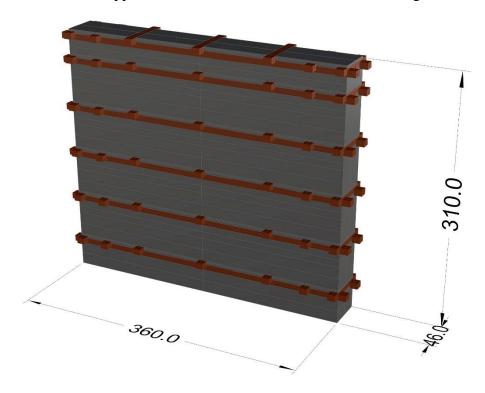


Figure 3-2 Modular unit - Orthogonal projection in cm

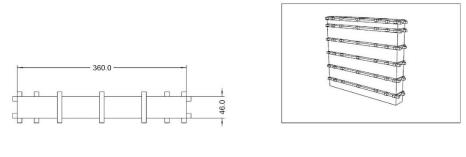
# 3.2 Single Wall

Using the modular unit it has been composed the largest wall suggested by the guide line. With a length of 12 feet it has been approximated to 3.6 m, width of 0,46 m and height of 3.1 m.



Figure~3-3 Largest~wall~possible~,~length~of~3.6m

# 3.2.1 Orthogonal projection



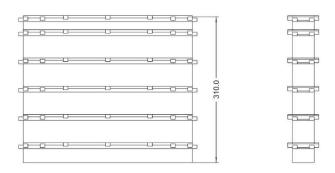


Figure 3-4 Wallt - Orthogonal projection in cm

# 3.3 One room box

Using four perimetric wall for a total around 12 modular units a room box have been defined. This room box is the largest single habitat unit which can be built with the use of the guide line. The one room box is composed by:

- Foundation and plinth band made of stones
- · First seismic band made of wood
- Dimensions (length: 3,60 m; Width: 3,60 m; Height 3,0 m)
- 1 door
- 2 window

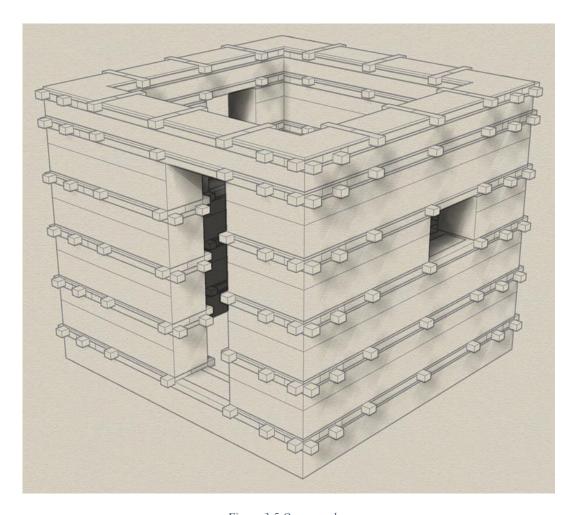
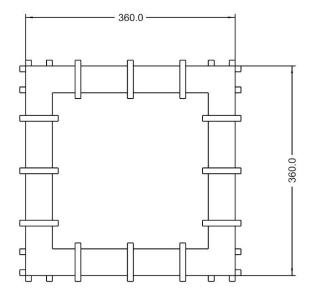
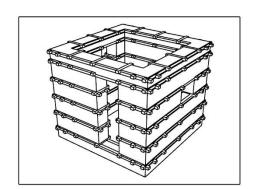
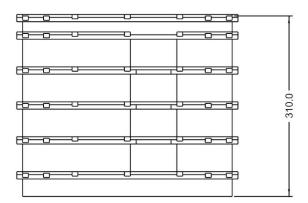


Figure 3-5 One room box

# 3.3.1 Orthogonal projection







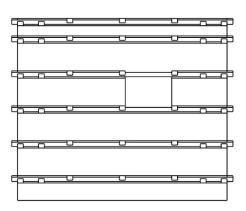


Figure 3-6 One room box orthogonal projections in cm

In the guide line is described the possibility of enlarging the structure adding walls in order to compose a second smaller habitat unit. The aim of the thesis is to understand the behavior of the basic structure thus all the studies regards the basic room box and in particular the behavior of the perimetric wall. As shown in the following figure the section AA represents the studied wall.

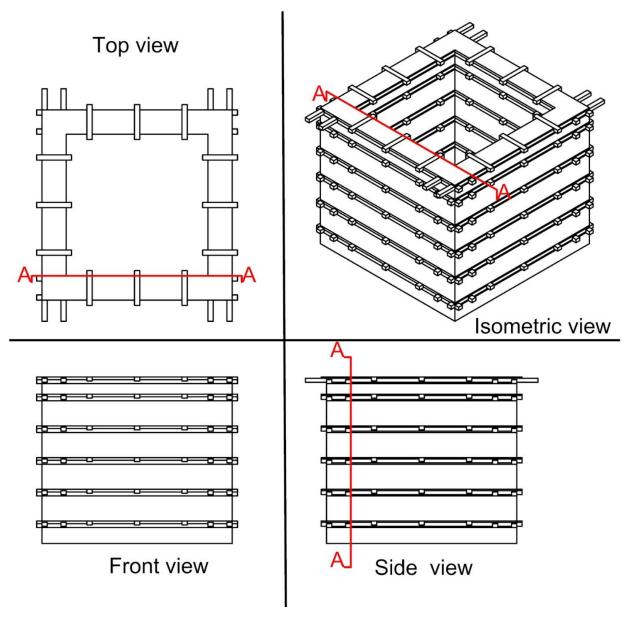
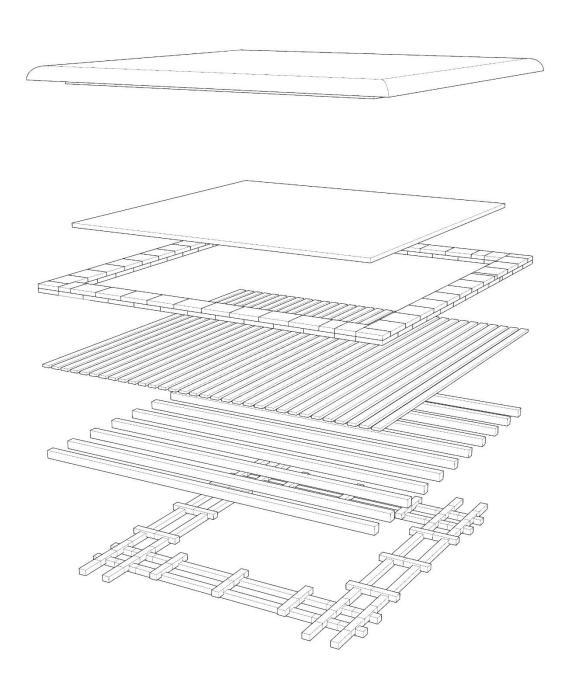


Figure 3-7 Section AA - studied wall

# **3.3.2** The Roof

The roof has been considered as flat heavy roof with earth cover which is composed , as show in the figure below, by (from the bottom):

- Last timber band
- Roof beams 10 cm height
- Planks 3 cm height
- Ring of flat stones 10 cm height
- Twigs 5 cm height
- Earth 20 cm height



Figure~3-8~Flat~earth~heavy~roof-exploded

# 4 MATERIALS

This chapter describe the two basic material, timber and stones used in the Nepal region.

# 4.1 Timber: Shorea Robusta

Thanks to the suggestions of Architect Martijn Schildkamp we know that exact timber traditionally used in Nepal to build Bhatar structures is the so called Shorea Robusta in Nepal language is called SAL.

#### 4.1.1 Botanic Characteristics

Below in the figure are reported the botanic characteristics.

Kingdom:	Plantae
Division	Magnoliophyta
Class:	Magnoliopsida
Order:	Malvales
Family:	Dipterocarpaceae
Genus:	Shorea
Species :	S. robusta
Scientific Name :	Shorea robusta



Figure 4-1 Shorea Robusta – SAL

# 4.1.2 Mechanical properties of Shorea Robusta

In order to find the proper mechanical properties of Shorea Robusta it has been necessary a bibliographic research. This reaserch ended with 4 important sources which are listed below:

- Source 1 : MECHANICAL PROPERTIES AND DURABILITY OF SOME SELECTED TIMBER SPECIES (M. Bellal Hossain1 and A.S.M. Abdul Awal2\*)
- Source 2: STUDIES ON TENSILE STRENGTH PROPERTY OF COMMERCIAL TIMBER SPECIES OF SOLAN DISTRICT (Himachal Pradesh SEEMA BHATT, BUPENDER DUTT, RAJESH KUMAR MEENA and TASRUF AHMAD\*)
- Source 3 : COMPARISON OF TEST RESULTS OF VARIOUS AVAILABLE NEPALESE TIMBERS FOR SMALL WIND TURBINE APPLICATIONS (R. Sharma1 1 1, R. Sinha, P. Acharya, L. Mishnaevsky Jr. 2, P. Freere3)
- Source 4: TECNOLOGIA DEL LEGNO (G. Giordano, UTET, Torino 1988.)

The different values found in the research have been averaged and they are reported in the following table.

Table 1 Shorea Robusta mechanical properties 1

Sal or Shorea robusta						
	Density ρ : ( Kg/m3)	Specific gravity SG = psubstance / pH2O	ultimate compressive strength σu (Mpa)	Tensile Ultimate stress longitudinal axis (MPa)		
source 1	921	0,84	48	/		
source 2	/	/	/	78,1		
source 3	913 or 950	/	/	/		
source 4	875	/	61	/		
Average	914,75	0,84	54,5	78,1		

Table 2 Shorea Robusta mechanical properties 2

	Sal or Shorea robusta						
	Young's Modulus E : (Gpa)	Bending Strength ( Mpa)	Minimum Static Bending Strength (Mpa)	Average Hardness (Mpa) Indentations : Incavatura	Minimum Hardness (Mpa)		
source 1	/	/	/	/	/		
source 2	/	/	/	/	/		
source 3	12,55	83,85	61,7	87,5 (+0- 42,5)	45		
source 4	15,6	121	/	medium/high	/		
Average	14,075	102,425	61,7	87,5 (+o- 42,5)	45		

# 4.1.3 Characteristic Values from EN 338

Comparison with Classification of timber in accordance with UNI EN 338 : 2009 Shorea Robusta is classified as D70 thus they have been used the following reference values.

Shorea Robusta		Hardwood species		
		D70		
Strength properties (in N/mm <sub>2</sub> )				
Bending	fm,k	70		
Tension parallel	ft,0,k	42		
Tension perpendicular	ft,90,k	0,6		
Compression parallel	fc,0,k	34		
Compression perpendicular	fc,90,k	13,5		
Shear	fv,k	5,0		
Stiffness properties (in kN/mm <sub>2</sub> )				
Mean modulus of elasticity parallel	E0,mean	20		
5 % modulus of elasticity parallel	E0,05	16,8		
Mean modulus of elasticity perpendicular	E90,mean	1,33		
Mean shear modulus	Gmean	1,25		
Density (in kg/m₃)				
Density	ρk	900		
Mean density	ρmean	1080		

# 4.1.4 Design Values from EC 5 and en.1995.1.1.2004 and NICOLE – Istruzioni CNR DT206 2007

Following the Eurocode 5 and the national codes for the design timber structure they have been selected and computed the following values.

Table 3 Design Value EC5-Nicole -1

DESIGN VALUE		
the partial factor for a material property	γm	1,5
Service class		2
modification factor taking into account the effect of the duration of load and moisture	Kmod permanent action	1,1
Depth factor	kh	From case

Table 4 Design Value EC5-Nicole -2

Strength properties (in N/mm2)				
Bending	fm,d	51,33		
Tension parallel	ft,0,d	30,80		
Tension perpendicular	ft,90,d	0,44		
Compression parallel	fc,0,d	24,93		
Compression perpendicular	fc,90,d	9,90		
Shear	fv,d	3,67		
Stiffness properties (in kN/mm2)				
Mean modulus of elasticity parallel	E0,d	13,33		
5 % modulus of elasticity parallel	E0,05 d	11,20		
Mean modulus of elasticity perpendicular	E90,d	0,89		
Mean shear modulus	Gd	0,83		
Density (in kg/m3)				
Density	ρk	900		
Mean density	ρmean	1080		

# 4.2 Stones: Main construction material since the Stone Age

In order to define the most probable stone largely used for the construction of bhatar the research has been started looking on which are the most common stones in the Nepal region taken as reference point. Thanks to Architect Martijn Schildkamp we know that people collect the stone from the ground and sometimes they take them directly to the quarries.

The most common rocks and their used in the Nepal region are listed below:

- marble, basalt, granite and red sandstones are cut into slabs and used in decoration;
- phyllite, slates, flaggy quartzite and schist are used for roofing;
- limestone, dolomite, quartzite, sandstone are used for aggregate in various construction works, road paving and flooring;

• vast quantities of river boulders, cobbles, pebbles and sands are mined as construction materials/ aggregates.

#### References:

DMG (Y.P. Sharma et al 1988) has evaluated such materials (boulders=347,006,000m3, cobbles=214,261,000m3 and pebbles=229,205,000m3) in the major rivers of Terai region.

MINERAL RESOURCES OF NEPAL AND THEIR PRESENT STATUS- Krishna P. Kaphle, Former Superintending Geologist, Department of Mines and Geology, Kathmandu, Nepal Former President, Nepal Geological Society

The world Housing Encyclopedia (WHE) specify that the rocks most used in wall and frame as rubble stones are Slates ,Limestone, Quartzite.

Architect Martijn Schildkamp collected pictures during the construction of a bhatar house. Comparing the pictures of the stones he sent and weaving togheter the possible material, it has been choosen the strongest one, limestone.





Figure 4-2 Architect Martijn Schildkamp - bhatar stones







Dolomite

Sandstone/Arenaria

Quartzite

Figure 4-3 Sedimentary rocks

The limestone/Calcarea has been choosen for the following steps of the thesis.

Limestone is good for building, and is generally the same either in masonry or building block. It is not a good fit for cobblestones because it is too soft.



Figure 4-4 Limestone/Calcarea

# 4.2.1 Limestone mechanical properties

In the context of this thesis the important parameters of the limestone are:

- Dry density
- Rebound Number with Schmidt hammer L-type (MATEST of Italy)
- Unconfined Compressive Strength (Miller's formula, 1972)
- JCS, joint compressive strength (Miller's formula, 1965)

The importance of these parameters will be explained in the Chapter 5 which will describe the surfaces behavior and the importance of the absence of the mortar.

In the table 5 are shown the results obtained by the research team of Dr. Ramli Nazir Faculty of Civil Engineering, Department of Geotechnics and Transportation, Universiti Teknologi Malaysia (Malaysia). The publication "Prediction of Unconfined Compressive Strength of Limestone Rock Samples Using L-Type Schmidt Hammer" has been really usefull in order to have preliminary laboratory data for the application of the Barton model which will be explained in the following Chapters.

Table 5 Rock characterization results

No.	Sample Type	Dry Density (kg/m3)	R : rebound number	UCS: Miller`s correlation(MPa)	UCS:Obtained in Laboratory(MPa)
1	Limestone	2817,0	36,0	72,0	72,9
2	Limestone	2748,0	35,9	76,0	72,9
3	Limestone	2646,0	31,5	55,0	58,5
4	Limestone	2777,0	31,5	60,0	60,6
5	Limestone	2671,0	28,9	49,0	52,2
6	Limestone	2773,0	30,4	56,0	56,4
7	Limestone	2676,0	37,7	79,0	76,7
8	Limestone	2683,0	36,8	76,0	75,7
9	Limestone	2748,0	34,8	71,0	72,5
10	Limestone	2707,0	35,6	72,0	69,6
11	Limestone	2759,0	36,6	79,0	78,1
12	Limestone	2704,0	33,9	66,0	63,5
13	Limestone	2726,0	35,1	71,0	75,7
14	Limestone	2796,0	37,9	88,0	83,3
15	Limestone	2822,0	36,4	82,0	85,6
16	Limestone	2730,0	36,0	74,0	76,2
17	Limestone	2720,0	36,0	71,0	74,8
18	Limestone	2887,0	35,0	72,0	70,5
19	Limestone	2699,0	39,0	81,0	83,6
20	Limestone	2679,0	37,0	76,0	73,4
Avarage		2738,4	35,1	71,3	71,6

Table 6 Miller's correlation 1972

Average data				
Miller's correlation, 1972:				
$UCS = \sigma_c = 12,83 * e^{0,0487*R_L}$				
Dry Density	2738,40	(kg/m3)		
R: rebound number	35,10	/		
UCS: Miller's correlation	71,30	(MPa)		
UCS:Obtained in Laboratory	71,64	(MPa)		

Table 7Miller's correlation 1965

Miller's correlation, 1965				
$Log10JCS = 0.00088*(\gamma)*(R) + 1.01$				
$JCS = 10^{\circ}$	0.00088*(γ)*(R) +	1.01		
γ 26,85 kN/m^3				
R 35,10 /				
JCS	69,10	MPa		

# 5 BARTON MODEL AND SHEAR STRENGTH OF ROCKFILL

One of the most peculiar aspect of the Bhatar system is the absence of mortar. This aspect is of great importance in the study of in plane behavior during an earthquake. The bhatar for its nature is already cracked. This means that micro displacements are possible. These micro movements must be considered as settlement. Micro slidings and displacements may be one of the reasons that allows the bhatar construction to dissipate energy.

From a safety engineering point of view in this thesis it has been studied the mechanism of resistance of the rock in the wall and the role of the absence of the mortar. This has meant to find a way to understand the behavior of rockfill. In order to do that the idea came reading the impressive work of BARTON, Nicholas R who studies the behavior of rock discontinuities in the field of Geotechnical engineering.

# 5.1 Interfaces between material: Timber-Stone and Stone-Stone

The behavior of the wall is strictly connected to the interfaces between the two main materials. The interaction stone-stone and timber-stone (see figure 5-1) is strongly related to the static frictional coefficients. The static frictional coefficient of the rocks is the most important for the aim of this work.



Figure 5-1 Stone layer (black box above) - Timber beam (black box below)

Due to the characteristics and dimensions of the rubble stones the behavior of the stone layers have been choosen as the peculiarity. The static frictional coefficient between the rocks is stongly higher than the static frictional coefficient between the stone and the timber. For this reason it has been made the hypothesis that the static frictional coefficient between the stone and the timber is negligible and the behavior of the wall in the layers where there are the timber bands has been studied using a reduction factor based on the areas of surfaces where the stones are in contact.

The reduction factor  $\xi$  has been computed as the ratio between the area of the section of the stones layer (Area) and the smaller area below the timber beam (Area\*).

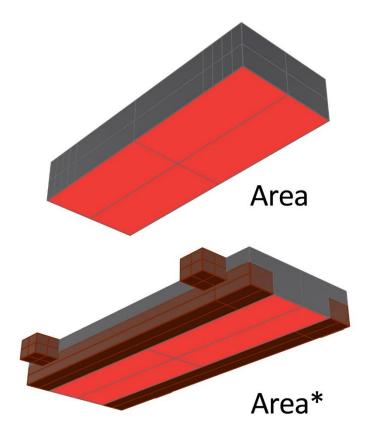


Figure 5-2 Contact surfaces

Table 8 Reduction factor due to the presence of the timber beam

Ratio between the areas
-------------------------

Contact Surface stones layer		
L module	1,2	m
Width	0,5	m
Area	0,6	m^2

Contact Surface below the timber beam		
L module	1,2	m
Width	0,3	m
Area*	0,3	m^2

Reduction factor $\xi = \text{Area/Area*}$	0,57
--	------

# 5.2 Shear strength of rock discontinuities

In the particular case of Bhatar it is necessary to evaluate the factors that control the shear strength of the discontinuities in a wall.

The following pages have the main intent to expose the principal theories and methods used in the analyses of stability for rock masses.

Starting from the Coulomb's law, it is shown how the behavior of a rock joint is described. Different authors defined their own methods to describe the rock joints behavior from more idealized scheme (linear) to more realistic scheme (non-linear).

The important aspect of the Barton's Method is the possibility to go from the rockjoint to the rockfill joint. The idea is to use the same approach of the rock masses analysis, with rock-fill joint, in the strength analysis of the in plane behavior of the wall.

# 5.3 Plane smooth joint

The first basic case is the most idealized one.

Hypothesis: plane and smooth joint surface

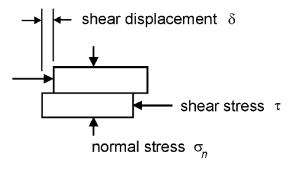


Figure 5-3 Plane and smooth joint surface

Observed mechanical behavior: shear stress quickly increases with deformation level, until a maximum value is reached; then, such value remains approximately constant.

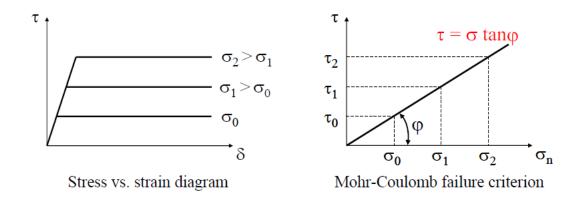


Figure 5-4 Stress vs Strain diagram and Mohr-Coulob failure criterion

- Linear friction model without cohesion:  $c^* = 0$
- Failure criterion (pure friction):  $\tau = \sigma_n * tan(\phi)$

Peak strength equal to residual strength

No dilatancy

# 5.4 Idealised rough joint (Patton, 1966)

Hypothesis: regular "saw-tooth" roughness (asperities with inclination i).

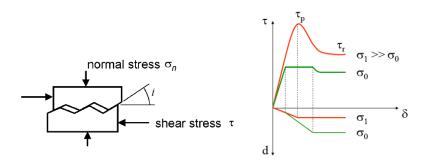


Figure 5-5 Rough joint surface

Observed mechanical behavior: shear stress quickly reaches a peak value. Then, increasing the deformation level, the shear stress stabilizes to a residual value.

#### **Dilatancy**

When a shear stress is applied on a rough surface joint, sliding occurs by climbing the asperities:

- to trigger a slide, it is at first required that the shear stress is capable to remove the embedding condition due to the asperities on the contact surface;
- the stress to apply is consequently higher than on a smooth surface.

The shear strength of the joint will consequently increase;

The material (rock) will expand

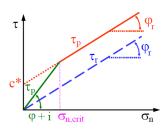
#### Sliding and dilatancy for low normal stresses

"Low" normal stresses:

- if the applied normal stress on remains below a critical value  $\sigma_{n,crit}$
- the upper rock block slides on the joint surface by climbing the asperity angle (in i direction)



- the peak strength during sliding  $\tau_p = \sigma_n * \tan(\phi + i)$
- the residual strength after sliding  $\tau_r = \sigma_n * \tan \phi_r$

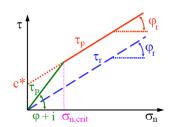


# "High" normal stresses:

- if the applied normal stress  $\sigma n$  is above the critical value  $\sigma n$ , crit
- the asperities are sheared and the upper rock block moves almost horizontally (no dilatancy)



- the peak strength before shearing  $\tau_p = \sigma_n * tan(\phi_r) + c *$
- the residual strength after shearing  $\tau_r = \sigma_n * \tan \phi_r$



« Low » normal stresses:

$$\sigma_n \leq \sigma_{n,criti}$$

$$\tau p = \sigma_n \tan (\phi + i)$$

- Friction angle  $(\phi + i)$
- Dilatancy d
- No cohesion



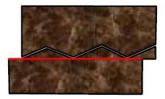
« High » normal stresses:

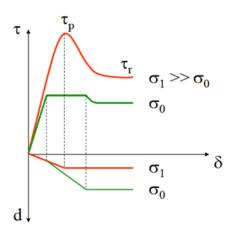
$$\sigma_n \geq \sigma_{n,criti}$$

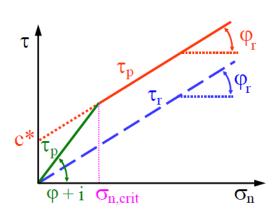
$$\tau_p = \sigma_n * tan(\phi_r) + c *$$

- Friction angle  $\phi_r$
- No dilatancy
- Cohesion *c* \*

with  $\sigma_{n,crit}$  the critical normal stress







Given:

 $\phi$  = friction angle on asperities surface

 $\phi_r$  = friction angle on the joint surface

it can be assumed:

$$\phi = \phi r$$

The residual strength after the shearing of the asperities is:

$$\tau_p = \sigma_n * tan(\phi_r)$$

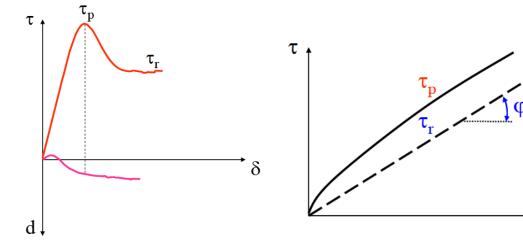
# 5.5 Real rough joint (Barton, 1973)

Hypothesis:

• the joint surface presents an irregular roughness (asperities with variable inclination i);

Observed mechanicalbehavior:

- progressive rupture of the asperities and some dilatancy
- The Mohr-Coulomb criterion is not fully applicable to describe the relation between shear strength and normal stress.



#### 5.5.1 Barton's failure criterion

Laboratory results obtained by means of a shear testing machine. The test is performed keeping a constant applied normal stress. The circles represent the peak value of the shear strength, while the crosses describe the residual strength level.

# 5.5.2 Barton's empirical model:

$$au_p = \sigma_n * \tan \left( JRC * log_{10} \left( \frac{JCS}{\sigma_n} \right) + \phi_r \right)$$

 $\tau p$  = peak shear strength

 $\sigma n$  = applied normal stress

JRC = Joint Roughness Coefficient

JCS = Joint wall Compressive Strength

 $\phi r$  = residual friction angle

#### 5.5.2.1 Joint Roughness Coefficient (JRC)

JRC is a number varying in the interval  $0 \div 20$  and represents the relevance of roughness in defining rocks' shear strength (smooth surfaces: JRC = 0; very rough surfaces: JRC = 20).

JRC can be estimated by:

- 1. comparing the real profile of the asperities with standard profiles:
  - « Barton comb » is used on site to reproduce the real roughness profile;



Figure 5-6 45-D0566/A Profilometer (Barton comb), 150 mm length. ControlsGroup.

• the obtained profile is compared with the standard profiles;

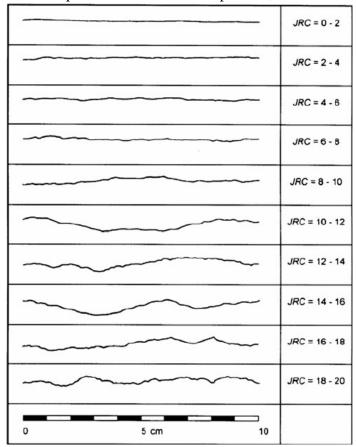


Figure 5-7 Roughness profiles and their corresponding JRC values (Barton and Choubey 1977)

• a value of JRC is assigned to evaluate the joint's roughness.

#### 2. performing a « tilt test »

• rock sample constituted by two parts separated by a joint;

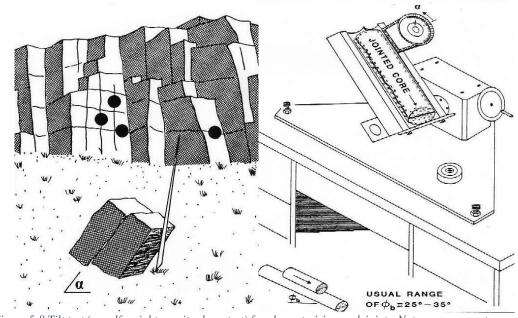


Figure 5-8 Tilt test (or self-weight gravity shear test) for characterizing rock joints. Note measurement

• the sample is placed on a plane, slowly tilted until sliding between the parts occurs;



Figure 5-9 Tilt Test apparatus

- the angle of inclination  $\alpha$  is measured;
- JRC is calculated by means of the equation:

$$JRC = (\alpha - \phi r) * \left(log_{10}\left(\frac{JCS}{\sigma_{n0}}\right)\right)^{-1}$$

where  $\sigma_{n0} = \gamma * h * cos^2(\alpha)$  is the normal stress in situ on a surface inclined by  $\alpha$ .

- 3. measuring length and amplitude of the asperity profile and using a graphic correlation with <u>JRC.</u>
  - the length of the asperity profile is measured;
  - the maximum amplitude of the asperity profile is measured;
  - a graphic correlation allows to determine the corresponding value of the Joint Roughness Coefficient (As shown below in figure 5-10).

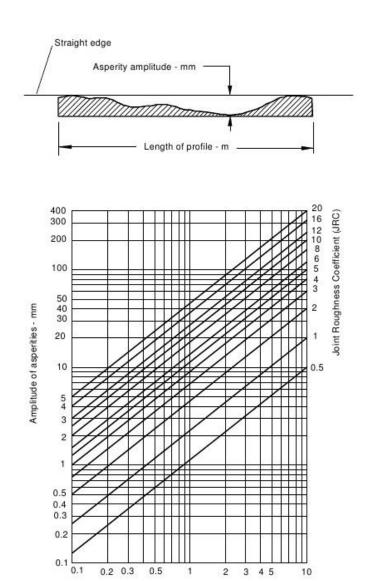


Figure 5-10 Alternative method for estimating JRC from Measuremens of surface roughness amplitude from a straight edge (Barton 1982).

#### 5.5.2.2 Joint wall Compressive Strength (JCS)

JCS represents the compressive strength of the joint, measured on the wall of the joint itself.

#### JCS can be estimated by:

1. comparing the alteration degree of the joint with the degree of alteration of the rock;

Length of profile - m

The degree of alteration of the joint is compared to the one of the rock. The value of JCS is then determined by means of a relation with the compressive strength of the intact rock.

Degree of alteration of the joint surface:

- - equal to rock:  $JCS = \sigma c$  (rock)
- - slighly higher than rock:  $JCS = 0.5 \sigma c$  (rock)
- - much higher than rock:  $JCS = 0.1 \sigma c$  (rock)

#### 2. performing on site measures with the Schmidt rebound hammer.

The Schmidt rebound hammer is used in field observations to evaluate the Joint Compressive Strength. Depending on the inclination of the hammer, the measure allows to know the Schmidt hardness. This parameter is combined with the unit weight of the rock to obtain the value of JCS.

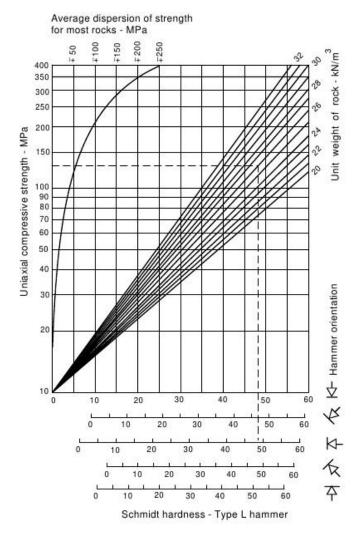


Figure 5-11 Estimate of joint wall compressive strength from Schmidt hardness

Barton's empirical model:

$$\tau_p = \sigma_n * \tan\left(\overline{JRC * log_{10}\left(\frac{JCS}{\sigma_n}\right)} + \phi_r\right)$$

- the first term in parentheses represents the dilation angle  $\delta$  (contribution of dilatancy to the shear strength)
- the more the joint surface is altered, the lower is the value of JRC and JCS and (as a consequence) of  $\tau_p$
- the less the joint's surfaces are embedded, the lower is the value of JRC (and  $\tau p$ )
- higher values of JRC give high dilation angles.

# 5.6 Shear Strength of Rockfill

The real contact stress levels are believed to be close to compressive failure where rock joint asperities and rockfill stones are in contact (e.g. Figure 5-12 for the case of rock joints). Therefore it is perhaps possible to use a common form of constitutive equation for extrapolating the strength measured at very low (index test) normal stress levels, to stress levels of engineering interest, as inside a large rockfill dam, inside a rock dump or under a rock slope formed of jointed rock.

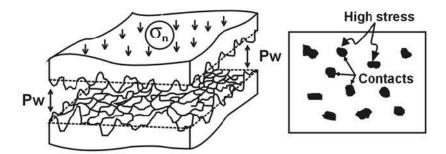


Figure 5-12 When peak shear strength is approached (joints and rockfill), the actual rock-to-rock contact stress levels are extremely high, due to small contact areas.

It is believed that the real ratios of  $\sigma cn$  /JCS (contact normal stress/joint wall compressive strength, in the case of rock joints) and  $\sigma cn$  /S (contact normal stress/particle strength,in the case of rockfill) are equal to the ratio A0 / A1 representing the ratio of true contact area/assumed contact area.

The terms JCS and S represent the joint compressive strength and the particle strength, respectively. In other words, contact area is a rock strength or particle strength regulated phenomenon at peak strength. Tilt tests are performed on a regular basis to characterise the roughness of rock joints.

The equation for back-calculating the effective roughness (R) of rockfill particles is shown in Figure 8 (diagram 5). Exactly the same format is used to back-calculate the joint roughness coefficient (JRC) for rock joints:

$$JRC = \frac{(\alpha_0 - \varphi_r)}{\log\left(\frac{JCS}{\sigma no}\right)} \tag{1}$$

where  $\sigma$ no represents the very low normal stress acting when sliding occurs between the two halves of a mating rock joint, at tilt angle  $\alpha$ o. In the case of tilt tests on laboratory-scale joint samples, the normal stress is often as low as 0.001 MPa

A schematic example of tilt testing for rock joints has been explained before, while a suggested method for testing rockfill at full scale (without needing parallel grading curves) is shown in Figure 5-13, from Barton and Kjærnsli (1981).

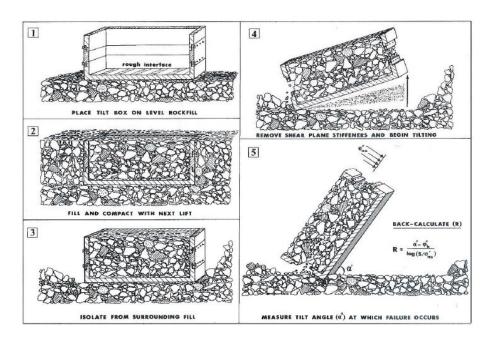


Figure 5-13 Illustration of the tilt test principle for rockfill (Barton and Kjærnsli, 1981)

# 5.6.1 The shear strength of rockfill as measured

Leps (1970) is responsible for assembling a significant number of large-scale triaxial shear test data for rockfills of various types.

The interpreted peak effective friction angles as a function of the estimated effective normal stress are shown in Figure 5-14 a.

We can 'fit' familiar values of JRC and JCS for rock joints (Figure 9b) that closely match the stress-dependent friction angles that (also) describe the shear strength of rockfills.

Mid-range JRC values (to correspond to an R-range of about 5 to 10, and low-to-high range JCS values (to correspond to an S-range of about 10 to 100 MPa) generated by medium weak to medium strong rock are seen to fit the test data.

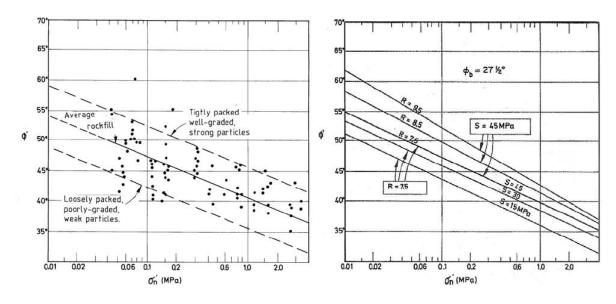


Figure 5-14 Leps ( 1970)

Left: Assembly of peak shear strength data for rockfills, from Leps (1970).

Right: Comparative JRC or R, and JCS or S values used to generate similar gradients to Leps 1970 data for rockfill. R = 5 to 10, and S = 10 to 100 MPa appear to cover the range of strengths assembled by Leps.

Less compacted rock dump materials will tend to have lower 'R-values' than the 'tightly-packed' particles, since there will generally be less interlocking.

The more conventionally plotted shear stress versus effective stress curves for rockfill, shown in Figure 5-15 from Marsal (1973), also confirm the similarities of the peak shear strength of rock joints and rockfill.

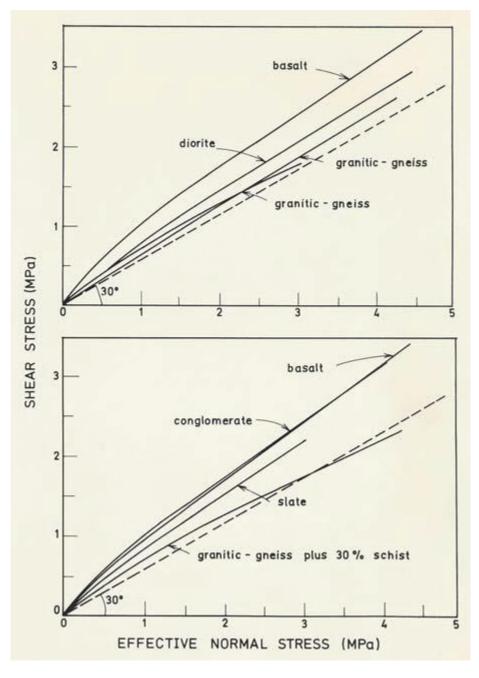


Figure 5-15 The peak shear strength envelopes for rockfill have remarkable similarity to those for medium rough, medium strength rock joints. Large-scale test data from Marsal (1973)

The large scale measurement of frictional strength of rock dump materials obtained from mines in the Chilean Andes shown in Figure 5-16 tend to further reinforce the idea of non-linear stress-dependent friction angles that are likely to apply to rock dumps in general (priv. comm., Sandra Linero, SRK).

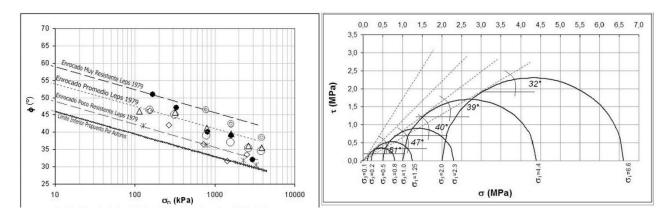


Figure 5-16 Large rock dumps are a familiar feature of mines in the Chilean Andes. Large-scale triaxial shear tests performed in Chile, with important results (black dots and Mohr circles) showing non-linear stress- ependent friction angles (Linero and Palma 2006)

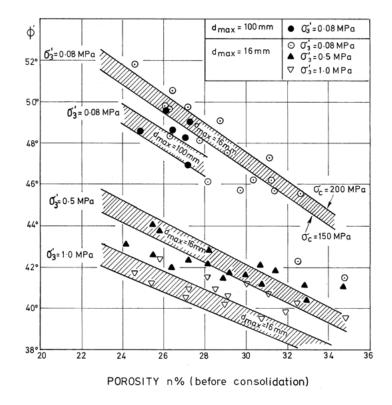


Figure 5-17 The same non-linearity with effective stress level is seen in large-scale triaxial tests performed at NGI (Strøm, 1974, 1975, 1978), with particle size-dependence, rock strength dependence, and porosity effects also indicated

For comparison, Figure 5-18 shows shear strength envelopes for rock joints that have been generated with the JRC-JCS model introduced in Figure 5-10. The strongly varying peak dilation angles, part of the reason for the non-linearity, are also shown on each envelope, except at lowest stress, where they may exceed  $30^{\circ}$ .

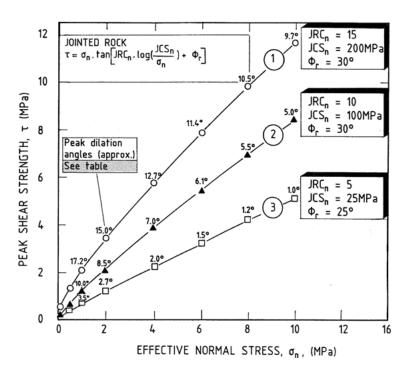


Figure 5-18 Shear strength envelopes (and peak dilation angles) predicted for rock joints, using the JRC-JCS non-linear model of Figure 5-10. Rockfill generally lies between curves #2 and #3

#### 5.6.2 Estimating the shear strength of rockfill

As emphasised in all reports of rockfill shear strength, including Barton and Kjærnsli (1981), the degree of compaction and porosity achieved when building a dam or when preparing relevant laboratory samples is all important. The particle roughness and smoothness is also fundamental. Figure 5-19 illustrates an empirical scheme developed by the writer, for estimating the likely R-value for rockfills, whether for rounded gravels or for rough quarried rock. The high (relatively uncompacted) porosities in mining rock dumps clearly places such dumps in the middle-to right-hand areas of this diagram, and even sharp angular particles (relevant for waste rock, but perhaps not always for tailings) are unlikely to generate 'R-values' above 5 to 7, as also suggested in Figure 5-14.

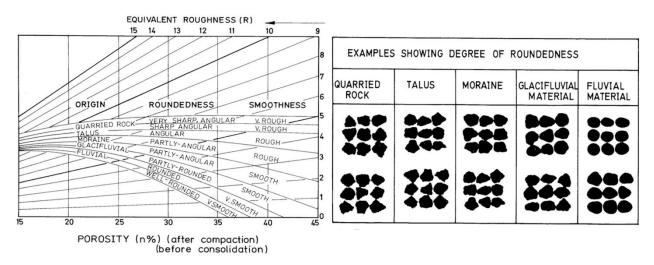


Figure 5-19 An empirical method for estimating the equivalent roughness R of rockfill as a function of porosity and particle origin, roundedness and smoothness. Barton and Kjærnsli (1981)

As a result of the literature survey of numerous rockfill test data, Barton, 1980 and Barton and Kjærnsli, 1981 developed a simple strength factoring scheme for estimating S as a function of UCS (or  $\sigma c$ ), when particle size (d50) varied over a wide range. The points A and B in Figure 15 were used to illustrate S-value estimation for a rock with UCS = 150 MPa, when d50 was 23 mm (S  $\approx$  0.3x150 = 50 MPa) and when d50 was 240 mm (S  $\approx$  0.2x150 = 30 MPa), in the case of interpreting triaxial strength data. Note the higher factors apparently needed when planar (and large-scale) shear is involved. Friction angles are typically several degrees higher (e.g. about 2° to 4°) when plane tests are compared with triaxial tests on the same material. There is noticeably less crushing of particles: hence the two empirical curves in Figure 5-20.

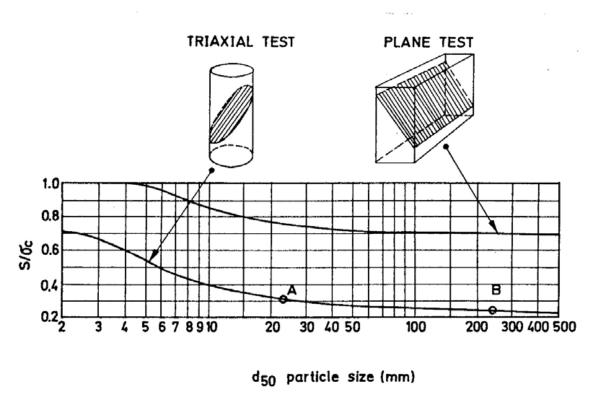


Figure 5-20 Particle size strongly effects the strength of contacts points in rockfill. Triaxial or plane shear also influences behavior.

Empirical S/UCS reduction factors for estimating S when evaluating equation 3.

#### **5.6.3** Interface shear strength

Interface shear strength, as between a (too smooth) rock foundation and a rockfill dam, seems to be governed by the 'weakest link' rule. If the roughness JRC of the interface, registered by amplitude/length profiling, is too low in relation to particle size (d50), the interface strength is controlled by JRC, and sliding occurs along the interface, as along the bottom face of a rock joint. If on the other hand, the interface roughness is sufficient to give good interlock to the rockfill particles, sliding will occur preferentially within the rockfill, in an 'R-controlled' particle smoothness or roughness dependent manner, with influence also of the porosity. A schematic illustration of the interface problem, and (probable) relevant controlling parameters is shown in Figure 5-21.

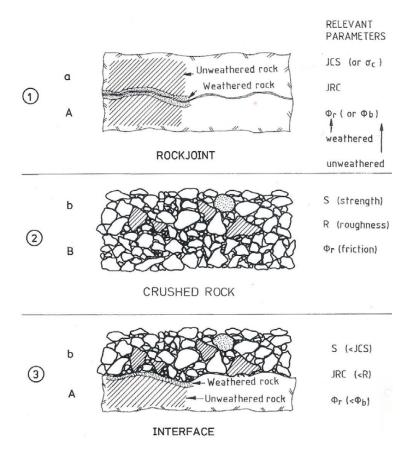


Figure 5-21 Asperity contact across stressed rock joints, and rockfill inter-particle contact, and rockfill lying on a rock foundation.

Asperity contact across stressed rock joints, and rockfill inter-particle contact, and rockfill lying on a rock foundation, are each examples of point-contact stress levels that are probably close to compressive failure, when peak shear strength is approached. For this reason the three cases have many points in common, including similar non-linear shear strength envelopes.

The peak shear strengths for rock joints, rockfill and interfaces are respectively:

Rock joints:

$$\tau_p = \sigma_n * \tan \left( JRC * \log_{10} \left( \frac{JCS}{\sigma_n} \right) + \phi_r \right)$$
 (2)

Rockfill:

$$\tau_p = \sigma_n * \tan\left(R * \log_{10}\left(\frac{S}{\sigma_n}\right) + \phi_r\right)$$
 (3)

Interface:

$$\tau_p = \sigma_n * \tan \left( JRC * \log_{10} \left( \frac{S}{\sigma_n} \right) + \phi_r \right) \tag{4}$$

If the rockfill particles are not weaker than the rock foundation, as assumed in equation 4, then S > JCS, and the strength is determined by the weak foundation. In the case of rockfill or waste rock

that is freshly blasted, the residual friction angle  $\phi_r$  assumed, can (initially) be replaced by  $\phi_b$ , which is usually a few degrees higher than the weathered value. Conservative, long-term design strength may nevertheless demand the use of  $\phi_r$  for 'permanent' rock dumps and rockfill dams, as suggested in all three equations.

#### 5.6.4 R-controlled or JRC-controlled behavior

As indicated above, the relative magnitudes of the interface parameters, and their possible contrast to the shear strength of the rockfill, will determine whether the interface (if very rough) causes 'R-controlled'behavior – meaning preferential failure through the rockfill, or 'JRC-controlled'behavior, meaning preferential shear along the interface. A review of interface tests, performed by Barton (1980) in response to doubts about the strength of a glacially-smoothed dam foundation in Norway, resulted in the separation of performance identified in Figure 5-22.

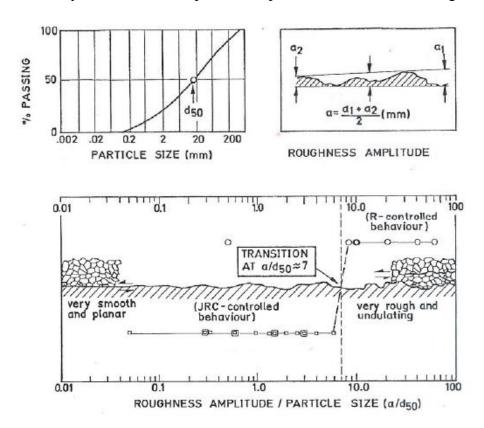


Figure 5-22 A review of interface shear tests was performed in response to concern over insufficient roughness for the rockfill dam foundation, in the glaciated mountain terrain in Norway.

# 5.7 Barton model applied on Bhatar system

The Barton model have been studied in two different configurations:

- Rockjoint
- Rockfill

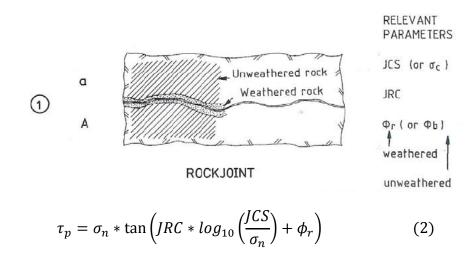
In order to applied the Barton model and its equations they have been used the results about vertical normal stresses obtained in the static analysis which will be explained in the following chapters.

#### 5.7.1 Rockjoint

In order to study the case of rockjoint interfaces of Bhatar construction we must refer to Case 1 of Figure 5-21 described by equation (2).

In this case we consider an ideal plane as a continuous joint. The whole interaction surface can be classified with the Barton parameters which are usually used to described a rock joint.

Different samples of stone may be tested and then It can be evaluated an average of the tanget:  $tan\left(JRC * log_{10}\left(\frac{JCS}{\sigma_n}\right) + \phi_r\right)$  which, for the Bhatar thesis, is the coefficient of friction.



where

 $\tau_p$  = peak shear strength

 $\sigma_n$  = applied normal stress

JRC = Joint Roughness Coefficient

JCS = Joint wall Compressive Strength

 $\phi_r$  = residual friction angle

#### 5.7.2 Rockfill

Recalling what has been written until here, in order to study the case of rockfill interfaces of Bhatar construction we must refer to Case 2 of Figure 5-21 described by equation (3):

$$\tau_p = \sigma_n * \tan\left(R * \log_{10}\left(\frac{S}{\sigma_n}\right) + \phi_r\right)$$
 (3)

where

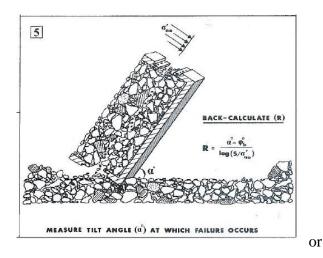
 $\tau p$  = peak shear strength  $\sigma n$  = applied normal stress R = Roughness

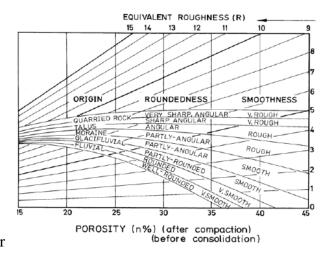
$$S = Strength$$
  
 $\phi r = residual friction angle$ 

In the case of rockfill or waste rock that is freshly blasted, the residual friction angle  $\varphi$ r assumed, can (initially) be replaced by  $\varphi$ b, which is usually a few degrees higher than the weathered value.

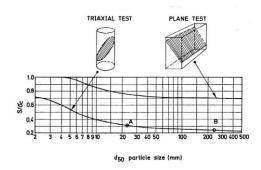
In order to estimate R:

$$R = \frac{\alpha^0 - \varphi_b}{\log_{10}\left(\frac{S}{\sigma_n}\right)}$$





In order to estimate S:



# 5.7.3 Voids ratio and Porosity

The *porosity* of the soil is the percent of void space.

$$n = 100 * \left(\frac{V_v}{V}\right)$$

Where

*n* is porosity (percentage)

 $V_{\!\scriptscriptstyle \mathcal{V}}$  is volume of the void space (L^3 ; cm^3 ; m^3 )

V is volume of the sample (L^3; cm^3; m^3)

The *void ratio* of the soil is the ratio of the volume of the voids to the volume of the solids.

$$e = \left(\frac{V_v}{V_s}\right)$$

Where

e is void ratio (percentage)

 $V_s$  is volume of the solids (L^3; cm^3; m^3)

The total volume is equal to the volume of the voids plus the volume of the solids.

$$V = V_{\nu} + V_{s}$$

The void ratio is closely related to the porosity if porosity is expressed as a ratio.

$$n = \left(\frac{e}{1+e}\right)$$
 and  $e = \left(\frac{n}{1-n}\right)$ 

# 5.7.4 Limestone Mechanical Properties for application of Barton model

In accordance to the previous chapter, they have been used the results of laboratory tests with the Schmidt hammer.

Table : Rock characterization results							
No. Sample Type (kg/m3) L	e Dry Density (kg/m3)	R: rebou number	and UCS: Miller`s correlation(MPa)	UCS:Obtained in Laboratory(MPa			
1 Limestone	2817,0	36,0	72,0	72,9			
2Limestone	2748,0	35,9	76,0	72,9			
3 Limestone	2646,0	31,5	55,0	58,5			
4Limestone	2777,0	31,5	60,0	60,6			
5 Limestone	2671,0	28,9	49,0	52,2			
6Limestone	2773,0	30,4	56,0	56,4			
7Limestone	2676,0	37,7	79,0	76,7			
8 Limestone	2683,0	36,8	76,0	75,7			
9 Limestone	2748,0	34,8	71,0	72,5			
10 Limestone	2707,0	35,6	72,0	69,6			
11 Limestone	2759,0	36,6	79,0	78,1			
12 Limestone	2704,0	33,9	66,0	63,5			
13 Limestone	2726,0	35,1	71,0	75,7			
14Limestone	2796,0	37,9	88,0	83,3			
15 Limestone	2822,0	36,4	82,0	85,6			
16Limestone	2730,0	36,0	74,0	76,2			
17Limestone	2720,0	36,0	71,0	74,8			
18Limestone	2887,0	35,0	72,0	70,5			
19Limestone	2699,0	39,0	81,0	83,6			
20 Limestone	2679,0	37,0	76,0	73,4			
Avarage	2738,4	35,1	71,3	71,6			

Average data							
Miller's correlation, 1972:							
$UCS = \sigma_c = 12,83 * e^{0.0487 * R_L}$							
Dry Density	2738,40	(kg/m3)					
Dry Density R: rebound number	2738,40 35,10	(kg/m3)					
or and the contraction of a		(kg/m3) / (MPa)					

Miller's correlation, 1965					
$Log10 JCS = 0.00088 * (\gamma) * (R) + 1.01$					
$JCS = 10^{0.00088*(\gamma)*(R) + 1.01}$					
γ 26,85 kN/m^3					
R 35,10 /					
JCS	69,10	MPa			

## 5.7.5 Rockjoint results

$$\tau_p = \sigma_n * \tan \left( JRC * log_{10} \left( \frac{JCS}{\sigma_n} \right) + \phi_r \right)$$
(2)

where

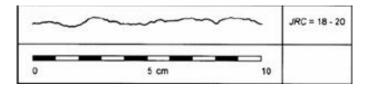
 $\tau p$  = peak shear strength

 $\sigma n$  = applied normal stress

JRC = Joint Roughness Coefficient

JCS = Joint wall Compressive Strength

 $\phi r$  = residual friction angle



#### **JCS**

Comparing the alteration degree of the joint with the degree of alteration of the rock; The degree of alteration of the joint is compared to the one of the rock. The value of JCS is then determined by means of a relation with the compressive strength of the intact rock. Degree of alteration of the joint surface:

- equal to rock:  $JCS = \sigma c$  (rock)
- slighly higher than rock:  $JCS = 0.5 \sigma c$  (rock)
- much higher than rock:  $JCS = 0.1 \sigma c$  (rock)

Table 9 Rockjoint data

Rockjoint					
Origin	Quarried rock				
Asperties	maximum				
JRC	20				
бс (from lab tests)	71,6	MPa			
JCS (Miller 1965)	69,1	MPa			
JCS comparison	71,6	MPa			
фr°	30	deg			

$$\tau_p = \sigma_n \tan\left(20\log\left(\frac{69.1}{\sigma_n}\right)\right)$$

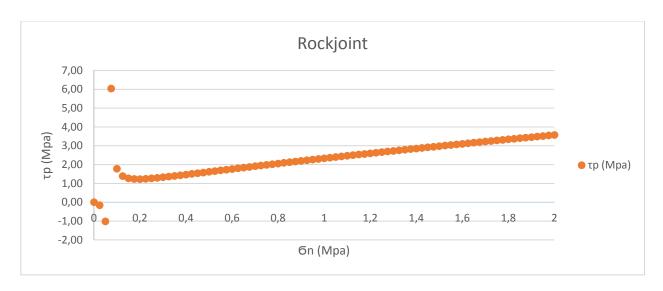


Figure 5-23 Rockjoint function for Bhatar

Table 10 Barton method for Rockjoint Bhatar results

Rockjoint							
бп (Mpa)		тр (Мра)	μ= τρ/ση	rad	deg		
	0,036	-0,354468	-9,939	-1,47052	-84,255		
	0,045	-0,671098	-15,046	-1,50443	-86,198		
	0,056	-1,817527	-32,267	-1,53981	-88,225		
	0,068	-29,4048	-432,086	-1,56848	-89,867		
	0,080	3,661616	45,897	1,549012	88,752		
	0,090	2,264381	25,235	1,53119	87,731		
	0,093	2,076058	22,379	1,52614	87,441		

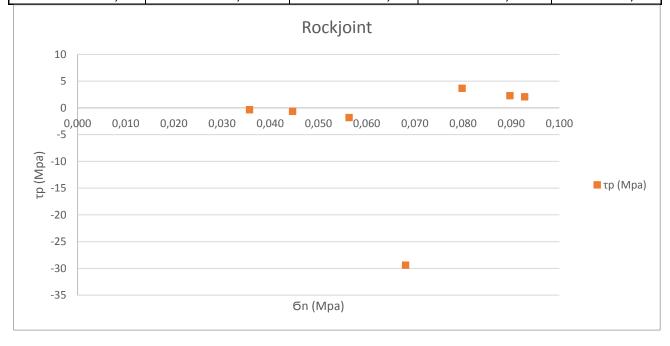


Figure 5-24 Rockjoint function for Bhatar range of interest

Asintoticbehavior, unreliable for our range of normal stress values.

#### 5.7.6 Rockfill results

$$\tau_p = \sigma_n * \tan\left(R * \log_{10}\left(\frac{S}{\sigma_n}\right) + \phi_r\right)$$

where

 $\tau p$  = peak shear strength

 $\sigma n$  = applied normal stress

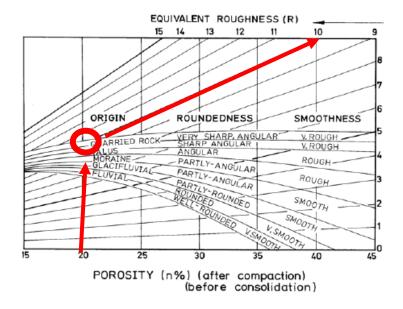
R = Roughness

S = Strength

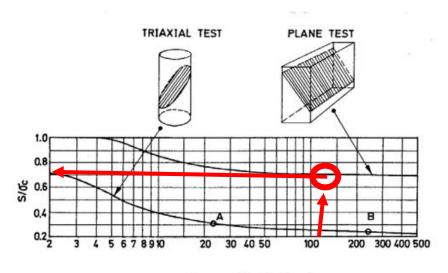
 $\phi r$  = residual friction angle

In the case of rockfill or waste rock that is freshly blasted, the residual friction angle  $\varphi$ r assumed, can (initially) be replaced by  $\varphi$ b, which is usually a few degrees higher than the weathered value.

#### In order to estimate R:



#### In order to estimate S:

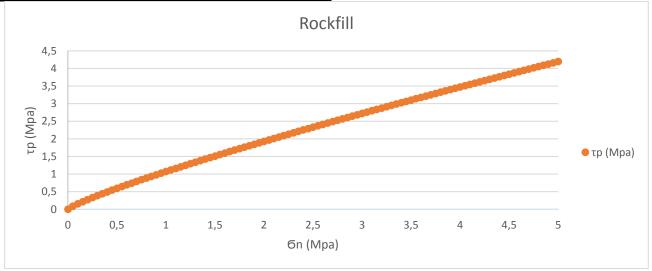


d<sub>50</sub> particle size (mm)

Table 11 Rockfill data

Rockfill					
n: porosity (%)	20				
Origin	Quarried rock				
R : Roughness =	10				
d50 Particle size	> 100 mm				
бс (from lab tests)	71,6	MPa			
S/бс	0,7				
S : Strength =	50,12	MPa			
φr °	30	deg			

$$\tau_p = \sigma_n \tan\left(10\log\left(\frac{50,12}{\sigma_n}\right)\right)$$



 $Figure \ 5\text{-}25 \ \textit{Rockfill function for Bhatar}$ 

#### Reliable for our range of normal stress values.

 $Table\ 12\ Barton\ method\ for\ Rockfill\ Bhatar\ results$ 

Normal Stress			Rockfill					
	KN/m^2	Kg/m^2	Kg/cm^2	бп (Mpa)	тр (Мра)	μ= τρ/ση	rad	deg
σ1	34,99	3566,30	0,36	0,036	0,066	1,840	1,073	61,478
σ2	43,75	4460,24	0,45	0,045	0,079	1,768	1,056	60,507
σ3	55,26	5632,78	0,56	0,056	0,096	1,697	1,038	59,493
σ4	66,76	6805,31	0,68	0,068	0,112	1,643	1,024	58,672
σ5	78,26	7977,84	0,80	0,080	0,128	1,599	1,012	57,981
σ6	88,03	8973,09	0,90	0,090	0,141	1,568	1,003	57,471
σ ground	91,01	9277,01	0,93	0,093	0,145	1,559	1,001	57,326

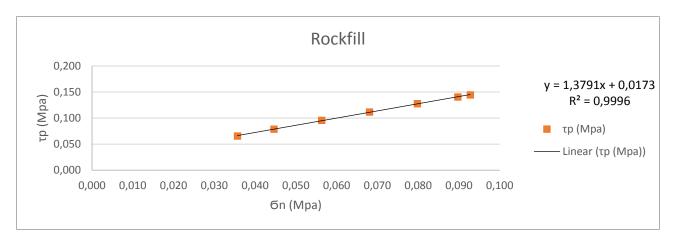


Figure 5-26 Rockfill function for Bhatar range of interest

Reliable for our range of normal stress values.

#### 5.8 Conclusions

The behavior has been descripted by using two different version of Barton's method:

- Barton's method for Rock-joint
- Barton's method for Rock-fill

In the first case the resulting formula is not useful due to the fact that in the range of our interest the equation shows an asymptote which distorts the reliability of the results.

In the second case, with the evaluation of the rock-fill, the behavior of the joint is properly described in the range of our interest and it seems to be correct.

This approach must be verified by proper laboratory test on the different samples or if possible by the use of large scale module as shown in Barton's in situ tests.

Other kind of stones which are used in Nepal or different regions are:

- Dolomite
- Slates
- Sandstone
- Ouartzite

The Barton's method for Rock-fill is one of the peculiar aspect of this thesis and it has been fundamental for the analysis of the in plane behavior of the Bhatar wall system.

# 6 TIMBER ELEMENTS AND CARPENTRY CONNECTIONS

## **6.1** Geometry of Timber elements

In accordance to the guidelines given in the Arch. Tom Schacher's manual they have been defined and studied all the timber elements Roof rafter beams, rafter beam and the cross piece. The modular unit is just an ideal module which allows us to study the static behavior. All the walls and the room box are built layer after layer with a vertical continuity from the plinth to the roof.



Figure 6-1 Continuous Bhatar wall



Figure 6-2 Carpentery connections

## **6.1.1 Rafter**

The rafter is the most common timber element which compose the all structure and it is laid down on the stones layer parallel to the ground.

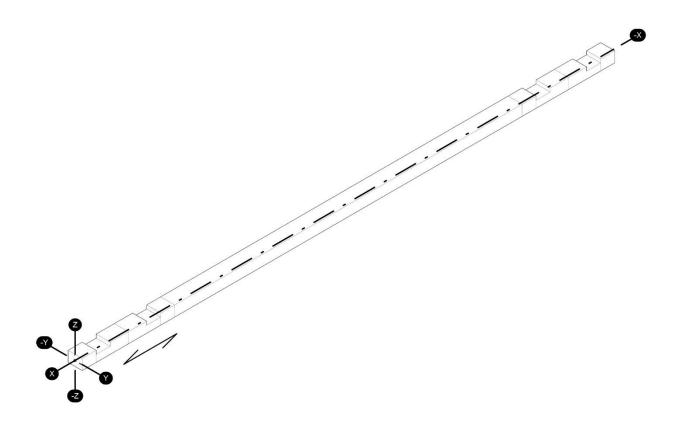


Figure 6-3 Rafter beam

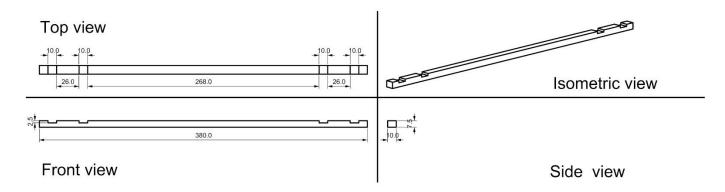
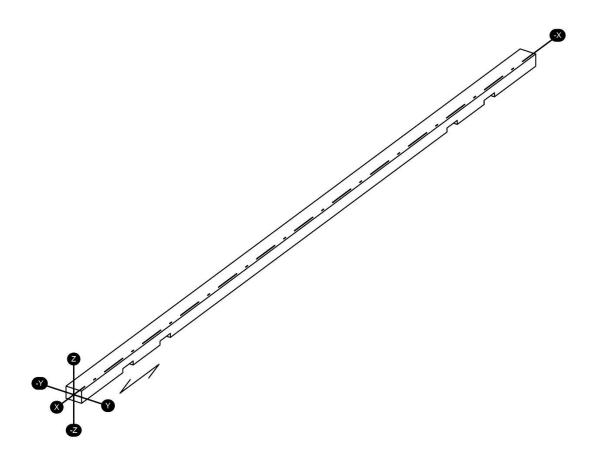


Figure 6-4 Rafter beam Orthogonal projections in cm

#### 6.1.2 Roof rafter

The roof rafter beam is used just at the roof level and the difference with respect to the rafter beam is just the lengths of the two ends. The extremities are longer in order to support the heavy flat roof of earth.



Figure~6--5~Roof~rafter~beam

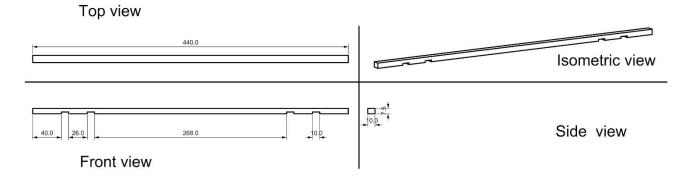


Figure 6-6 Roof rafter beam Orthogonal projections in cm

## 6.1.3 Cross piece

The cross pieces are the elements which assure stability. Cross pieces help to hold the beams and walls together. You need notches only on the cross pieces, but not on the main beams.

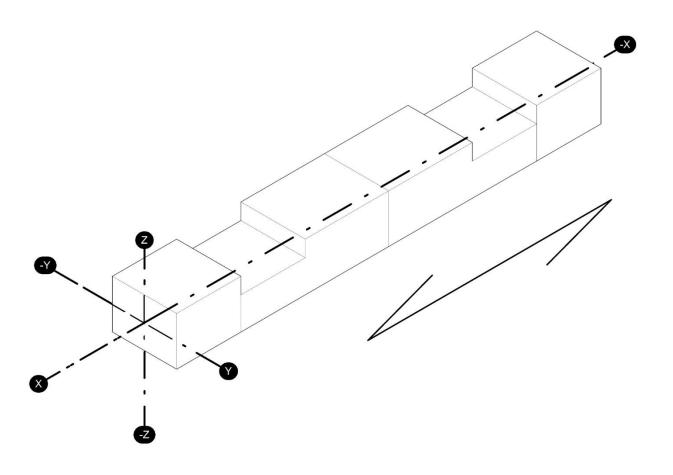


Figure 6-7Cross piece

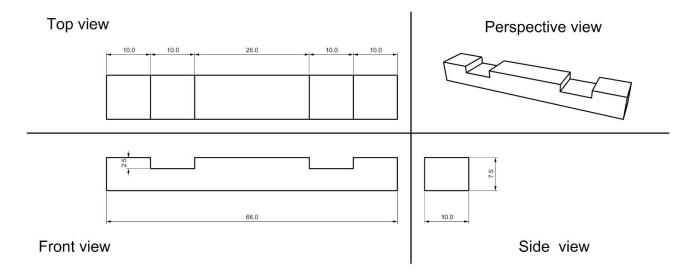


Figure 6-8 Cross Piece Orthogonal projections in cm

# 6.2 Assembling

Assembled timber band composed by Rafters and cross pieces

#### 6.2.1 Timber Band

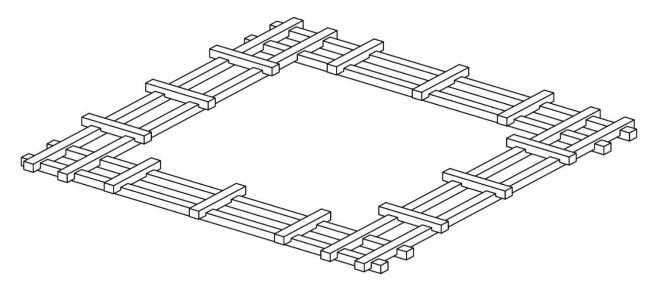


Figure 6-9 Timber Band

## 6.2.1.1 Rafter exploded

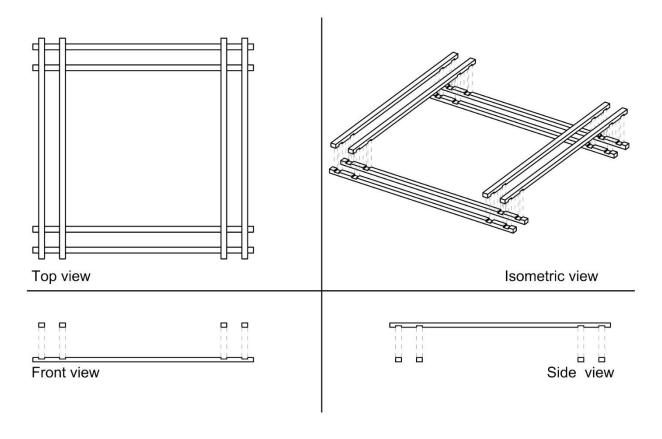


Figure 6-10 Timber band Rafter exploded

## 6.2.1.2 Cross pieces exploded

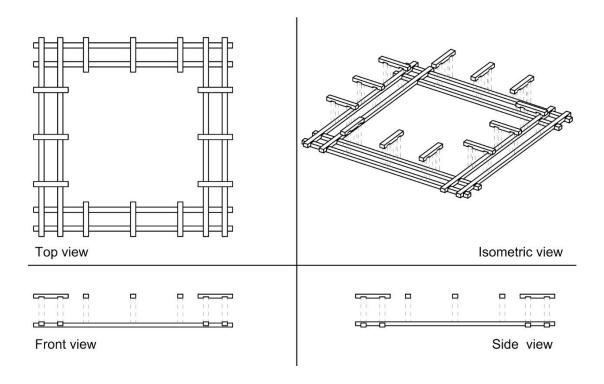


Figure 6-11 Timber band Cross pieces exploded

#### 6.2.1.3 All exploded

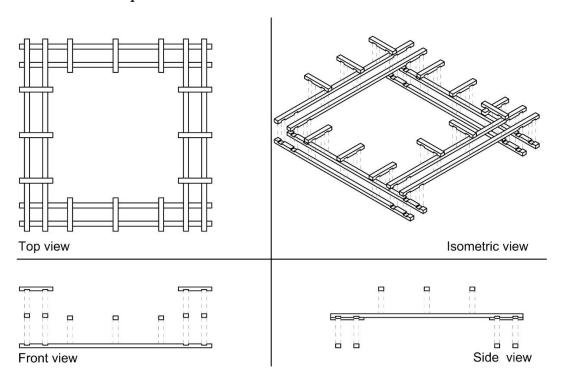


Figure 6-12 Timber band All exploded

# 6.2.2 Roof Timber Band

Assembled timber band composed by Roof rafters, Rafters and cross pieces.

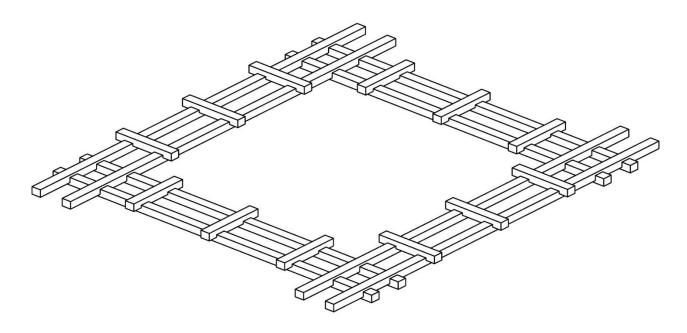


Figure 6-13 Roof Timber band

#### 6.2.2.1 Roof Rafter explosed

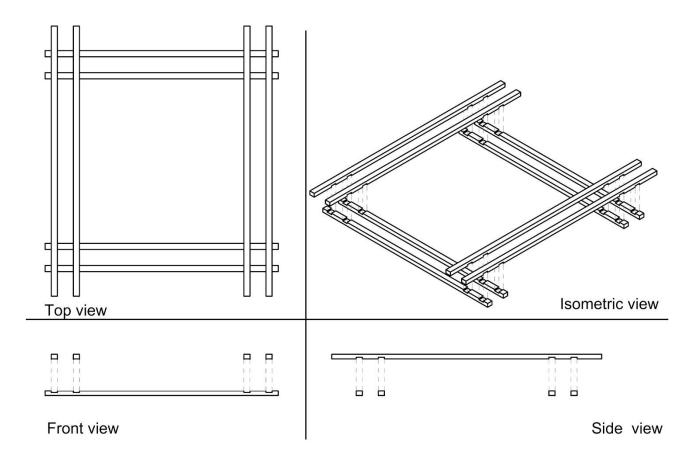


Figure 6-14Roof timber band Roof rafter exploded

## 6.2.2.2 Cross pieces explosed

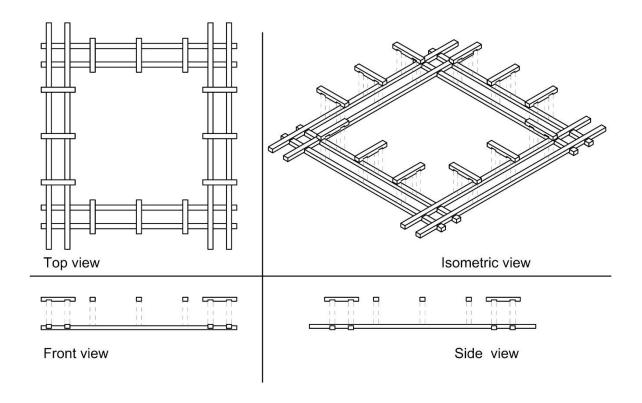
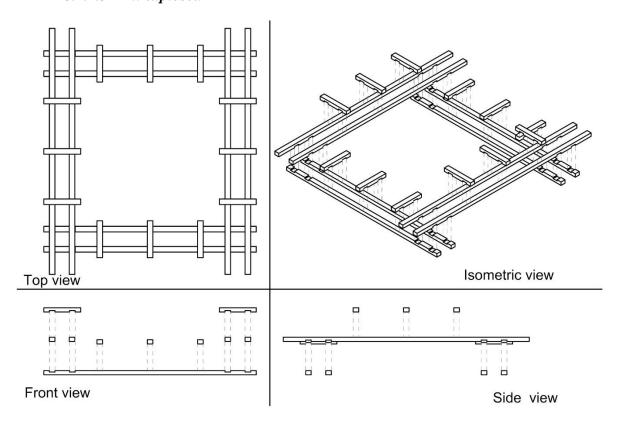


Figure 6-15 Roof timber band Cross pieces exploded

# 6.2.2.3 All explosed

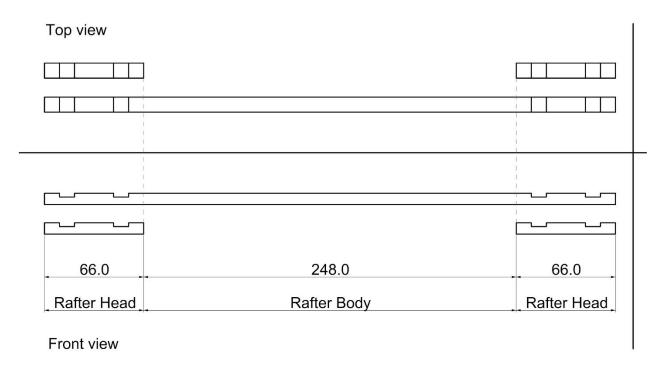


 $Figure\ 6\text{-}16\ Roof\ Timber\ band\ All\ \ exploded$ 

#### **6.3** Portions of Rafter and Roof rafter

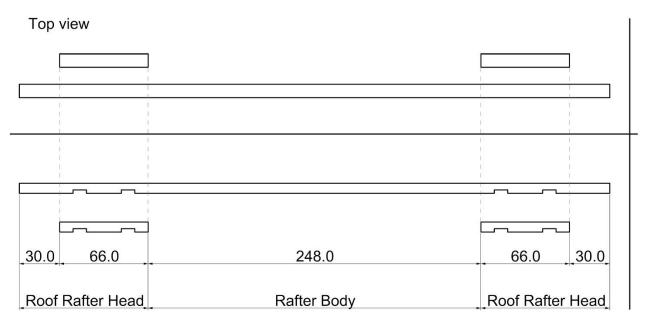
The rafters are composed by a central part which has been named rafter body and two ends which are exactly the same as the cross pieces. For the Roof rafter the heads are longer.

## 6.3.1 Rafter Head + Rafter Body + Rafter Head



 $Figure\ 6\text{-}17\ 6.3.1\quad Rafter\ Head+Rafter\ Body+Rafter\ Head$ 

#### 6.3.2 Roof Rafter Head + Rafter Body + Roof Rafter Head



Front view

 $\textit{Figure 6-18 6.3.2} \quad \textit{Roof Rafter Head} + \textit{Rafter Body} + \textit{Roof Rafter Head}$ 

## **6.3.3** Subdivisions of the timber elements

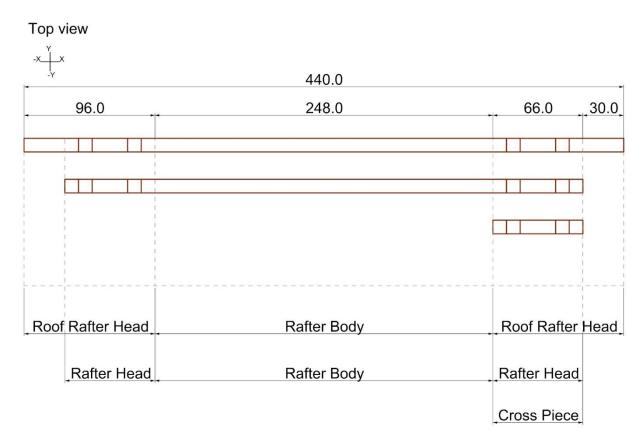


Figure 6-19 Subdivisions of the timber elements

## 6.4 Area under stresses

## 6.4.1 Cross Piece

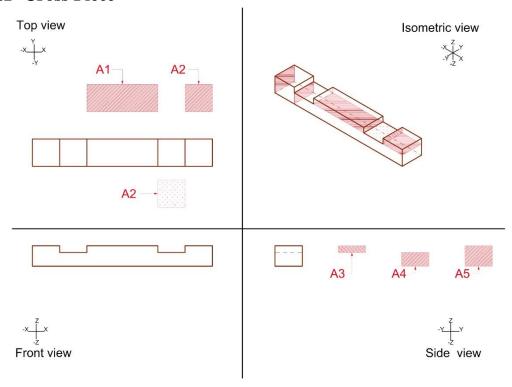


Figure 6-20 Area under stresses - Cross piece

## **6.4.1 Rafter**

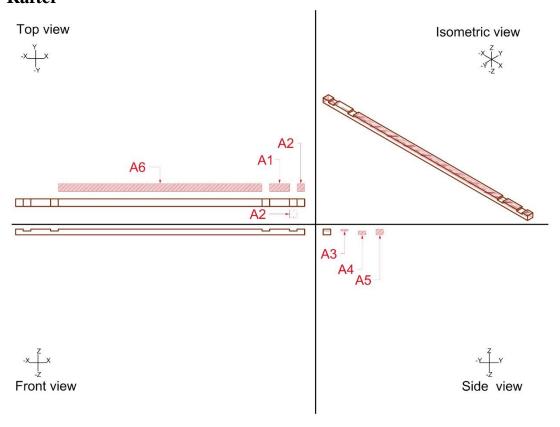


Figure 6-21 Area under stresses - Rafter

## 6.4.2 Roof Rafter

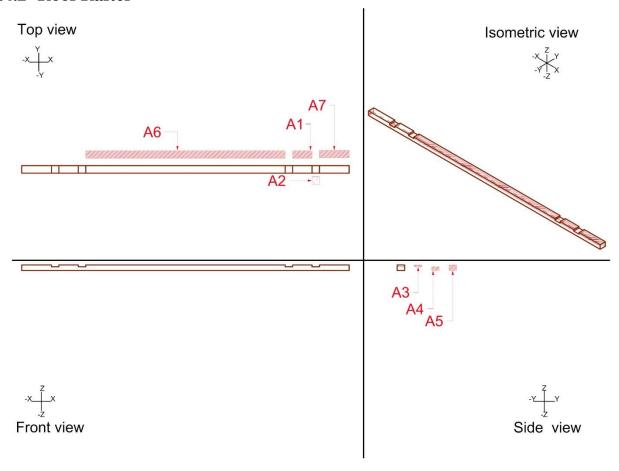


Figure 6-22 Area under stresses - Roof Rafter

# **6.4.3** Measures for area under stresses

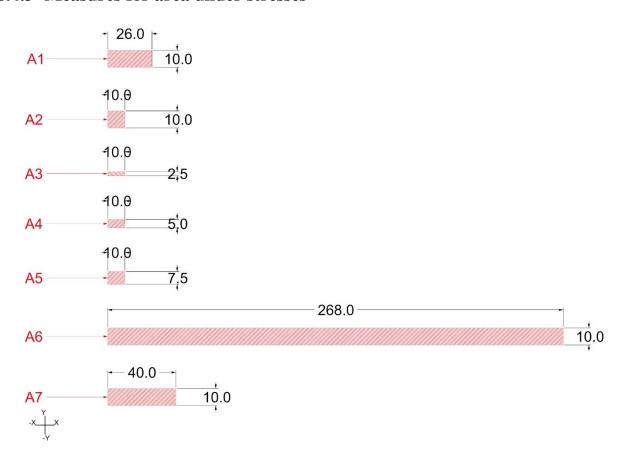


Figure 6-23 All areas under stresses

Table 13 Measures for all areas under stresses

	b	h	AREA		
	cm	cm	cm^2	m^2	mm^2
A1	26	10	260	0,026	26000
A2	10	10	100	0,01	10000
A3	10	2,5	25	0,0025	2500
A4	10	5	50	0,005	5000
A5	10	7,5	75	0,0075	7500
A6	268	10	2680	0,268	268000
A7	40	10	400	0,04	40000

## **6.5** Saint Venant for Timber elements

The distribution of the stresses in the section is mainly the same for the two roof rafter and for the normal rafter, the real different is in the notch of the cross piece as well in the rafter heads and in the roof rafter head.

#### **6.5.1** Rafter

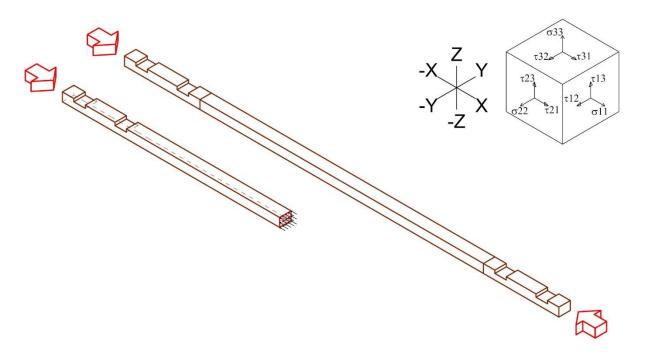


Figure 6-24 Rafter -Compression along X axis

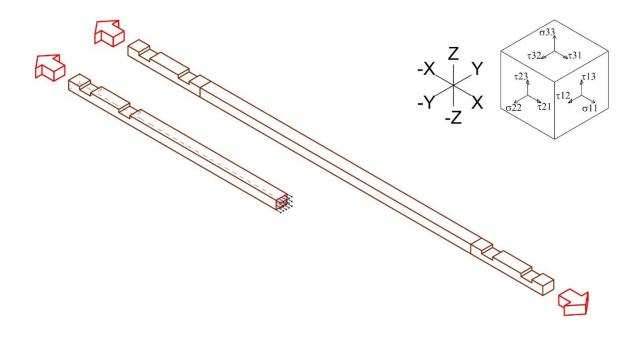


Figure 6-25 Rafter -Tension along X axis

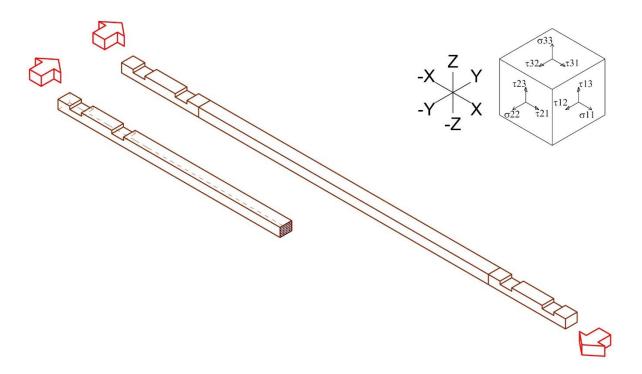


Figure 6-26 Rafter -Shear on Y axis

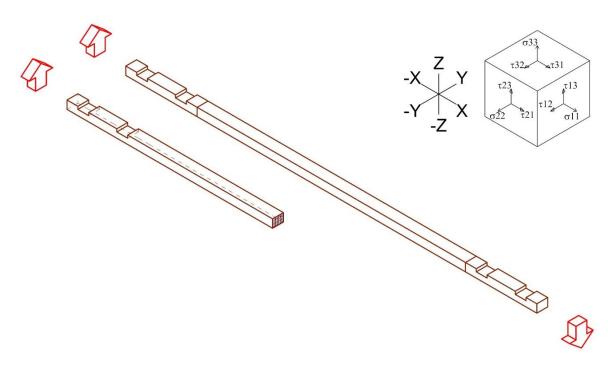


Figure 6-27 Rafter -Shear on Z axis

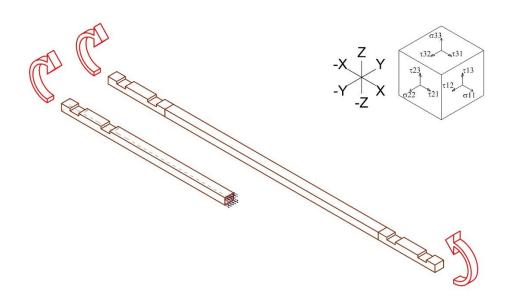


Figure 6-28 Rafter - Bending Moment My on Y axis

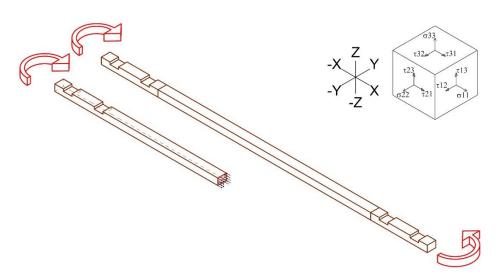


Figure 6-29 Rafter - Bending Moment Mz on  $\ Z$  axis

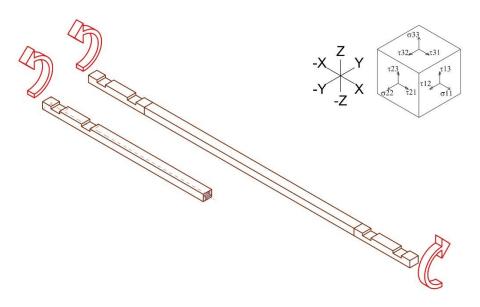


Figure 6-30 Rafter - Torsion: Mx on x axis

## 6.5.2 Roof rafter

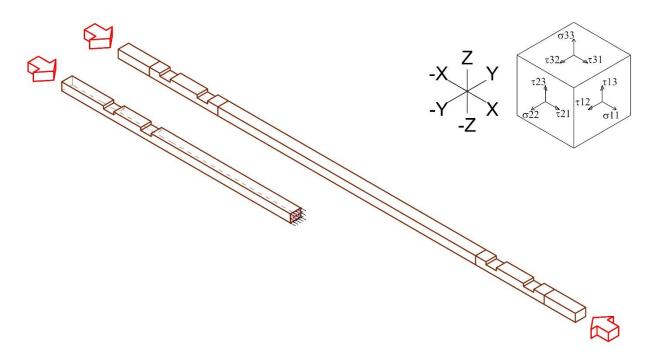


Figure 6-31 Roof Rafter -Compression along X axis

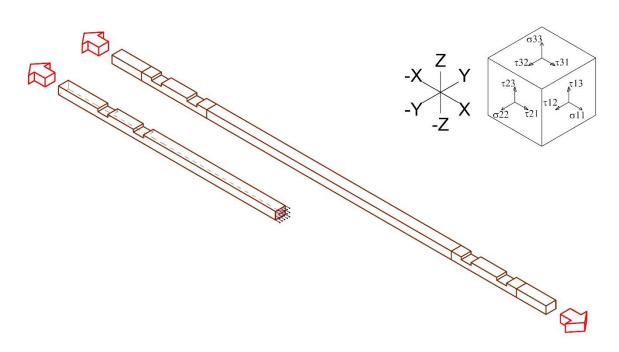


Figure 6-32 Roof Rafter -Tension along X axis

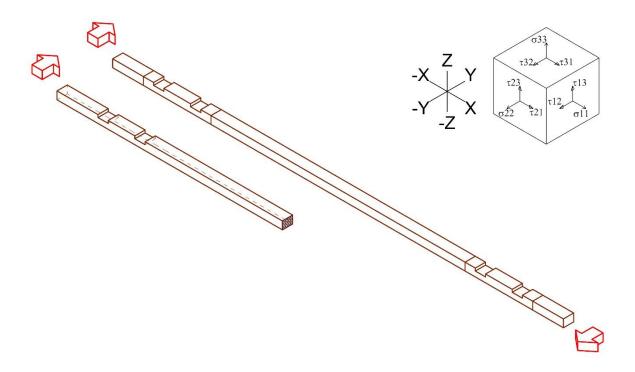


Figure 6-33 Roof Rafter -Shear on Y axis

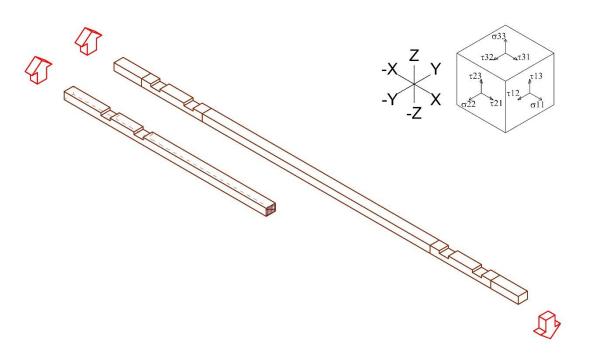


Figure 6-34 Roof Rafter -Shear on Zaxis

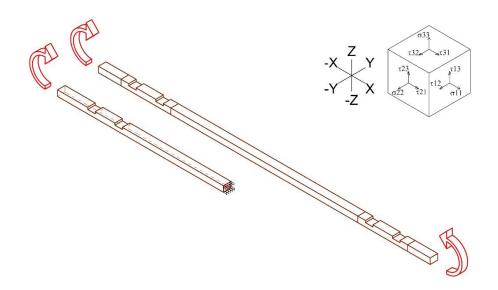


Figure 6-35 Roof Rafter - Bending Moment My on Y axis

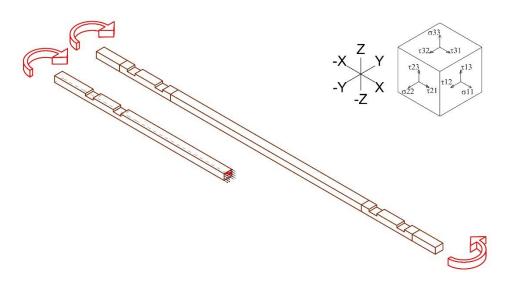


Figure 6-36 Roof Rafter - Bending Moment Mz on Z axis

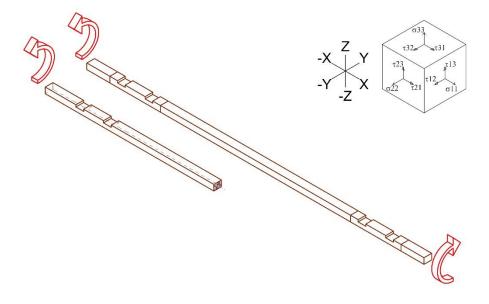


Figure 6-37 Roof Rafter - Torsion: Mx on x axis

# 6.5.3 Cross piece

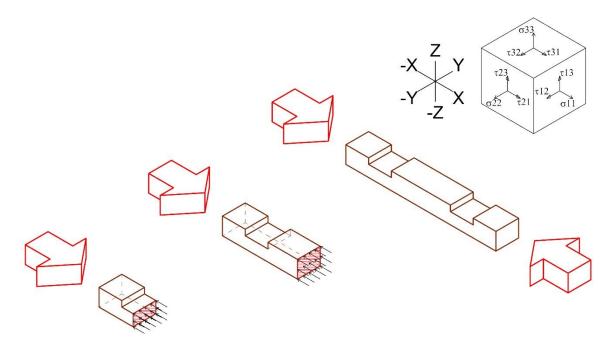


Figure 6-38 Cross Piece -Compression along X axis

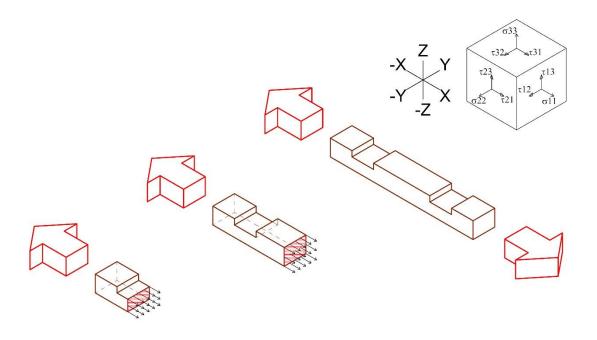


Figure 6-39 Cross Piece -Tension along X axis

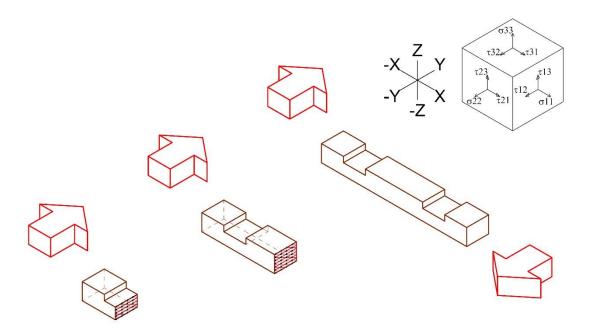


Figure 6-40 Cross Piece -Shear on Y axis

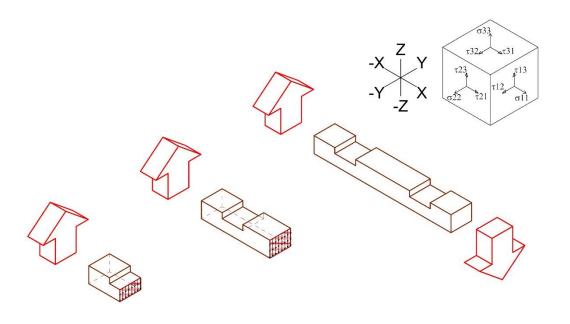


Figure 6-41 Cross Piece -Shear on  $\ Z$  axis

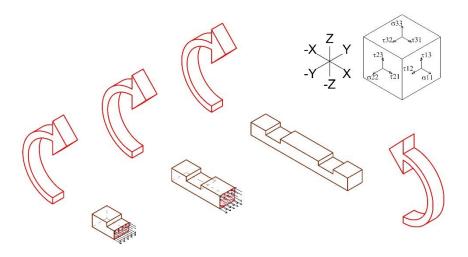


Figure 6-42 Cross Piece -Bending Moment My on Y axis

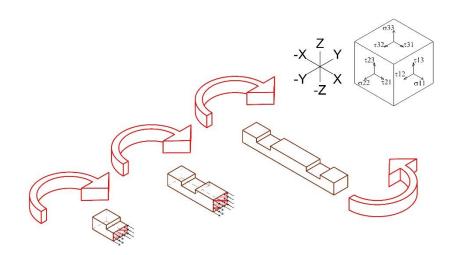


Figure 6-43 Cross Piece -Bending Moment Mz on Z axis

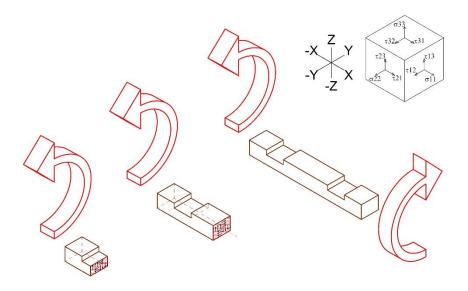


Figure 6-44 Cross Piece -Torsion: Mx on x axis

#### 6.6 Eurocode 5 : EN 1995-1-1 :2004+A 1

In order to study the behavior of the timber elements connections the have been followed the verifications required in the Eurocode 5 and some additional verification required by the Italian code Nicole.

#### 6.6.1 Tension parallel to the grain

$$\sigma_{t,0,d} \leq f_{t,0,d}$$

 $\sigma_{t,0,d}$  is the design tensile stress along the grain

 $f_{t,0,d}$  is the design tensile strength along the grain

$$\sigma_{t0d} = \frac{N_{0d}}{A_{net,t}}$$

 $N_{0d}$  is the design axial force parallel to the grain

 $A_{net,t}$  is the net cross-sectional area perpendicular to the grain

 $A_{net,v}$  is the net shear area in the parallel to grain direction

#### 6.6.2 Tension parallel to the grain with keyed scarf joint

$$\sigma_{t,0,d} \leq f_{t,0,d} * R_{scarf}$$

 $\sigma_{t,0,d}$  is the design tensile stress along the grain

 $f_{t,0,d}$  is the design tensile strength along the grain

$$\sigma_{t0d} = \frac{N_{0d}}{A_{net,t}}$$

 $N_{0d}$  is the design axial force parallel to the grain

 $A_{net,t}$  is the net cross-sectional area perpendicular to the grain

 $A_{net,v}$  is the net shear area in the parallel to grain direction

 $R_{scarf} = 0.11$  is the reduction factor for the presence of the keyed scarf joint.

## **6.6.3** Compression parallel to the grain

$$\sigma_{c,0,d} \leq f_{c,0,d}$$

 $\sigma_{c,0,d}$  is the design tensile stress along the grain

 $f_{c,0,d}$  is the design tensile strength along the grain

$$\sigma_{c0d} = \frac{N_{0d}}{A_{net.t}}$$

 $N_{0d}$  is the design axial force parallel to the grain

 $A_{net,t}$  is the net cross-sectional area perpendicular to the grain

#### 6.6.4 Compression perpendicular to the grain

$$\sigma_{c,90,d} \le k_{c,90} f_{c,90,d}$$

$$\sigma_{c,90,d} \le \frac{F_{c,90,d}}{A_{ef}}$$

 $\sigma_{c,90,d}$  is the design compressive stress in the effective contact

 $F_{c,90,d}$  is the design compressive load perpendicular to the grain

 $A_{ef}$  is the effective contact area in compression perpendicular

 $f_{c,90,d}$  is the design compressive strength perpendicular to the grain

 $k_{c,90} = 1,25 \ or \ 1,5$  is a factor taking into account the load configuration, the possibility of splitting and the degree of compressive deformation

#### 6.6.5 Tension perpendicular to the grain

This stress is not consistent for the bhatar constructions.

$$\sigma_{t90d} = \frac{N_{90d}}{A_{net,t}}$$

#### **6.6.6 Bending**

$$\frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

 $\sigma_{m,y,d}$  and  $\sigma_{m,z,d}$  are the design bending stresses about the principal axes y and z.

 $f_{m,v,d}$  and  $f_{m,z,d}$  are the corresponding design bending strengths.

$$\sigma_{m,y,d} = \frac{M_{y,d}}{W_y}$$

 $M_{v,d}$  is the design bending moment on y axis

 $W_y = \frac{b*h^2}{6}$  is the moment of resistance of the section around y axis

$$\sigma_{m,z,d} = \frac{M_{z,d}}{W_z}$$

 $M_{z,d}$  is the design bending moment on z axis

 $W_z = \frac{h*b^2}{6}$  is the moment of resistance of the section around z axis

NOTE: The factor km makes allowance for re-distribution of stresses and the effect of inhomogeneities of the material in a cross-section.

- 2) The value of the factor should be taken as follows:
  - for rectangular sections: km = 0.7
  - otherwise km = 1

#### **6.6.7** Shear

For shear with a stress component parallel to the grain, see Figure 6.45(a), as well as for shear with both stress components perpendicular to the grain, see Figure 6.45(b), the following expression shall be satisfied:

$$\tau_d \leq f_{v,d}$$

 $\tau_d$  is the design shear stress

 $f_{v,d}$  is the design shear strength for the actual condition

$$\tau_d = \frac{3}{2} * \frac{V_{ad}}{b * h}$$

 $V_{ad}$  is the design shear force and "a" means the parallel axis

b is the width of the section

h is the height of the section

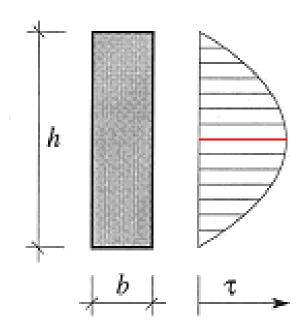


Figure 6-45 Jourawky stress distribution

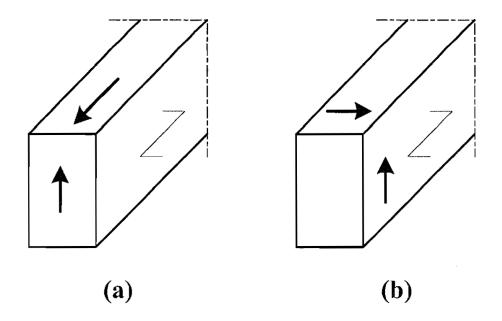


Figure 6-46 (a) Member with a shear stress component parallel to the grain (b) Member with both stress components perpendicular to the grain (rolling shear)

NOTE: The shear strength for rolling shear is approximately equal to twice the tensile strength perpendicular to grain. (2) For the verification of shear resistance of members in bending, the influence of cracks should be taken into account using an effective width of the member given as:

$$b_{ef} = k_{cr} * b$$

where b is the width of the relevant section of the member. $k_{cr} = 0.67$  for solid timber.

#### 6.6.8 Torsion

The following expression shall be satisfied:

$$\tau_{tor,d} \leq k_{shape} * f_{v,d}$$

$$k_{shape} = \left\{ min \left\{ 1 + 0,15 * \frac{h}{b}; 2,0 \right\} \right\}$$

 $\tau_{tor.d}$  is the design torsional stress

 $f_{v,d}$  is the design shear strength

 $k_{shape}$  is a factor depending on the shape of the cross-section;

h is the larger cross-sectional dimension;

b is the smaller cross-sectional dimension.

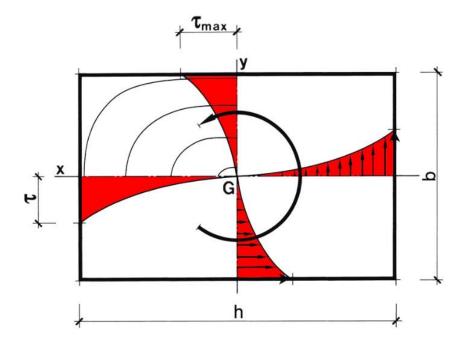


Figure 6-47 Torsional stress distribution

$$\tau_{tor,d} = \alpha * \frac{M_{x,d}}{h * b^2}$$

 $M_{x,d}$  is the design torsional moment along x axis

 $\alpha$  is a semi empirical coefficient which take into account the polar inertia of the section, the where h > b

$$\alpha = 3 + 1.8 * \frac{b}{h}$$

#### 6.6.9 Combined bending and axial tension

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0.7
- otherwise km = 1

### 6.6.10 Combined bending and axial compression

$$\left(\frac{\sigma_{c,0,d}}{f_{c,0,d}}\right)^{2} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_{m} * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\left(\frac{\sigma_{c,0,d}}{f_{c,0,d}}\right)^{2} + k_{m} * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0.7
- otherwise km = 1

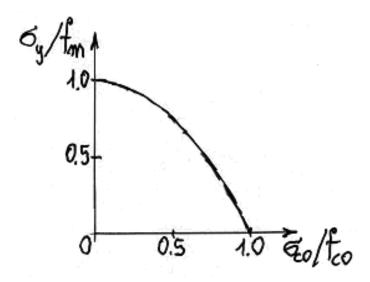
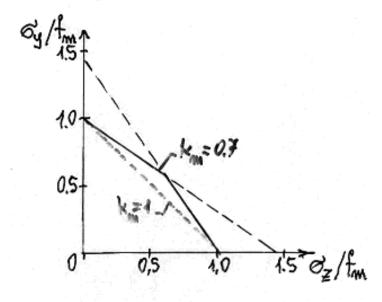


Figure 6-48 Combined bending with axial compression/tension



Figure~6-49~Combined~biaxial~bending~with~axial~compression/tension:

### 6.6.11 Combined Torsion and Shear - CNR-DT 206/2007

$$\frac{\tau_{tor,d}}{k_{shape}*f_{v,d}} + \left(\frac{\tau_d}{f_{v,d}}\right)^2 \leq 1$$

 $\tau_d$  is the design shear stress

 $f_{v,d}$  is the design shear strength for the actual condition for a rectangular cross section :

$$k_{shape} = \left\{ min\left\{1 + 0,15 * \frac{h}{b}; 2,0\right\} \right\}$$

 $au_{tor,d}$  is the design torsional stress;

 $f_{v,d}$  is the design shear strength;

 $k_{shape}$  is a factor depending on the shape of the cross-section;

h is the larger cross-sectional dimension;

b is the smaller cross-sectional dimension.

## 6.7 Resistances - Rafter Body

Appling the verification they have been computed the maximum resistance possible with the Arch.Tom Schacher's manual dimensioning.

## 6.7.1 Longitudinal to the grain

### 6.7.1.1 Rafter Body Compression: RB0comp and Rafter Body Tension: RB0tens

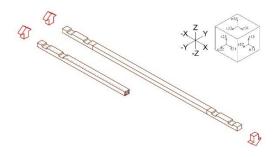


RB0comp		
N_0d	?	N
b	100,00	mm
h	75,00	mm
A_(net)	7500,00	mm
σ_(c,0,d)	#VALUE!	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	#VALUE!	
N_(0d)max	187,00	kN

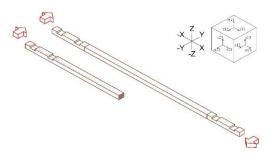


RB0tens		
N_0d	?	N
b	100,00	mm
h	75,00	mm
A_(net)	7500,00	mm
σ_(t,0,d)	#VALUE!	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	#VALUE!	
N_(0d)max	231,00	kN

6.7.1.1 Rafter Body Shear in Z: RB0shearZ and Rafter Body Shear in Y: RB0shearY



RB0shearZ with bending		
V_zd	?	Ν
K_cr	0,67	
A_(net)	5025,00	mm
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_zd max	12,28	kN

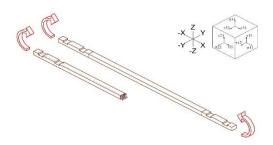


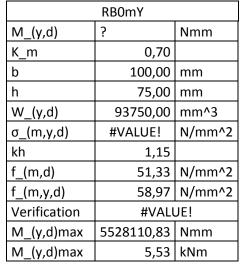
RBOshearY with bending		
V_yd	?	N
K_cr	0,67	
A_(net)	5025,00	mm
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_yd max	12,28	kN

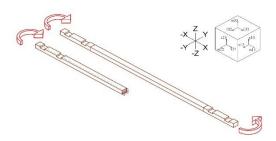
RB0shearZ		
V_zd	?	N
A_(net)	7500,00	mm
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_zd max	18,33 kN	

RB0shearY		
V_yd	?	Ν
A_(net)	7500,00	mm
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_yd max	18,33	kN

# 6.7.1.2 Rafter Body bending moment in Y: RB0mY and Rafter Body bending moment in Z: RB0mZ

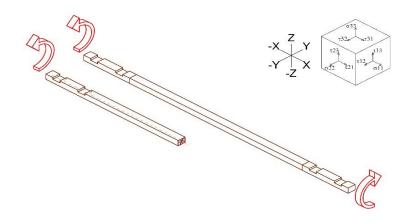






RB0mZ		
M_(z,d)	?	Nmm
K m	0,70	
b	100,00	mm
h	75,00	mm
W_(z,d)	125000,00	mm^3
σ_(m,z,d)	#VALUE!	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	#VALU	JE!
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

### 6.7.1.3 Rafter Body torsional bending moment in X: RB0mX

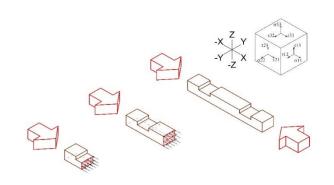


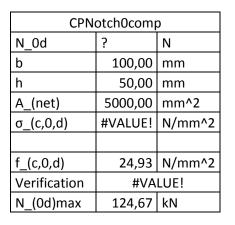
RB0mX		
M_(x,d)	?	Nmm
b	100,00	mm
h	75,00	mm
α	4,35	
T_(tor,d)	#VALUE!	N/mm^2
K_shape	1,02	
f_(v,d)	3,67	N/mm^2
k_shape*f_(v,d)	3,74	N/mm^2
Verification	#VALUE!	
M_(x,d) max	483620,69	Nmm
M_(x,d) max	0,48	kNm

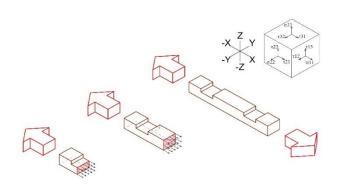
# 6.8 Resistances - Cross piece Notch

## 6.8.1 Longitudinal to the grain

6.8.1.1 Cross piece notch compression: CPNotch0comp and Cross piece notch tension: CPNotch0tens

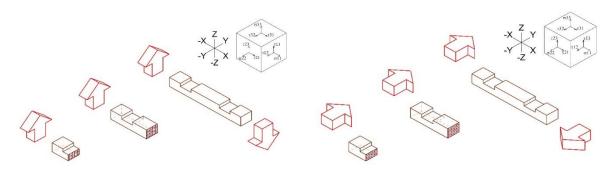






CPNotch0tens		
N_0d	?	Ν
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
σ_(t,0,d)	#VALUE!	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	#VA	LUE!
N_(0d)max	154,00	kN

# 6.8.1.1 Cross piece notch shear in Z: CPNotch0shearZ and Cross piece notch shear in Y: CPNotch0shearY



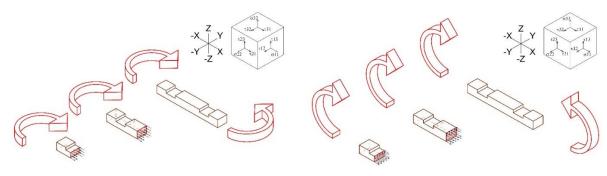
CPNotch0shearZ with bending		
V_zd	?	N
K_cr	0,67	
A_(net)	3350,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_zd max	8,19	kN

CPNotch0shearZ		
V_zd	?	Ν
A_(net)	5000,00	mm
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_zd max	12,22 kN	

CPNotch0shearY with bending		
V_yd	?	N
K_cr	0,67	
A_(net)	3350,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_yd max	8,19	kN

CPNotch0shearY		
V_yd	?	Ν
A_(net)	5000,00	mm
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_yd max	12,22 kN	

# 6.8.1.2 Cross piece notch bending moment in Y: CPNotch0mY and Cross piece notch bending moment in Z: CPNotch0mZ



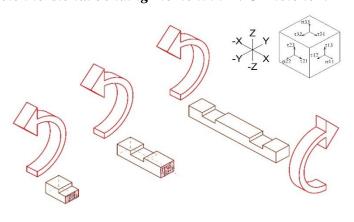
CPNotch0mY		
M_(y,d)	?	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(y,d)	41666,67	mm^3

CPNotch0mZ		
M_(z,d)	?	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(z,d)	83333,33	mm^3

σ_(m,y,d)	#VALUE!	N/mm^2
kh	1,25	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	63,95	N/mm^2
Verification	#VALU	JE!
M_(y,d)max	2664480,07	Nmm
M_(y,d)max	2,66	kNm

σ_(m,z,d)	#VALUE!	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	#VALUE!	
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

### 6.8.1.3 Cross piece notch torsional bending moment in X: CPNotch0mX



CPNotch0mX		
M_(x,d)	?	Nmm
b	100,00	mm
h	50,00	mm
α	3,90	
T_(tor,d)	#VALUE!	N/mm^2
K_shape	1,03	
f_(v,d)	3,67	N/mm^2
k_shape*f_(v,d)	3,78	N/mm^2
Verification	#VALUE!	
M_(x,d) max	242094,02	Nmm
M_(x,d) max	0,24	kNm

# 6.9 Activation of the chains

The possible failure mechanisms, which will be described in the following chapters, have shown the necessity of defining and naming the reaction in the corner joints at the roof level. In this chapter it is shown the overturning mechanism in order to make the reader understand the specific elemnts involved in the description.

# **6.9.1 Overturning Mechanism**

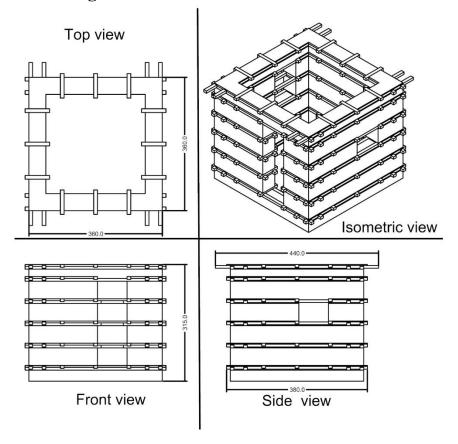


Figure 6-50 Overview of the room box

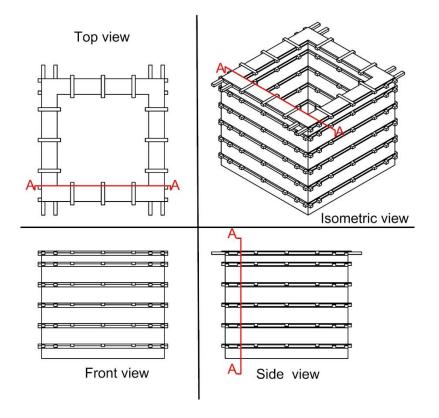


Figure 6-51Section of the studied wall

During the seismic event it may happen the overturning mechanism due to the inertia of the wall, because the mass is subjected to the movement of the ground defined as the peak ground acceleration. This phenomenon is shown in the following figures.

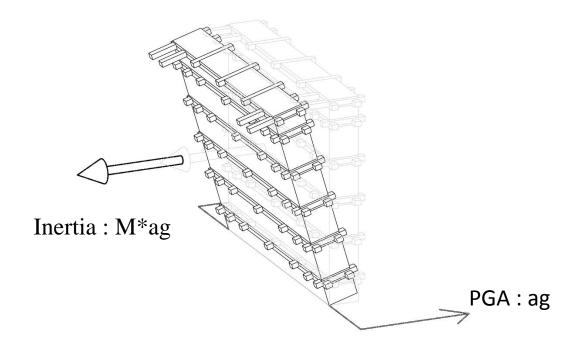


Figure 6-52 Overturning mechanism

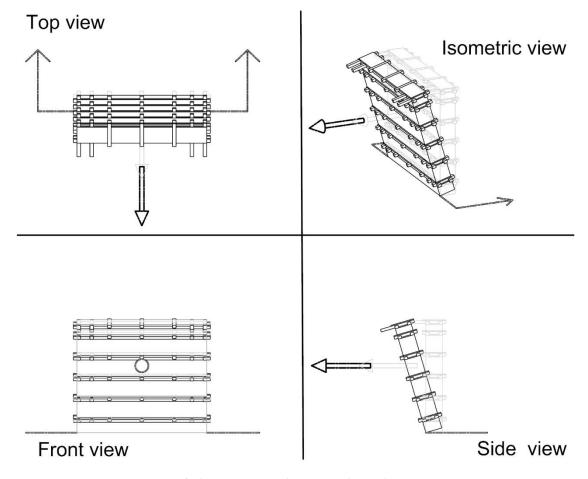


Figure 6-53Overturning mechanism - Orthogonal projections

### 6.9.2 Activation of the chain along Roof Rafter Head

The failure mechanism activate the timber elements at the roof level. The Roof rafters start to work as chain. In the following pictures the two red arrow show the forces developed by the chains.

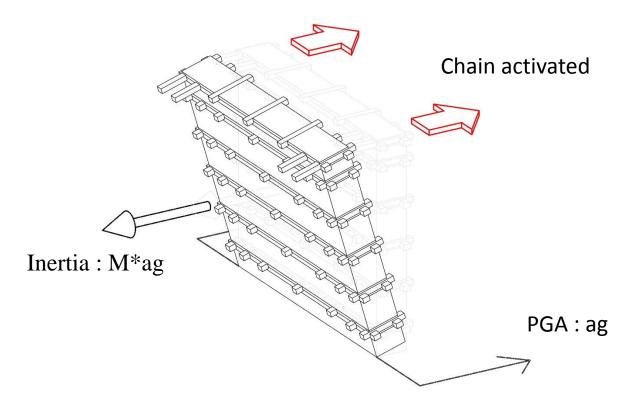


Figure 6-54 - activation of the chains Overturning mechanism

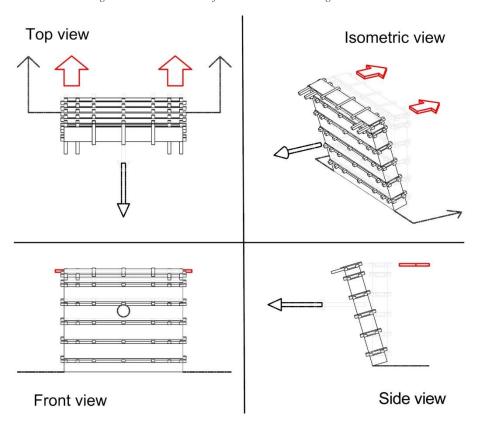


Figure 6-55 Figure 6 53Overturning mechanism - Orthogonal projections activation of the chains

The forces developed in the chains are sheared in the rafters, this repartitions will be described in the following chapters because it is different for the different failure mechanisms.

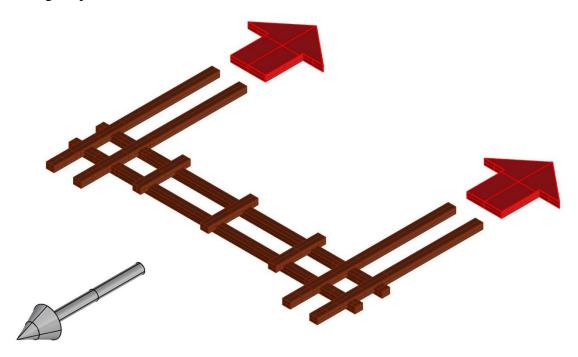


Figure 6-56 Roof timber beam subjected to seismic actions

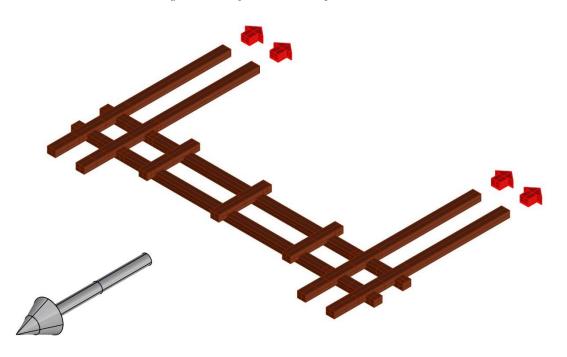
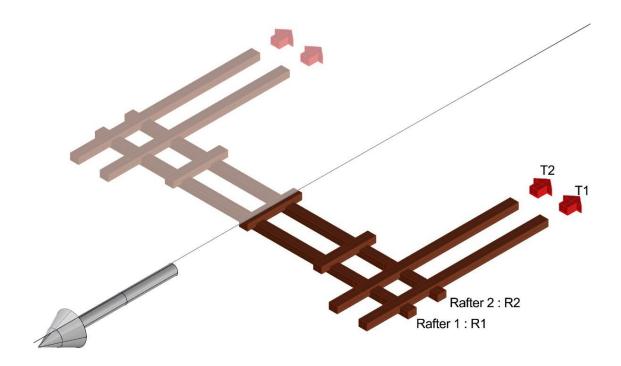


Figure 6-57 Repartitions of forces - Roof timber beam subjected to seismic actions

In order to describe the behavior of the corner joint at the roof level they have been named the four rafters crossed in the corner joint, considering that the behavior of the structure in symmetric. The numbering always start from the external element to the internal.

The elements belonging to the failing wall have been named R# due to the fact that they are passive resisting elements. The rafter belonging to the timber beam working as a chain have been named T# due to the fact that they are subjected mainly to tension.



 $Figure\ 6\text{--}58\ Descriptions\ of\ the\ rafters\ crossed\ at\ the\ roof\ timber\ beam$ 

Each intersection have been recolled as the summations of the names of the crossing rafters, as shown in the picture below.

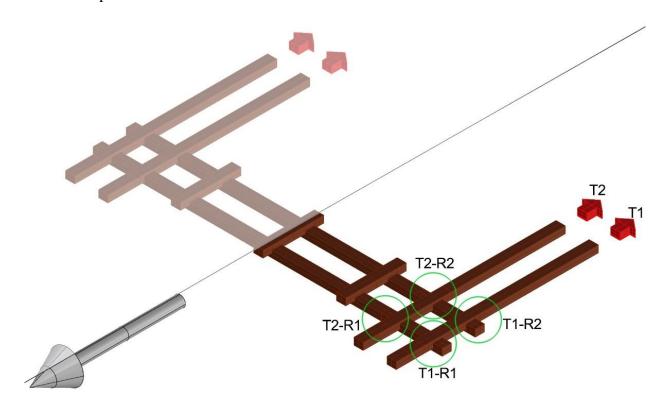


Figure 6-59 Description of the crossing rafters at roof level

#### 6.9.2.1 Axial forces

In the following pages they are described the stressed part of the timber elements and the hierarchy of the forces due to the activation of the chains .

#### 6.9.2.1.1 Axial stresses

The following tables reports the values of the maximum allowed stresses and their position are shown in the following figures.

To be clear names must be read like these examples:

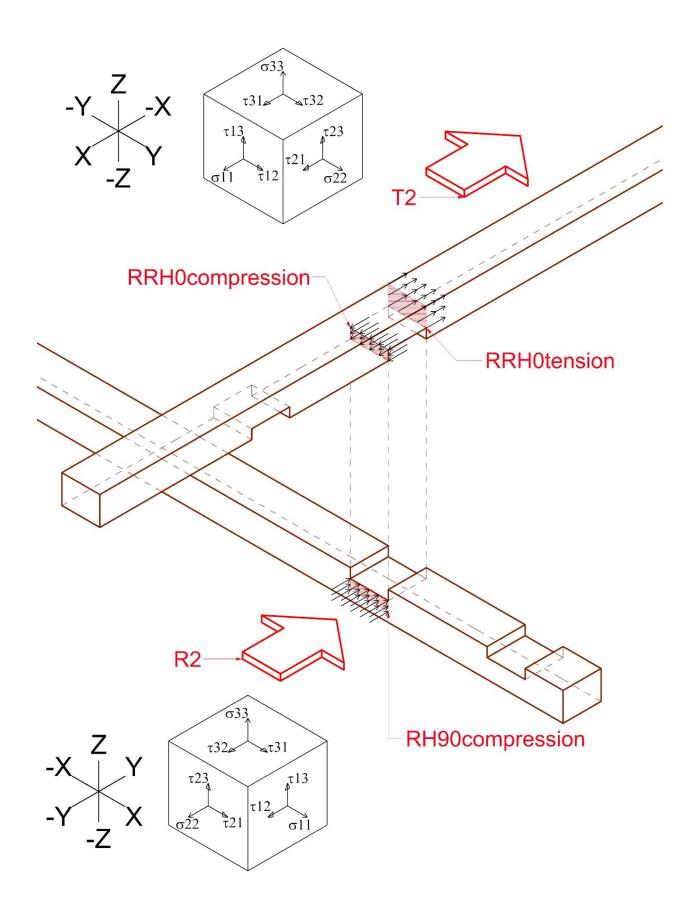
- RRH0tension : Roof Rafter Head 0 =along the fibers tens : in tension A# stressed area
- RRH0compression : Roof Rafter Head 0 = along the fibers compression : in compression A# stressed area
- RRH90compression: Roof Rafter Head 90 = perpendicular to the fibers compression: in compression A# stressed area

Table 14 Roof rafter Head and Rafter head

RRH0tension	A4	
N_0d	?	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
σ_(t,0,d)	#VALUE!	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	#VALUE!	
N_(od)max	154,00	kN

RH90compression	A3	
N_90d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	#VALUE!	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	#VALUE!	
N_(90d)max	16,50	kN

RRH0compression	А3	
N_0d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,0,d)	#VALUE!	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	#VALUE!	
N_(od)max	62,33	kN



Figure~6-60~Roof~Rafter~Head~Axial~stresses:~Crossing~rafters~T2-R2

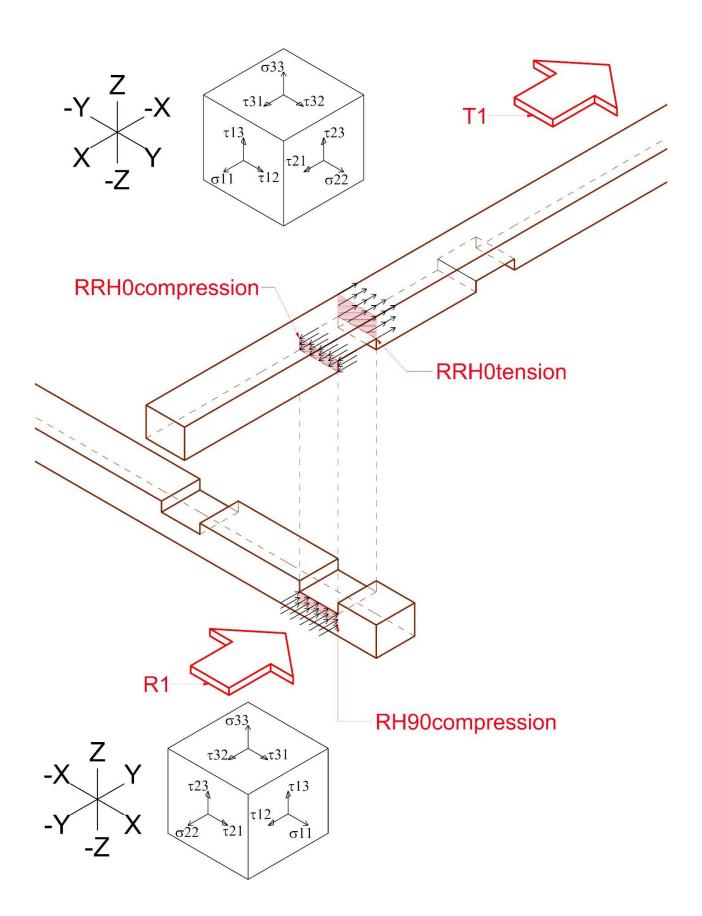


Figure 6-61 Roof Rafter Head Axial stresses: Crossing rafters T1-R1

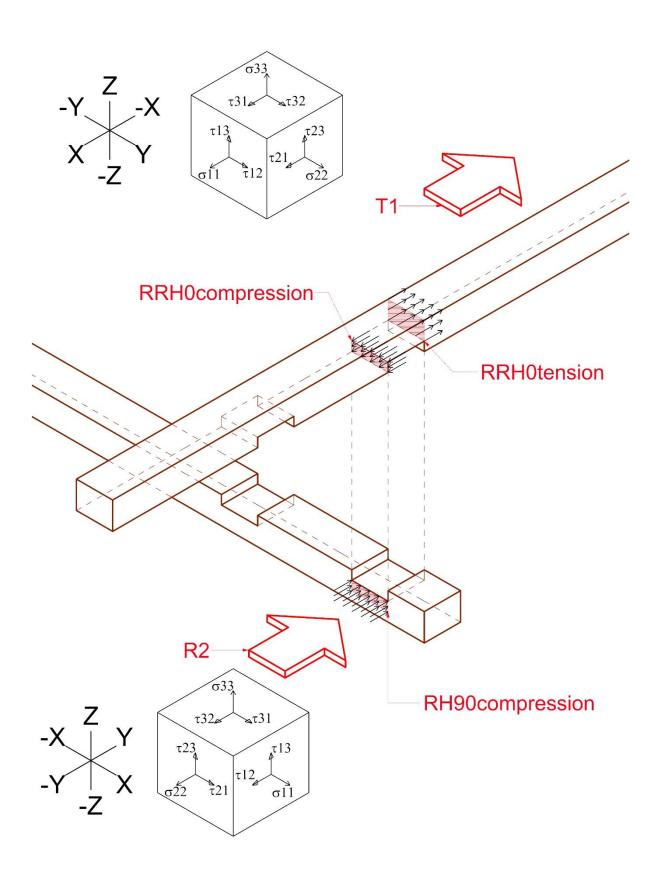


Figure 6-62 Roof Rafter Head Axial stresses : Crossing rafters T1-R2

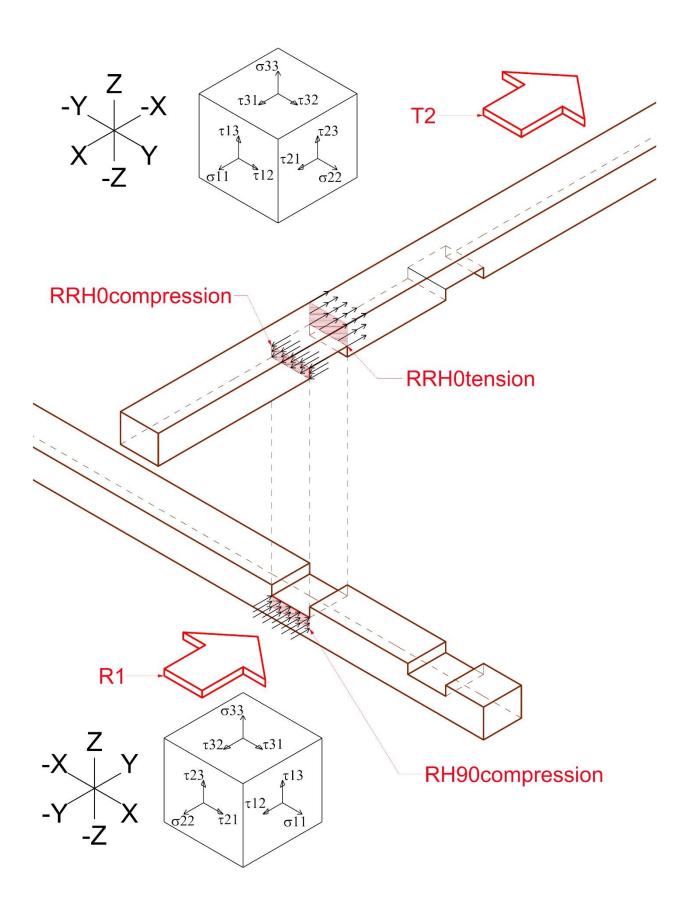


Figure 6-63 Roof Rafter Head Axial stresses: Crossing rafters T2-R1

### **6.9.2.1.2** Tangential stresses

As before the names are related to the timber elements and to the subjected stresses.

- RRH0shearEXT : Roof Rafter Head 0 = along the fibers shear : under shear force EXT : external surface A# stressed area
- RRH0shearINT : Roof Rafter Head 0 = along the fibers shear : under shear force INT : internal surface A# stressed area
- RH90shear : Rafter Head 90 = perpendicular to the fibers shear : under shear force A# stressed area
- RRH0shearEXT, RRH0shearINT, and RH90shear: are computed in case of just pure shear force or shear force and bending moment acting at the same time.

RRH0shearEXT	A7	
with	n bending	
V_0d	?	Ν
K_cr	0,67	
A_(net)	26800,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VAL	.UE!
V_0d max	65,51	kN

RRH0shearINT	A1	
with bending		
V_0d	?	N
K_cr	0,67	
A_(net)	17420,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	42,58	kN

RRH0shearEXT	A7	
V_0d	?	N
A_(net)	40000,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VAL	.UE!
V_0d max	97,78	kN

RRH0shearINT	A1	
V_0d	?	N
A_(net)	26000,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	63,56	kN

RH90shear	A5		
with bending			
V_90d	?	N	
K_cr	0,67		
A_(net)	5025,00	mm^2	
τ_(d)	#VALUE!	N/mm^2	
ft,90,d	0,44	N/mm^2	
f_(v,d)	0,88	N/mm^2	
Verification	#VALUE!		
V_90d max	2,95	kN	

RH90shear	A5	
V_90d	?	N
A_(net)	7500,00	mm^2
τ_(d)	#VALUE!	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	#VALUE!	
V_90d max	4,40	kN

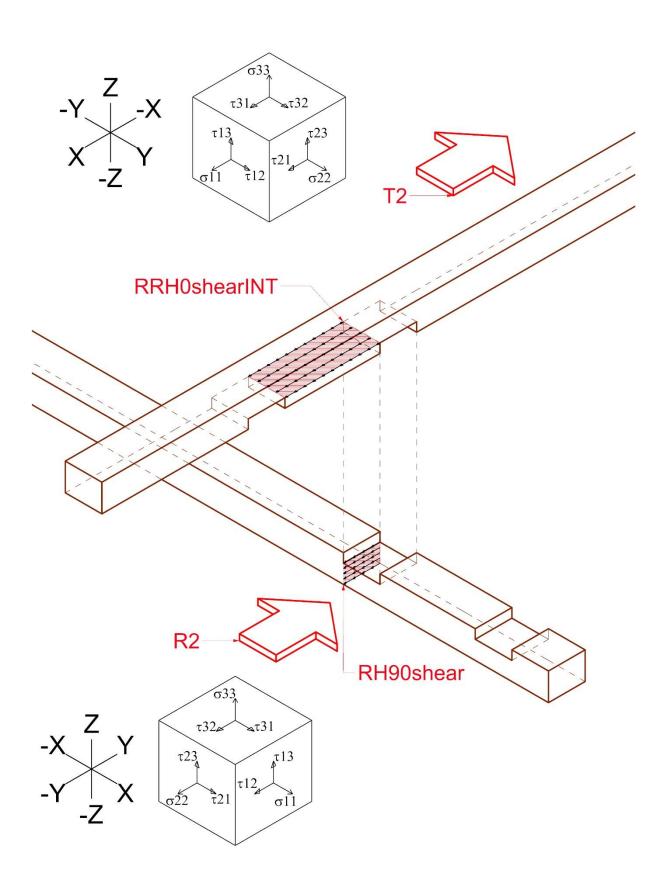


Figure 6-64 Roof Rafter Head Tangential stresses: Crossing rafters T2-R2

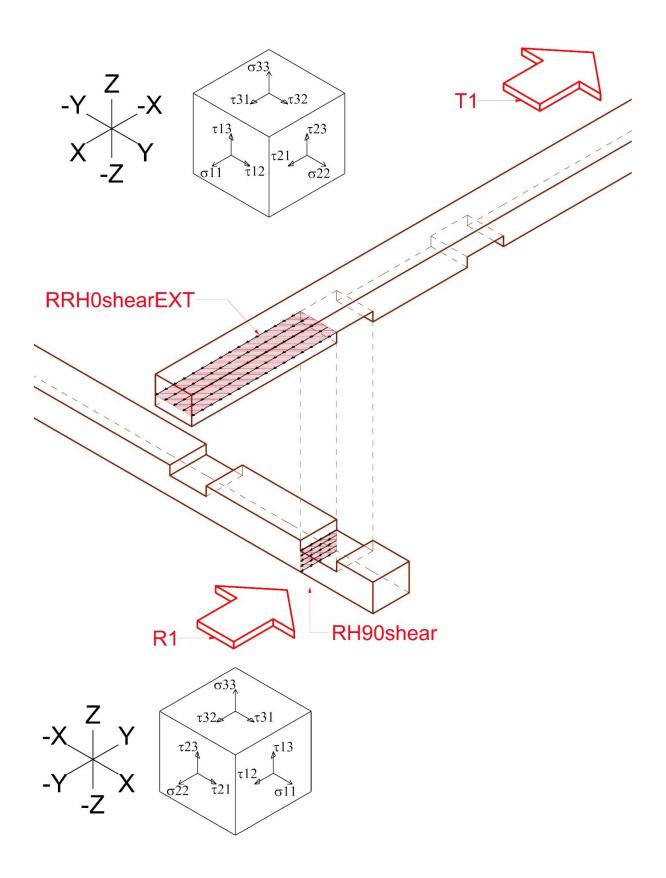


Figure 6-65 Roof Rafter Head Tangential stresses: Crossing rafters T1-R1

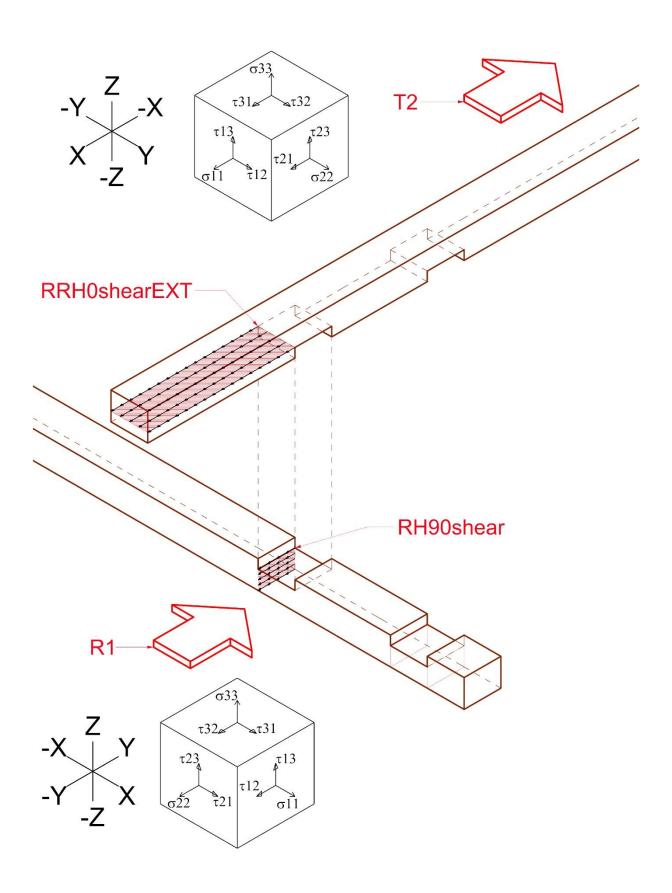


Figure 6-66 Roof Rafter Head Tangential stresses: Crossing rafters T1-R2

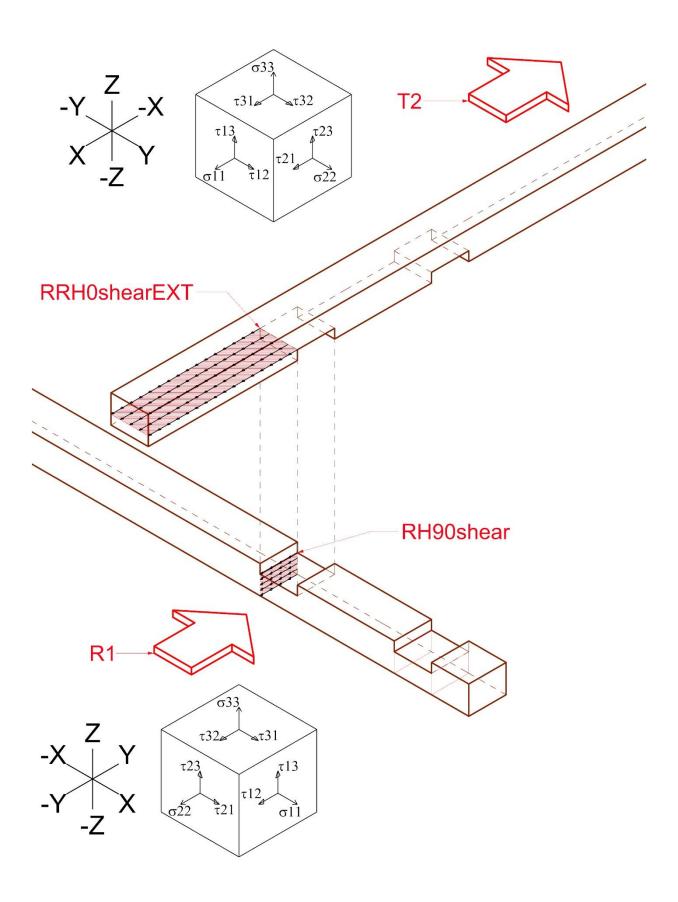


Figure 6-67 Roof Rafter Head Tangential stresses: Crossing rafters T2-R1

### 6.9.2.2 Bending Moments

In the following figures are shown the "parasitic" bending moments developed congruently with the geometry of the timber elements and the applied actions. In the following paragraph it will be shown how the values of each bending moment has been computed.

### **6.9.2.2.1** Bending Moments: Axial stresses

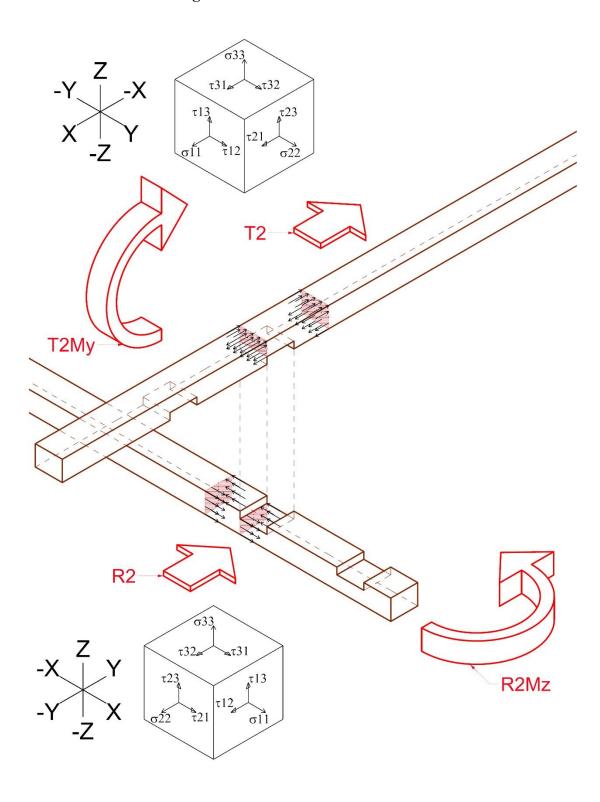


Figure 6-68 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T2-R2

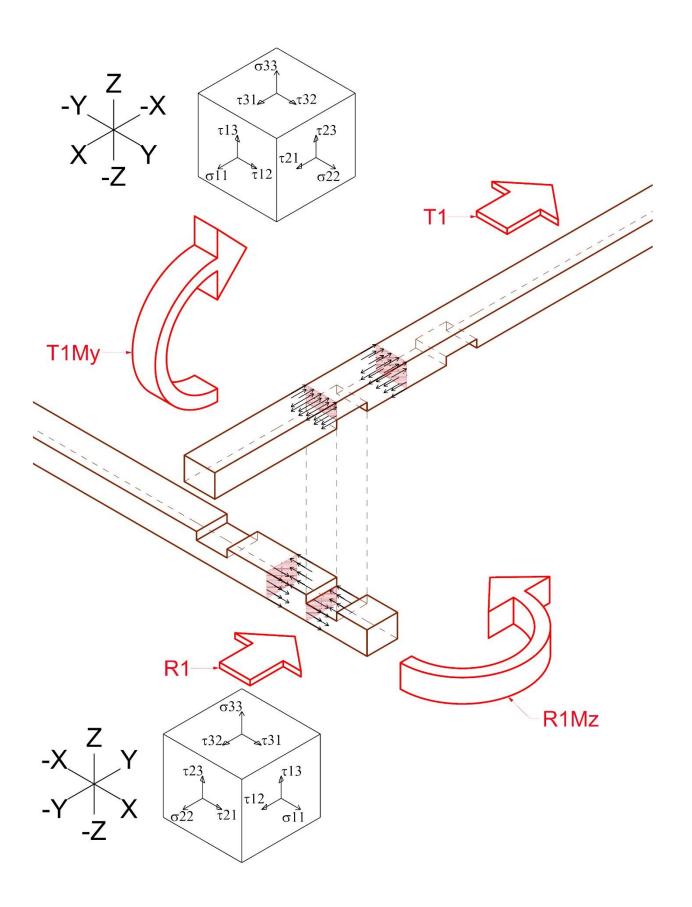


Figure 6-69 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T1-R1

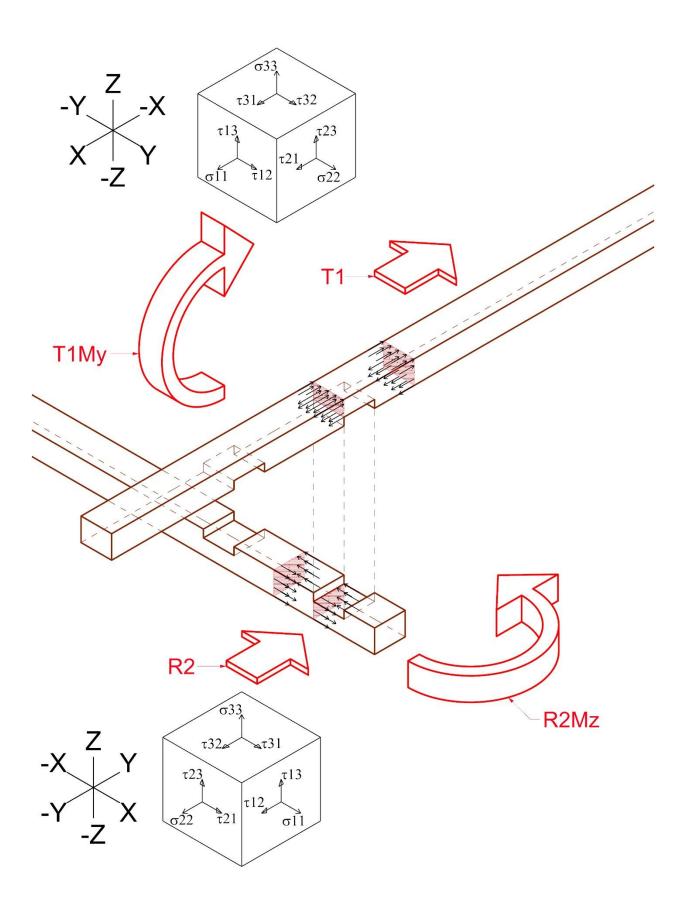


Figure 6-70 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T1-R2

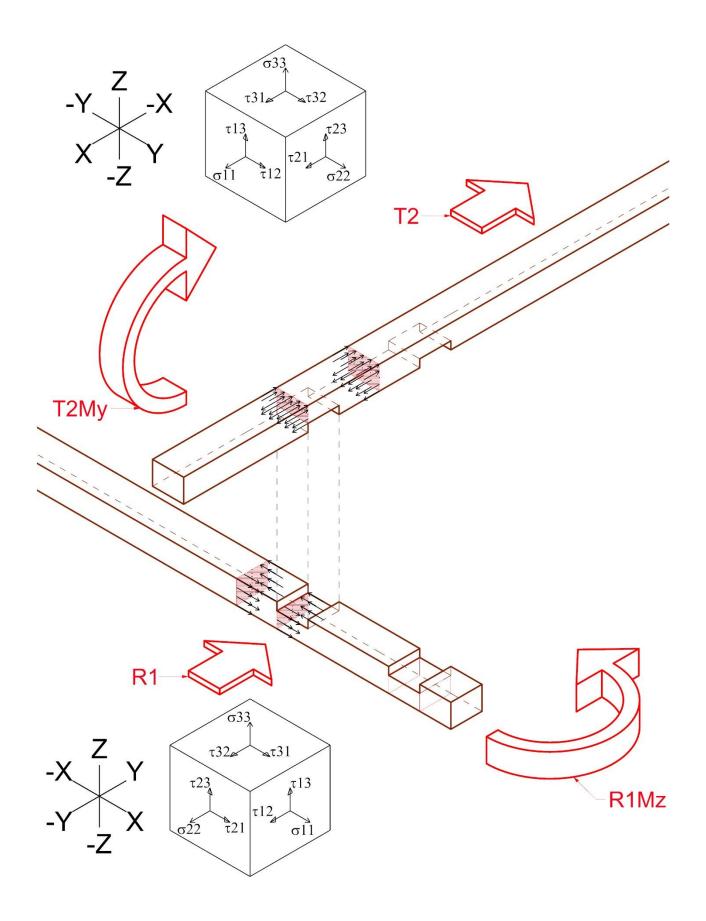


Figure 6-71 Roof Rafter Head Bending moments: Axial stresses: Crossing rafters T2-R1

# **6.9.2.2.2** Torsnion: Tangential stresses

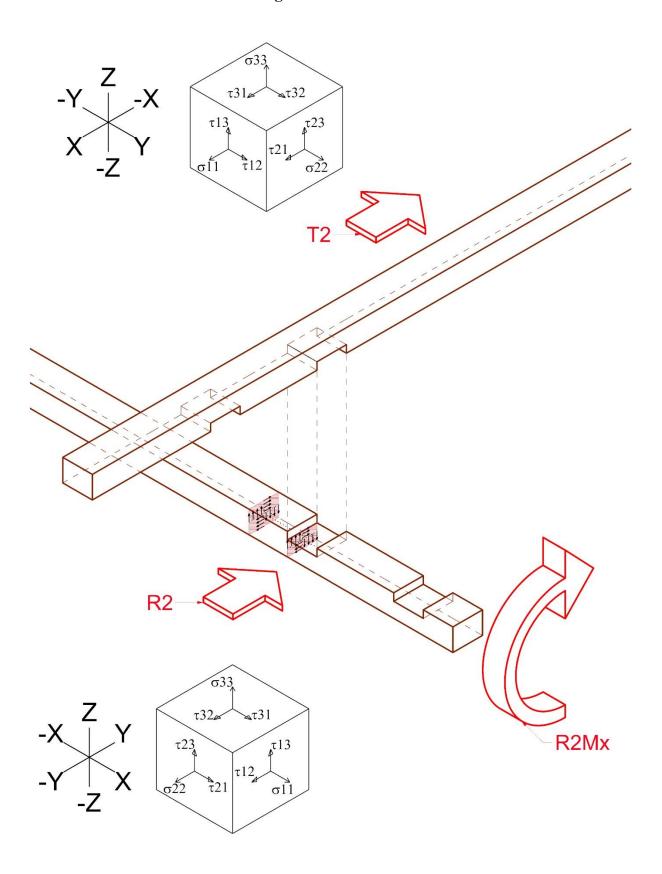


Figure 6-72 Roof Rafter Head Bending moments: Torsnion: Tangential stresses: Crossing rafters T2-R2

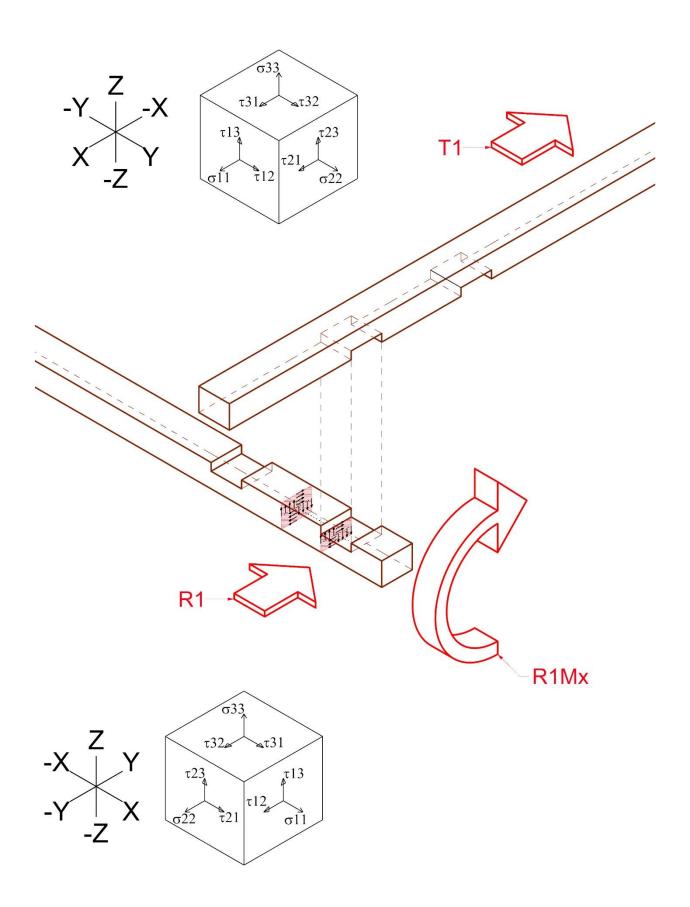


Figure 6-73 Roof Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters T1-R1

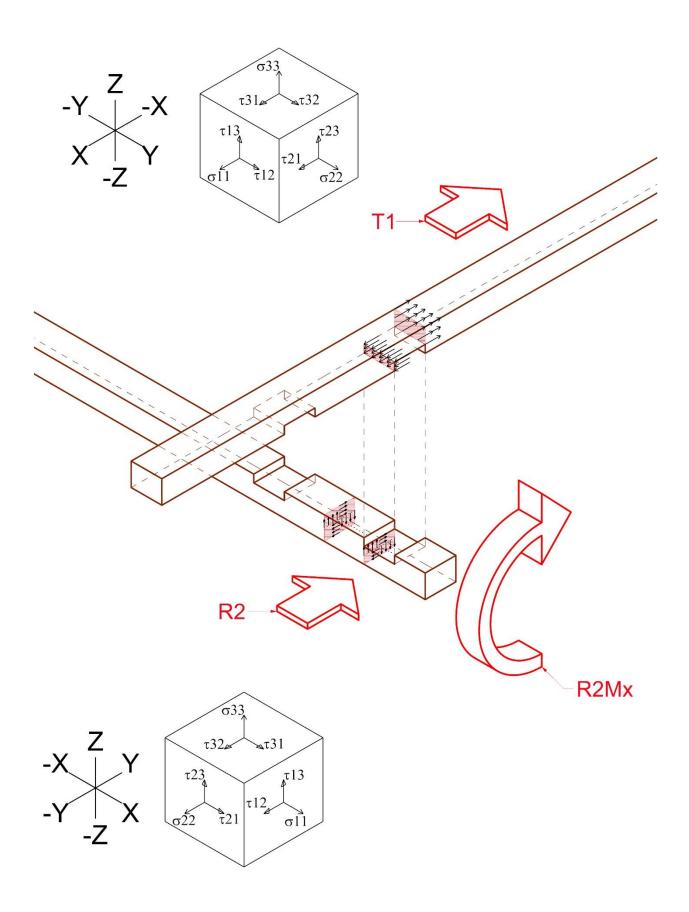


Figure 6-74 Roof Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters T1-R2

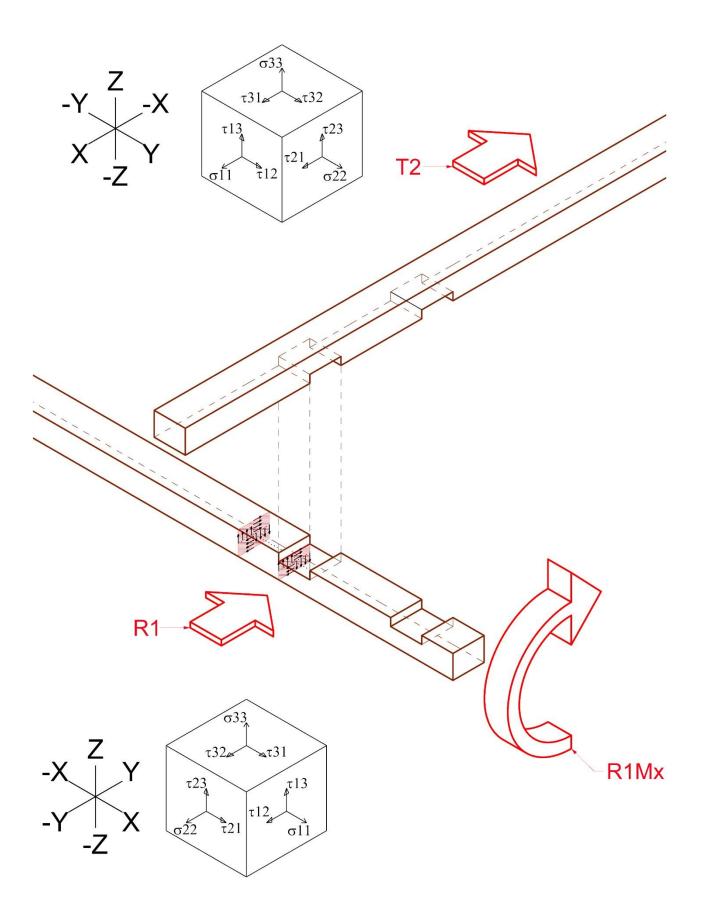


Figure 6-75 Roof Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters T2-R1

### 6.9.3 Activation of the chain along Rafter Head

In accordance to what has been written for the roof rafter head it has been studied in the same way the behavior of the rafter head because the failure mechanism may be activated in the perpendicular direction studied for the roof rafter head.

The forces developed in the chains are sheared in the rafters, this repartitions will be described in the following chapters because it is different for the different failure mechanisms.

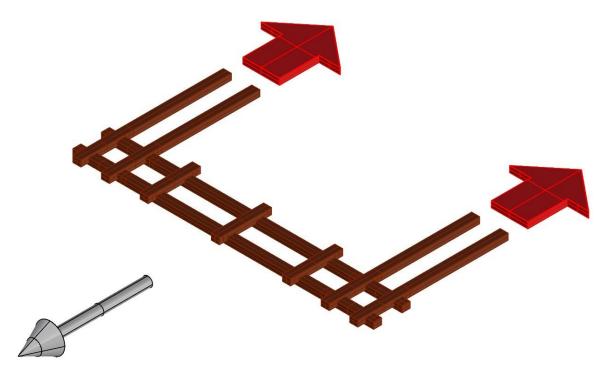


Figure 6-76 Roof timber beam subjected to seismic actions (normal rafter)

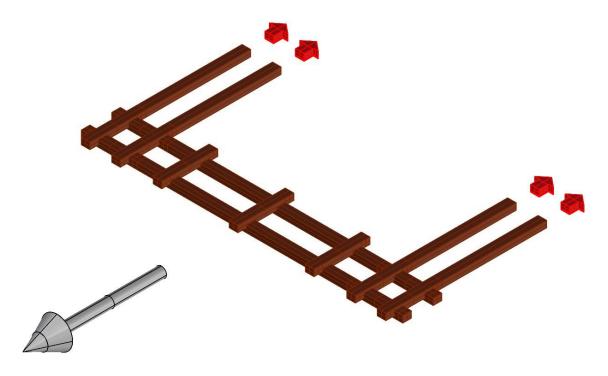


Figure 6-77 Repartitions of forces - Roof timber beam subjected to seismic actions (normal rafter)

In order to describe the behavior of the corner joint at the roof level they have been named the four rafters crossed in the corner joint, considering that the behavior of the structure in symmetric. The numbering always start from the external element to the internal.

The elements belonging to the failing wall have been named R# due to the fact that they are passive resisting elements. The rafter belonging to the timber beam working as a chain have been named T# due to the fact that they are subjected mainly to tension.

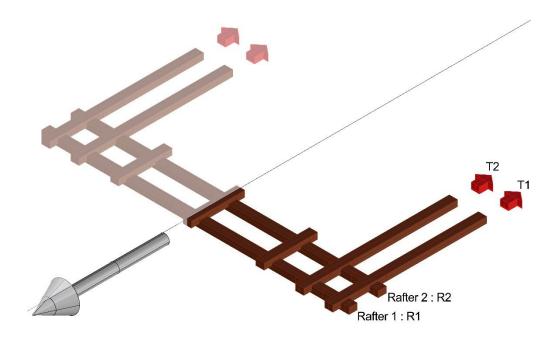


Figure 6-78 Descriptions of the rafters crossed at the roof timber beam actions (normal rafter)

Each intersection have been recolled as the summations of the names of the crossing rafters, as shown in the picture below.

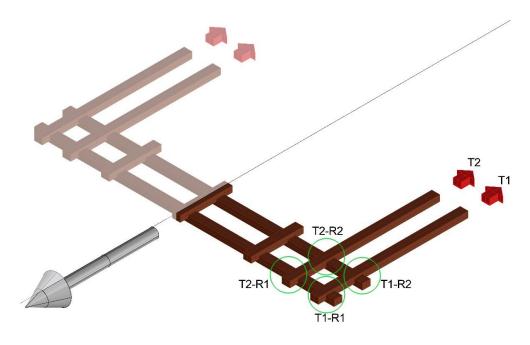


Figure 6-79 Description of the crossing rafters at roof level actions (normal rafter)

#### 6.9.3.1 Axial forces

In the following pages they are described the stressed part of the timber elements and the hierarchy of the forces due to the activation of the chains .

#### 6.9.3.1.1 Axial stresses

The following tables reports the values of the maximum allowed stresses and their position are shown in the following figures.

To be clear names must be read like these examples:

- RRH0tension : Roof Rafter Head 0 =along the fibers tens : in tension A# stressed area
- RRH0compression : Roof Rafter Head 0 = along the fibers compression : in compression A# stressed area
- RRH90compression: Roof Rafter Head 90 = perpendicular to the fibers compression: in compression A# stressed area

Table 15 Roof rafter Head and Rafter head

RH0tension	A4	
N_0d	?	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
σ_(t,0,d)	#VALUE!	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	#VALUE!	
N_(od)max	154,00	kN

RH90compression	A3	
N_90d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	#VALUE!	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	#VALUE!	
N_(90d)max	16,50	kN

RH0compression	A3	
N_0d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,0,d)	#VALUE!	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	#VALUE!	
N_(od)max	62,33	kN

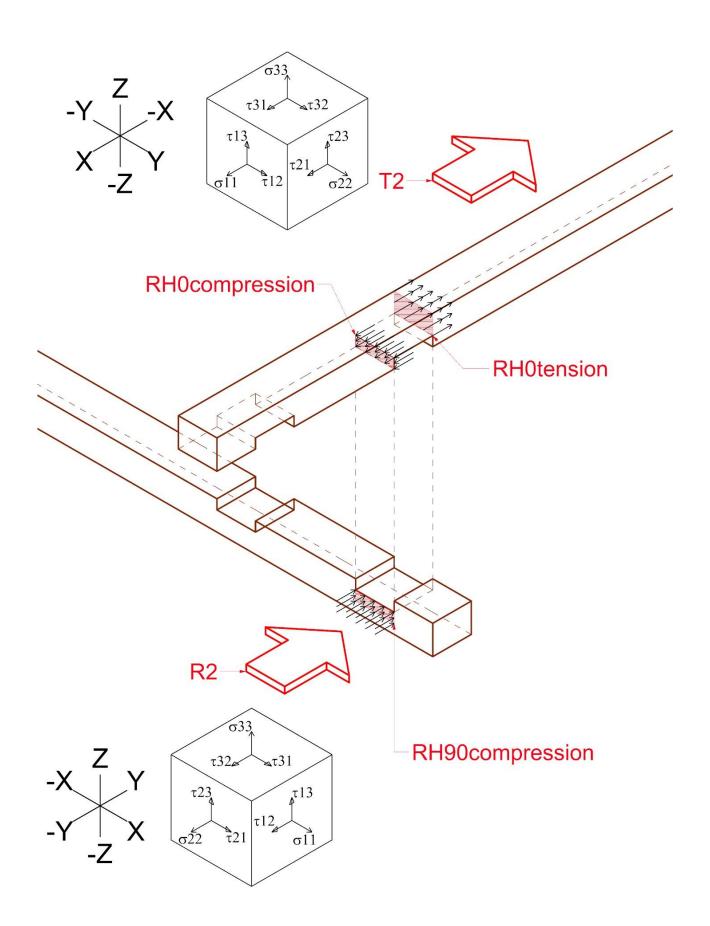


Figure 6-80 Rafter Head Axial stresses: Crossing rafters T2-R2

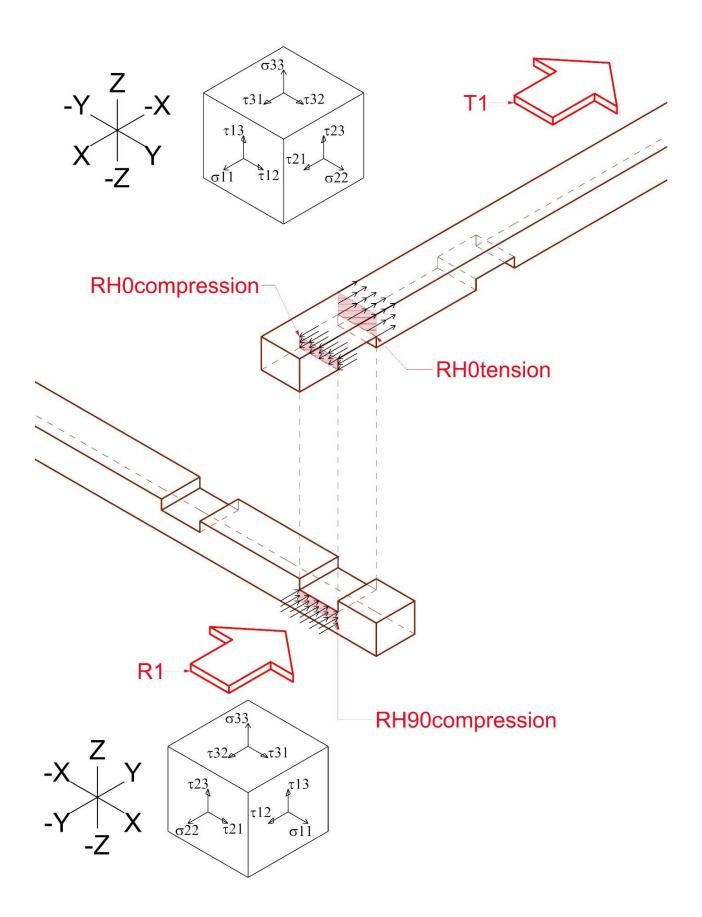


Figure 6-81 Rafter Head Axial stresses: Crossing rafters T1-R1

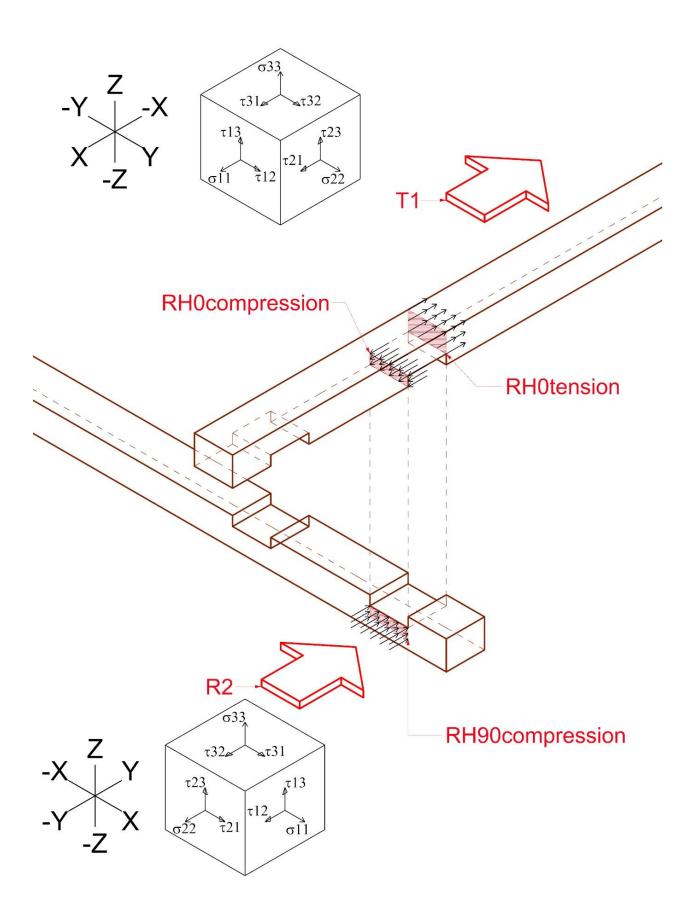


Figure 6-82 Rafter Head Axial stresses: Crossing rafters T1-R2

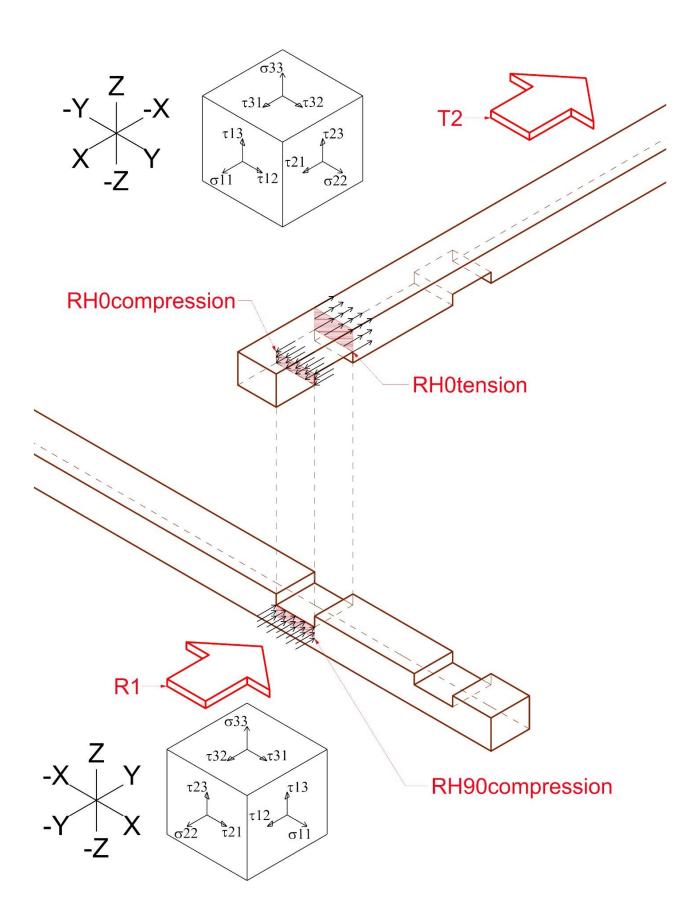


Figure 6-83 Rafter Head Axial stresses: Crossing rafters T2-R1

### **6.9.3.1.2** Tangential stresses

As before, the names are related to the timber elements and to the subjected stresses.

- RH0shearEXT : Rafter Head 0 = along the fibers shear : under shear force EXT : external surface - A# stressed area
- RH0shearINT : Rafter Head 0 = along the fibers shear : under shear force INT : internal surface - A# stressed area
- RH90shear: Head 90 = perpendicular to the fibers shear: under shear force A# stressed area
- RH0shearEXT, RH0shearINT, and RH90shear: are computed in case of just pure shear force or shear force and bending moment acting at the same time.

RH0shearEXT	A2	
with	n bending	
V_0d	?	N
K_cr	0,67	
A_(net)	6700,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	16,38	kN

RH0shearEXT	A2	
V_0d	?	N
A_(net)	10000,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	24,44	kN

RH90shear	A5	
W	ith bendin	g
V_90d	?	N
K_cr	0,67	
A_(net)	5025,00	mm^2
τ_(d)	#VALUE!	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	#VALUE!	
V_90d max	2,95	kN

RH90shear	A5	
V_90d	?	N
A_(net)	7500,00	mm^2
τ_(d)	#VALUE!	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	#VALUE!	
V_90d max	4,40	kN

RH0shearINT	A1	
with bending	, " = RRH0sl	hearINT "
V_0d	?	N
K_cr	0,67	
A_(net)	17420,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	42,58	kN

RH0shearINT	A1	
V_0d	?	N
A_(net)	26000,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	63,56	kN

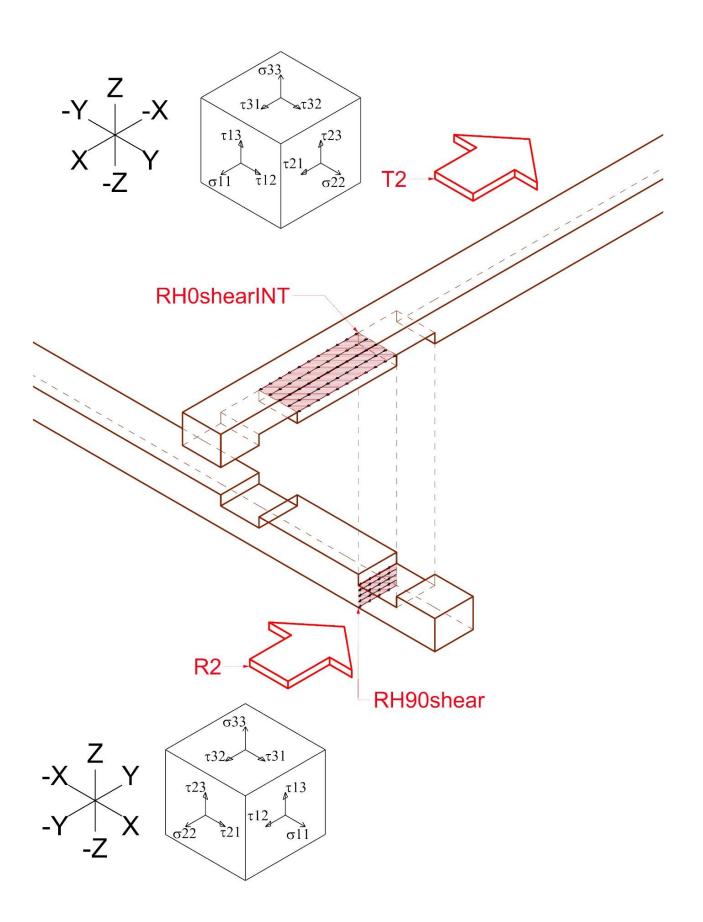


Figure 6-84 Rafter Head Tangential stresses: Crossing rafters T2-R2

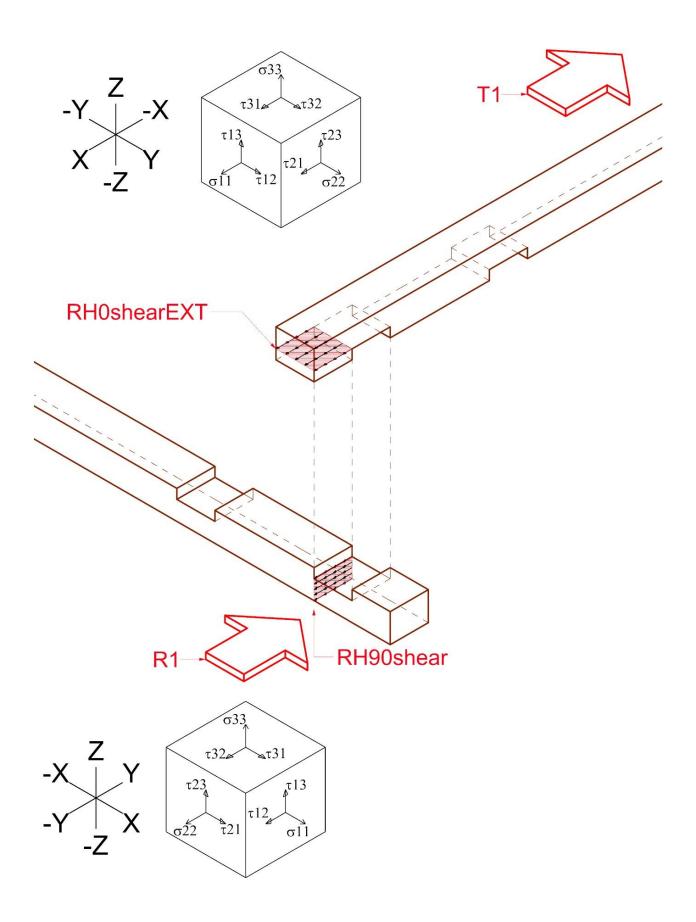


Figure 6-85 Rafter Head Tangential stresses: Crossing rafters T1-R1

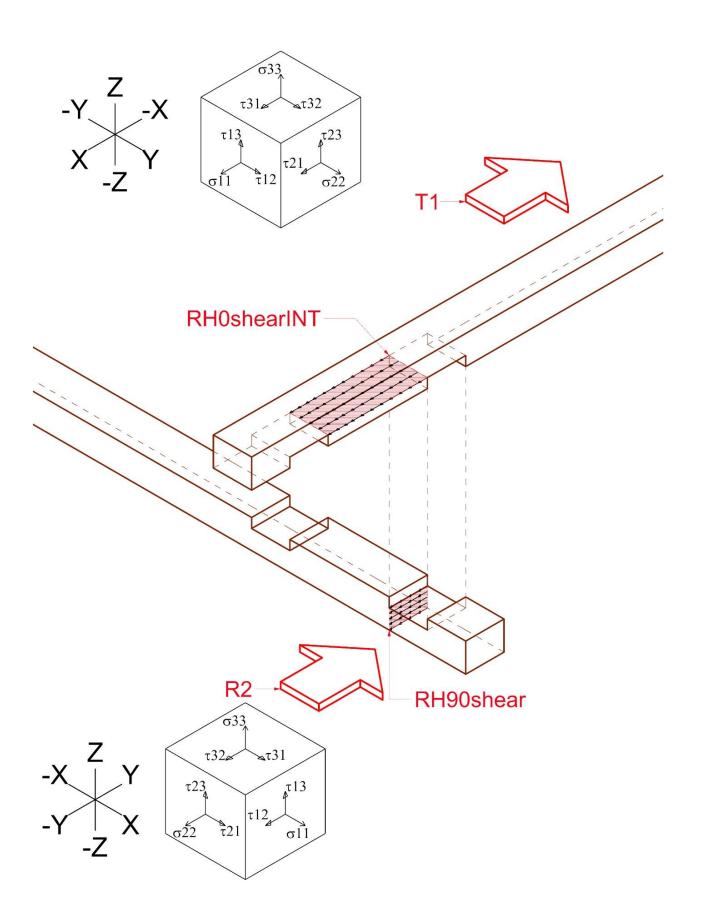


Figure 6-86 Rafter Head Tangential stresses: Crossing rafters T1-R2

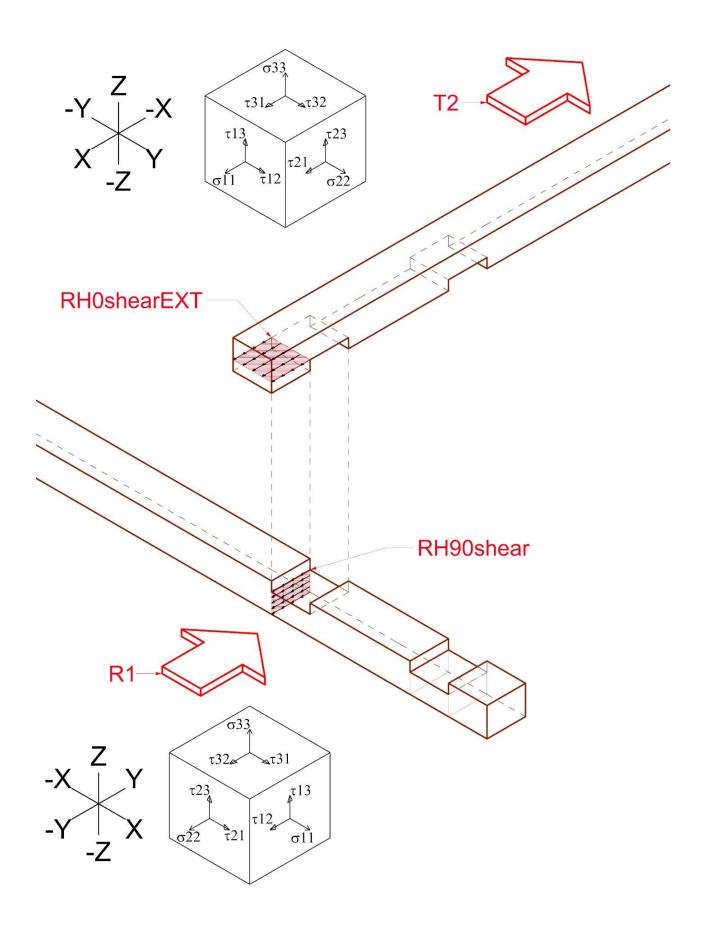


Figure 6-87 Rafter Head Tangential stresses: Crossing rafters T2-R1

# 6.9.3.2 Bending Moments

In the following figures are shown the "parasitic" bending moments developed congruently with the geometry of the timber elements and the applied actions. In the following paragraph it will be shown how the values of each bending moment has been computed.

# **6.9.3.2.1** Bending Moments: Axial stresses

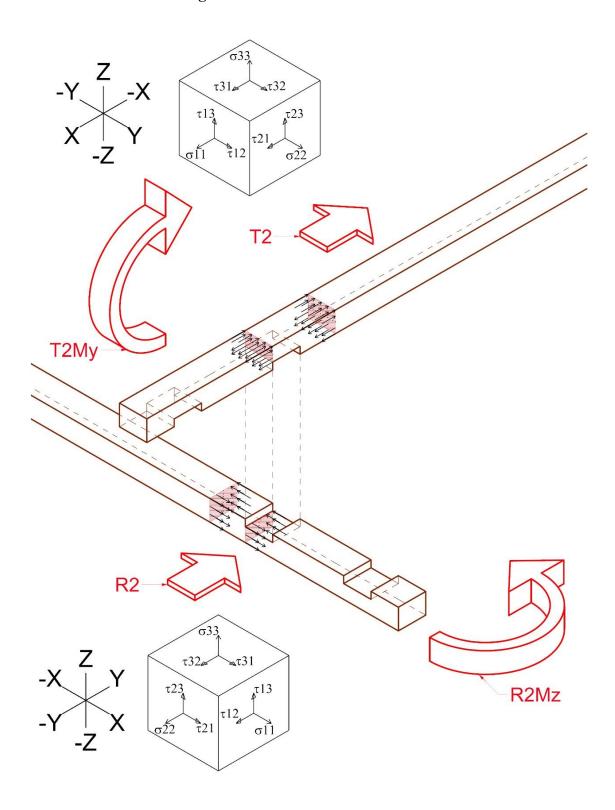


Figure 6-88 Rafter Head Bending moments: Axial stresses: Crossing rafters T2-R2

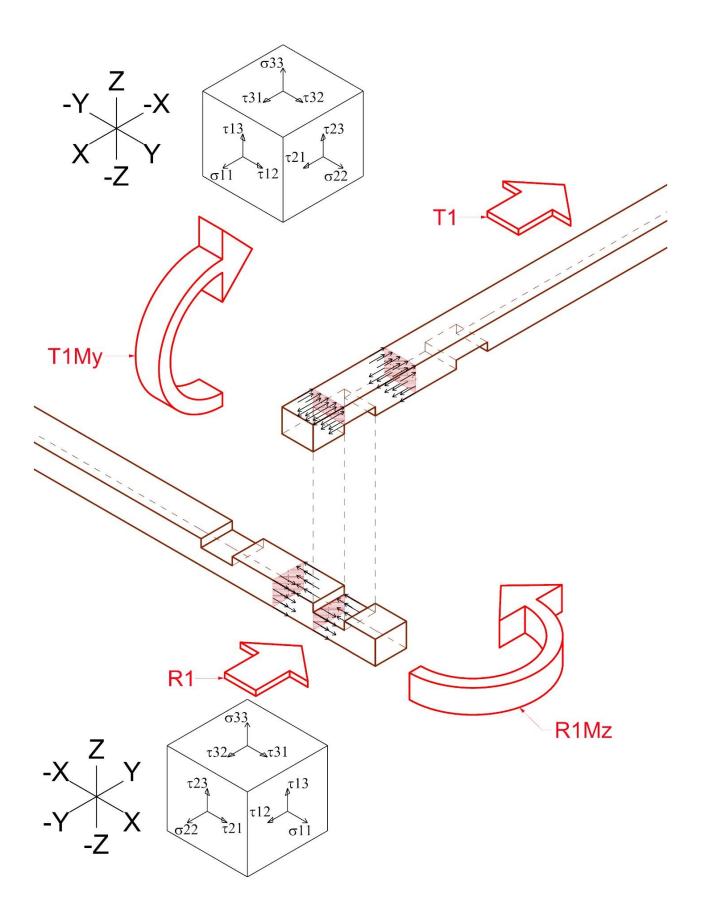


Figure 6-89 Rafter Head Bending moments: Axial stresses: Crossing rafters T1-R1

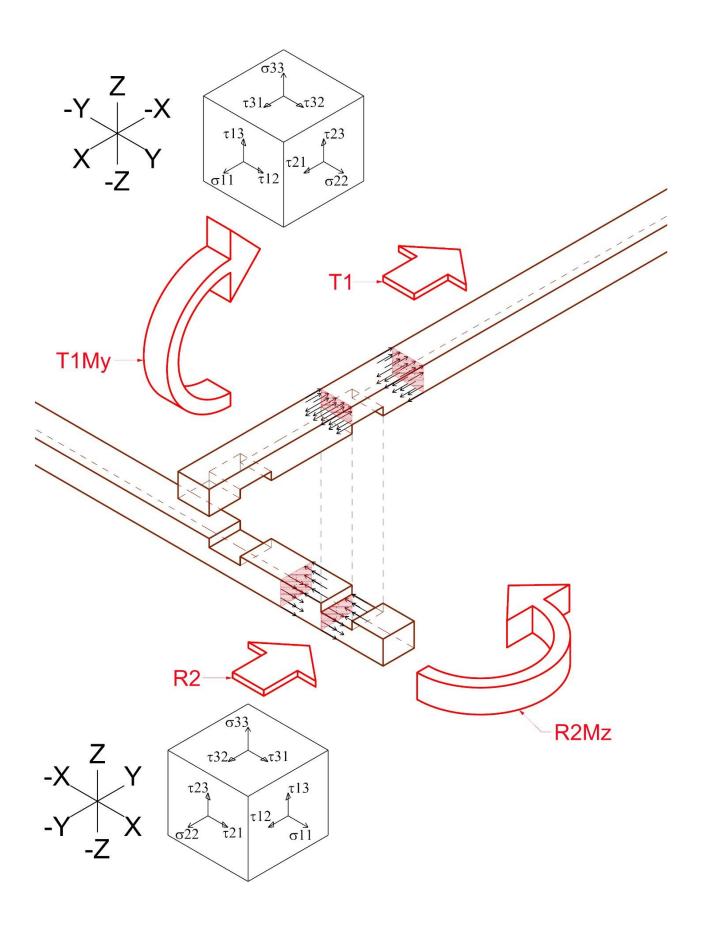


Figure 6-90 Rafter Head Bending moments: Axial stresses: Crossing rafters T1-R2

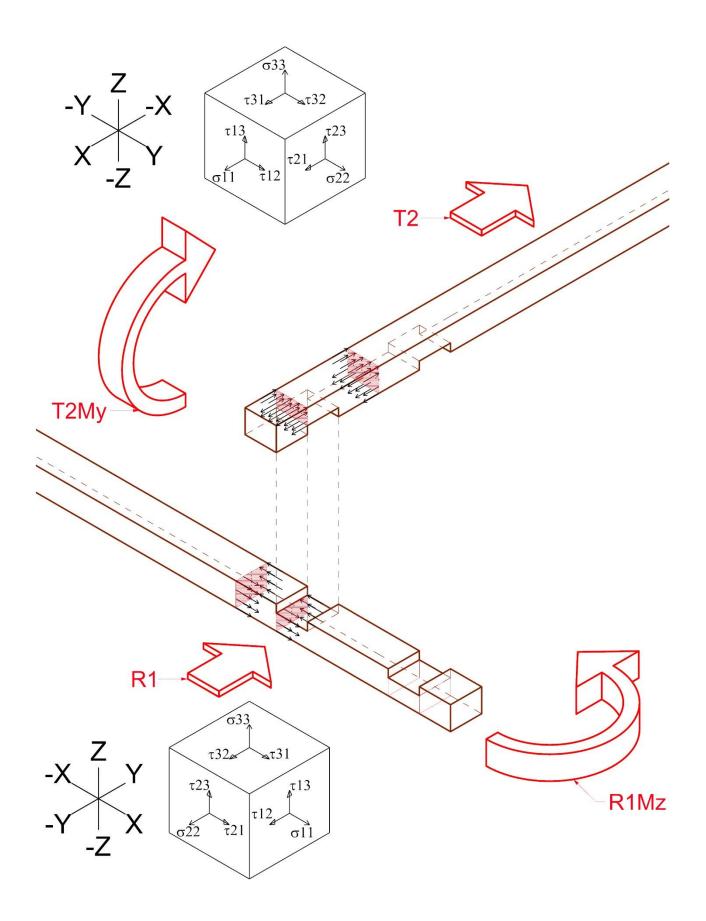


Figure 6-91 Rafter Head Bending moments: Axial stresses: Crossing rafters T2-R1

# **6.9.3.2.2** Torsnion: Tangential stresses

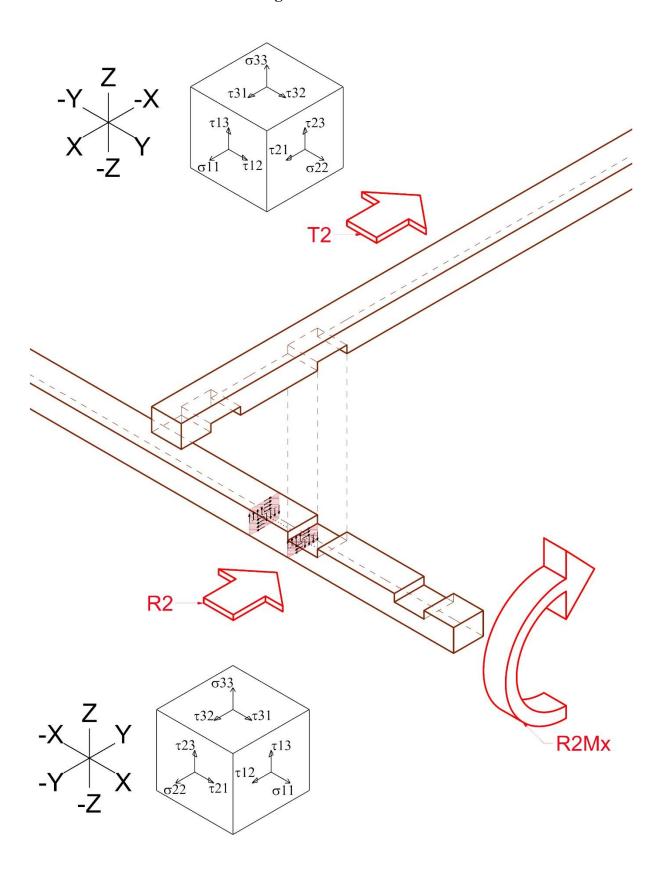


Figure 6-92 Rafter Head Bending moments: Torsnion: Tangential stresses: Crossing rafters T2-R2

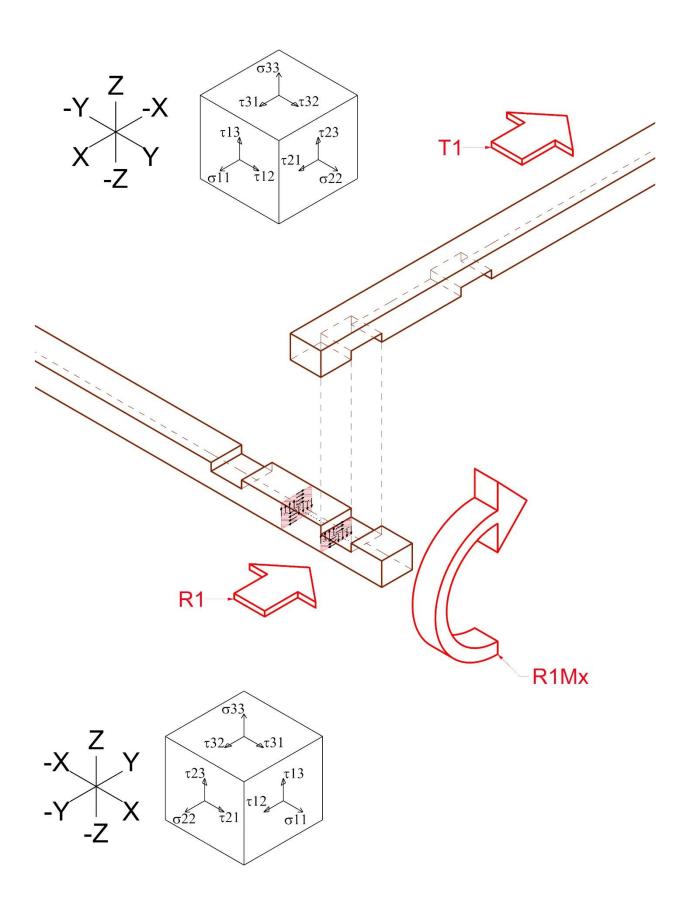


Figure 6-93 Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters T1-R1

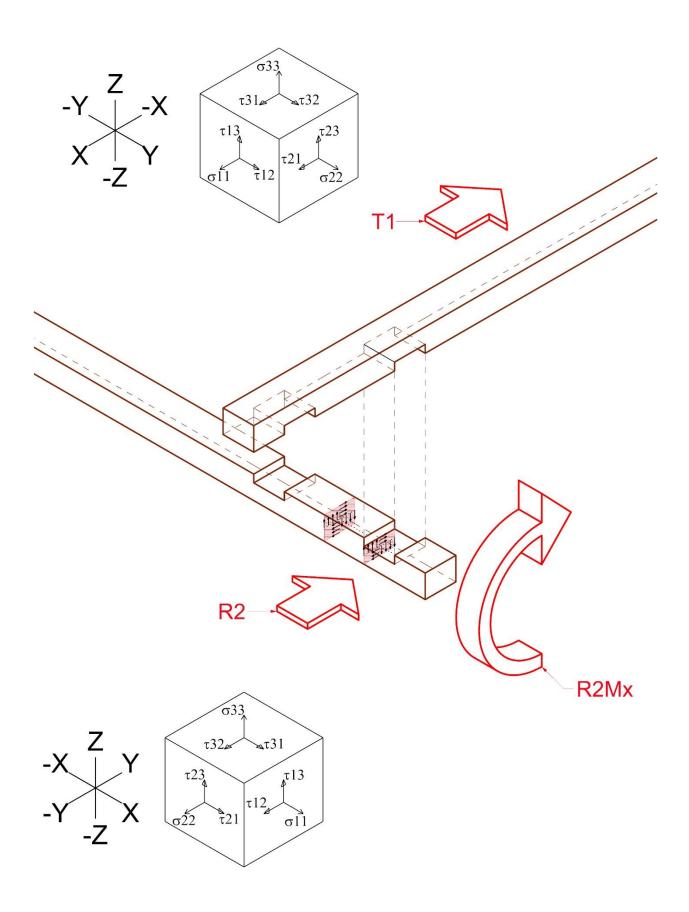


Figure 6-94 Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters T1-R2

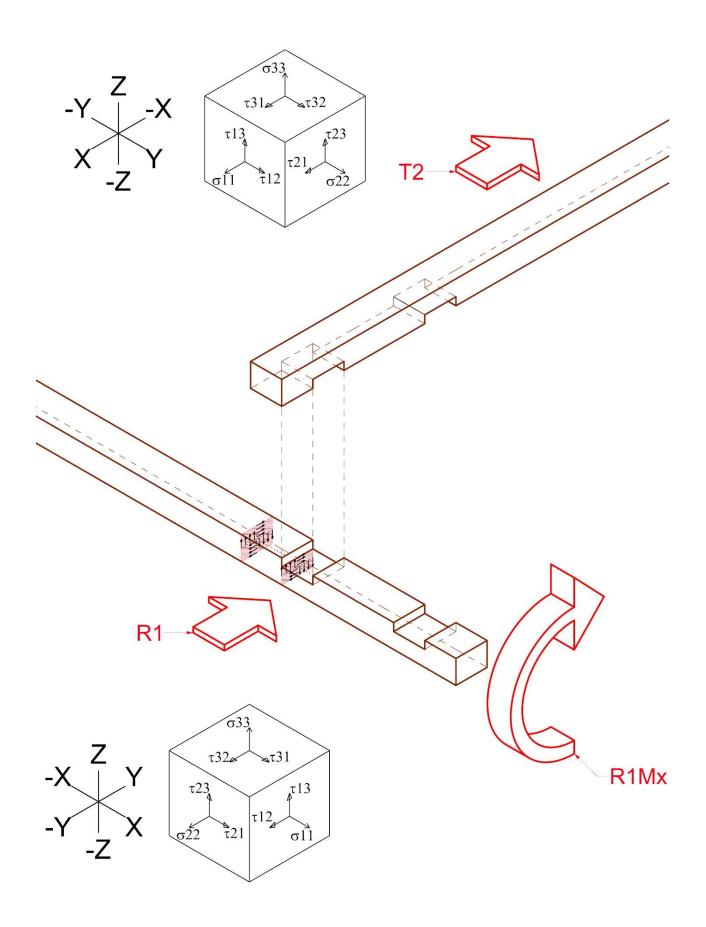


Figure 6-95 Rafter Head Bending moments: Torsion: Tangential stresses: Crossing rafters T2-R1

### **6.9.1** Possible actions along cross pieces

For further information they have been studied three possible actions along the cross pieces. The cross pieces may be subjected to compression, to tension or to a third case which has named "friction/inertia" and which subject the cross piece to a shear stress due to contrast of the timber beam and the inertia of the stones layer.

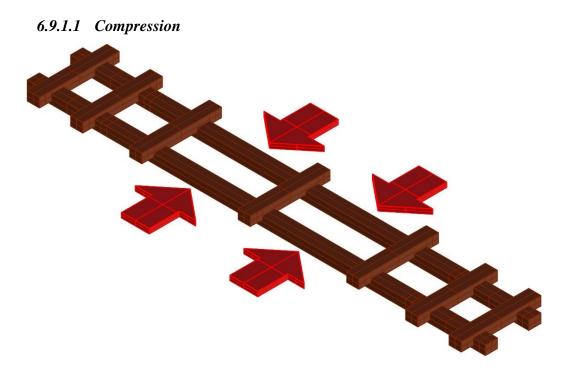


Figure 6-96 Cross Piece – Compression

As before the names are related to the timber elements and to the subjected stresses.

- CP0compression : Cross piece 0 = along the fibers compression : under compression A# stressed area
- RB90compression : Rafter Body 90 = perpendicular to the fibers compression : under compression A# stressed area
- RB90shear : Rafter Body 90 = perpendicular to the fibers shear : under shear force A# stressed area
- RB90shear: is computed in case of just pure shear force or shear force and bending moment acting at the same time.

CP0compression	A3	
N_0d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,0,d)	#VALUE!	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	#VALUE!	
N_(0d)max	62,33	kN

RB90compression	A3	
N_90d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	#VALUE!	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	#VALUE!	
N_(90d)max	16,50	kN

RB90shear	A5	
	with be	nding
V_90d	?	N
K_cr	0,67	
A_(net)	5025,00	mm^2
τ_(d)	#VALUE!	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	#VALUE!	
V_90d max	2,95	kN

RB90shear	A5	
V_90d	#VALUE!	Ν
A_(net)	7500,00	mm^2
τ_(d)	#VALUE!	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	#VALUE!	
V_90d max	4,40	kN

# 6.9.1.1.1 Axial stresses

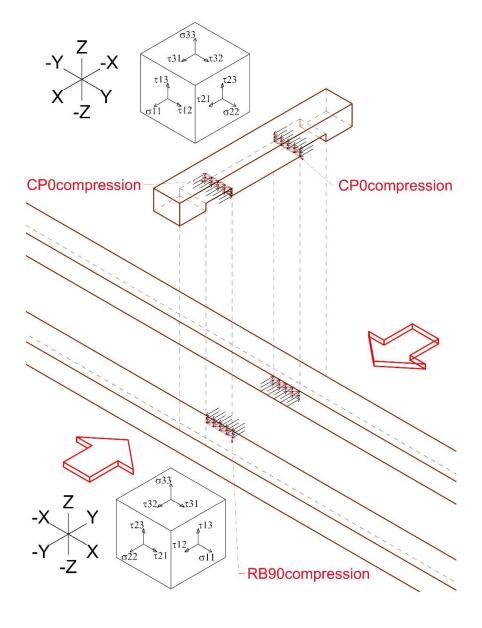


Figure 6-97 Cross Piece - Compression - Axial stresses

# **6.9.1.1.2** Tangential stresses

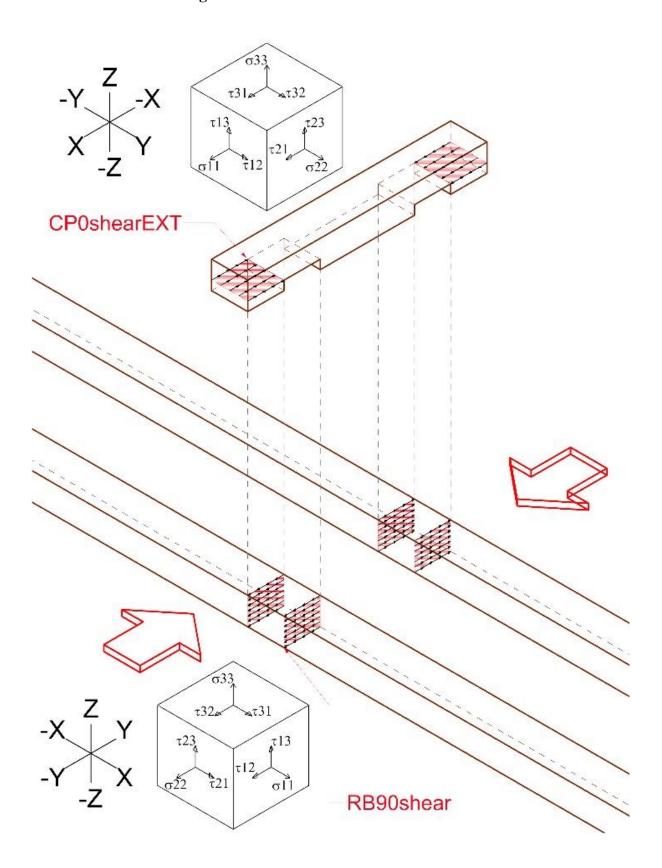


Figure 6-98 Cross Piece - Compression - Tangential stresses

#### 6.9.1.2 Tension

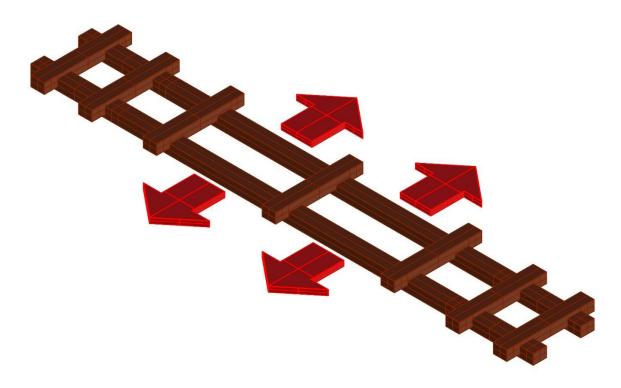


Figure 6-99 Cross Piece – Tension

- CP0tension: Cross piece 0 = along the fibers tension: under tension A# stressed area
- CP0compression : Cross piece 0 = along the fibers compression : under compression A# stressed area
- RB90compression : Rafter Body 90 = perpendicular to the fibers compression : under compression A# stressed area
- CP0shearEXT:Cross piece 0=along the fibers shear : under shear force EXT : external surface A# stressed area
- RB90shear : Rafter Body 90 = perpendicular to the fibers shear : under shear force A# stressed area
- CP0shearEXT and RB90shear: are computed in case of just pure shear force or shear force and bending moment acting at the same time.

CP0tension	A4	
N_0d	?	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
σ_(t,0,d)	#VALUE!	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	#VALUE!	
N_(0d)max	154,00	kN

CP0compression	A3	
N_0d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,0,d)	#VALUE!	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	#VALUE!	
N_(0d)max	62,33	kN

RB90compression	А3	
N_90d	?	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	#VALUE!	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	#VALUE!	
N_(90d)max	16,50	kN

CP0shearEXT	A2		
	with bend	ding	
V_0d	?	N	
K_cr	0,67		
A_(net)	6700,00	mm^2	
τ_(d)	#VALUE!	N/mm^2	
f_(v,d)	3,67	N/mm^2	
Verification	#VALUE!		
V_0d max	16,38	kN	

CP0shearEXT	A2	
V_0d	#VALUE!	N
A_(net)	10000,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	24,44	kN

RB90shear	A5	
wi	th bending	
V_90d	?	N
K_cr	0,67	
A_(net)	5025,00	mm^2
τ_(d)	#VALUE!	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	#VALUE!	
V_90d max	2,95	kN

RB90shear	A5	
V_90d	#VALUE!	N
A_(net)	7500,00	mm^2
τ_(d)	#VALUE!	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	#VALUE!	
V_90d max	4,40	kN

### 6.9.1.2.1 Axial stresses

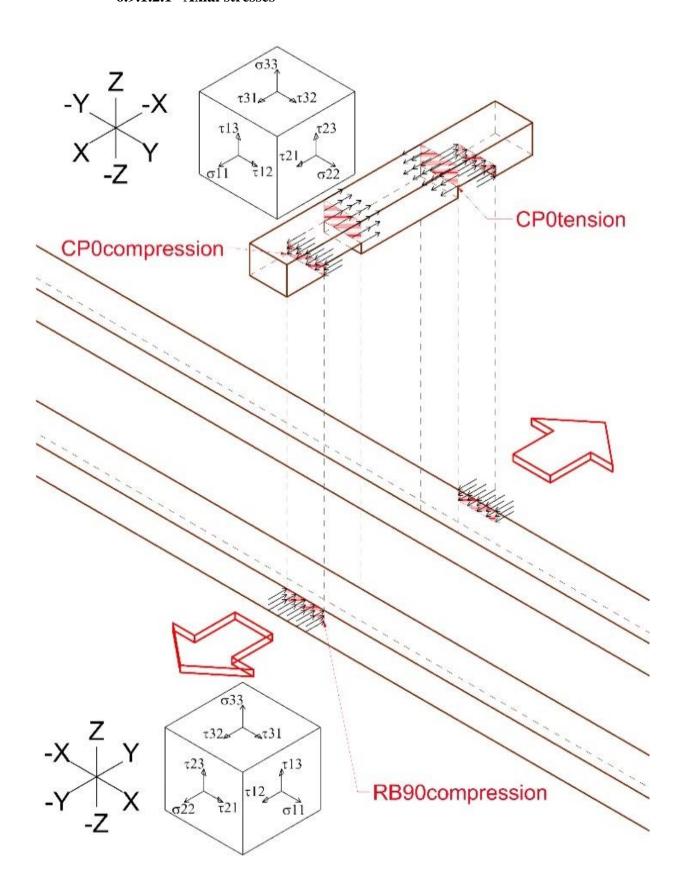


Figure 6-100 Cross Piece - Tension - Axial stresses

# **6.9.1.2.2** Tangential stresses

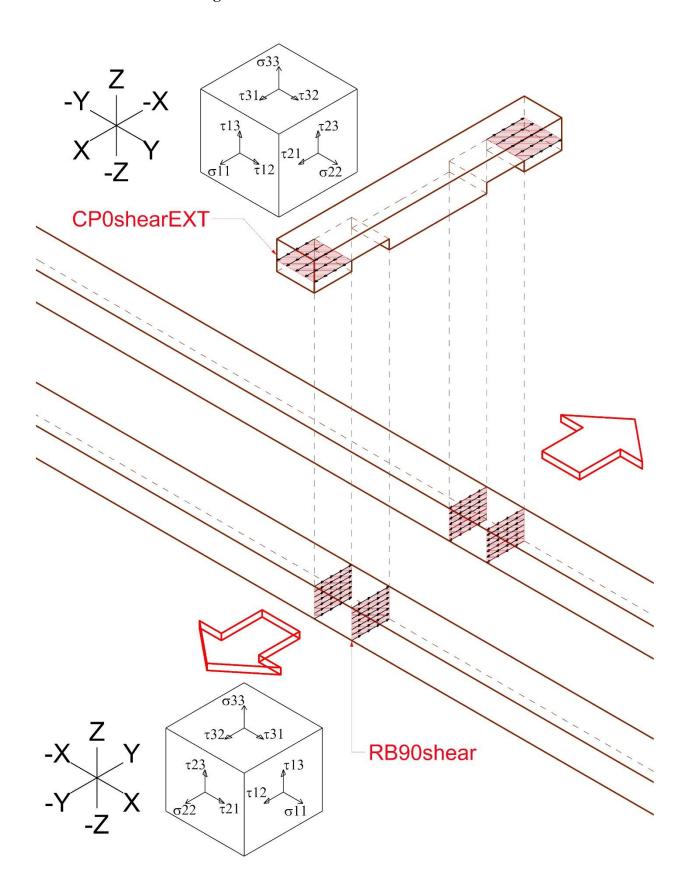


Figure 6-101 Cross Piece - Tension - Tangential stresses

#### 6.9.1.3 Friction/Inertia

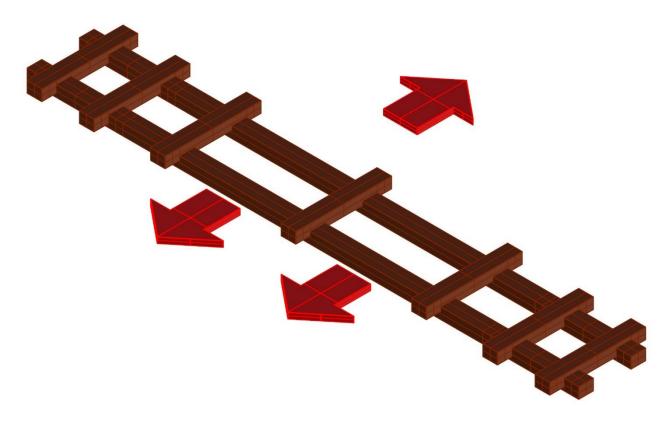


Figure 6-102 Cross Piece – Friction/Inertia

The names are reported in the previous paragraphs except for:

- CP0shearINT:Cross piece 0=along the fibers shear : under shear force INT : internal surface A# stressed area
- CP0shearINT, CP0shearEXT and RB90shear: are computed in case of just pure shear force or shear force and bending moment acting at the same time.

CP0shearINT	A1	
with b	ending	
V_0d	?	Ν
K_cr	0,67	
A_(net)	17420,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	42,58	kN

CP0shearINT	A1	
V_0d	#VALUE!	N
A_(net)	26000,00	mm^2
τ_(d)	#VALUE!	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	#VALUE!	
V_0d max	63,56	kN

### 6.9.1.3.1 Axial stresses

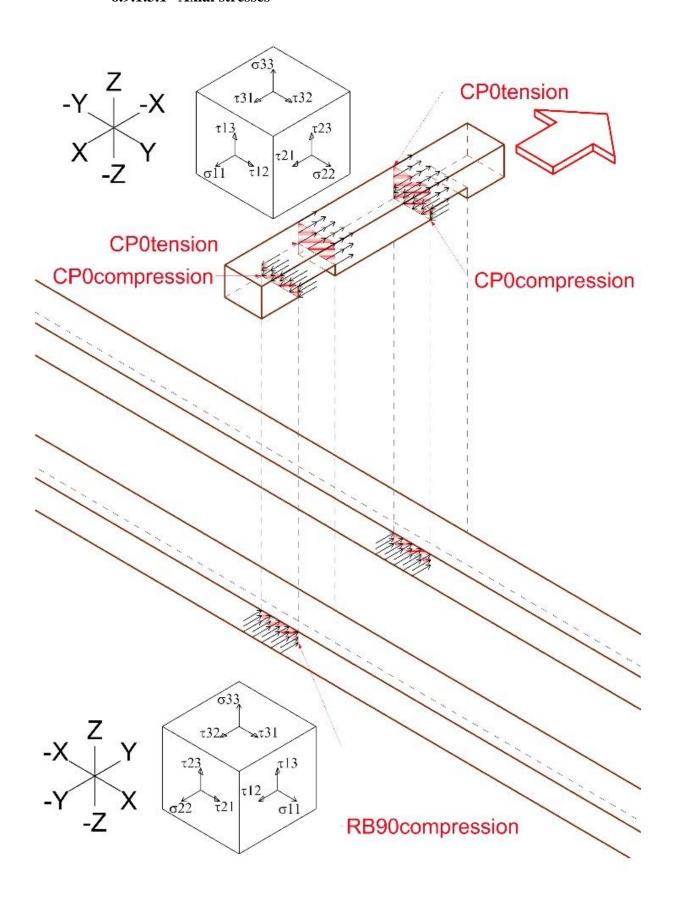


Figure 6-103 Cross Piece - Friction/Inertia - Axial stresses

# **6.9.1.3.2** Tangential stresses

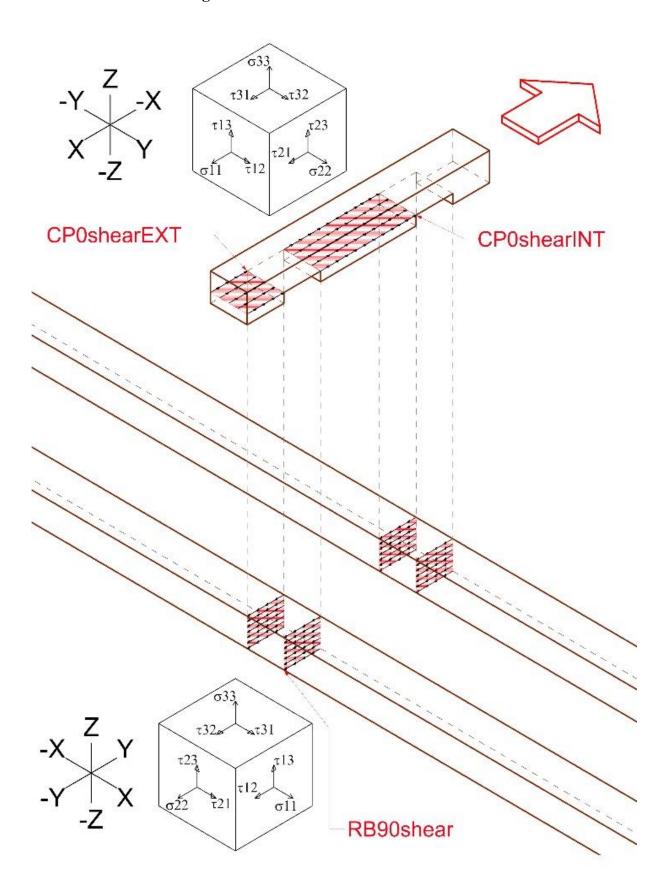


Figure 6-104 Cross Piece - Friction/Inertia - Tangential stresses

# **6.10Internal developed bending moments**

In the following paragraphs are reported the considerations about the "parasitic" bending moments developed internally to the timber elements. The parasitic bending moments are computed in correspondence of the notch and they come from equilibrium considerations. The positions of the parasitic bending moments have been reported before, look at the figure 6-88 to figure 6-95.

In the excel computations it has been used the following nomenclature:

- Body0mY: Body: section of the body 0 =along the fibers mY: bending moment around Y axis
- Body0mZ : Body : section of the body 0 =along the fibers mZ : bending moment around Z axis
- Body0mX : Body : section of the body 0 =along the fibers mX : bending moment around X axis
- Notch0mY : Notch = section of the notch − 0 =along the fibers − mY : bending moment around Z axis
- Notch0mZ: Notch = section of the notch -0 =along the fibers mZ: bending moment around Y axis
- Notch0mX : Notch = section of the notch 0 = along the fibers mX : bending moment around X axis

### 6.10.1 Mytf bending moment due to tension

Mytf: TENSIONS+FLEXION

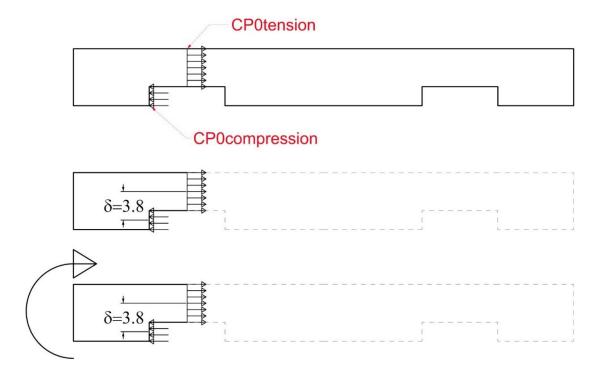


Figure 6-105 Parasitic Bending moment along Y axis due to tension and flexion

Depenings on the position and on the tension force applied the calculations have been based on the following formulas where  $\delta$  is the lever arm, or the distance between the two geometric centroids of the areas where the stresses are applied:

• M<sub>v1</sub> Parasitic Bending moment along Y axis due to tension and flexion on external notch

$$M_{v1} = T_1 * \delta = T_1 * 0.038m$$

• M<sub>y2</sub> Parasitic Bending moment along Y axis due to tension and flexion on internal notc

$$M_{v2} = T_2 * \delta = T_2 * 0.038m$$

In the figure 6-105 CP0tension and CP0compression are the stresses which develop T1 force.

### 6.10.2 Mycf bending moment due to compression

The same considerations have been adopted to the compression case.

M<sub>vcf</sub>: Parasitic Bending moment along Y axis due to tension and flexion

Mycf: COMPRESSION+FLEXION

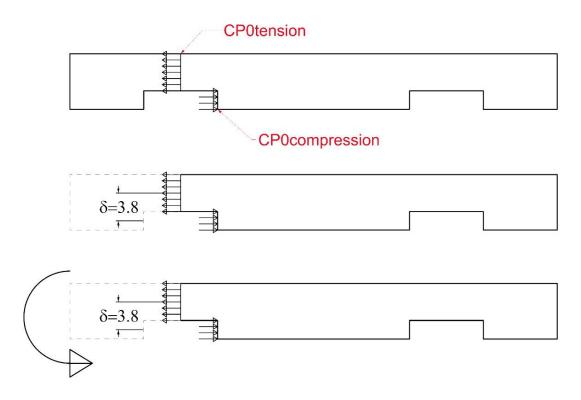


Figure 6-106 Parasitic Bending moment along Y axis due to compression and flexion

The same consideration of CP0tension and CP0compression are the same explained before.

# 6.10.3Mz bending moment



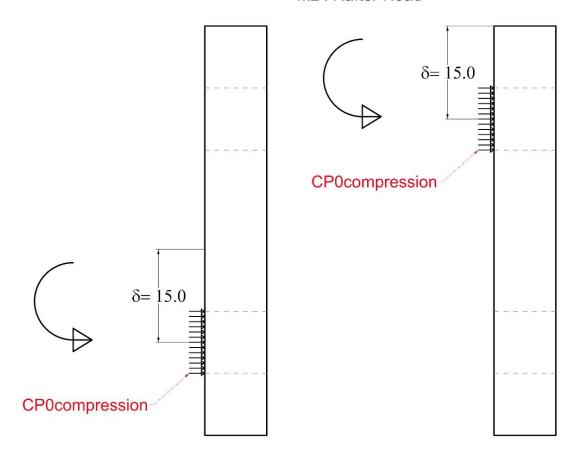


Figure 6-107 Parasitic Bending moment along Z axis due to compression and flexion

Depenings on the position and on the tension force applied the calculations have been based on the following formulas where  $\delta$  is the lever arm, or the distance between the two geometric centroids of the areas where the stresses are applied:

• M<sub>z1</sub> Parasitic Bending moment along Y axis due to tension and flexion

$$M_{Z1} = T_1 * \delta = T_1 * 0.015m$$

•  $M_{z2}$  Parasitic Bending moment along Y axis due to tension and flexion (considering the addition of  $M_{z1}$ )

$$M_{Z2} = M_{Z1} + T_2 * \delta = M_{Z1} + T_2 * 0.015m$$

In the figure 6-107 CP0compression is the stress which develop T1 and T2 forces.

#### 6.10.4 Torsional Mx

The torsional moments have been studied as the effect of compression of the rafter T, acting as a chain, on the R rafter. The instantaneous torsional moment develops on the notch. In the larger section of the rafter head the torsional moment will be verified as consequence of the moment on the notch. The lever arm  $\delta_{tor}$  is the distance between the centroid of the notch section and the point where the force is applied. This distance is 12mm.

#### 6.10.4.1 Mx Notch

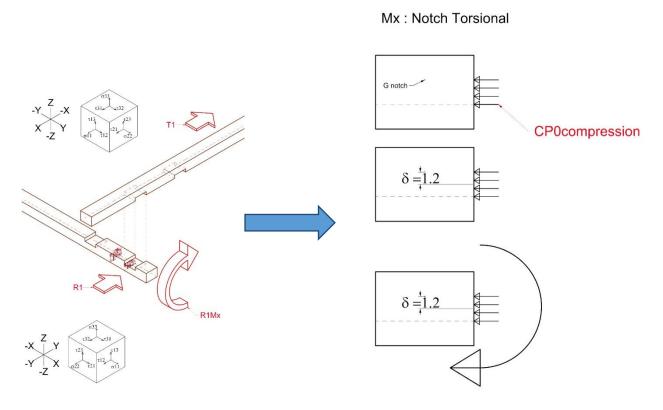


Figure 6-108 Parasitic Torsional Bending moment along X axis due to compression on the notch

•  $M_{x1}$  Parasitic Bending moment along X axis due to compression and flexion

$$M_{x1} = R_1 * \delta_{tor} = R_1 * 0.012m$$

• M<sub>x2</sub> Parasitic Bending moment along X axis due to compression and flexion

$$M_{x2} = R_2 * \delta_{tor} = R_2 * 0.012m$$

#### 6.10.4.2 Mx Rafter Head

In the case of of the body section the lever arm  $\delta_{tor}$ , the distance between the centroid of the notch section and the point where the force is applied, is null then the verification will be done just on the notch section which is the weaker.

# Mx: Rafter Head Torsional

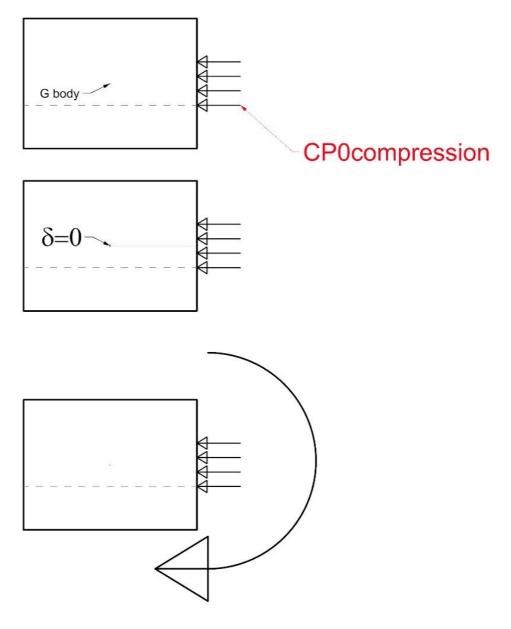


Figure 6-109 Parasitic Torsional Bending moment along X axis due to compression on the body section

# 6.10.4.3 Mx Rafter Body

For further information it has been studied the parasitic torsional bending moment due to the friction inertia case, and the proper lever arm which is  $\delta_{tor} = 0.025$ mm.

Mx: Rafter Body Torsional

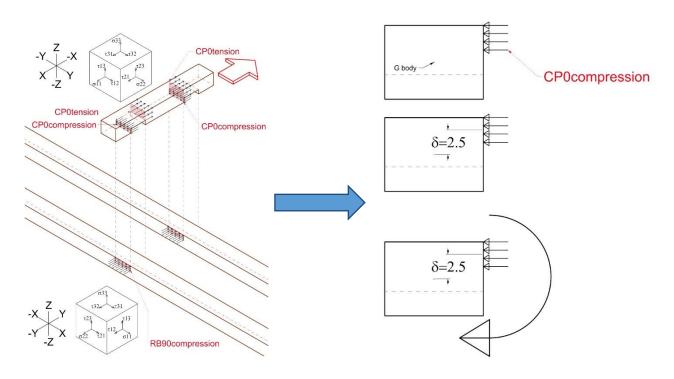


Figure 6-110 Figure 6 109 Parasitic Torsional Bending moment along X axis due to Friction/Inertia case on the body section

# 6.11 Keyed scarf joint

### 6.11.1 Geometry and resistance

The presence of a keyed scarf joint on a timber element under tension affects the resistance of the element, from laboratory tests it has been defined the reduction factor Rscarf., references from Strutture in legno .Piazza M., Tomasi R, Modena R.

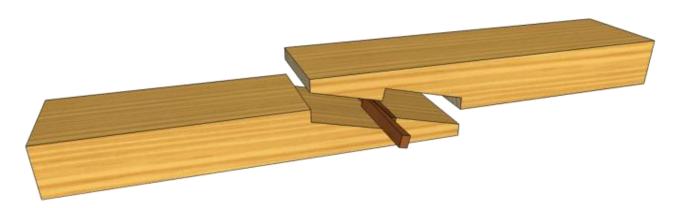


Figure 6-111 Kashmir Joint or Keyed Scarf Joint

From Laboratory test Rscarf has been defined as:

$$R_{scarf} = \frac{Stength\ of\ the\ element\ with\ joint}{Stength\ of\ the\ element} = 0.11$$

# 6.11.2Influence of keyed scarf joint on element subjected to tension

Recalling the Eurocode paragraph it has been reported the maximum tension force allowed on the timber rafter without and with a keyed scarf joint.

$$\sigma_{t,0,d} \leq f_{t,0,d} * R_{scarf}$$

 $\sigma_{t,0,d}$  is the design tensile stress along the grain

 $f_{t,0,d}$  is the design tensile strength along the grain

$$\sigma_{t0d} = \frac{N_{0d}}{A_{net,t}}$$

 $N_{0d}$  is the design axial force parallel to the grain

 $A_{net,t}$  is the net cross-sectional area perpendicular to the grain

 $A_{net,v}$  is the net shear area in the parallel to grain direction

 $R_{scarf} = 0.11$  is the reduction factor for the presence of the keyed scarf joint.

RB0tens		
N_0d	?	N
b	100,00	mm
h	75,00	mm
A_(net)	7500,00	mm
σ_(t,0,d)	0,00	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	Not defined	
N_(0d)max	231,00	kN

Influence of keyed scarf joint		
Verification Not defined		
N_(0d)max	25,41	kN

# 7 STATIC ANALYSIS

# 7.1 Aim of static analysis

The aim of the static analysis is to understand thebehavior of the structure subjected to the vertical loads due to the roof and the self-weight.

In order to be consistent with the preliminary considerations shown in the previous chapter The analysis has been conducted in two peculiar positions:

- the criticalbehavior in the middle of the stones layers;
- the critical behavior immediately below the timber band .

The wall has been decomposed in modular unit with dimensions 1,20m x 0,46m x 3,0 m as shown in Study Case chapter.

Each module has been decomposed in subparts.

# 7.2 Single modular unit

In order to study the single modular unit in the static analysis they have been defined the volumes of each element which composes each layer, they have been used the material properties described in the chapter 4 thus they have been obtained the weights of the elements.

# 7.2.1 Material properties

Table 16 Material Properties for single modular unit

SPECIFIC WEIGHT - DENSITY		
KN/m^3 Kg/m^3		
Ystone	26,86	2738,02
Y rubble stone	19,88	2026,14
Ytimber	8,97	914,75

e: void ratio	0,26
n : porosity (%)	0,20

Contact Surface		
L module	1,20	m
Width	0,46	m
Area	0,55	m^2

#### **7.2.2 Volumes**

The model have been drawn with the Rhinoceros 3D computer graphics and computer-aided design (CAD) application software. This software allowed to have information about the geometrical properties of the designed 3d models.

The single modular unit has been decomposed in its elementary parts as shown in the figure below.

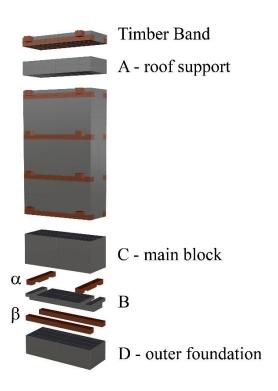


Figure 7-1 Single modular unit – Decomposed

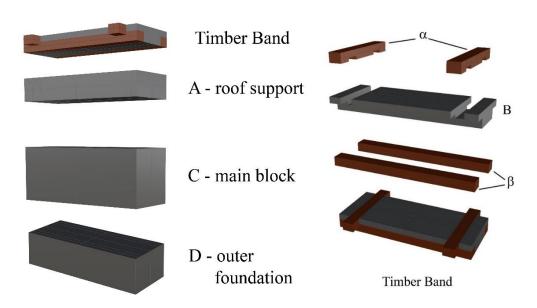


Figure 7-2 Single modular unit – Large -Decomposed

Table 17 Elementary parts of single modular unit - Volumes

VOLUMES	
	m^3
A - roof support	0,11040
B - rubble stone band	0,04510
C - main block	0,26220
D - outer foundation	0,16560
α - cross pieces	0,00900
β- timber rafter	0,00445

Timber Band		
volume percentage	%	
timber	37,36	
rubble stones	62,64	

### 7.2.3 Weights and stresses

They have been computed the weights of the elementary parts multiplying each volume for the proper specific weight, then the stresses immediately below each layers have been obtained dividing the mass over the contact surface.

Table 18 Elementary parts of single modular unit - Weights and stresses

WEIGHTS Timber Band		
	KN	Kg
Wtb	1,14	115,99

WEIGHTS A - roof		
support		
	KN	Kg
Wrs	2,19	223,69

WEIGHTS C - main block		
	KN	Kg
Wmb	5,21	531,25

WEIGHTS D - outer		
foundation		
	KN	Kg
Wof	3,29	335,53

Stress under Timber		
Band		
	KN/m^2	Kg/m^2
σtb	2,06	210,12

Stress under A - roof		
support		
	KN/m^2	Kg/m^2
σrs	3,98	405,23

Stress under C - main		
block		
	KN/m^2	Kg/m^2
σmb	9,44	962,41

Stress under C - main		
block		
	KN/m^2	Kg/m^2
σof	5,96	607,84

### **7.1 Roof**

The same approach used for the single modular unit has been adopted for the roof. In order to study the roof in the static analysis they have been defined the volumes of each element which composes each layer, they have been used the material properties described in the chapter 4 thus they have been obtained the weights of the elements.

### 7.1.1 Material properties

In the following table are reported some new properties of materials which have not been described before like the earth/clay and the twigs. For these two material the properties have been choosen roughly in respect to the others because the uncertainties of which are the exact material used in the chosen Nepal regions by the way still suggested by Arch Tom Schacher's guide.

Table 19 Material Properties for roof

SPECIFIC WEIGHT - DENSITY		
KN/m^3 Kg/m^3		
Ystone	26,86	2738,02
Ytimber	8,97	914,75
Yearth/clay	22,56	2300,00
Ytwigs	0,50	50,97

e : void ratio	0,00
n: porosity (%)	0,00

Contact Surface roof plane			
L roof 3,90 m			
Width	3,90	m	
Area	15,21	m^2	

Contact Surface roof module			
L roof 1,20 m			
Width	0,46	m	
Area	0,55	m^2	

#### **7.1.2 Volumes**

As for the single modular unit Rhinoceros 3D has been used to have information about the geometrical properties of the designed 3d models.

The heavy flat roof has been decomposed in its elementary parts as shown in the figure below.

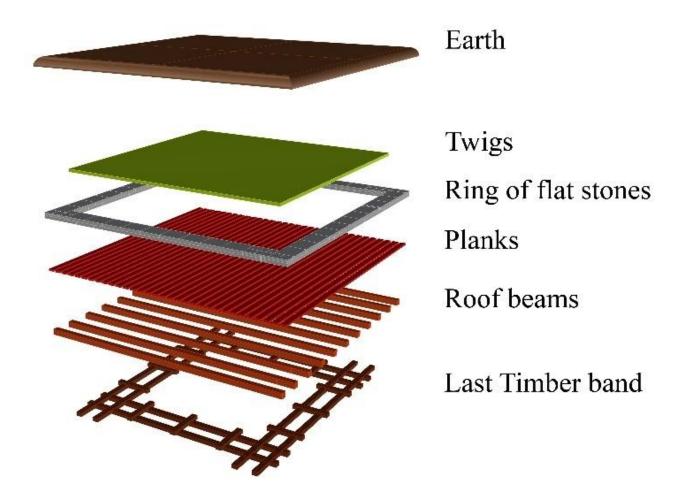


Figure 7-3 Roof – Decomposed

Table 20 Elementary parts of Roof - Volumes

VOLUMES	
	m^3
Earth/clay	3,58
Twigs	0,73
Ring of stones	0,49
Planks	0,42
Roof beams	0,48

### 7.1.3 Weights and linear load

They have been computed the weights of the elementary parts multiplying each volume for the proper specific weight, then the linear load immediately below each layers have been obtained dividing the mass over the roof area .

Table 21 Elementary parts of Roof - Weights and linear load

WEIGHTS Earth/clay		
	KN	Kg
Wearth	80,68	8224,63

WEIGHTS Twigs			
KN Kg			
Wtwigs	0,36	37,19	

WEIGHTS Ring of stones		
	KN	Kg
Wringstones	13,28	1353,68

Load Earth/clay		
KN/m Kg/m		
Wearth linear	5,30	540,74

Load Twigs		
KN/m Kg/m		
Wtwigs linear	0,02	2,44

Load Ring of stones		
	KN/m	Kg/m
Wringstones linear	0,87	89,00

WEIGHTS Planks			
KN Kg			
Wplanks	3,79	386,39	

WEIGHTS Roof beams			
	KN	Kg	
Wrb	4,34	442,74	

Load Planks			
KN/m Kg/m			
Wplanks linear	0,25	25,40	

Load Roof beams		
	KN/m	Kg/m
Wrb linear	0,29	29,11

Looking at the Arch. Tom Schacher's manual, the roof is sustained by the roof beams, which are supported by two walls. The total weight of the roof have been studied applied to the two wall perpendicular to the roof beams. Successively the roof weight has been counted to a single modular unit.

Table 22 Total weight of roof on wall and on module

Roof total weight	
Kooi totai weigit	

WEIGHT ROOF		
	KN	Kg
Wroof	102,46	10444,63

WEIGHT ON 1 WALL (3,6m)		
	KN	Kg
Wroof wall	51,23	5222,31

WEIGHT ON 1 MODULE (1,2m)		
	KN	Kg
Wroof module	17,08	1740,77

# 7.2 Normal Stresses

The normal stresses due to the vertical load have been studied in two peculiar position. The first case is the surfaces in the middle of each the stones layers, this because the rubble stones are not confined and thus it is a weaker position. The second case is the surfaces immediately below the timber beam, this because the contact surfaces between the stones is smaller due to the reduction factor obtained by the ratio between the areas described in the chapter 5 paragraph 1.

# 7.2.1 Normal Stress inside stones layer

Considering the hypothesis made about interfaces between stones, they have been computed the stresses on each layer as shown in the figure.

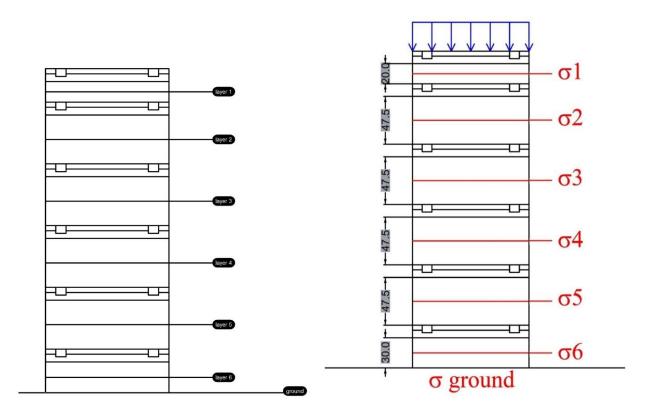


Figure 7-4 Normal stresses - Inside stones layers - Studied surfaces Figure 7-5 Normal stresses - Inside stones layers - Sigma Stresses

Table 23 Normal stresses - Inside stones layers - Sigma Stresses

Normal Stress						
	KN/m^2	Kg/m^2	Kg/cm^2	MPa		
σ1	34,99	3566,30	0,36	0,036		
σ2	43,75	4460,24	0,45	0,045		
σ3	55,26	5632,78	0,56	0,056		
σ4	66,76	6805,31	0,68	0,068		
σ5	78,26	7977,84	0,80	0,080		
σ6	88,03	8973,09	0,90	0,090		
$\sigma$ ground	91,01	9277,01	0,93	0,093		

# 7.2.2 Normal Stress below timber beam

They have been computed the stresses on each layer below the timber beam as shown in the picture.

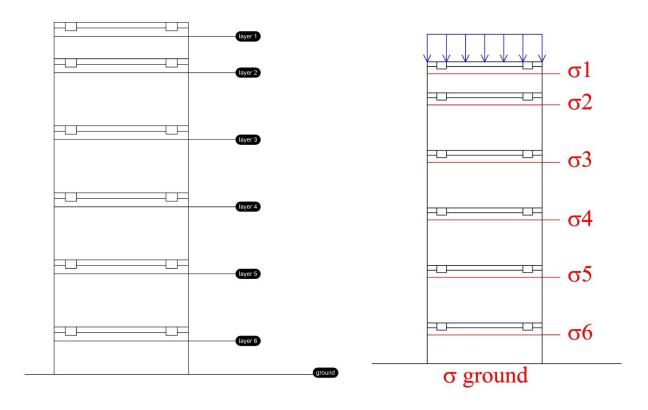


Figure 7-6 Normal stresses - Below timber beam - Sigma Stresses Figure 7-7 Normal stresses - Below timber beam - Studied surfaces

Table 24 Normal stresses - Below timber beam - Sigma Stresses

Normal Stress						
	KN/m^2	Kg/m^2	Kg/cm^2	MPa		
σ1	33,00	3363,69	0,34	0,034		
σ2	39,03	3979,04	0,40	0,040		
σ3	50,54	5151,57	0,52	0,052		
σ4	62,04	6324,10	0,63	0,063		
σ5	73,54	7496,64	0,75	0,075		
σ6	85,04	8669,17	0,87	0,087		
$\sigma$ ground	91,01	9277,01	0,93	0,093		

# 8 SEISMIC ANALYSIS IN PLANE

The effect of an earthquake on a structure is schematize as an horizontal action. The seismic actions are proportional to the mass of the structure and to the peak ground acceleration at the base of the structure. There are different schematization for representing the distribution of these forces on the building. In this thesis they have been selected the three case possible by hand calculation:

- Force applied at the top of the wall;
- Triangular lateral distribution over the height of the wall;
- Uniform lateral distribution over the weight of the wall.

As it has been explained in the initial chapters the bhatar system is composed by rubble stones masonry, this means that the structure is already cracked. The seismic analysis in plane is based on the application of the Barton's empirical model for the rock fill presented in the chapter 5. This non-linear model permits to have values of the friction coefficient  $\mu$ , which develops at each studied layer due to the vertical load and the self-weight. These friction coefficients have been used to described the resisting behavior of each layers subjected to the horizontal forces due to the seismic event.

# 8.1 Shear strength for rockfill with Barton empirical model

The Barton model is described by a function which described the shear stress in function of the normal stresses and others parameters described in the chapter 5. The peculiar thing is that the non-linearity of the function strictly depends on the normal stresses. For each studied surface the normal stresses have been taken from the static analysis described in the chapter 7. Thus they have been obtained values of friction coefficient  $\mu$  for each critical surface. They have been reported the corresponding friction angles.

# 8.1.1 Normal Stress and Coefficients of friction inside stones layer

Normal Stress			Barton empirical model for rockfill - Friction coefficient μ						
	KN/m^2	Kg/m^2	Kg/cm^2	MPa	бп (Mpa)	тр (Мра)	$\mu = \tau p/\sigma n$	rad	deg
σ1	34,99	3566,30	0,36	0,036	0,04	0,07	1,84	1,07	61,52
σ2	43,75	4460,24	0,45	0,045	0,04	0,08	1,77	1,06	60,58
σ3	55,26	5632,78	0,56	0,056	0,06	0,09	1,70	1,04	59,59
σ4	66,76	6805,31	0,68	0,068	0,07	0,11	1,65	1,03	58,78
σ5	78,26	7977,84	0,80	0,080	0,08	0,12	1,61	1,01	58,11
σ6	88,03	8973,09	0,90	0,090	0,09	0,14	1,58	1,01	57,60
$\sigma$ ground	91,01	9277,01	0,93	0,093	0,09	0,14	1,57	1,00	57,46

### 8.1.2 Normal Stress and Coefficients of friction below timber beam

Table 26 Normal Stress and Coefficients of friction below timber beam

Normal Stress			Barton empirical model for rockfill - Friction coefficient μ				-		
	KN/m^2	Kg/m^2	Kg/cm^2	MPa	бп (Mpa)	τp (Mpa)	$\mu = \tau p/\sigma n$	rad	deg
σ1	33,00	3363,69	0,34	0,034	0,034	0,06	1,86	1,08	61,73
σ2	39,03	3979,04	0,40	0,040	0,040	0,07	1,81	1,07	61,06
σ3	50,54	5151,57	0,52	0,052	0,052	0,09	1,73	1,05	59,97
σ4	62,04	6324,10	0,63	0,063	0,063	0,11	1,67	1,03	59,10
σ5	73,54	7496,64	0,75	0,075	0,075	0,12	1,62	1,02	58,37
σ6	85,04	8669,17	0,87	0,087	0,087	0,14	1,58	1,01	57,75
σ ground	91,01	9277,01	0,93	0,093	0,093	0,14	1,57	1,00	57,46

# 8.2 Seismic load multiplier

The aim of this analysis is to understand the point at which the structure shows a critical behavior, which may cause a failure. The failure happens when the the seismic force is larger than the resisting shear force of the wall. The seismic force has been defined with the term Fs and the resisting shear force with the term Rs. In order to study the problem it has been introduced the seismic load multiplier  $\alpha$  applied to the seismic force Fs. The subscript term i is to specify the considered layer.

# 8.2.1 Critical multiplier for inside stones layer

The analyzed layers are shown in the following figure.

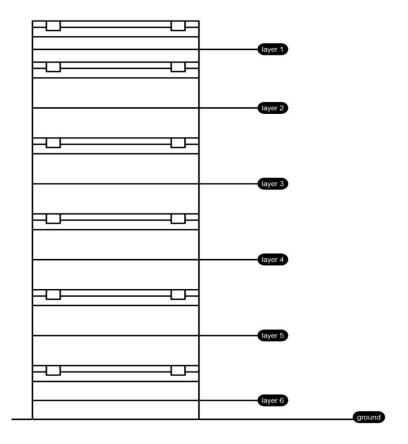


Figure 8-1 Analyzed layers for inside stones layer case

### 8.2.1.1 Force applied at the top of the wall



Figure 8-2 Force applied at the top of the wall

The system has a safe behavior if

Where the forces can be written as:

$$\alpha * \frac{Wtot}{2} * PGA < \mu i * Wi$$

 $\alpha$  is the load multiplier

Wtot is the total weight of the box structure and of the roof

PGA is the peak ground acceleration

 $\mu i$  is the friction coefficient of the i<sup>th</sup> layer

Wi is the pertinent weight on the i<sup>th</sup> layer

The critical load multiplier for each surfaces can be written as:

$$\alpha \leq \frac{(\mu i * W i)}{(W tot/2 * PGA)}$$

Table 27 Force applied at the top of the wall - Data

Total Weight of the box+roof						
		KN	Kg			
Wbox		313,91	31999,28			
Wroof		102,46		10444,63		
Wtot		416,37		42443,90		
Peak Ground Accelleration						
PGA 1 g						

In the following table are reported values of the limit values of load multiplier for a safe behavior,

$$\alpha \ such that \qquad \frac{Fs}{Rs} < 1$$

The resisting shear forces Rsi have been obtained multiplying the normal force acting on the layer time the pertinent friction coefficient.

Table 28 Safe limit multipliers - Force applied at the top of the wall -inside stones layer case

	Wi = Ni	$Rsi = \tau i = Ni*\mu i$	α <
	kN	kN	
layer1	57,94	106,81	0,51
layer2	72,46	128,49	0,62
layer3	91,51	155,91	0,75
layer4	110,55	182,43	0,88
layer5	129,60	208,26	1,00
layer6	145,77	229,71	1,10
ground	150,71	236,19	1,13

### 8.2.1.2 Triangular lateral distribution over the height of the wall

In the case of a triangular lateral distribution they have been defined the distributions factors depending on the mass and the height of the analyzed layer.

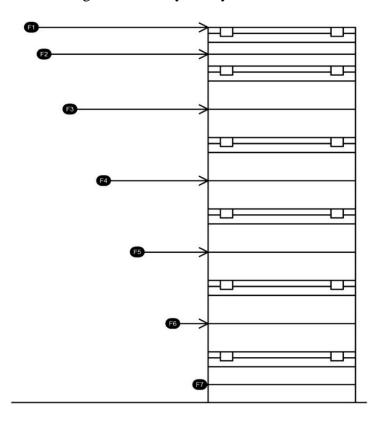


Figure 8-3 Triangular lateral distribution over the height of the wall for inside stones layer

The seismic force Fs has been computed as in the first case, thus it has been distributed multiplying by the distribution factor  $\beta$ . The distribution factors have been obtained using the following formula:

$$\beta_{j} = \frac{W_{j} * h_{j}}{\sum_{i=1}^{N} W_{i} * h_{i} + W_{roof} * H}$$

Where

 $\beta_j$ : is the distribution factor corresponding to the analyzed layer

 $W_i$ : is the weight corresponding to the analyzed layer

 $h_i$ : is the height corresponding to the analyzed layer

 $\sum_{i=1}^{N} W_i * h_i + W_{roof} * H$ : is the summation of all the masses times the corresponding heights

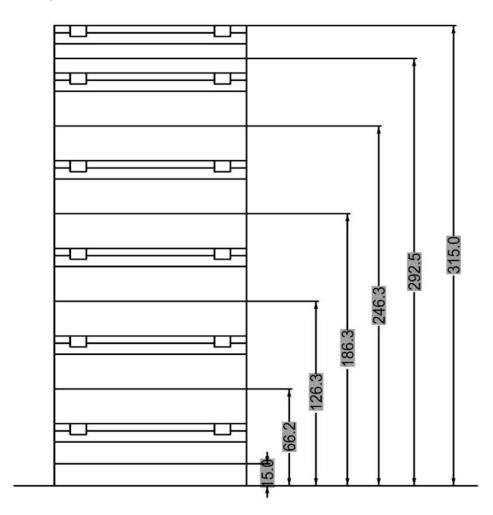


Figure 8-4 Triangular lateral distribution over the height of the wall for inside stones layer- Heights (cm)

The force Fj applied to each layer has been obtained using the following formula:

$$F_j = F_s * \beta_j = F_s * \frac{W_j * h_j}{\sum_{i=1}^{N} W_i * h_i + W_{roof} * H}$$

	Seismic Force	
Fs= Wtot/2*PGA	201,24	kN

Table 29 Triangular distribution of the forces - inside stones layer case

	Distribution of the forces					
	W	Height of Force : hi	Wi * hi			
Wi	kN	m	kN*m			
Wroof	17,08	3,15	53,79			
W1	2,23	3,15	7,04			
W2	4,84	2,93	14,16			
W3	6,35	2,46	15,64			
W4	6,35	1,86	11,83			
W5	6,35	1,26	8,02			
W6	5,39	0,66	3,57			
W7	1,65	0,15	0,25			
		Distribution factors and Forces				
Fj		Bj .	Fj=Fs*βj			
		/	kN			
F1	0,	53	110,81			
F2	0,	12	25,79			
F3	0,	14	28,49			
F4	0,	10	21,55			
F5		07	14,61			
F6	0.	03	6,50			
F7	0,0	002	0,45			

The system has a safe behavior if

Where the forces can be written as:

$$\alpha * PGA * \frac{1}{2} * (W_T + W_{roof}) * \sum_{i=1}^{j} \frac{W_j * h_j}{\sum_{i=1}^{N} W_i * h_i + W_{roof} * H} \le \mu s_j * Nj$$

 $\alpha$  is the load multiplier

Wtot is the total weight of the box structure and of the roof

PGA is the peak ground acceleration

 $\mu i$  is the friction coefficient of the i<sup>th</sup> layer

Wi is the pertinent weight on the i<sup>th</sup> layer

 $\mu s_j$  is the friction coefficient obtained by the Barton models for rockfill corresponding to the analyzed layer

Nj is the pertinent normal force acting on the on the analyzed layer

The critical load multiplier for each surfaces can be written as:

$$\alpha \leq \frac{\mu s_{j} * Nj}{PGA * \frac{1}{2} * \left(W_{T} + W_{roof}\right) * \sum_{i=1}^{j} \frac{W_{j} * h_{j}}{\sum_{i=1}^{N} W_{i} * h_{i} + W_{roof} * H}}$$

In the following table are reported values of the limit values of load multiplier for a safe behavior,

$$\alpha \, such \, that \, \frac{Fs}{Rs} < 1$$

Table 30 Safe limit multipliers- Triangular lateral distribution-inside stones layer case

	Computing α such tha	t Fs< Rshear	
Layer	Nj	Rshear = $N*\mu s$	α <
	kN	kN	
Layer1	57,94	106,81	0,96
Layer2	72,46	128,49	0,94
Layer3	91,51	155,91	0,94
Layer4	110,55	182,43	0,98
Layer5	129,60	208,26	1,03
Layer6	145,77	229,71	1,11
Layer_ground	150,71	236,19	1,13

#### 8.2.1.1 Uniform lateral distribution over the height of the wall

In the case of a triangular lateral distribution they have been defined the distributions factors depending on the mass and the height of the analyzed layer.

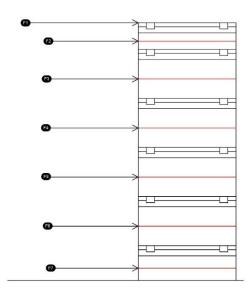


Figure 8-5 Uniform lateral distribution over the height of the wall for inside stones layer

The seismic force Fs has been computed as in the first case, thus it has been distributed multiplying by the distribution factor  $\beta$ . The distribution factors have been obtained using the following formula:

$$\beta_j = \frac{W_j}{\sum_{i=1}^N W_i + W_{roof}}$$

Where

 $\beta_j$  : is the distribution factor corresponding to the analyzed layer

 $W_i$ : is the weight corresponding to the analyzed layer

 $\sum_{i=1}^{N} W_i + W_{roof}$ : is the summation of all the masses

The force Fj applied to each layer has been obtained using the following formula:

$$F_{j} = F_{s} * \beta_{j} = F_{s} * \frac{W_{j}}{\sum_{i=1}^{N} W_{i} + W_{roof}}$$

Seismic Force				
Fs= Wtot/2*PGA	201,24	kN		

Table 31 Uniform Distribution of the forces - inside stones layer case

Distribution of the forces		
	W	
Wi	kN	
Wroof	17,08	
W1	2,23	
W2	4,84	
W3	6,35	
W4	6,35	
W5	6,35	
W6	5,39	
W7	1,65	
Sum	50,24	

Distribution factors and Forces		
Fj	βj	Fj=Fs*βj (kN)
	/	
F1	0,38	80,03
F2	0,10	20,06
F3	0,13	26,31
F4	0,13	26,31
F5	0,13	26,31
F6	0,11	22,33
F7	0,03	6,82

The system has a safe behavior if

Where the forces can be written as:

$$\alpha * PGA * \frac{1}{2} * (W_T + W_{roof}) * \sum_{i=1}^{j} \frac{W_j}{\sum_{i=1}^{N} W_i + W_{roof}} \le \mu s_j * Nj$$

 $\alpha$  is the load multiplier

Wtot is the total weight of the box structure and of the roof

PGA is the peak ground acceleration

μi is the friction coefficient of the i<sup>th</sup> layer

Wi is the pertinent weight on the i<sup>th</sup> layer

 $\mu s_j$  is the friction coefficient obtained by the Barton models for rockfill corresponding to the analyzed layer

Nj is the pertinent normal force acting on the on the analyzed layer

The critical load multiplier for each surfaces can be written as:

$$\alpha \leq \frac{\mu s_{j} * Nj}{PGA * \frac{1}{2} * (W_{T} + W_{roof}) * \sum_{i=1}^{j} \frac{W_{j}}{\sum_{i=1}^{N} W_{i} + W_{roof}}}$$

In the following table are reported values of the limit values of load multiplier for a safe behavior,

 $\alpha \text{ such that } \frac{Fs}{Rs} < 1$ 

Table 32 Safe limit multipliers- Triangular lateral distribution-inside stones layer case

Computing α such that Fs< Rshear					
Layer	Layer Nj Rshear = $N*\mu s$				
	kN	kN			
Layer1	57,94	106,81	1,33		
Layer2	72,46	128,49	1,28		
Layer3	91,51	155,91	1,23		
Layer4	110,55	182,43	1,19		
Layer5	129,60	208,26	1,16		
Layer6	145,77	229,71	1,14		
Layer_ground	150,71	236,19	1,13		

### 8.2.2 Critical Multiplier below the timber band

Following the same procedure described in the previous paragraph, they have been analyzed the surfaces on the layers immediately below the timber bands. The positions of the new layers are shown in the figure below.

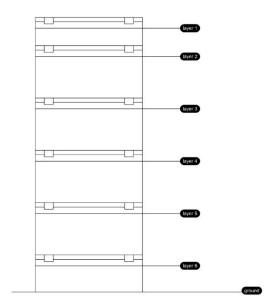


Figure 8-6 Analyzed layers for the below timber bands case

### 8.2.2.1 Force applied at the top of the wall

The resisting shear forces Rsi have been obtained multiplying the normal force acting on the layer by the pertinent friction coefficient and by the reduction factor  $\xi$ .

In the following table are reported values of the limit values of load multiplier for a safe behavior,

$$\alpha$$
 such that  $\frac{Fs}{Rs} < 1$ 

Table 33 Safe limit multipliers - Force applied at the top of the wall -below timber band case

	337: NI:	D -:: _ N1*: **		
	Wi = Ni	$Rsi = \tau i = N*\mu i *\xi$		α <
	kN	kN		
layer1	54	,64	57,44	0,28
layer2	64	,64	66,08	0,32
layer3	83	,69	81,83	0,39
layer4	102	,74	97,01	0,47
layer5	121	,79	111,76	0,54
layer6	140	,83	126,16	0,61
ground	150	,71	133,50	0,64

### 8.2.2.2 Triangular lateral distribution over the height of the wall

Following the same procedure described in the previous paragraph.

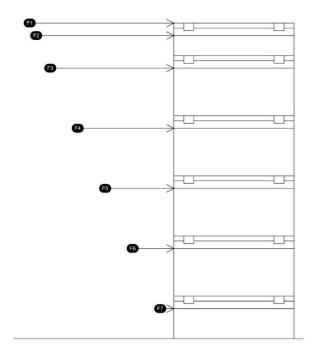
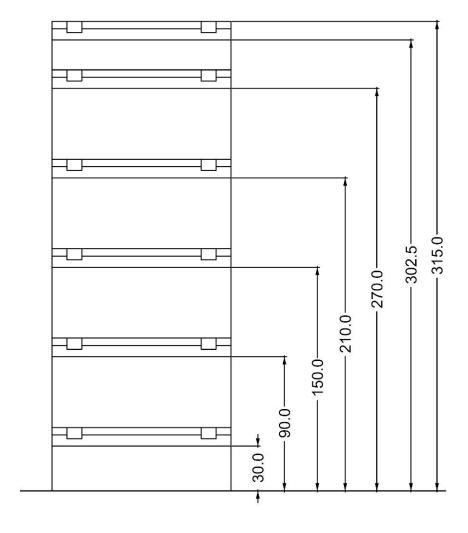


Figure 8-7 Triangular lateral distribution over the height of the wall for below timber bands



Seismic Force		
Fs= Wtot/2*PGA	201,24	kN

Table 34 Triangular distribution of the forces – below the timber bands case

	Distribution of the forces			
	Weight Heigth of Force : hi Wi * hi			
Wi	kN	m	kN*m	
Wroof	17,08	3,150	53,79	
W1	1,14	3,150	3,58	
W2	3,33	3,025	10,08	
W3	6,35	2,700	17,14	
W4	6,35	2,100	13,33	
W5	6,35	1,500	9,52	
W6	6,35	0,900	5,71	
W7	3,29	0,300	0,99	

	Distribution factors and Forces		
Fj	Fj βj Fj=Fs*βj		
	/		
F1	0,50	104,63	
F2	0,09	18,38	
F3	0,15	31,26	
F4	0,12	24,32	
F5	0,08	17,37	
F6	0,05	10,42	
F7	0,01	1,80	

In the following table are reported values of the limit values of load multiplier for a safe behavior,

$$\alpha \, such \, that \qquad \frac{Fs}{Rs} < 1$$

Table 35 Safe limit multipliers- Triangular lateral distribution- below timber bands case

Computing α such that Fs< Rshear			
Layer	Layer Nj Rshear = $N*\mu s*\xi$		
	kN	kN	
Layer1	54,64	57,44	0,55
Layer2	64,64	66,08	0,54
Layer3	83,69	81,83	0,53
Layer4	102,74	97,01	0,54
Layer5	121,79	111,76	0,57
Layer6	140,83	126,16	0,61
Layer_ground	150,71	133,50	0,64

# 8.2.2.1 Uniform lateral distribution over the height of the wall

Following the same procedure described in the previous paragraph.

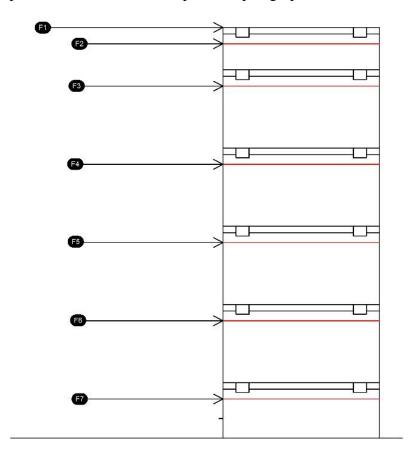


Figure 8-9 Triangular lateral distribution over the height of the wall for below timber bands

Seismic Force		
Fs= Wtot/2*PGA	201,24	kN

Table 36 Uniform distribution of the forces – below the thimber bands case

Distribution of the forces		
	Weight	
Wi	kN	
Wroof	17,08	
W1	1,14	
W2	3,33	
W3	6,35	
W4	6,35	
W5	6,35	
W6	6,35	
W7	3,29	
Sum	50,24	

Distribution factors and Forces		
Fj	βj	Fj=Fs*βj
	/	kN
F1	0,36	75,49
F2	0,07	13,81
F3	0,13	26,31
F4	0,13	26,31
F5	0,13	26,31
F6	0,13	26,31
F7	0,07	13,64

In the following table are reported values of the limit values of load multiplier for a safe behavior,

$$\alpha \, such \, that \qquad \frac{Fs}{Rs} < 1$$

Table~37~Safe~limit~multipliers-~Triangular~lateral~distribution-~below~timber~bands~case

Computing α such that Fs< Rshear				
Layer	Layer Nj Rshear = $N^*\mu s^*\xi$			
	kN	kN		
Layer1	54,64	57,44	0,55	
Layer2	64,64	66,08	0,54	
Layer3	83,69	81,83	0,53	
Layer4	102,74	97,01	0,54	
Layer5	121,79	111,76	0,57	
Layer6	140,83	126,16	0,61	
Layer_ground	150,71	133,50	0,64	

### 8.2.1 Conclusions on seismic analysis in-plane

Recalling the results, they have been identified the critical layers for the in plane seismic analysis. The color red identified the critical load multiplier smaller than the Nepal peak ground acceleration, which is 0,5 g.

Table 38 Summary of results for the in-plane seismic analysis

	Critical Multiplier for in stones layer case	Critical Multiplier below timber band case	w the	
doa	Layer	α <	Layer	α <
ne (	layer1	0,51	layer1	0,28
all all	layer2	0,62	layer2	0,32
Force applied at the top of the wall	layer3	0,75	layer3	0,39
plie	layer4	0,88	layer4	0,47
ap]	layer5	1,00	layer5	0,54
rce	layer6	1,10	layer6	0,61
F01	Layer_ground/Foundation	1,13	Layer_ground/Foundation	0,64
e e	Layer	α <	Layer	α <
ral all	Layer1	0,96	Layer1	0,55
ate ver e we	Layer2	0,94	Layer2	0,54
Triangular lateral distribution over the height of the wall	Layer3	0,94	Layer3	0,53
gulk ntio t of	Layer4	0,98	Layer4	0,54
ang ibr	Layer5	1,03	Layer5	0,57
Tri istr hei	Layer6	1,11	Layer6	0,61
7	Layer_ground/Foundation	1,13	Layer_ground/Foundation	0,64
e	Layer	α <	Layer	α <
r th	Layer1	1,33	Layer1	0,76
iter ovej e w	Layer2	1,28	Layer2	0,74
n la on c	Layer3	1,23	Layer3	0,71
orn atio	Layer4	1,19	Layer4	0,68
Uniform lateral distribution over the height of the wall	Layer5	1,16	Layer5	0,66
U istr he	Layer6	1,14	Layer6	0,65
p	Layer_ground/Foundation	1,13	Layer_ground/Foundation	0,64

### 8.2.1.1 Critical Multiplier for inside stones layer case

Considering the Nepal peak ground acceleration given PGA = 0.5 g the seismic force results smaller than resisting shear force in both the sliding configurations.

The most critical one is the first configuration of «Force applied at the top of the wall» at the roof level but still on the safe side with a critical multiplier  $\alpha = 0.51$  thus resisting to Nepal PGA.

### 8.2.1.2 Critical Multiplier below the timber band case

Considering the Nepal peak ground acceleration given PGA = 0.5 g the behavior shown is different in the sliding configurations examined .

The most critical one is the first configuration of «Force applied at the top of the wall", which shows problems at the following layers:

- Layer 1
- Layer 2
- Layer 3
- Layer 4

As shown in the picture below

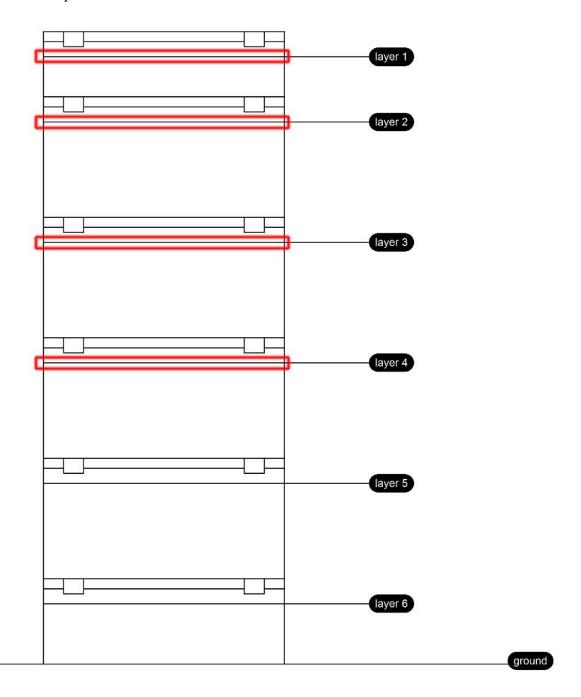


Figure 8-10 Critical layers for the in-plane seismic analysis

# 8.2.1.3 Safety Factor

In-plane analysis have been conducted without the use of any safty factors. The designer can only analyze for what he knows to be true. Philosophically and practically speaking it is impossible to know everything. Using a safety factor is an admission to this ignorance. Thus it has been applied to the results a safety factor  $\gamma_b = 1.5$  in order to amplify the seismic actions.

Table 39 Summary of results for the in-plane seismic analysis Reduced by Safety factor  $\gamma_b = 1.5$ 

	Critical Multiplier for inside layer case	stones	Critical Multiplier below the band case	timber
do	Layer	α <	Layer	α <
le t	layer1	0,34	layer1	0,18
	layer2	0,41	layer2	0,21
Force applied at the top of the wall	layer3	0,50	layer3	0,26
plie	layer4	0,58	layer4	0,31
ap]	layer5	0,67	layer5	0,36
rce	layer6	0,74	layer6	0,40
Fol	Layer_ground/Foundation	0,76	Layer_ground/Foundation	0.76
ى ن	Layer	α <	Layer	α <
ral all	Layer1	0,64	Layer1	0,37
ate ver	Layer2	0,63	Layer2	0,36
Triangular lateral distribution over the height of the wall	Layer3	0,63	Layer3	0,35
guls itio t of	Layer4	0,65	Layer4	0,36
ang ribu ight	Layer5	0,69	Layer5	0,38
Tri istr hei	Layer6	0,74	Layer6	0,41
۵	Layer_ground/Foundation	0,76	Layer_ground/Foundation	0.76
e	Layer	α <	Layer	α <
all all	Layer1	0,89	Layer1	0,51
ter vei	Layer2	0,86	Layer2	0,49
l la on o th	Layer3	0,82	Layer3	0,47
Uniform lateral distribution over the height of the wall	Layer4	0,80	Layer4	0,46
nife ribu igh	Layer5	0,78	Layer5	0,44
U istr he	Layer6	0,76	Layer6	0,43
р	Layer_ground/Foundation	0,76	Layer_ground/Foundation	0,43

# 9 SEISMIC ANALYSIS OUT OF PLANE – OVERTURNING RIGID BEHAVIOR

The failure mechanisms due to the seismic action may happen in the perpendicular direction in respect to the length of the wall. In this chapter will be explained what is defined as the overturning mechanism of the wall.

# 9.1 Hypothesis of rigid body behavior

The wall has been considered as it was composed by rigid blocks, which may overturn around ideal hinges set in different positions over the height of the wall. The figure below is recalled from the chapter 6 where they were explained the timber tie-beam chain activation.

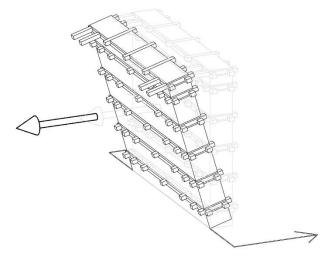


Figure 9-1 Overturning mechanism – example scheme

The stabilizing traction is equally divided between the to parallel tie-timber beam chains which are composed by 2 roof rafters or 2 rafters.

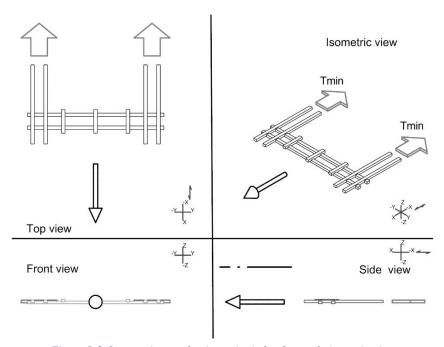


Figure 9-2 Overturning mechanism - tie-timber beam chains activation

# 9.2 Rigid body over rigid soil by Equilibrium – Tmin as function of $\alpha$ load multiplier - Hand calculation

Using the equilibrium method, they have been analyzed the horizontal and rotational equilibrium. They have been obtained equations in function of the load multiplier in order to get the values of the applied to the rafters. These tensions are the limit values needed to avoid the failures.

The main data used for this aim are reported below.

Table 40 Masses of each analyzed layer

Mass for each Force 3 module					
	KN	Kg			
Wroof module	51,23	5222,31			
W1	6,70	683,48			
W2	14,52	1480,36			
W3	19,05	1941,72			
W4	19,05	1941,72			
W5	19,05	1941,72			
W6	16,17	1648,13			
W7	4,94	503,29			

Table 41 Total weight and mass of the wall composed by 3 single modular unit

Total Weight 1 wall (3 module)				
KN Kg				
Wtot	150,71	15362,73		

Table 42 Heights of the considered rafters

Height of Force : hi					
Hi	m				
Hroof	3,10				
H1	3,10				
H2	2,78				
Н3	2,18				
H4	1,58				
Н5	0,98				
Н6	0,38				
Htchain	3,0875				

Where Htchain is the height from the ground of the centroid of the roof rafter beam.

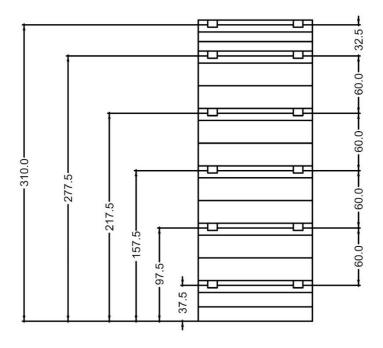


Figure 9-3 Heights of the rafters and distances between the timber beams bands

# 9.2.1 Horizontal equilibrium

The horizontal equilibrium has been computed for completeness and it is unlikely. It is the only one case where all the rafters have been considered as working at the same time. For the analysis the tensions defined in the drawing as T1 until T6 has the identic values then it will be named with just a single name like Tmin.

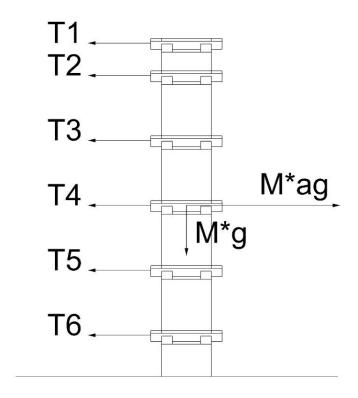


Figure 9-4 Horizontal equilibrium – equilibrium method

#### 9.2.1.1 Minimum Traction dependent on a

Equations for computing the Minimum tensions as function of  $\alpha$  load multiplier are shown below:

$$T_{min} * n = M_{tot} * a_g * \alpha$$

$$a_g = g$$
 then

$$T_{min} = \frac{M_{tot} * a_g}{n} * \alpha = \frac{W_{tot}}{n} * \alpha$$

Where

 $T_{min}$  is the minimum tension allowed for resisting to the seismic action

n is the total number of tie timber beams, for 3,6 m length wall = 12 (Each timber tie-beam is composed by 2 rafters)

 $M_{tot}$  is the total mass of the 3,6 m length wall

 $a_q$  is the seismic acceleration in g

g is the gravity acceleration constant =  $9.81 \text{ m/s}^2$ 

Results are reported in the following table.

Table 43 Horizontal equilibrium - minimum tensions

Minimum Tension dependent on α				
α	Tmin (kN)			
0	0,00			
0,1	1,26			
0,2	2,51			
0,3	3,77			
0,4	5,02			
0,5	6,28			
0,6	7,54			
0,7	8,79			
0,8	10,05			
0,9	11,30			
1	12,56			

# 9.2.2 Rotational equilibrium

The rotational equilibrium has been computed by the equilibrium method, in this case they have been considered just the two tie timber beam at the roof level, as shown in the figure below.

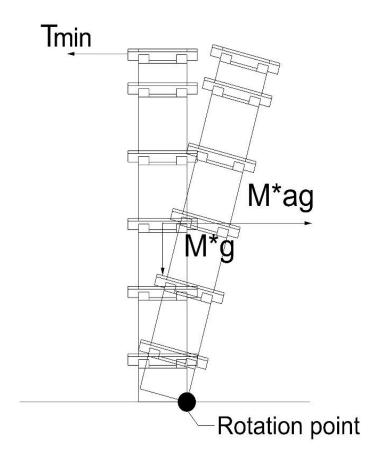


Figure 9-5 Rotational equilibrium – equilibrium method

### 9.2.2.1 Minimum Traction dependent on a

Equations for computing the minimum tensions as function of  $\alpha$  load multiplier are shown below:

$$2 * T_{min} * Ht_{chain} = M_{tot} * a_g * \frac{H}{2} * \alpha - M_{tot} * g * \frac{B}{2}$$

$$a_g = g$$

$$T_{min} = \frac{M_{tot} * a_g * \frac{H}{2} * \alpha - M_{tot} * g * \frac{B}{2}}{2 * Ht_{chain}}$$

Where the new parameters in respect to the horizontal equilibrium are :

H is the height of the centroid of the section of the wall

B is the horizontal component of the centroid of the section of the wall

Table 44 Centroid of the section of the wall - data

H centroid	1,583	m
B centroid	0,23	m

Table 45 Rotational equilibrium - minimum tensions

Minimum Tension dependent on $\alpha$				
α	Tmin (kN)			
0	-5,61			
0,1	-1,75			
0,2	2,11			
0,3	5,98			
0,4	9,84			
0,5	13,70			
0,6	17,57			
0,7	21,43			
0,8	25,29			
0,9	29,16			
1	33,02			

# 9.3 Rigid body over rigid soil by PVW - $\alpha$ load multiplier - Hand calculation

The overturning mechanism has been studied with the principle of virtual work method, which consider the relations about the energies involved in the mechanism. In order to ensure the equilibrium the external energies must be equal to the internal energies developed by the mechanism. The principle of virtual work method has been used for different configurations of the forces as well for different configurations of the position of the hinges (or rotation points).

In the following paragraphs are reported two configuration of forces:

- Unique seismic force on the top;
- Roof force + Wall force, this means that the force developed from the inertia of the roof mass has been considered distinguished and applied at the roof level while the wall mass has been set applied at the centroid of the section of the wall.

The two force configurations have been studied for different cases:

- α critical for this case the activation of the tie-timber beam chains has been neglected with the aim of understanding the behavior of the free wall;
- Tmin for this case they have been applied all the value of the seismic load multiplier from 0,0 to 1,0 with steps of 0,1 in order to know the values of the tension applied to the rafters by the seismic event.

The analysis has been conducted considering a rigid block behavior, for accuracy, it is important to underline that in the following pages, the behavior of the whole wall is the last case of the blocks analysis but it has been described apart.

# 9.3.1 Unique seismic force on the top

The main data used for this aim are reported below.

Table 46 Weights and masses pertinent to studied blocks

Weights in 3 modulus					
	KN	Kg			
Wroof on timber 1	57,94	5905,80			
W2 on timber 2	72,46	7386,16			
W3 on timber 3	91,51	9327,88			
W4 on timber 4	110,55	11269,59			
W5 on timber 5	129,60	13211,31			
W6 on timber 6	145,77	14859,44			
W7 on ground	150,71	15362,73			

Table 47 Heights and ratios for  $\Delta$  proportional multiplier between 0 and 1

Heights and ratios for $\Delta$ proportional multiplier between 0 and 1							
Hinge	inge H $\Delta$ hi $\Delta$ H*(1- $\Delta$ H)						
2	3,15	0,325	0,103175	2,825			
3	3,15	0,925	0,293651	2,225			
4	3,15	1,525	0,484127	1,625			
5	3,15	2,125	0,674603	1,025			
6	3,15	2,725	0,865079	0,425			

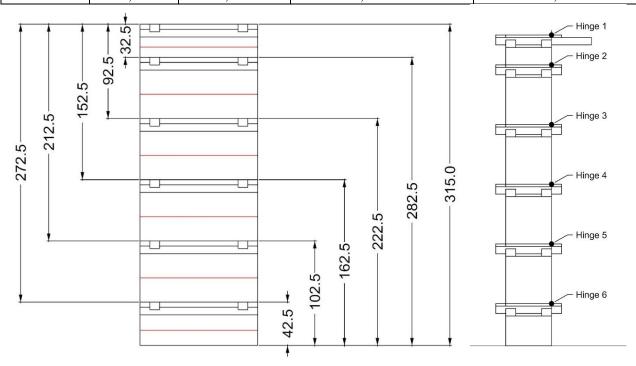


Figure 9-6 Hinges posotions

Figure 9-7 Hinges heights and Blocks Heights

# 9.3.1.1 Overturning Wall - a critical

Equations for computing the critical  $\alpha$  load multiplier are shown below:

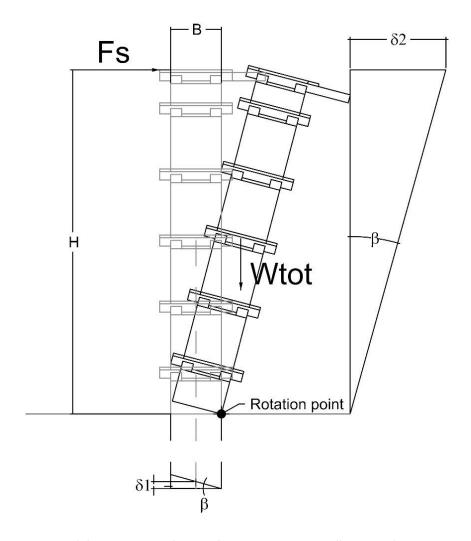


Figure 9-8 Unique seismic force on the top - Overturning Wall -  $\alpha$  critical

$$E_{ext} - E_{int} = 0$$

$$F_s * \delta 2 - W_{tot} * \delta 1 = 0$$

$$\alpha*(W_{tot})*H*\beta-W_{tot}*\frac{B}{2}*\beta=0$$

$$\alpha = \frac{W_{tot} * B}{W_{tot} * 2 * H}$$

$$\alpha = \frac{B}{2 * H}$$

$$\alpha = \frac{0.46}{2 * 3,15} = 0.073$$

Where

 $E_{ext}$  is the external energy

 $E_{int}$  is the internal energy

 $\beta$  is the rotation angle for the overturning mechanism

 $\delta 1$  is the displacement of the centroid

 $\delta 2$  is the displacement of the application point of the considered seismic force

### 9.3.1.2 Overturning Blocks - a critical

The critical load multiplier for the configuration of the unique seismic force applied on the top of the wall for the analysis of the blocks is the same of the entire wall case. This is due to the fact that the computation end up with a ratio of the same geometrical component.

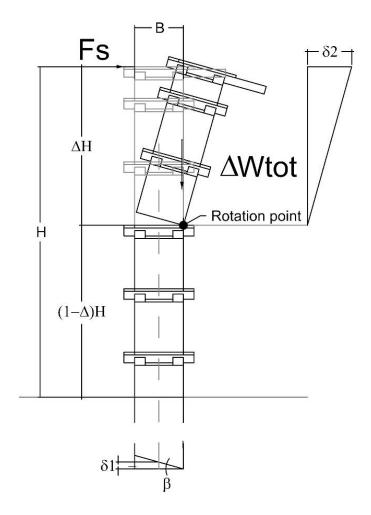


Figure 9-9 Unique seismic force on the top - Overturning Blocks - a critical

$$F_{s} * \delta 2 - \Delta W_{tot} * \delta 1 = 0$$

$$\alpha * (W_{tot}) * \Delta * H * \beta - \Delta * W_{tot} * \frac{B}{2} * \beta = 0$$

$$\alpha = \frac{\Delta * W_{tot} * B}{\Delta * W_{tot} * 2 * H}$$

$$\alpha = \frac{B}{2 * H}$$

$$\alpha = \frac{0.46}{2 * 3.15} = 0.073$$

where

 $\Delta$ = proportional multiplier between 0 and 1 based on the position of the hinges and the heights of the blocks.

### 9.3.2 Roof force + Wall force

### 9.3.2.1 Overturning Wall - a critical

Equations for computing the critical  $\alpha$  load multiplier are shown below:

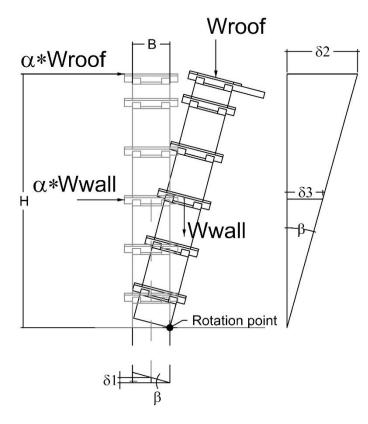


Figure 9-10 Roof force + Wall force - Overturning Wall - α critical

$$\alpha * W_{roof} * \delta 2 + \alpha * W_{wall} * \delta 3 - (W_{roof} + W_{wall}) * \delta 1 = 0$$

$$\alpha * W_{roof} * H * \beta + \alpha * W_{wall} * \frac{H}{2} * \beta - (W_{roof} + W_{wall}) * \frac{B}{2} * \beta = 0$$

$$\alpha * H * (W_{roof} + \frac{W_{wall}}{2}) = \frac{(W_{roof} + W_{wall}) * B}{2}$$

$$\alpha = \frac{B}{2 * H} * \frac{(W_{roof} + W_{wall})}{(W_{roof} + \frac{W_{wall}}{2})}$$

$$\alpha = 0.1055$$

 $W_{roof}$  is the weight of the roof

 $W_{wall}$  is the weight of the wall

 $\delta 1$  is the displacement of the centroid

 $\delta 2$  is the displacement of the application point of the considered seismic force of the roof  $\delta 3$  is the displacement of the application point of the considered seismic force of the wall

### 9.3.2.2 Overturning Blocks - a critical

Equations for computing the critical  $\alpha$  load multiplier are shown below:

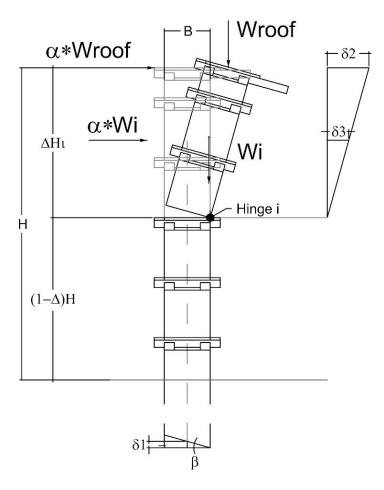


Figure 9-11 Roof force + Wall force - Overturning Blocks - a critical

$$\begin{split} \alpha * W_{roof} * \delta 2 + \alpha * W_i * \delta 3 - \left(W_{roof} + W_i\right) * \delta 1 &= 0 \\ \alpha * W_{roof} * \Delta H_i * \beta + \alpha * W_i * \frac{\Delta H_i}{2} * \beta - \left(W_{roof} + W_i\right) * \frac{B}{2} * \beta &= 0 \\ \alpha &= \frac{B}{2 * \Delta H_i} * \frac{\left(W_{roof} + W_i\right)}{\left(W_{roof} + \frac{W_i}{2}\right)} \end{split}$$

Where

# $W_i$ is the weights of the pertinent block.

Table 48 Roof force + Wall force - Overturning Blocks - α critical multipliers

	Weights in 3 modulus							
hinge i	Wi	KN	Kg	Н	Δhi	Δ	H*(1-ΔH)	α
hinge 1	Wroof on timber 1	57,94	5905,80	/	/	/	/	/
hinge 2	W2 on timber 2	72,46	7386,16	3,15	0,325	0,103	2,825	0,79
hinge 3	W3 on timber 3	91,51	9327,88	3,15	0,925	0,294	2,225	0,30
hinge 4	W4 on timber 4	110,55	11269,59	3,15	1,525	0,484	1,625	0,20
hinge 5	W5 on timber 5	129,60	13211,31	3,15	2,125	0,675	1,025	0,15
hinge 6	W6 on timber 6	145,77	14859,44	3,15	2,725	0,865	0,425	0,12

The  $\alpha$  critical load multipliers reported must be read as maximum limit value beyond which the failure mechanism happens.

# 9.3.3 Unique seismic force on the top with timber tie-beams - Minimum Tension dependent on $\boldsymbol{\alpha}$

In the following pages, it is reported the calculus procedure used to obtain the minimum tension acting on the tie-timber beam chains due to different  $\alpha$  load multipliers.

The main data used for this aim are reported below.

Table 49 Weights and masses pertinent to studied blocks - Tmin

Weights in 3 modulus					
	KN	Kg			
Wroof on timber 1	57,94	5905,80			
W2 on timber 2	72,46	7386,16			
W3 on timber 3	91,51	9327,88			
W4 on timber 4	110,55	11269,59			
W5 on timber 5	129,60	13211,31			
W6 on timber 6	145,77	14859,44			
W7 on ground	150,71	15362,73			

Table 50 Heights and ratios for  $\Delta$  proportional multiplier between 0 and 1 - Tmin

Table 30 Heights and ratios for 2 proportional multiplier between 0 and 1 - 1 min											
Heights of the mass and forces											
		Heights of the mas	Height of Force : Ht								
Block	Δhi	Δ	H of hinge : $H^*(1-\Delta)$	Hti	m	Δhti (m)					
roof	0	0	3,15	Htroof	3,100	/					
Timber1	0	0	3,15	Ht1	3,100	/					
Timber2	0,325	0,103174603	2,825	Ht2	2,775	0,2625					
Timber3	0,925	0,293650794	2,225	Ht3	2,175	0,8625					
Timber4	1,525	0,484126984	1,625	Ht4	1,575	1,4625					
Timber5	2,125	0,674603175	1,025	Ht5	0,975	2,0625					
Timber6	2,725	0,865079365	0,425	Ht6	0,375	2,6625					
Wall	3,15	1	0	Htchain	0,000	3,0875					
				Sum	10,975	3,0875					

### 9.3.3.1 Overturning Wall – Tmin

Equations for computing the minimum tensions as function of  $\alpha$  load multiplier are shown below:

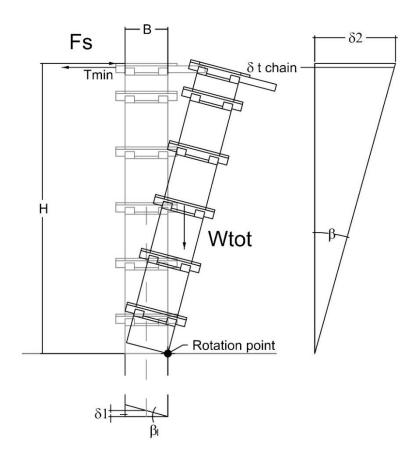


Figure 9-12 Unique seismic force on the top - Overturning Wall - Tmin

$$F_{s} * \delta 2 - W_{tot} * \delta 1 - 2 * T_{min} * \delta t_{chain} = 0$$

$$\alpha * (W_{tot}) * H * \beta - W_{tot} * \frac{B}{2} * \beta - 2 * T_{min} * H t_{chain} * \beta = 0$$

$$T_{min} = \frac{W_{tot} * H * \alpha - W_{tot} * \frac{B}{2}}{2 * H t_{chain}}$$

Where

 $T_{min}$  is the minimum tension due to the seismic event on the roof tie timber beam  $\delta t_{chain}$  is the displacement of the application point of the roof timber beams acting as a chain  $Ht_{chain}$  is the height of the roof timber beams acting as a chain

Table 51 Unique seismic force on the top - Overturning Wall - Tmin

Minimum Tension dependent on α							
α	Tmin (kN)						
0	-5,61						
0,1	1,95						
0,2	9,52						
0,3	17,08						
0,4	24,65						
0,5	32,22						
0,6	39,78						
0,7	47,35						
0,8	54,91						
0,9	62,48						
1	70,05						

### 9.3.3.2 Overturning Blocks – Tmin

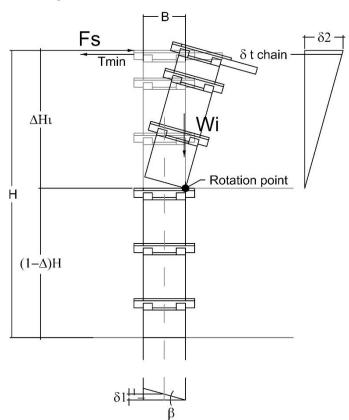


Figure 9-13 Unique seismic force on the top - Overturning Blocks - Tmin

$$\begin{split} F_{s}*\delta 2 - W_{on\;timber}*\delta 1 &- 2*T_{min}*\delta t_{chain} = 0 \\ \alpha_{i}*(W_{i})*\Delta H_{i}*\beta - W_{i}*\frac{B}{2}*\beta - 2*T_{min}*Ht_{chain}*\beta = 0 \\ \\ T_{min}* &= \frac{\alpha_{i}*W_{i}*\Delta H_{i} - W_{i}*\frac{B}{2}}{2*Ht_{i-chain}} \end{split}$$

Table 52 Unique seismic force on the top - Overturning Blocks - Tmin -data and results

Tmin for Unique seismic force on top														
hinge i	Wi		Δhi	Δhti- chain	α=									
		kN	(m)	(m)	0,1	0,2	0,3	0,4	0,5	0,6	0,7	0,8	0,9	1
hinge 1	Wroof on timber 1	57,94	/	/	/	/	/	/	/	/	/	/	/	/
hinge 2	W2 on timber 2	72,46	0,325	0,26	-27,26	-22,77	-18,29	-13,80	-9,32	-4,83	-0,35	4,14	8,63	13,11
hinge 3	W3 on timber 3	91,51	0,925	0,86	-7,29	-2,39	2,52	7,43	12,33	17,24	22,15	27,05	31,96	36,87
hinge 4	W4 on timber 4	110,55	1,525	1,46	-2,93	2,83	8,60	14,36	20,13	25,89	31,65	37,42	43,18	48,95
hinge 5	W5 on timber 5	129,60	2,125	2,06	-0,55	6,13	12,80	19,48	26,16	32,83	39,51	46,19	52,86	59,54
hinge 6	W6 on timber 6	145,77	2,725	2,66	1,16	8,62	16,08	23,54	31,00	38,46	45,92	53,38	60,84	68,30

Table 53 Unique seismic force on the top - Overturning Blocks - Tmin

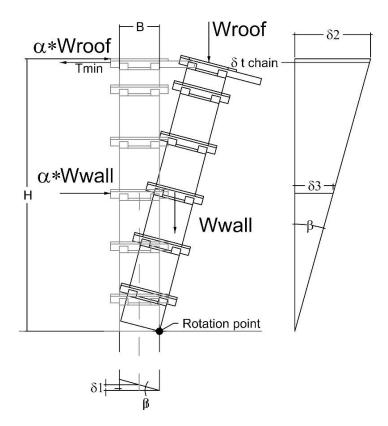
Tmin= $f(\alpha)$ [kN]	$\alpha =$										
111111 1(0) [1111]	0,1	0,2	0,3	0,4	0,5	0,6	0,7	0,8	0,9	1	
hinge 1	/	/	/	/	/	/	/	/	/	/	
hinge 2	-27,26	-22,77	-18,29	-13,80	-9,32	-4,83	-0,35	4,14	8,63	13,11	
hinge 3	-7,29	-2,39	2,52	7,43	12,33	17,24	22,15	27,05	31,96	36,87	
hinge 4	-2,93	2,83	8,60	14,36	20,13	25,89	31,65	37,42	43,18	48,95	
hinge 5	-0,55	6,13	12,80	19,48	26,16	32,83	39,51	46,19	52,86	59,54	
hinge 6	1,16	8,62	16,08	23,54	31,00	38,46	45,92	53,38	60,84	68,30	

The negative values must be considered with no physical meanings.

# 9.3.4 Roof force + Wall force with timber tie-beams - Minimum Traction dependent on $\boldsymbol{\alpha}$

# 9.3.4.1 Overturning Wall – Tmin

Equations for computing the minimum tensions as function of  $\alpha$  load multiplier are shown below:



Figure~9-14~Roof force~+~Wall~force~-~Overturning~Wall~-~Tmin

$$\alpha * W_{roof} * \delta 2 + \alpha * W_{wall} * \delta 3 - \left(W_{roof} + W_{wall}\right) * \delta 1 - 2 * T_{min} * \delta t_{chain} = 0$$

$$\alpha * W_{roof} * H * \beta + \alpha * W_{wall} * \frac{H}{2} * \beta - \left(W_{roof} + W_{wall}\right) * \frac{B}{2} * \beta - 2 * T_{min} * Ht_{chain} * \beta = 0$$

$$\alpha * H * \left(W_{roof} + \frac{W_{wall}}{2}\right) = \frac{(W_{roof} + W_{wall})^{*B}}{2} + 2 * T_{min} * Ht_{chain}$$

$$T_{min} = \frac{\alpha * H * \left(W_{roof} + \frac{W_{wall}}{2}\right) - \left(W_{roof} + W_{wall}\right) * \frac{B}{2}}{2 * Ht_{chain}}$$

Table 54 Roof force + Wall force - Overturning Wall - Tmin

	Minimum Tension dependent on α						
α	Tmin (kN)						
0	-7,77						
0,1	-1,08						
0,2	5,61						
0,3	12,30						
0,4	18,99						
0,5	25,69						
0,6	32,38						
0,7	39,07						
0,8	45,76						
0,9	52,45						
1	59,14						

#### 9.3.4.2 Overturning Blocks – Tmin

Equations for computing the minimum tensions as function of  $\alpha$  load multiplier are shown below:

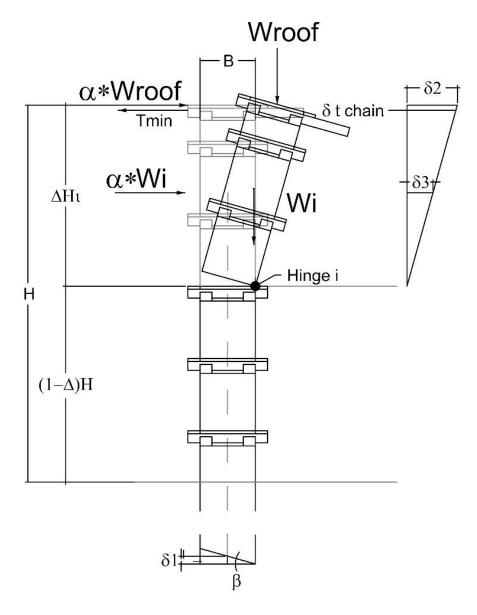


Figure 9-15 Roof force + Wall force - Overturning Blocks - Tmin

$$\begin{split} &\alpha_{i} * W_{roof} * \delta 2 + \alpha_{i} * W_{i} * \delta 3 - \left(W_{roof} + W_{i}\right) * \delta 1 - 2 * T_{min} * \delta t_{i-chain} = 0 \\ &\alpha_{i} * W_{roof} * \Delta H_{i} * \beta + \alpha_{i} * W_{i} * \frac{\Delta H_{i}}{2} * \beta - \left(W_{roof} + W_{i}\right) * \frac{B}{2} * \beta - 2 * T_{min} * H t_{i-chain} * \beta = 0 \\ &T_{min} = \frac{\alpha_{i} * \Delta H_{i} * \left(W_{roof} + \frac{W_{i}}{2}\right) - \left(W_{roof} + W_{i}\right) * \frac{B}{2}}{2 * H t_{i-chain}} \end{split}$$

Table 55 Roof force + Wall force - Overturning Blocks - Tmin -data and results

	Tmin for Seismic force due to wall( in the centroide) and roof (on top)													
hinge i	hinge i Wi Δhi Δhti-chain α=													
		KN	(m)	(m)	m) 0,1 0,2 0,3 0,4 0,5 0,6 0,7 0,8 0,9 1					1				
hinge 1	Wroof timber 1	57,94	/	/	/	/	/	/	/	/	/	/	/	/
hinge 2	W2 on timber 2	72,46	0,325	0,26	-27,71	-23,67	-19,64	-15,60	-11,56	-7,53	-3,49	0,54	4,58	8,62
hinge 3	W3 on timber 3	91,51	0,925	0,86	-8,19	-4,19	-0,18	3,83	7,83	11,84	15,85	19,85	23,86	27,87
hinge 4	W4 on timber 4	110,55	1,525	1,46	-4,30	0,09	4,48	8,88	13,27	17,66	22,05	26,45	30,84	35,23
hinge 5	W5 on timber 5	129,60	2,125	2,06	-2,40	2,43	7,27	12,10	16,93	21,76	26,59	31,42	36,25	41,08
hinge 6	W6 on timber 6	145,77	2,725	2,66	-1,08	4,13	9,34	14,55	19,76	24,98	30,19	35,40	40,61	45,83

Table 56 Roof force + Wall force - Overturning Blocks - Tmin

Tmin= $f(\alpha)$ [kN]	α=									
$[IIIIII-I(\alpha)]$	0,1	0,2	0,3	0,4	0,5	0,6	0,7	0,8	0,9	1
hinge 1	/	/	/	/	/	/	/	/	/	/
hinge 2	-27,71	-23,67	-19,64	-15,60	-11,56	-7,53	-3,49	0,54	4,58	8,62
hinge 3	-8,19	-4,19	-0,18	3,83	7,83	11,84	15,85	19,85	23,86	27,87
hinge 4	-4,30	0,09	4,48	8,88	13,27	17,66	22,05	26,45	30,84	35,23
hinge 5	-2,40	2,43	7,27	12,10	16,93	21,76	26,59	31,42	36,25	41,08
hinge 6	-1,08	4,13	9,34	14,55	19,76	24,98	30,19	35,40	40,61	45,83

# 9.4 Conclusions about the highest required tension strength Tmin

#### 9.4.1 Horizontal equilibrium and Rotational equilibrium – Tmin

The values reported on the Rotational equilibrium , focused on Minimum Traction dependent on  $\alpha$ , show how much the chain at the roof level needs to bear.

In order to resist a PGA of 0,5 g the chain must bear at least 13,70 kN.

Minimum Tension dependent on $\alpha$					
α	Tmin (kN)				
0	-5,61				
0,1	-1,75				
0,2	2,11				
0,3	5,98				
0,4	9,84				
0,5	13,70				
0,6	17,57				
0,7	21,43				
0,8	25,29				
0,9	29,16				
1	33,02				

#### 9.4.2 Unique seismic force on the top - $\alpha$ critical

In the kinetic approach both the studied cases, entire wall mechanism and block by block mechanism, show the same critical seismic multiplier. This multilpier is quite low but it seems to be correct due to the fact of the absence of the mortar and neither any other stabilizing devices.

 $\alpha = 0.073$ 

#### 9.4.3 Roof force + Wall force - α critical

In the kinetic approach both the studied cases, entire wall mechanism and block by block mechanism, show the different critical seismic multipliers. In the case of block by block mechanism:

	Weights in							
hinge i	Wi	KN	Kg	Н	Δhi	Δ	H*(1-ΔH)	α
hinge 1	Wroof on timber 1	57,94	5905,80	/	/	/	/	/
hinge 2	W2 on timber 2	72,46	7386,16	3,15	0,325	0,103	2,825	0,79
hinge 3	W3 on timber 3	91,51	9327,88	3,15	0,925	0,294	2,225	0,30
hinge 4	W4 on timber 4	110,55	11269,59	3,15	1,525	0,484	1,625	0,20
hinge 5	W5 on timber 5	129,60	13211,31	3,15	2,125	0,675	1,025	0,15
hinge 6	W6 on timber 6	145,77	14859,44	3,15	2,725	0,865	0,425	0,12

The most critical case is the one of the entire wall mechanism:

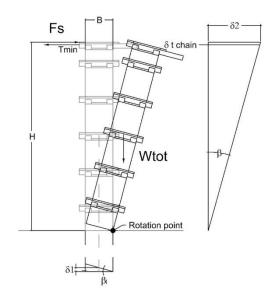
$$\alpha=0.1055$$

#### 9.4.4 Unique seismic force on the top with timber tie-beams - Tmin

In the kinetic approach the entire wall mechanism with the unique seismic force at the top of the wall is the most critical. This is due to the facts that the whole mass of the wall takes part to the mechanism and the lever arm is the maximum possible. Minimum Tension dependent on  $\alpha$  considering the chain only at the roof level.

In order to resist a PGA of 0,5 g the chain must bear at least 32,22 kN.

Minimum	Minimum Tension dependent on α					
α	Tmin (kN)					
0	-5,61					
0,1	1,95					
0,2	9,52					
0,3	17,08					
0,4	24,65					
0,5	32,22					
0,6	39,78					
0,7	47,35					
0,8	54,91					
0,9	62,48					
1	70,05					



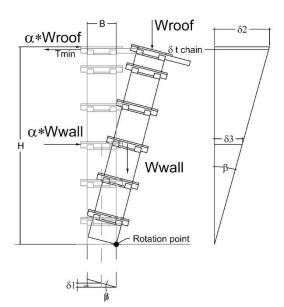
#### 9.4.5 Roof force + Wall force with timber tie-beams - Tmin

Wall mechanism and block by block mechanism have been studied by the kinetic approach. The most critical case is the entire wall mechanism where the minimum traction of the chain in function of  $\alpha$  isreported in the recall table

In order to resist a PGA of 0,5 g the chain bear at least 25,69 kN.

	Minimum Tension dependent on $\alpha$						
α	Tmin (kN)						
0	-7,77						
0,1	-1,08						
0,2	5,61						
0,3	12,30						
0,4	18,99						
0,5	25,69						
0,6	32,38						
0,7	39,07						
0,8	45,76						
0,9	52,45						
1	59,14						

must

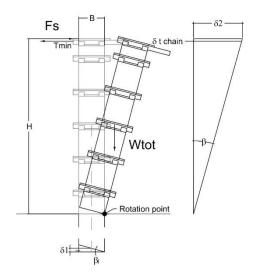


# 9.5 Verifications for Overturning Rigidbehavior

The verifications have been performed considering the worst case with load seismic multiplier  $\alpha = 1$  thus Tmin= 70,05 kN and considering that the reactions in the joint are equally distributed between the Tie-timber chain and the tie-timber beam of the failing wall.

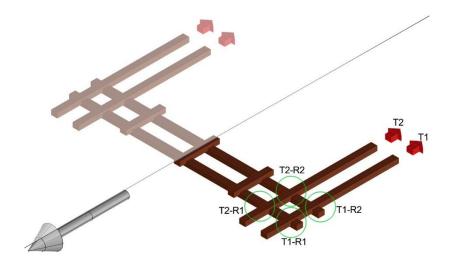
# 9.5.1 Analyzing the worst case: Unique seismic force on the top with timber tie-beams - Tmin

Minim	Minimum Tension dependent on α					
α	Tmin (kN)					
0	-5,61					
0,1	1,95					
0,2	9,52					
0,3	17,08					
0,4	24,65					
0,5	32,22					
0,6	39,78					
0,7	47,35					
0,8	54,91					
0,9	62,48					
1	70,05					



#### 9.5.2 Equal distribution of the reactions T1=T2 and R1=R2

In order to be clear they are recalled the hypothesis asserted in the chapter 6, and the equal distribution of the reactions on corner joint.



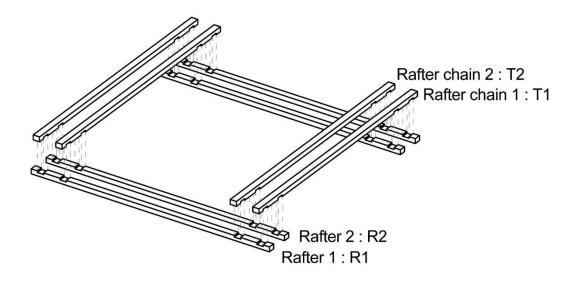
$$T_1 = T_2 = \frac{T_{min}}{2}$$

$$(T_1R_1) = (T_1R_2) = (T_2R_1) = (T_2R_2) = \frac{T_{min}}{4}$$

Thebehavior of the rafters chain 1 and 2 is the same thus the verifications on T1 is equal to T2.

Thebehavior of rafters belonging to the overturning wall is the same thus the verifications on R1 is equal to R2.

The verifications have been performed on the biggest section of the rafters, the body, which refers to a section of area equal to A5. The same verifications have been performed considering the smallest section, the notch, of area equal to A4.



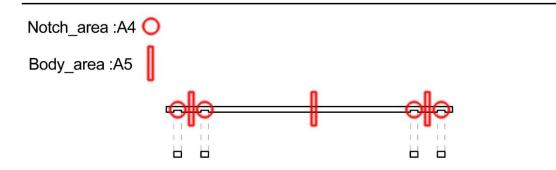


Table 57 Geometric dimensions for Notch and Body Areas

	b	h	AREA net				
	mm	mm	mm^2	cm^2	m^2		
A4	100	50	5000	50	0,005		

	b	h	AREA net				
	mm	mm	mm^2	cm^2	m^2		
A5	100	75	7500	75	0,0075		

All the verifications have been done considering the highest load multiplier, thus  $\alpha$ =1 .In the cases where the verification is not satisfied the load multiplier has been reduced until the verification was verified.

#### 9.5.3 Verifications T1=T2

All the verifications have been

#### 9.5.3.1 Body







RB0tens						
N_0d	35022,92	N				
b	100,00	mm				
h	75,00	mm				
A_(net)	7500,00	mm				
$\sigma_{-}(t,0,d)$	4,67	N/mm^2				
kh	1,08					
f_(t,0,d)	30,80	N/mm^2				
Verification	VERI	FIED				
N_(0d)max	231,00	kN				

RB0mY					
M_(y,d)	1330870,94	Nmm			
K_m	0,70				
b	100,00	mm			
h	75,00	mm			
W_(y,d)	93750,00	mm^3			
$\sigma_{\underline{}}(m,y,d)$	14,20	N/mm^2			
kh	1,15				
f_(m,d)	51,33	N/mm^2			
f_(m,y,d)	58,97	N/mm^2			
Verification	VERIF	IED			
M_(y,d)max	5528110,83	Nmm			
M_(y,d)max	5,53	kNm			

RB0mZ		
M_(z,d)	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
$W_{-}(z,d)$	125000,00	mm^3
$\sigma_{-}(m,z,d)$	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	TED
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

Influence of keyed scarf joint		
Verification	NOT VERIFIED	
N_(0d)max	25,41	kN

RB0tens is satisfied for a load seismic multiplier  $\alpha = 0.7$ 

#### 9.5.3.2 Body Combinations

Combined bending and axial tension

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

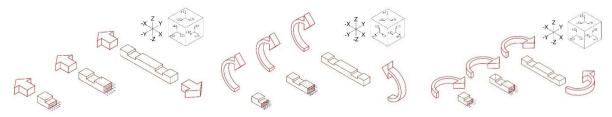
For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0,7
- otherwise km = 1

Combination of RB0tens and RB0mY are satisfied for a load multiplier  $\alpha = 1\,$ 

Combination of RB0tens and RB0mZ are satisfied for a load multiplier  $\alpha=1$ 

#### 9.5.3.3 Notch



CPNotch0tens		
N_0d	35022,92	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
σ_(t,0,d)	7,00	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERI	FIED
N_(0d)max	154,00	kN

CPNotch0mY		
M_(y,d)	1330870,94	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(y,d)	41666,67	mm^3
$\sigma_{m,y,d}$	31,94	N/mm^2
kh	1,25	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	63,95	N/mm^2
Verification	VERIFIED	
M_(y,d)max	2664480,07	Nmm
M_(y,d)max	2,66	kNm

CPNotch0mZ		
M_(z,d)	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(z,d)	83333,33	mm^3
σ_(m,z,d)	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	IED
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

It is important to underline that in the notch section, the keyed scarf joint has not been considered.

#### 9.5.3.1 Notch Combinations

Combined bending and axial tension

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

For solid timber, glued laminated timber and LVL:

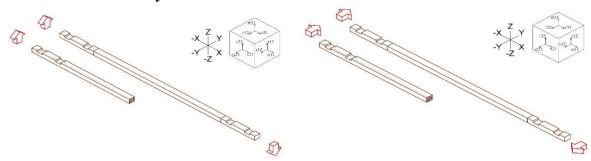
- for rectangular sections: km = 0,7
- otherwise km = 1

Combination of CPNotch0tens and CPNotch0mY are satisfied for a load multiplier  $\alpha = 1$ 

Combination of CPNotch0tens and CPNotch0mZ are satisfied for a load multiplier  $\alpha = 1$ 

# 9.5.4 Verifications R1=R2

# 9.5.4.1 Body

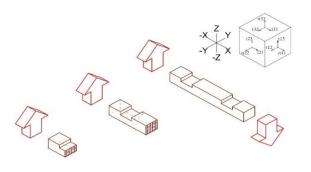


RB0shearZ with bending		
V_zd	0,00	N
K_cr	0,67	
A_(net)	5025,00	mm
τ_(d)	0,00	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VERIFIED	
V_zd max	12,28	kN
R	B0shearZ	
V_zd	0,00	N
A_(net)	7500,00	mm
τ_(d)	0,00	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VER	IFIED
V_zd max	18,33	kN

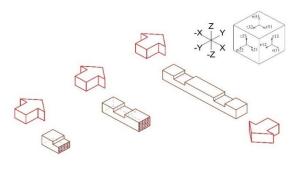
RB0shearY with bending		
V_yd	35022,92	N
K_cr	0,67	
A_(net)	5025,00	mm
τ_(d)	10,45	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	NOT VE	RIFIED
V_yd max	12,28	kN
I	RB0shearY	
V_yd	35022,92	N
A_(net)	7500,00	mm
τ_(d)	7,00	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	NOT VE	RIFIED
V_yd max	18,33	kN

RB0shearY with bending and RB0shearY are satisfied for a load seismic multiplier  $\alpha = 0.35$ 

#### 9.5.4.2 Notch



CPNotch0shearZ with bending			
V_zd	0,00	Ν	
K_cr	0,67		
A_(net)	3350,00	mm^2	
τ_(d)	0,00	N/mm^2	
f_(v,d)	3,67	N/mm^2	
Verification	VERIFIED		
V_zd max	8,19	kN	

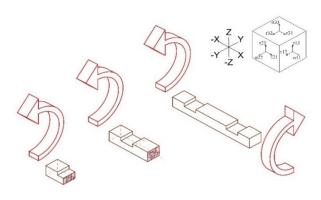


CPNotch0shearY with bending			
V_yd	35022,92	N	
K_cr	0,67		
A_(net)	3350,00	mm^2	
τ_(d)	15,68	N/mm^2	
f_(v,d)	3,67	N/mm^2	
Verification	NOT VERIFIED		
V_yd max	8,19	kN	

CPNotch0shearZ		
V_zd	0,00	N
A_(net)	5000,00	mm
τ_(d)	0,00	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VERIFIED	
V_zd max	12,22	kN

CPNotch0shearY		
V_yd	35022,92	N
A_(net)	5000,00	mm
τ_(d)	10,51	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	NOT VE	ERIFIED
V_yd max	12,22	kN

CPNotch0shearY with bending and CPNotch0shearY are satisfied for a load seismic multiplier  $\alpha = 0.2$ 



CPNotch0mX		
M_(x,d)	420275,03	Nmm
b	100,00	mm
h	50,00	mm
α	3,90	
$\tau_{\text{(tor,d)}}$	6,56	N/mm^2
K_shape	1,03	
f_(v,d)	3,67	N/mm^2
k_shape*f_(v,d)	3,78	N/mm^2
Verification	NOT VE	
M_(x,d) max	242094,02	Nmm
M_(x,d) max	0,24	kNm

CPNotch0mX is satisfied for a load seismic multiplier  $\alpha = 0.5$ 

#### 9.5.4.3 Notch Combinations

Combined Torsion and Shear - CNR-DT 206/2007

$$\frac{\tau_{tor,d}}{k_{shape} * f_{v,d}} + \left(\frac{\tau_d}{f_{v,d}}\right)^2 \leq 1$$

Combination of CPNotch0mX and CPNotch0shearZ are satisfied for a load multiplier  $\alpha=0.5$ 

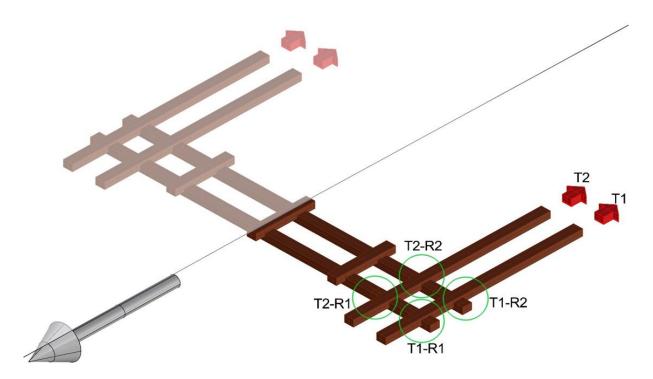
Combination of CPNotch0mX and CPNotch0shearY are satisfied for a load multiplier  $\alpha=0.2$ 

#### 9.5.5 Verifications on corner joint, seismic event parallel to Roof Rafter

In order to report the verification on the corner joint at the roof level they have been recalled the hypothesis done in the chapter 6.

#### 9.5.5.1 Scheme

As well they have been reported all the data collected in the previous chapter to verify all the sections.

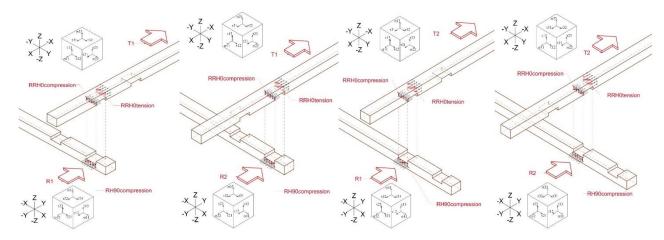


Tmin	70045,84	N	70,05	kN
α	1,00			
T1R1=T1R2=T2R1=T2R2=Tmin/4=	17511,46	Ν	17,51	kN

		Design Actions				
Compression	N_0d		17511,46	N	17,51	kN
Tension	N_0d		17511,46	N	17,51	kN
Shear Z	V_zd		17511,46	N	17,51	kN
Shear Y	V_yd		17511,46	N	17,51	kN
Bend.MY	M_(y,d)		665435,47	Nmm	0,67	kNm
Bend.MZ	M_(z,d)		5253437,94	Nmm	5,25	kNm
Torsion.MX	M_(x,d)		210137,52	Nmm	0,21	kNm

lever arm	δ for My	38,00	mm
lever arm	δ1 for Mz1	150,00	mm
lever arm	δ2 for Mz2	150,00	mm
lever arm	δnotch for Mxnotch	12,00	mm
lever arm	δbody for Mxbody	25,00	mm

# 9.5.5.2 Axial stresses: Compression and Tension

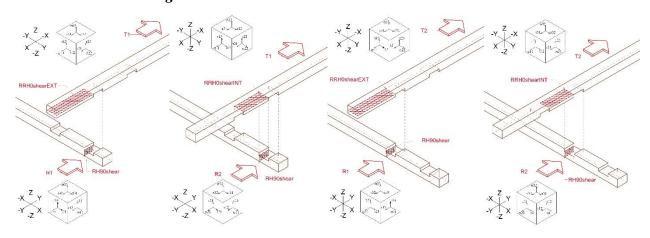


RRH0tension	A4	
N_0d	17511,46	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
σ_(t,0,d)	3,50	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERI	FIED
N_(od)max	154,00	kN

RRH0compression	A3	
N_0d	17511,46	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,0,d)	7,00	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	VERI	IFIED
N_(od)max	62,33	kN

RH90compression	A3	
N_90d	17511,46	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	7,00	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	VERI	FIED
N_(90d)max	16,50	kN

# 9.5.5.3 Tangential stresses: Shear



RRH0shearEXT	A7	
with	bending	
V_0d	17511,46	N
K_cr	0,67	
A_(net)	26800,00	mm^2
τ_(d)	0,98	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VERI	FIED
V_0d max	65,51	kN

RRH0shearINT	A1			
Wi	with bending			
V_0d	17511,46	N		
K_cr	0,67			
A_(net)	17420,00	mm^2		
τ_(d)	1,51	N/mm^2		
f_(v,d)	3,67	N/mm^2		
Verification	VE	RIFIED		
V_0d max	42,58	kN		

RH90shear	A5	
wi	th bending	
V_90d	17511,46	N
K_cr	0,67	
A_(net)	5025,00	mm^2
τ_(d)	5,23	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VE	RIFIED
V_90d max	2,95	kN

RRH0shearEXT	A7	
V_0d	17511,46	N
A_(net)	40000,00	mm^2
τ_(d)	0,66	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VERI	FIED
V_0d max	97,78	kN

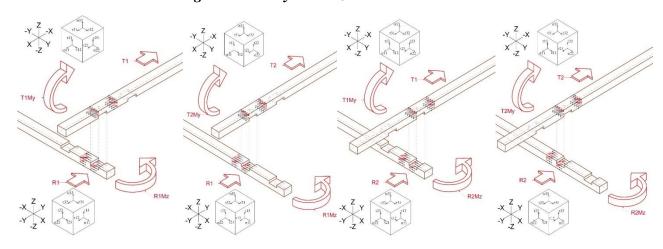
RRH0shearINT	A1	
V_0d	17511,46	N
A_(net)	26000,00	mm^2
τ_(d)	1,01	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VE	RIFIED
V_0d max	63,56	kN

RH90shear	A5	
V_90d	17511,46	N
A_(net)	7500,00	mm^2
τ_(d)	3,50	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification		
V_90d max	4,40	kN

RH90shear with bending is satisfied for a load multiplier  $\alpha = 0.15$ 

RH90shear is satisfied for a load multiplier  $\alpha = 0.2\,$ 

# 9.5.5.4 Bending moments My and Mz



Notch0mY		
N_0d	17511,46	N
δ for My	38,00	mm
$M_{(y,d)}$	665435,47	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
$W_{-}(y,d)$	41666,67	mm^3
$\sigma_{\text{_}}(m,y,d)$	15,97	N/mm^2
kh	1,25	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	63,95	N/mm^2
Verification	VERIF	IED
M_(y,d)max	2664480,07	Nmm
M_(y,d)max	2,66	kNm

Notch0mZ1		
V_90d	17511,46	N
δ1 for Mz1	150,00	mm
$M_{-}(z,d)$	2626718,97	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(z,d)	83333,33	mm^3
$\sigma_{-}(m,z,d)$	31,52	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	IED
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

Notch0mZ2		
V_90d	35022,92	N
δ2 for Mz2	150,00	mm
$M_{}(z,d)$	5253437,94	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
$W_{-}(z,d)$	83333,33	mm^3
$\sigma_{m,z,d}$	63,04	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	NOT VEF	RIFIED
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

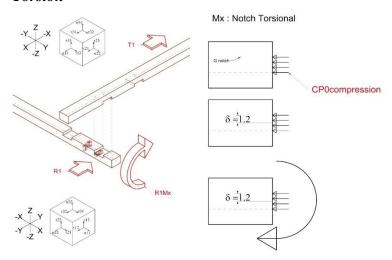
Body0mY		
N_0d	17511,46	N
δ for My	38,00	mm
M_(y,d)	665435,47	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(y,d)	93750,00	mm^3
σ_(m,y,d)	7,10	N/mm^2
kh	1,15	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	58,97	N/mm^2
Verification	VERIF	IED
M_(y,d)max	5528110,83	Nmm
M_(y,d)max	5,53	kNm

Body0mZ1		
V_90d	150,00	N
δ1 for Mz1	510,00	mm
M_(z,d)	76500,00	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(z,d)	125000,00	mm^3
$\sigma_{-}(m,z,d)$	0,61	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	IED
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

Body0mZ2		
V_90d	35022,92	N
δ2 for Mz2	150,00	mm
M_(z,d)	5253437,94	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(z,d)	125000,00	mm^3
$\sigma_{-}(m,z,d)$	42,03	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	TED
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

# Notch0mZ2 is satisfied for a load multiplier $\alpha = 0.8\,$

#### 9.5.5.5 Torsion



Notch0mX			
V_90d	1,75E+04	N	
δnotch for Mxnotch	12,00	mm	
$M_{-}(x,d)$	210137,52	Nmm	
b	100,00	mm	
h	50,00	mm	
α	3,90		
τ_(tor,d)	3,28	N/mm^2	
K_shape	1,03		
f_(v,d)	3,67	N/mm^2	
k_shape*f_(v,d)	3,78	N/mm^2	
Verification	VERII	FIED	
M_(x,d) max	242094,02	Nmm	
M_(x,d) max	0,24	kNm	

Body0mX			
V_90d	1,75E+04	N	
δnotch for Mxnotch	12,00	mm	
$M_{\perp}(x,d)$	210137,52	Nmm	
b	100,00	mm	
h	75,00	mm	
α	4,35		
$\tau_{\text{(tor,d)}}$	1,63	N/mm^2	
K_shape	1,02		
f_(v,d)	3,67	N/mm^2	
k_shape*f_(v,d)	3,74	N/mm^2	
Verification	VERII	FIED	
$M_{\perp}(x,d)$ max	483620,69	Nmm	
$M_{\perp}(x,d)$ max	0,48	kNm	

#### 9.5.5.6 Combinations

Combined bending and axial tension

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \leq 1$$

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0,7
- otherwise km = 1

Combination of Notch0tens and Notch0mY are satisfied for a load multiplier  $\alpha=1$  Combination of Notch0tens and Notch0mZ2 are satisfied for a load multiplier  $\alpha=1$  Combination of Body0tens and Body0mY are satisfied for a load multiplier  $\alpha=1$  Combination of Body0tens and Body0mZ2 are satisfied for a load multiplier  $\alpha=1$ 

Combined bending and axial compression

$$\left(\frac{\sigma_{c,0,d}}{f_{c,0,d}}\right)^{2} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_{m} * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\left(\frac{\sigma_{c,0,d}}{f_{c,0,d}}\right)^{2} + k_{m} * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0,7
- otherwise km = 1

Combination of Notch0comp and Notch0mY are satisfied for a load multiplier  $\alpha=1$  Combination of Notch0comp and Notch0mZ2 are satisfied for a load multiplier  $\alpha=1$  Combination of Body0comp and Body0mY are satisfied for a load multiplier  $\alpha=1$  Combination of Body0comp and Body0mZ2 are satisfied for a load multiplier  $\alpha=1$ 

Combined Torsion and Shear - CNR-DT 206/2007

$$\frac{\tau_{tor,d}}{k_{shape} * f_{v,d}} + \left(\frac{\tau_d}{f_{v,d}}\right)^2 \le 1$$

Combination of Notch0mX and Notch90shearY are satisfied for a load multiplier  $\alpha=0.8$  Combination of Body0mX and Body90shearY are satisfied for a load multiplier  $\alpha=1$ 

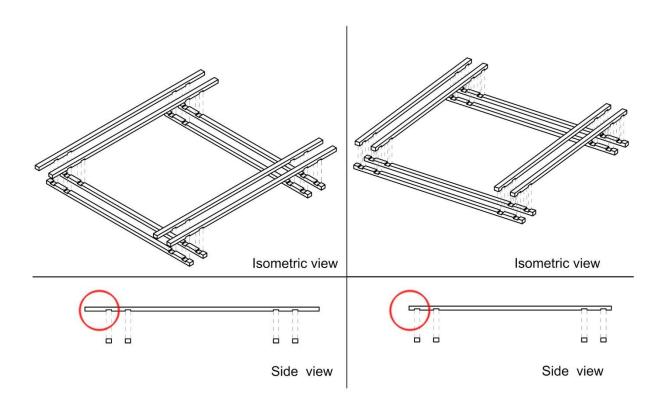
# 9.5.6 Verifications on corner joint, seismic event parallel to Rafter

#### 9.5.6.1 Scheme

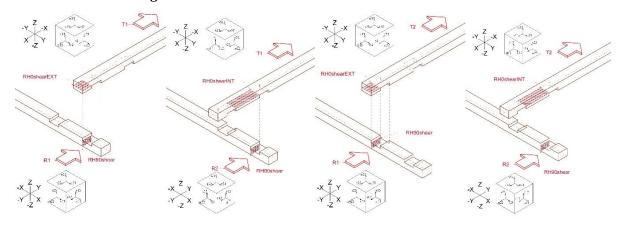
The difference between the roof rafter and the rafter is the length of the head.

The length of the normal rafter is shorter and the difference affects the longitudinal shear resistance of the element.

For all the other verifications nothing changes, that is why in the following, they are reported only the verifications about the shear resistance.



#### 9.5.6.2 Tangential stresses: Shear



RH0shearEXT	A2		
with bending			
V_0d	17511,46	N	
K_cr	0,67		
A_(net)	6700,00	mm^2	
τ_(d)	3,92	N/mm^2	
f_(v,d)	3,67	N/mm^2	
Verification	NOT VE	RIFIED	
V 0d max	16,38	kN	

RH0shearINT	A1			
with bending	with bending , " = RRHOshearINT "			
V_0d	17511,46	N		
K_cr	0,67			
A_(net)	17420,00	mm^2		
τ_(d)	1,51	N/mm^2		
f_(v,d)	3,67	N/mm^2		
Verification	VERIFIED			
V 0d max	42,58	kN		

RH90shear	A5	
	with bending	
V_90d	17511,46	N
K_cr	0,67	
A_(net)	5025,00	mm^2
τ_(d)	5,23	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VERIFIED	
V_90d max	2,95	kN

RH0shearEXT	A2	
V_0d	17511,46	N
A_(net)	10000,00	mm^2
τ_(d)	2,63	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VERI	FIED
V_0d max	24,44	kN

RH0shearINT	A1		
" = RRHOshearINT "			
V_0d	17511,46	N	
A_(net)	26000,00	mm^2	
τ_(d)	1,01	N/mm^2	
f_(v,d)	3,67	N/mm^2	
Verification	VERIFIED		
V_0d max	63,56	kN	

RH90shear	A5	
V_90d	17511,46	N
A_(net)	7500,00	mm^2
τ_(d)	3,50	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VERIFIED	
V_90d max	4,40	kN

RH0shearEXT with bending is satisfied for a load multiplier  $\alpha = 0.9$ 

RH90shear with bending is satisfied for a load multiplier  $\alpha = 0.15$ 

RH90shear is satisfied for a load multiplier  $\alpha = 0.25$ 

#### 9.6 Conclusions on seismic analysis out of plane – Overturning

#### 9.6.1 Safetybehavior under seismic multiplier α=0,15

All the verifications have been computed in function of the seismic load multiplier  $\alpha$ .

Summing up the results it can be noticed that the timber elements with the function of chain is not particularly affected by the keyed scarf joint and it has an high strengthbehavior, this is due to the shorea robust properties.

The most critical section is in the rafters of the timber beam belonging to overturning wall.

This section has been named RH90shear and it is shown in the figure below.

The verification of this section is satisfied for a seismic load multiplier  $\alpha = 0.15$ 

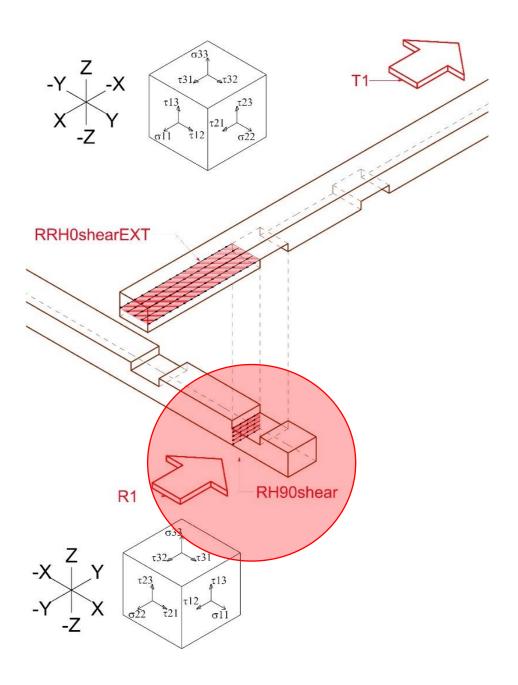


Figure 9-16 Overturning - RH90Shear most critical section

# 10 SEISMIC ANALYSIS OUT OF PLANE -FLEXIBLE RESPONSE BENDING BEHAVIOR

#### 10.1 Hypothesis of Flexible response – Bending behavior

The wall has been considered as it was composed by flexible layers, which may bend in the plane parallel to the ground. The figure below shows the analyzed failure mechanism in the flexible configuration .

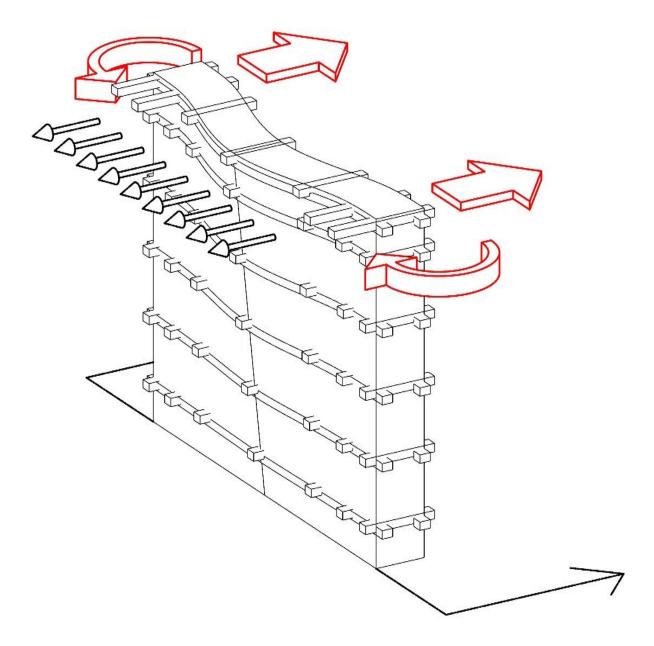


Figure 10-1 Flexible mechanism – example scheme

In this configuration, the activation of the tie-timber beam chain has been analyzed in a different way respect to the overturning configuration.

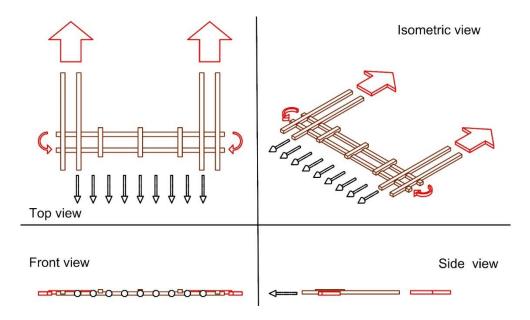


Figure 10-2 Flexible mechanism - tie-timber beam chains activation

#### 10.1.1 Hypothesis of Flexible behavior

The load is distributed along the tie-timber beam and it is due to a portion of the total mass around each tie-timber beam.

In order to know the reactions of each timber beam it is necessary to study the end connections composed by 2 rafters perpendicular to others 2 roof rafter or 2 rafters.

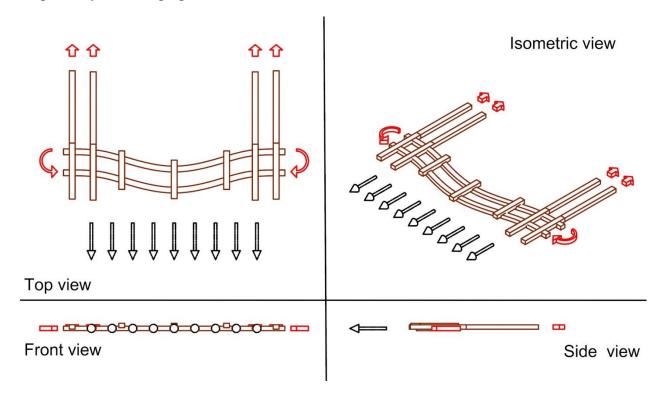


Figure 10-3 Flexible mechanism - deformed tie-timber beam chains and activation

#### 10.1.2Static scheme of the timber tie- beam

For each timber band in the wall, it has been defined an equivalent static scheme (clamped-clamped) and they have been defined the masses involved for each mechanism.

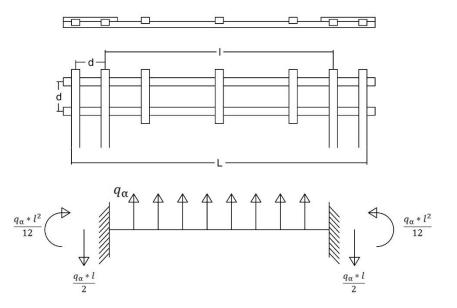


Figure 10-4 Static scheme of the timber tie-beam (clamped ends)

#### Reactions used in the scheme:

• Seismic load:

$$q_{\alpha} = \frac{Mass_{i} * \alpha * g}{L}$$

• T chain, shear force:

$$T_{chain} = \frac{q_{\alpha} * l}{2}$$

• M chain, bending moment:

$$M_{chain} = \frac{q_{\alpha} * l^2}{12}$$

All the this terms will be explained in the following sub-chapters.

# 10.1.3 Hyperstatic scheme of the corner joint and actions from static scheme of the timber tie- beam

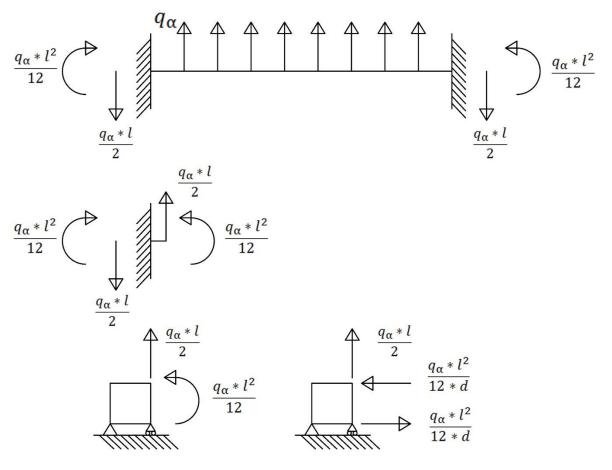


Figure 10-5 Hyperstatic scheme of the corner joint and actions from static scheme of the timber tie- beam

The new reaction in the scheme is the couple (from the bending moment Mchain ):

$$Couple = \frac{q_{\alpha} * l^2}{12 * d}$$

#### 10.1.4 Hyperstatic rigid-jointed frame

In order to study the behavior of the corner joint composed by four crossed rafters it has been solved the following frame with the listed nomenclature of the forces.

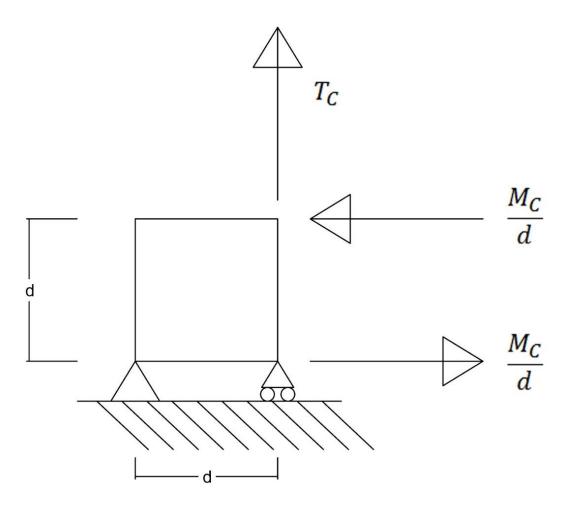


Figure 10-6 Hyperstatic rigid-jointed frame

#### Reactions used in the scheme:

• Seismic load:

$$q_{\alpha} = \frac{Mass_{i} * \alpha * g}{L}$$

• T chain, shear force:

$$T_c = \frac{q_\alpha * l}{2}$$

• M chain, bending moment:

$$M_c = \frac{q_\alpha * l^2}{12}$$

• Couple, bending moment

$$\frac{M_C}{d} = \frac{q_\alpha * l^2}{12 * d}$$

#### 10.2Force method with Müller-Breslau equations

The frame has been solved by the use of the force method using the Müller-Breslau equations. In the following sub-chapters they have been reported the steps for solving the structure and the solutions which have been fundamental for the flexible configuration analysis and the verifications of the timber elements.

#### 10.2.1 Force method

#### Method Procedure:

- 1. Determine the degree of static indeterminacy.
  - Number of releases\* equal to the degree of static indeterminacy are applied to the structure.
  - Released structure is referred to primary structure.
  - Primary structure must be chosen such that it is geometrically stable and statically determinate.
- 2. Calculate "errors" (displacements) at the primary structure redundants. These displacements are calculated using the method of virtual forces.
- 3. Determine displacements in the primary structure due to unit values of redundants (method of virtual forces). These displacements are required at the same location and in the same direction as the displacement errors determined in step 2.
- 4. Calculate redundant forces to eliminate displacement errors.
  - Use superposition equations (Müller-Breslau equations ) in which the effects of the separate redundants are added to the displacements of the released structure.
  - Displacement superposition results in a set of n linear equations (n = number of releases) that express the fact that there is zero relative displacement at each release.
  - These compatibility equations guarantee a final displaced shape consistent with known support conditions, i.e., the structure fits together at the n releases with no relative displacements.
- 5. Hence, we find the forces on the original indeterminate structure. They are the sum of the correction forces (redundants) and forces on the released structure.

#### 10.2.2 Degree of indeterminacy Rigid-Jointed Frame

Description of the Rigid-Jointed Frame

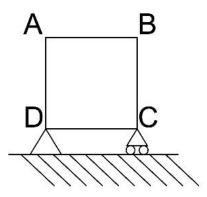


Figure 10-7 Rigid-Jointed Frame - names of the corners

The structure is externally statically determinate but internally statically indeterminate.

n: number of rigid joints n = 4

m: number members m = 4

r: support reactions r = 3

i: degree of indeterminacy i = ?

$$i = [(3 * m) + r] - 3 * n$$

$$i = [(3 * 4) + 3] - 3 * 4 = 3$$

The internal degree of indeterminacy is i = 3.

#### 10.2.2.1 Primary structure, from indeterminate system to a determinate one

Conversion of the indeterminate structure to a determinate one by removing 3 unknown forces and replacing them with (assumed) known / unit forces.

#### Static System

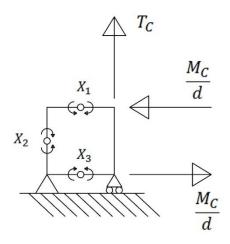
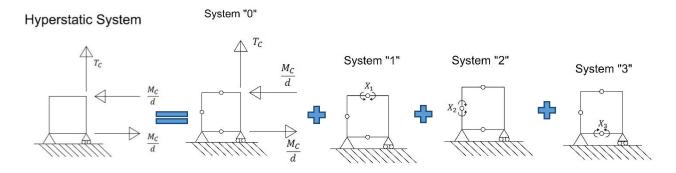


Figure 10-8 Primary structure - Static system



 $Figure\ 10-9\ Decomposition\ of\ the\ redundant\ frame$ 

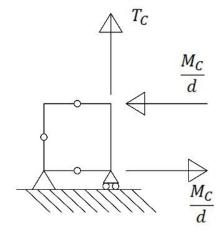
#### 10.2.3 Solved released systems

#### 10.2.3.1 System 0

The primary structure is the released structure shown in the figure below and it is named System0.

They have been computed the reaction

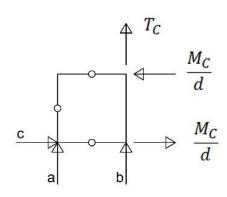
# System "0"



#### Figure 10-10 System

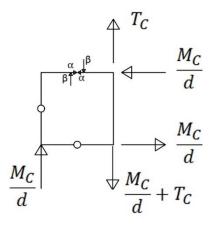
# $\begin{cases} a+b=-Tc \\ c=0 \\ Tc*d+b*d+Mc=0 \end{cases}$ $\begin{cases} a=\frac{Mc}{d} \\ c=0 \\ b=-\frac{Mc}{d}-Tc \end{cases}$

# External Equilibrium System "0"



0 Figure 10-11 External Equilibrium System "0"

# Internal Equilibrium System "0"



Figure~10-12~Internal~~Equilibrium~System~"0"

$$\begin{cases} -\alpha * \frac{d}{2} + \beta \frac{d}{2} = 0\\ \alpha * d + Tc * \frac{d}{2} + Mc - \frac{Mc}{2} - Tc * \frac{d}{2} = 0 \end{cases}$$

$$\begin{cases} \alpha = -\frac{Mc}{2d} \\ \beta = -\frac{Mc}{2} \end{cases}$$

Normal force System "0"

Shear force System "0" Bending Moment System "0"

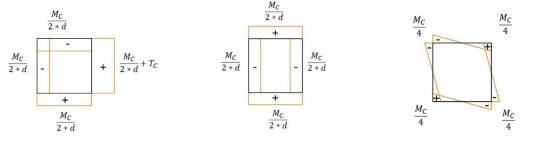


Figure 10-13 Internal reactions System "0"

#### 10.2.3.2 System 1

The released structure with the addition of the the redoundant  $X_1 = 1$  structure shown in the figure below and it is named System 1.

They have been computed the reaction

# System "1"

# External Equilibrium System "1"

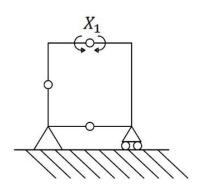


Figure 10-14 System 1

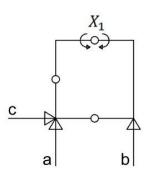


Figure 10-15 External Equilibrium System "1"

$$\begin{cases} a = 0 \\ b = 0 \\ c = 0 \end{cases}$$

# Internal Equilibrium System "1"

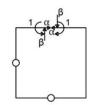


Figure 10-16 Internal Equilibrium System "1"

$$\begin{cases} \alpha*d-1=0\\ -\alpha*\frac{d}{2}+\beta*\frac{d}{2}=0 \end{cases} \qquad \begin{cases} \alpha=\frac{1}{d}\\ \beta=-\frac{1}{d} \end{cases}$$

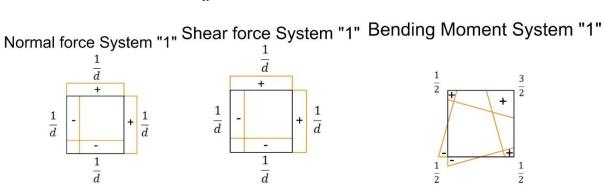


Figure 10-17 Internal Reactions System "1"

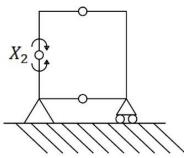
#### 10.2.3.3 System 2

The released structure with the addition of the the redoundant  $X_2 = 1$  structure shown in the figure below and it is named System 2.

They have been computed the reactions

# System "2"

# External Equilibrium System "2"



a

Figure 10-18 System 2

Figure 10-19 External Equilibrium System "2"

$$\begin{cases}
a = 0 \\
b = 0 \\
c = 0
\end{cases}$$

# Internal Equilibrium System "2"

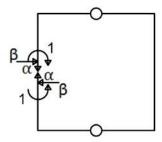
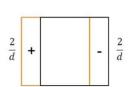


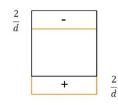
Figure 10-20 Internal Equilibrium System "2"

$$\begin{cases} \alpha * \frac{d}{2} - \beta * \frac{d}{2} - 1 = 0 \\ -\alpha * \frac{d}{2} - \beta * \frac{d}{2} + 1 = 0 \end{cases} \begin{cases} \alpha = \frac{2}{d} \\ \beta = 0 \end{cases}$$

Bending Moment System "2"

Normal force System "2" Shear force System "2"





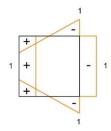


Figure 10-21 Internal Reactions System "2"

#### 10.2.3.4 System 3

The released structure with the addition of the the redoundant  $X_3 = 1$  structure shown in the figure below and it is named System 3.

They have been computed the reaction

System "3"

### External Equilibrium System "3"

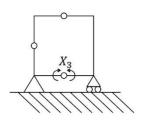


Figure 10-22 System "3"

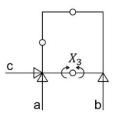


Figure 10-23 External Equilibrium System "3"

$$\begin{cases} a = 0 \\ b = 0 \\ c = 0 \end{cases}$$

# Internal Equilibrium System "3"

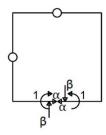


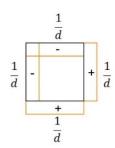
Figure 10-24 Internal Equilibrium System "3"

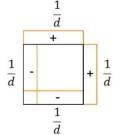
$$\begin{cases} \alpha * \frac{d}{2} - \beta * \frac{d}{2} - 1 = 0 \\ -\alpha * \frac{d}{2} - \beta * \frac{d}{2} + 1 = 0 \end{cases}$$

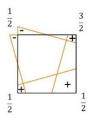
$$\begin{cases} \alpha = \frac{2}{d} \\ \beta = 0 \end{cases}$$

Normal force System "3" Shear force System "3"  $\frac{1}{d}$ 

Bending Moment System "3"



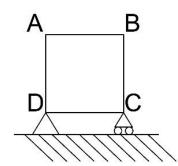




Figure~10-25~Internal~reactions~System~"3"

#### 10.2.4 Functions of the diagrams

For each system, they have been written the functions that describes the behavior of the forces, focus on each member.



$$N_{AB}^{0} = -\frac{Mc}{2d} \quad N_{BC}^{0} = \frac{Mc}{2d} + Tc \qquad N_{CD}^{0} = \frac{Mc}{2d} \qquad N_{DA}^{0} = -\frac{Mc}{2d}$$

$$M_{AB}^{0} = \left(-\frac{1}{4} + \frac{s}{2d}\right) * Mc \qquad M_{BC}^{0} = \left(\frac{1}{4} - \frac{s}{2d}\right) * Mc \qquad M_{CD}^{0} = \left(-\frac{1}{4} + \frac{s}{2d}\right) * Mc \qquad M_{DA}^{0} = \left(\frac{1}{4} - \frac{s}{2d}\right) * Mc$$

$$MC$$

#### 10.2.4.2 System 1

$$\begin{split} N_{AB}^1 &= \frac{1}{d} & N_{BC}^1 = \frac{1}{d} & N_{CD}^1 = -\frac{1}{d} & N_{DA}^1 = -\frac{1}{d} \\ M_{AB}^1 &= \left(\frac{1}{2} + \frac{s}{d}\right) & M_{BC}^1 &= \left(\frac{3}{2} - \frac{s}{d}\right) & M_{CD}^1 &= \left(\frac{1}{2} - \frac{s}{d}\right) \\ M_{DA}^2 &= \left(-\frac{1}{2} + \frac{s}{d}\right) & M_{DA}^2 &= \left(-\frac{1}{2} + \frac{s}{d}\right) \end{split}$$

#### 10.2.4.3 System 2

$$N_{AB}^2 = 0$$
  $N_{BC}^2 = -\frac{2}{d}$   $N_{CD}^2 = 0$   $N_{DA}^2 = \frac{2}{d}$   $M_{AB}^2 = \left(1 - \frac{2s}{d}\right)$   $M_{BC}^2 = -1$   $M_{CD}^2 = \left(-1 + \frac{2s}{d}\right)$   $M_{DA}^2 = 1$ 

#### 10.2.4.4 System 3

$$N_{AB}^{1} = -\frac{1}{d} \qquad N_{BC}^{1} = \frac{1}{d} \qquad N_{CD}^{1} = +\frac{1}{d} \qquad N_{DA}^{1} = -\frac{1}{d}$$

$$M_{AB}^{1} = \left(-\frac{1}{2} + \frac{s}{d}\right) \qquad M_{BC}^{1} = \left(\frac{1}{2} + \frac{s}{d}\right) \qquad M_{CD}^{1} = \left(\frac{3}{2} - \frac{s}{d}\right) \qquad M_{DA}^{1} = \left(\frac{1}{2} - \frac{s}{d}\right)$$

#### 10.2.5 Müller-Breslau equations

Base on the linearity of the problem they have been used the Müller-Breslau equations for the compatibility.

 $\eta_i$ : is the effective displacement in the effective structure

 $\eta_{i0}$  : is the displacement due to the primary system on the i released

 $X_i$ : is the unitary force in the position of the i released

 $\eta_{ik}$ : is the displacement of the point of application of the released  $X_i$  due to the redoundant  $X_k = 1$ 

n: is the number of the released equal to the degree if indeterminacy i

$$\eta_i = \eta_{i0} + \sum_{1}^{n} \eta_{ik} * X_k$$

Thus, the 3 equation of Müller-Breslau that assures the compatibility are:

$$\begin{cases} \eta_1 = \eta_{10} + \eta_{11} * X_1 + \eta_{12} * X_2 + \eta_{13} * X_3 \\ \eta_2 = \eta_{20} + \eta_{21} * X_1 + \eta_{22} * X_2 + \eta_{23} * X_3 \\ \eta_3 = \eta_{30} + \eta_{31} * X_1 + \eta_{32} * X_2 + \eta_{33} * X_3 \end{cases}$$

Using the theorem of virtual work, it is possible to compute all the displacements as follows.

$$\eta_{i0} = \int \left( \frac{N_i * N_0}{EA} + \frac{T_i * T_0}{GK} + \frac{M_i * M_0}{EJ} \right) ds$$

$$\eta_{ik} = \int \left( \frac{N_i * N_k}{EA} + \frac{T_i * T_k}{GK} + \frac{M_i * M_k}{EI} \right) ds$$

The contribution of the shear forces are negligible with the assumption that  $GK = \infty$ .

$$\eta_{i0} = \int \left(\frac{N_i * N_0}{EA} + \frac{M_i * M_0}{EJ}\right) ds$$

$$\eta_{ik} = \int \left(\frac{N_i * N_k}{EA} + \frac{M_i * M_k}{EI}\right) ds$$

It is important to underline the following observations:

$$\eta_{ii} = \int \left(\frac{N_i * N_i}{EA} + \frac{M_i * M_i}{EJ}\right) ds = \int \left(\frac{N_i^2}{EA} + \frac{M_i^2}{EJ}\right) ds > 0$$

 $\eta_{ik} = \eta_{ki}$  due to Maxwell Theorem

Then the coefficient matrix is symmetric and all the diagonal elements are positive.

#### 10.2.5.1 Displacement coefficients

$$\begin{split} \eta_{10} &= \int \left(\frac{N_1 * N_0}{EA} + \frac{M_1 * M_0}{EJ}\right) ds \\ &= \int \left(\frac{N_{AB}^1 * N_{AB}^0 + N_{BC}^1 * N_{BC}^0 + N_{CD}^1 * N_{CD}^0 + N_{DA}^1 * N_{DA}^0}{EA} \right. \\ &+ \frac{M_{AB}^1 * M_{AB}^0 + M_{BC}^1 * M_{BC}^0 + M_{CD}^1 * M_{CD}^0 + M_{DA}^1 * M_{DA}^0}{EJ} \right) ds = \frac{Tc}{EA} \end{split}$$

$$\begin{split} \eta_{20} &= \int \left(\frac{N_2 * N_0}{EA} + \frac{M_2 * M_0}{EJ}\right) ds \\ &= \int \left(\frac{N_{AB}^2 * N_{AB}^0 + N_{BC}^2 * N_{BC}^0 + N_{CD}^2 * N_{CD}^0 + N_{DA}^2 * N_{DA}^0}{EA} \right. \\ &+ \frac{M_{AB}^2 * M_{AB}^0 + M_{BC}^2 * M_{BC}^0 + M_{CD}^2 * M_{CD}^0 + M_{DA}^2 * M_{DA}^0}{EJ} \right) ds = -\frac{2}{EA} * \left(Tc + \frac{Mc}{d}\right) \end{split}$$

$$\begin{split} \eta_{30} &= \int \left(\frac{N_3 * N_0}{EA} + \frac{M_3 * M_0}{EJ}\right) ds \\ &= \int \left(\frac{N_{AB}^3 * N_{AB}^0 + N_{BC}^3 * N_{BC}^0 + N_{CD}^3 * N_{CD}^0 + N_{DA}^3 * N_{DA}^0}{EA} \right. \\ &+ \frac{M_{AB}^3 * M_{AB}^0 + M_{BC}^3 * M_{BC}^0 + M_{CD}^3 * M_{CD}^0 + M_{DA}^3 * M_{DA}^0}{EJ} ds = \frac{1}{EA} * \left(Tc + \frac{2Mc}{d}\right) \end{split}$$

$$\eta_{11} = \int \left(\frac{N_1^2}{EA} + \frac{M_1^2}{EJ}\right) ds 
= \int \left(\frac{N_{AB}^{1}^2 + N_{BC}^{1}^2 + N_{CD}^{1}^2 + N_{DA}^{1}^2}{EA} + \frac{M_{AB}^{1}^2 + M_{BC}^{1}^2 + M_{CD}^{1}^2 + M_{DA}^{1}^2}{EJ}\right) ds 
= \frac{4}{d} * \frac{1}{EA} + \frac{7}{3} * \frac{d}{EJ}$$

$$\eta_{22} = \int \left(\frac{N_2^2}{EA} + \frac{M_2^2}{EJ}\right) ds 
= \int \left(\frac{N_{AB}^2 + N_{BC}^2 + N_{CD}^2 + N_{DA}^2}{EA} + \frac{M_{AB}^2 + M_{BC}^2 + M_{CD}^2 + M_{DA}^2}{EJ}\right) ds 
= \frac{8}{d} * \frac{1}{EA} + \frac{8}{3} * \frac{d}{EJ}$$

$$\eta_{33} = \int \left(\frac{N_3^2}{EA} + \frac{M_3^2}{EJ}\right) ds 
= \int \left(\frac{N_{AB}^3^2 + N_{BC}^3^2 + N_{CD}^3^2 + N_{DA}^3^2}{EA} + \frac{M_{AB}^3^2 + M_{BC}^3^2 + M_{CD}^3^2 + M_{DA}^3^2}{EJ}\right) ds 
= \frac{4}{d} * \frac{1}{EA} + \frac{7}{12} * \frac{d}{EJ}$$

$$\begin{split} \eta_{12} &= \eta_{21} = \int \left( \frac{N_1 * N_2}{EA} + \frac{M_1 * M_2}{EJ} \right) ds \\ &= \int \left( \frac{N_{AB}^1 * N_{AB}^2 + N_{BC}^1 * N_{BC}^2 + N_{CD}^1 * N_{CD}^2 + N_{DA}^1 * N_{DA}^2}{EA} \right. \\ &\quad + \frac{M_{AB}^1 * M_{AB}^2 + M_{BC}^1 * M_{BC}^2 + M_{CD}^1 * M_{CD}^2 + M_{DA}^1 * M_{DA}^2}{EJ} \right) ds = -\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ} \\ \eta_{13} &= \eta_{31} = \int \left( \frac{N_1 * N_3}{EA} + \frac{M_1 * M_3}{EJ} \right) ds \\ &= \int \left( \frac{N_{AB}^1 * N_{AB}^3 + N_{BC}^1 * N_{BC}^3 + N_{DD}^1 * N_{CD}^3 + N_{DA}^1 * N_{DA}^3}{EA} \right. \\ &\quad + \frac{M_{AB}^1 * M_{AB}^3 + M_{BC}^1 * M_{BC}^3 * M_{BC}^3 + M_{CD}^1 * M_{CD}^3 + M_{DA}^1 * M_{DA}^3}{EJ} \right) ds \\ &= \int \left( \frac{N_2 * N_3}{EA} + \frac{M_2 * M_3}{EJ} \right) ds \\ &= \int \left( \frac{N_{AB}^2 * N_{AB}^3 + N_{BC}^2 * N_{BC}^3 + N_{CD}^2 * N_{CD}^3 + N_{DA}^2 * N_{DA}^3}{EA} \right. \\ &\quad + \frac{M_{AB}^2 * M_{AB}^3 + M_{BC}^2 * M_{BC}^3 * M_{BC}^3 + M_{CD}^2 * M_{DA}^3 + N_{DA}^2 * M_{DA}^3}{EA} \\ &\quad + \frac{M_{AB}^2 * M_{AB}^3 + M_{BC}^2 * M_{BC}^3 + M_{CD}^2 * M_{CD}^3 + M_{DA}^2 * M_{DA}^3}{EI} \right) ds = -\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EI} \end{split}$$

$$\begin{cases} \eta_1 = \frac{Tc}{EA} + \left(\frac{4}{d} * \frac{1}{EA} + \frac{7}{3} * \frac{d}{EJ}\right) * X_1 + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_2 + \left(\frac{1}{2} * \frac{d}{EJ}\right) * X_3 \\ \eta_2 = -\frac{2}{EA} * \left(Tc + \frac{Mc}{d}\right) + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_1 + \left(\frac{8}{d} * \frac{1}{EA} + \frac{8}{3} * \frac{d}{EJ}\right) * X_2 + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_3 \\ \eta_3 = \frac{1}{EA} * \left(Tc + \frac{2Mc}{d}\right) + \left(\frac{1}{2} * \frac{d}{EJ}\right) * X_1 + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_2 + \left(\frac{4}{d} * \frac{1}{EA} + \frac{7}{12} * \frac{d}{EJ}\right) * X_3 \end{cases}$$

They have been released internal actions this means the displacements are null because they are mutual.

$$\eta_{1} = 0 \qquad \eta_{2} = 0 \qquad \eta_{3} = 0$$

$$0 = \frac{Tc}{EA} + \left(\frac{4}{d} * \frac{1}{EA} + \frac{7}{3} * \frac{d}{EJ}\right) * X_{1} + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_{2} + \left(\frac{1}{2} * \frac{d}{EJ}\right) * X_{3}$$

$$0 = -\frac{2}{EA} * \left(Tc + \frac{Mc}{d}\right) + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_{1} + \left(\frac{8}{d} * \frac{1}{EA} + \frac{8}{3} * \frac{d}{EJ}\right) * X_{2} + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_{3}$$

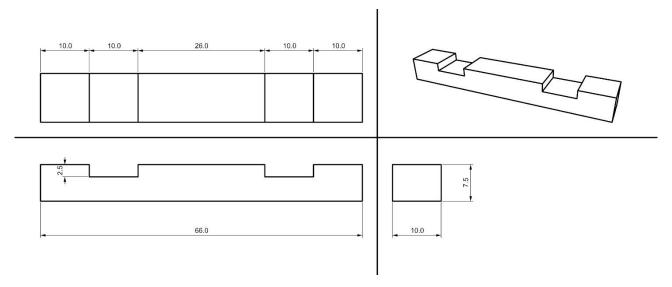
$$0 = \frac{1}{EA} * \left(Tc + \frac{2Mc}{d}\right) + \left(\frac{1}{2} * \frac{d}{EJ}\right) * X_{1} + \left(-\frac{4}{d} * \frac{1}{EA} - \frac{4}{3} * \frac{d}{EJ}\right) * X_{2} + \left(\frac{4}{d} * \frac{1}{EA} + \frac{7}{12} * \frac{d}{EJ}\right) * X_{3}$$

### 10.2.5.2 Axial rigidity (EA) and flexural rigidity (EJ)

The Young modulus considered is the design Young modulus parallel to the fibers for the Shorea robusta timber.

$$E_{0,d} = 15.38 \left[ \frac{kN}{mm^2} \right] = 15384615 \left[ \frac{kN}{m^2} \right] = 15384615 * 10^3 \left[ \frac{N}{m^2} \right]$$

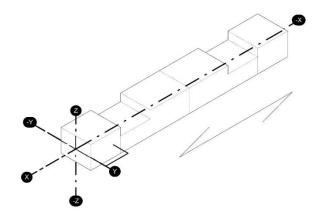
It can be assumed that the rigid joint frames studied is composed by cross pieces. Following measures of the cross piece are in cm.



Data of the studied section:

$$b = 0.075 m$$

$$h = 0.1 \, m$$



Moment of inertia around Z axis:

$$J_Z = \frac{b * h^3}{12} = \frac{0.075m * (0.1m)^3}{12} = 6.25 * 10^{-6}m^4$$

Area of the considered section A:

$$A = b * h = 0.075m * 0.1m = 7.5 * 10^{-3}m^2$$

## 10.2.5.3 Seismic distributed load $q_{\alpha}$

The uniformly distributed load considered for the overturning with flexible body has been computed as following.

$$q_{\alpha} = \frac{Mass_{i} * g * \alpha}{L} \qquad \left[\frac{N}{m}\right]$$

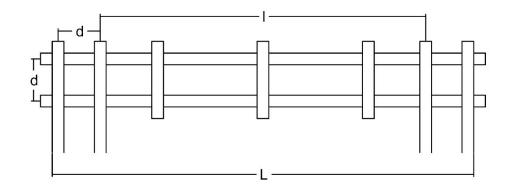
Where

 $Mass_i$ : is the mass involved for the specific tie-timber beam

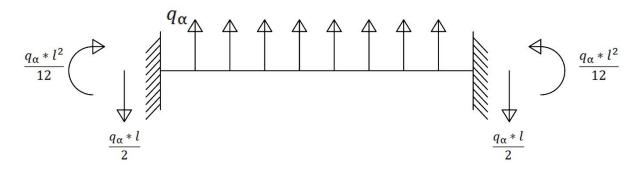
g: is the gravity accelleration

 $\alpha$ : is the seismic load multiplier

L: is the length of the wall



10.2.5.4 Seismic shear force (Tc) and Seismic bending moment (Mc)



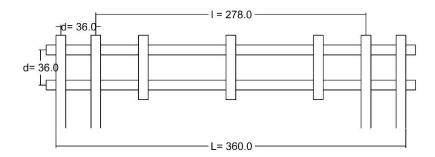
Seismic shear force is named Tc

$$T_C = \frac{q_{\alpha} * l}{2} \qquad [N]$$

Seismic bending moment is named Mc

$$M_C = \frac{q_\alpha * l^2}{12} \quad [Nm]$$

In order to have the homogenous coefficients depending on the seismic uniformly distributed load  $q_{\alpha}$ , the actions have been written substituting the proper wall geometrical value.



L: is the length of the wall equal to 3.6 m

l: is the length of the wall where the load is distributed, equal to 2.78 m

d: is the distances between all the timber elements, equal to 0.36 m

Thus

$$T_C = \frac{q_{\alpha} * l}{2} = q_{\alpha} \frac{2,78m}{2} = q_{\alpha} \frac{139}{100} [N]$$

$$M_C = \frac{q_{\alpha} * l^2}{12} = q_{\alpha} \frac{(2,78m)^2}{12} = q_{\alpha} \frac{19321}{30000} [Nm]$$

### 10.2.5.5 Solutions of Müller-Breslau equations

Substituting all the known values it has been obtained the following linear system in function of the seismic load.

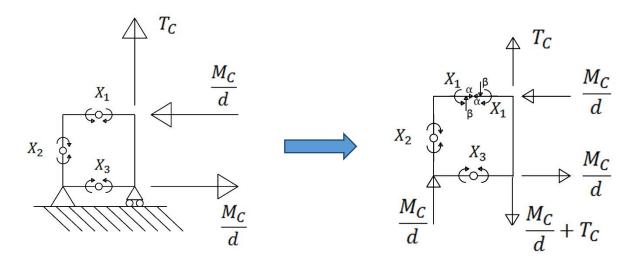
$$\begin{cases} 0 = -185.33 * q_{\alpha} + (135881,48) * X_1 + (-78281,48) * X_2 + (28800,00) * X_3 \\ 0 = 847.73 * q_{\alpha} + (-78281,48) * X_1 + (156562,96) * X_2 + (-78281,48) * X_3 \\ 0 = -662.4 * q_{\alpha} + (28800,00) * X_1 + (-78281,48) * X_2 + (35081,48) * X_3 \end{cases}$$

The solutions of the system depends on the seismic load  $q_{\alpha}$ .

$$\begin{cases} X_1 = 6.87 * 10^{-3} * q_{\alpha} \\ X_2 = 2.95 * 10^{-2} * q_{\alpha} \\ X_3 = 4.12 * 10^{-2} * q_{\alpha} \end{cases}$$

### 10.2.6 Solutions of the complete isostatic structure

## Static System



They have been obtained the internal forces by equilibrium.

$$\begin{cases} X_3 - \frac{Mc}{d} * \frac{d}{2} - Tc * \frac{d}{2} + Mc + Tc * \frac{d}{2} + \alpha * d - X_1 = 0 \\ -X_2 + X_1 - \alpha * \frac{d}{2} + \beta * \frac{d}{2} = 0 \end{cases}$$

$$\begin{cases} \alpha = \frac{X_1}{d} - \frac{X_3}{d} - \frac{Mc}{2d} \\ \beta = -\frac{X_1}{d} + 2 * \frac{X_2}{d} - \frac{X_3}{d} - \frac{Mc}{2d} \end{cases}$$

$$\begin{cases} \alpha = -0.99 * q_{\alpha} \\ \beta = -0.86 * q_{\alpha} \end{cases}$$

10.2.6.1 Normal, shear and bending moment diagrams in function of seismic load  $q_{\alpha}$ 

$$T_C = \frac{139}{100} * q_{\alpha} [N]$$

$$M_C = \frac{19321}{30000} * q_{\alpha} [Nm]$$

$$X_1 = 6.87 * 10^{-3} * q_\alpha [Nm]$$

$$X_2 = 2.95 * 10^{-2} * q_{\alpha} [Nm]$$

$$X_3 = 4.12 * 10^{-2} * q_{\alpha} [Nm]$$

$$\alpha = -0.99 * q_{\alpha} [N]$$

$$\beta = -0.86 * q_{\alpha} [N]$$

Normal forces

$$N_{AB} = \alpha = -0.99 * q_{\alpha} [N]$$

$$N_{BC} = Tc - \beta = \frac{139}{100} * q_{\alpha} + 0.86 * q_{\alpha} = 2.25 * q_{\alpha}[N]$$

$$N_{CD} = -\alpha = 0.99 * q_{\alpha}[N]$$

$$N_{DA} = \beta = -0.86 * q_{\alpha}[N]$$

Shear forces

$$T_{AB} = -\beta = 0.86 * q_{\alpha}[N]$$

$$T_{BC} = -\alpha - \frac{Mc}{d} = 0.99 * q_{\alpha} - \frac{19321}{30000} * \frac{100}{36} * q_{\alpha} = -0.8 * q_{\alpha}[N]$$

$$T_{CD} = \beta + \frac{Mc}{d} = -0.86 * q_{\alpha} + \frac{19321}{30000} * \frac{100}{36} * q_{\alpha} = 0.92 * q_{\alpha}[N]$$

$$T_{DA} = \alpha = -0.99 * q_{\alpha}[N]$$

Bending moments (in the corner rigid joints)

$$M_{inA} = X_1 + \beta * \frac{d}{2} = 6.87 * 10^{-3} * q_{\alpha} - 0.86 * \frac{36}{100} * \frac{1}{2} * q_{\alpha} = -0.15 * q_{\alpha}[Nm]$$

$$M_{inB} = X_1 - \beta * \frac{d}{2} = 6.87 * 10^{-3} * q_{\alpha} + 0.86 * \frac{36}{100} * \frac{1}{2} * q_{\alpha} = 0.16 * q_{\alpha}[Nm]$$

$$M_{inc} = X_1 - \beta * \frac{d}{2} - \alpha * d - Mc$$

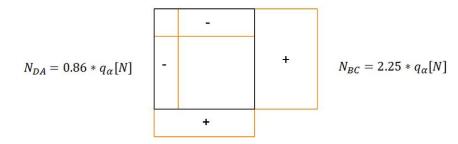
$$= 6.87 * 10^{-3} * q_{\alpha} + 0.86 * \frac{36}{100} * \frac{1}{2} * q_{\alpha} + 0.99 * \frac{36}{100} * q_{\alpha} - \frac{19321}{30000} * q_{\alpha}$$

$$= -0.13 * q_{\alpha}[Nm]$$

$$\begin{split} M_{inD} &= X_1 + \beta * \frac{d}{2} - \alpha * d = 6.87 * 10^{-3} * q_{\alpha} - 0.86 * \frac{36}{100} * \frac{1}{2} * q_{\alpha} + 0.99 * \frac{36}{100} * q_{\alpha} \\ &= 0.21 * q_{\alpha}[Nm] \end{split}$$

## Normal force

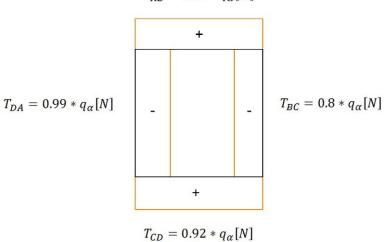
$$N_{AB} = 0.99 * q_{\alpha} [N]$$



$$N_{CD} = 0.99 * q_{\alpha}[N]$$

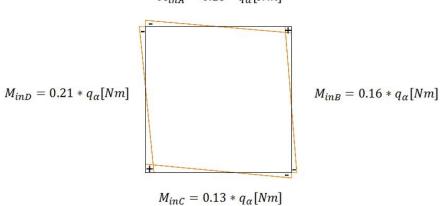
## Shear force

$$T_{AB} = 0.86 * q_{\alpha}[N]$$



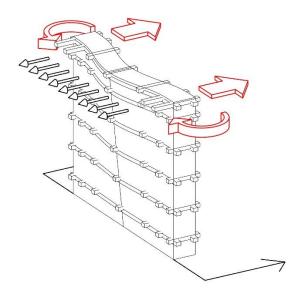
# **Bending Moment**

$$M_{inA} = 0.15 * q_{\alpha}[Nm]$$



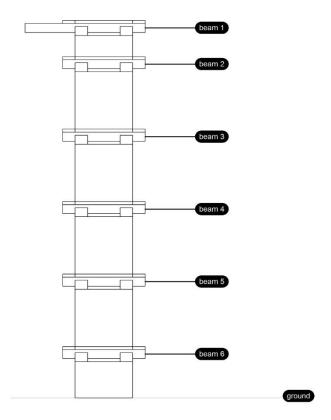
## 10.3 Triangular distribution of seismic load $q_{\alpha}$

In the case of the bending behavior wall, the seismic forces has been set with the only triangular distribution over the height. This configuration is the only one possible for the hand calculation, other possible distribution would be possible with the modal analysis method but they would need a numerical approach.



## 10.3.1 Scheme of wall Flexible response – Bendingbehavior flexible behavior

They have been named and numbered the analyzed beam in a similar way used for the rigid behavior.



 $Figure~10\text{-}26~Flexible~response-Bending behavior} \text{-} Analyzed~beams$ 

## 10.3.2 Masses involved and heights of each timber beam

They have been assigned the pertinent masses at each timber band.

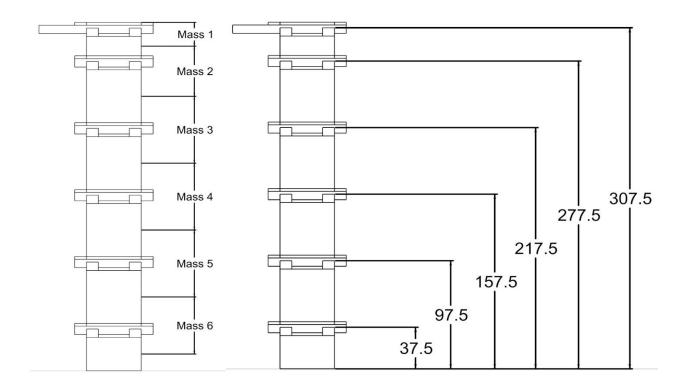


Figure 10-27 Bending behavior Pertinent masses for each timber band Figure 10-28 Bending behavior Heights of each timber band Table 58 Pertinent masses foe each timber bands

	Mass for each Force 3 module				
		KN	Kg		
Mass 1	Wroof module	51,23	5222,31		
	W1	6,70	683,48		
Mass 2	W2	14,52	1480,36		
Mass 3	W3	19,05	1941,72		
Mass 4	W4	19,05	1941,72		
Mass 5	W5	19,05	1941,72		
Mass 6	W6	16,17	1648,13		
	W7	4,94	503,29		

## 10.3.3 Seismic load $q_{\alpha}$ and Distribution factor $\beta_{j}$

The distribution factor for the triangular distribution has been obtained with the procedure descripted in the chapter 8.2.1.2 . Here are reported the main equations to compute the distribution factors and the corresponding seismic loads.

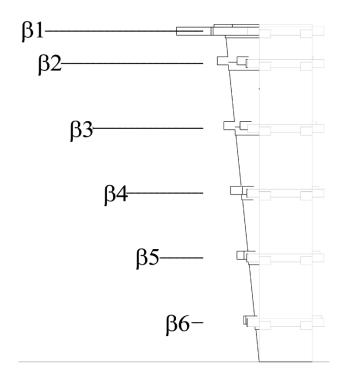


Figure 10-29 Bending behavior -Distribution factors

$$q = \frac{Mtot *g}{L} = \frac{Wtot}{L}$$

$$q_{\alpha} = q * \alpha = \frac{Mtot * g}{L} * \alpha = \frac{Wtot * g}{L} * \alpha$$

$$\beta_{j} = \frac{W_{j} * h_{j}}{\sum_{i=1}^{N} W_{i} * h_{i} + W_{roof} * H}$$

$$q_{\alpha j} = q_{\alpha} * \beta_{j} = q_{\alpha} * \frac{W_{j} * h_{j}}{\sum_{i=1}^{N} W_{i} * h_{i} + W_{roof} * H}$$

Table 59 Bending behavior - Distribution of the weight over the height

Distribution of the weight over the height						
	W	Height of Force : hi	Wi * hi			
Wi	kN	m	kN*m			
Wroof	51,23	3,075	157,53			
W1	6,70	3,075	20,62			
W2	14,52	2,775	40,30			
W3	19,05	2,175	41,43			
W4	19,05	1,575	30,00			
W5	19,05	0,975	18,57			
W6	16,17	0,375	6,06			

Table 60 Bending behavior - Wall lenght

Wall length		
L	3,60	m

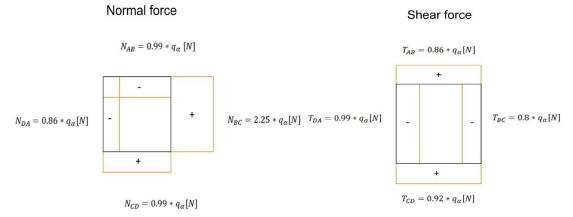
Table 61 Bending behavior - Distribution factors and seismic loads

Distribution factors and seismic loads		
qαj	βj	qαj=q*βj
qa1	0,57	23,71
qa2	0,13	5,36
qa3	0,13	5,51
qα4	0,10	3,99
qα5	0,06	2,47
<b>q</b> α6	0,02	0,81

## 10.4 Reactions for each beam

In this sub-chapter they have been recalled the results obtained by the force method and the seismic loads depending on the load multiplier in order to obtain the reactions on all the heads of the rafters which cross in the corner joint.

## 10.4.1 Rafter body reactions for each beam in the corner joint



#### **Bending Moment**

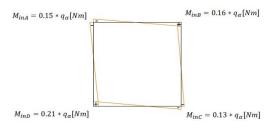


Table 62 Flexible behavior - Rafter body reactions for each beam

			Normal	force		Shear force			Bending Moment				
	q(a)j with	Nab	Nbc	Ncd	Nda	Tab	Tbc	Tcd	Tda	BM in A	BM in B	BM in C	BM in D
Beam	$\alpha=1 [kN/m]$	[kN]	[kN]	[kN]	[kN]	[kN]	[kN]	[kN]	[kN]	[kNm]	[kNm]	[kNm]	[kNm]
1	23,71	-23,47	53,46	23,47	-20,50	20,50	-18,95	21,92	-23,47	-3,53	3,85	-2,97	4,92
2	5,36	-5,31	12,09	5,31	-4,64	4,64	-4,29	4,96	-5,31	-0,80	0,87	-0,67	1,11
3	5,51	-5,46	12,43	5,46	-4,77	4,77	-4,41	5,10	-5,46	-0,82	0,90	-0,69	1,15
4	3,99	-3,95	9,00	3,95	-3,45	3,45	-3,19	3,69	-3,95	-0,59	0,65	-0,50	0,83
5	2,47	-2,45	5,57	2,45	-2,14	2,14	-1,98	2,29	-2,45	-0,37	0,40	-0,31	0,51
6	0,81	-0,80	1,82	0,80	-0,70	0,70	-0,64	0,75	-0,80	-0,12	0,13	-0,10	0,17

## 10.4.2 Rigid- jointed frame reactions for each beam

In this configuration, the computed values regard the whole longitudinal tie-timber beam composed by 2 rafters body. In order to have the values for 1 timber rafter body, 2 must divide the values in the following tables.

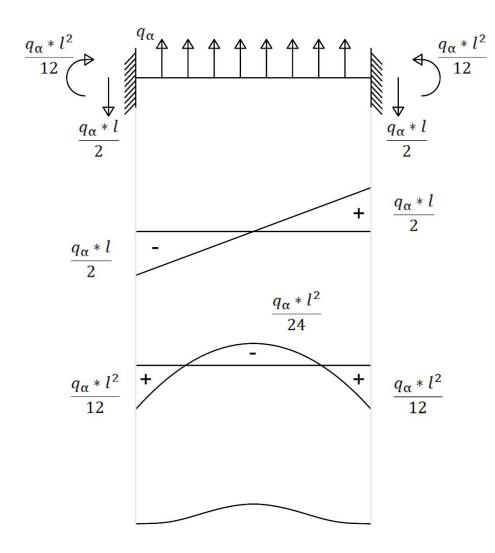


Figure 10-30 Bending behavior - Rigid- jointed frame reactions for each beam-

Table 63 Bending behavior - Rigid- jointed frame reactions for each beam

	Shear forc		rce [kN]	ce [kN] Bending Moment [kNi		Nm]
		end left	end right	end left	midpoint	end right
Beam	$q(\alpha)j$ with $\alpha=1$ [kN/m]	$q(\alpha)*(1)/2$	$q(\alpha)*(1)/2$	$q(\alpha)*(1^2)/12$	$q(\alpha)*(1^2)/24=$	$q(\alpha)*(1^2)/12$
1	23,71	-32,96	32,96	15,27	-7,64	15,27
2	5,36	-7,46	7,46	3,45	-1,73	3,45
3	5,51	-7,67	7,67	3,55	-1,78	3,55
4	3,99	-5,55	5,55	2,57	-1,29	2,57
5	2,47	-3,44	3,44	1,59	-0,80	1,59
6	0,81	-1,12	1,12	0,52	-0,26	0,52

For the verifications of the rafters, belonging to the failing wall it has been used the following table

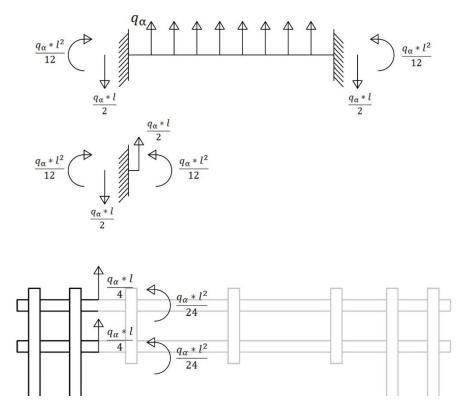


Figure 10-31 Bending behavior - Rigid- jointed frame reactions for each beam - Resisting rafters R

Table 64 Bending behavior - Rigid- jointed frame reactions for each beam - Resisting rafters R

		Shear force [kN]		Bending Moment [kNm]		
		end left	end right	end left	midpoint	end right
Beam	$q(\alpha)j$ with $\alpha=1$ [kN/m]	$q(\alpha)*(1)/4$	$q(\alpha)*(1)/4$	$q(\alpha)*(1^2)/24$	$q(\alpha)*(1^2)/48$	$q(\alpha)*(1^2)/24$
1	23,71	-16,48	16,48	7,64	-3,82	7,64
2	5,36	-3,73	3,73	1,73	-0,86	1,73
3	5,51	-3,83	3,83	1,78	-0,89	1,78
4	3,99	-2,78	2,78	1,29	-0,64	1,29
5	2,47	-1,72	1,72	0,80	-0,40	0,80
6	0,81	-0,56	0,56	0,26	-0,13	0,26

## 10.4.3 T1 in compression & T2 in tension

In the flexible configuration the external Rafter T1 is in compression, T1 = -(Mc/d) and T2 = Tc + Mc/d.

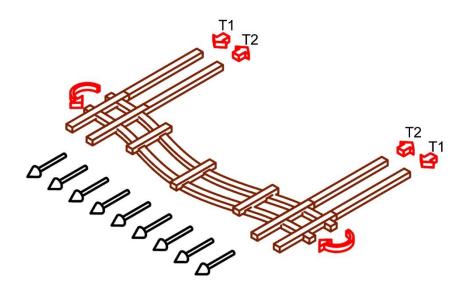


Figure 10-32 Bending behavior - Distribution of the forces on the rafters

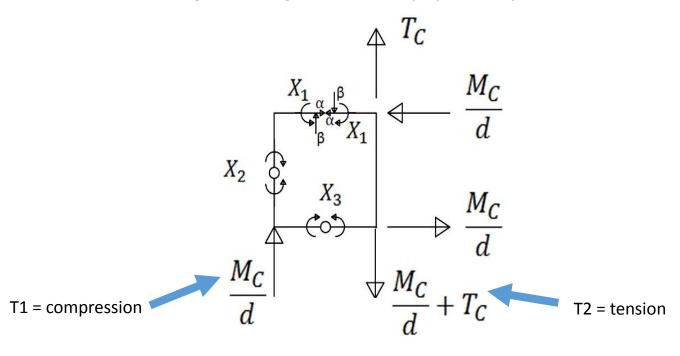


Figure 10-33 Bending behavior - Distribution of the forces on the rafters- corner joint

$$T_C = \frac{139}{100} * q_{\alpha} [N]$$

$$M_C = \frac{19321}{30000} * q_{\alpha} [Nm]$$

$$X_1 = 6.87 * 10^{-3} * q_{\alpha} [Nm]$$

$$X_2 = 2.95 * 10^{-2} * q_{\alpha} [Nm]$$

$$X_3 = 4.12 * 10^{-2} * q_{\alpha} [Nm]$$

In the flexible configuration the external Rafter T1 is in compression, T1 = -(Mc/d) and T2 = Tc + Mc/d .

Table 65 Bending behavior - External rafter T1 - compression

Rafter body T1		
Beam	$q(\alpha)j$ with $\alpha=1$ [kN/m]	-Mc/d [kN/]
1	23,71	-42,42
2	5,36	-9,60
3	5,51	-9,87
4	3,99	-7,14
5	2,47	-4,42
6	0,81	-1,44

Table 66 Bending behavior - Internal rafter T2 - tension

Rafter body T2		
Beam	$q(\alpha)j$ with $\alpha=1$ [kN/m]	Mc/d+Tc [kN]
1	23,71	75,38
2	5,36	17,05
3	5,51	17,53
4	3,99	12,69
5	2,47	7,86
6	0,81	2,57

## 10.5 Verifications for Flexible response – Bendingbehavior

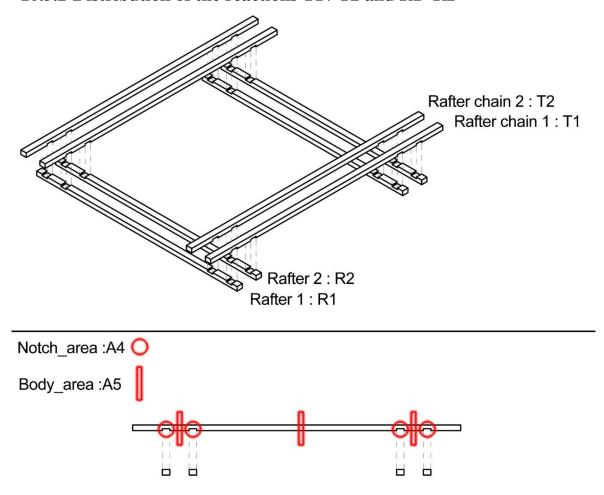
# 10.5.1 Analyzing the worst case : Roof level with maximum Seismic load $q\alpha$ ( $\alpha$ =1)

In the sub-chapter "10.3.3 Seismic load  $q_{\alpha}$  and Distribution factor  $\beta_{j}$ " they have been computed the maximum distribution factor and seismic load which belong to the beam 1, the one at the roof level.

All the beam bands are geometrically equal, thus the satisfied verifications on the most stressed beam ensure that the verification on the other beam bands subjected to smaller actions are satisfied as well.

The analysis have been made considering a seismic direction parallel to the roof rafter as well in the perpendicular direction in respect to the roof rafter.

10.5.2 Distribution of the reactions T1≠T2 and R1=R2



Thebehavior of the rafters chain 1 and 2 is different, thus the verifications on T1 and T2 have been performed separately. Thebehavior of rafters belonging to the bending wall is the same thus the verifications on R1 is equal to R2. The verifications have been performed on the biggest section of the rafters, the body, which refers to a section of area equal to A5. The same verifications have been performed considering the smallest section, the notch, of area equal to A4.

It has been reported the table Tab 51 Geometric dimensions for Notch and Body Areas:

	b	h	AREA net		
	mm	mm	mm^2	cm^2	m^2
A4	100	50	5000	50	0,005

	b	h	AREA net		
	mm	mm	mm^2	cm^2	m^2
A5	100	75	7500	75	0,0075

All the verifications have been done considering the highest load multiplier, thus  $\alpha=1$ . In the cases where the verification is not satisfied the load multiplier has been reduced until the verification was verified.

## 10.5.3 Verifications T1 - compression

10.5.3.1 Body



RB0comp						
N_0d	42421,64	N				
b	100,00	mm				
h	75,00	mm				
A_(net)	7500,00	mm				
σ_(c,0,d)	5,66	N/mm^2				
f_(c,0,d)	24,93	N/mm^2				
Verification	VERIFIED					
N_(0d)max	187,00	kN				

RB0mY		
M_(y,d)	1612022,38	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(y,d)	93750,00	mm^3
σ_(m,y,d)	17,19	N/mm^2
kh	1,15	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	58,97	N/mm^2
Verification	VERIFIED	
M_(y,d)max	5528110,83	Nmm
M_(y,d)max	5,53	kNm

RB0mZ		
M_(z,d)	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(z,d)	125000,00	mm^3
σ_(m,z,d)	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIFIED	
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

### 10.5.3.2 Body Combinations

Combined bending and axial compression

$$\left(\frac{\sigma_{c,0,d}}{f_{c,0,d}}\right)^{2} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_{m} * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\left(\frac{\sigma_{c,0,d}}{f_{c,0,d}}\right)^{2} + k_{m} * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

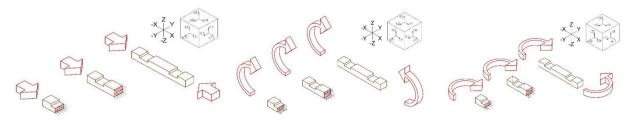
For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0,7 otherwise km = 1

Combination of RB0comp and RB0mY are satisfied for a load multiplier  $\alpha = 1$ 

Combination of RB0comp and RB0mZ are satisfied for a load multiplier  $\alpha = 1$ 

### 10.5.3.3 Notch



CPNotch0comp		
N_0d	42421,64	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
$\sigma_{(c,0,d)}$	8,48	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	VERIFIED	
N_(0d)max	124,67	kN

CPNotch0mY		
M_(y,d)	1612022,38	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(y,d)	41666,67	mm^3
σ_(m,y,d)	38,69	N/mm^2
kh	1,25	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	63,95	N/mm^2
Verification	VERIF	TED
M_(y,d)max	2664480,07	Nmm
M_(y,d)max	2,66	kNm

CPNotch0mZ		
M_(z,d)	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(z,d)	83333,33	mm^3
σ_(m,z,d)	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	TED
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

### 10.5.3.4 Notch Combinations

Combined bending and axial compression

$$\left( \frac{\sigma_{c,0,d}}{f_{c,0,d}} \right)^2 + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\left( \frac{\sigma_{c,0,d}}{f_{c,0,d}} \right)^2 + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

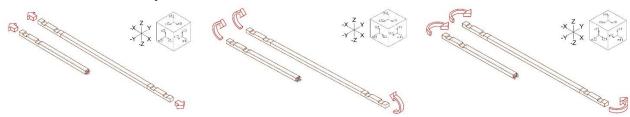
For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0,7
   otherwise km = 1
- Combination of CPNotch0comp and CPNotch0mY are satisfied for a load multiplier  $\alpha=1$

Combination of CPNotch0comp and CPNotch0mZ are satisfied for a load multiplier  $\alpha = 1$ 

## 10.5.4 Verifications T2 - tension

### 10.5.4.1 Body



RB0tens		
N_0d	75382,34	N
b	100,00	mm
h	75,00	mm
A_(net)	7500,00	mm
σ_(t,0,d)	10,05	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERIFIED	
N_(0d)max	231,00	kN

Influence of keyed scarf joint		
Verification	NOT VE	RIFIED
N (0d)max	25.41	kN

RB0mY		
M_(y,d)	2864528,98	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(y,d)	93750,00	mm^3
σ_(m,y,d)	30,55	N/mm^2
kh	1,15	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	58,97	N/mm^2
Verification	VERIFIED	
M_(y,d)max	5528110,83	Nmm
M_(y,d)max	5,53	kNm

RB0mZ		
M_(z,d)	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(z,d)	125000,00	mm^3
σ_(m,z,d)	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	TED
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

The verification about "Influence of keyed scarf joint" is satisfied for a load multiplier  $\alpha = 0.3$ .

### 10.5.4.2 Body Combinations

Combined bending and axial tension

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \leq 1$$

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

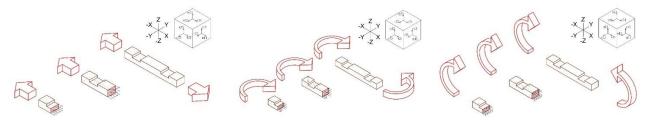
For solid timber, glued laminated timber and LVL:

- for rectangular sections: km = 0,7
- otherwise km = 1

Combination of RB0tens and RB0mY are satisfied for a load multiplier  $\alpha=1$ 

Combination of RB0tens and RB0mZ are satisfied for a load multiplier  $\alpha = 1$ 

### 10.5.4.3 Notch



CPNotch0tens		
N_0d	75382,34	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
σ_(t,0,d)	15,08	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERI	FIED
N_(0d)max	154,00	kN

CPNotch0mY		
M_(y,d)	2864528,98	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(y,d)	41666,67	mm^3
σ_(m,y,d)	68,75	N/mm^2
kh	1,25	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	63,95	N/mm^2
Verification	NOT VEF	RIFIED
M_(y,d)max	2664480,07	Nmm
M_(y,d)max	2,66	kNm

CPNotch0mZ		
M_(z,d)	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(z,d)	83333,33	mm^3
σ_(m,z,d)	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	IED
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

CPNotch0mY is satisfied for a load multiplier  $\alpha = 0.9$ 

### 10.5.4.4 Notch Combinations

Combined bending and axial tension

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \leq 1$$

$$\frac{\sigma_{t,0,d}}{f_{t,0,d}} + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \leq 1$$

For solid timber, glued laminated timber and LVL:

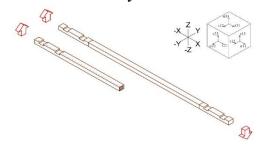
- for rectangular sections: km = 0,7
   otherwise km = 1

Combination of CPNotch0tens and CPNotch0mY are satisfied for a load multiplier  $\alpha=0.9$ 

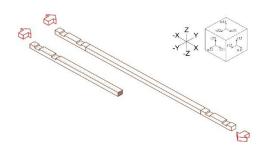
Combination of CPNotch0tens and CPNotch0mZ are satisfied for a load multiplier  $\alpha = 0.9$ 

## 10.5.5 Verifications R1=R2

10.5.5.1 Body ends



RB0shearZ with bending		
V_zd	0,00	N
K_cr	0,67	
A_(net)	5025,00	mm
τ_(d)	0,00	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VERIFIED	
V_zd max	12,28	kN

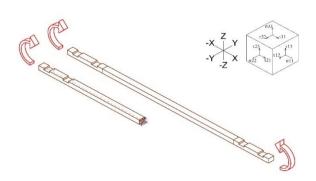


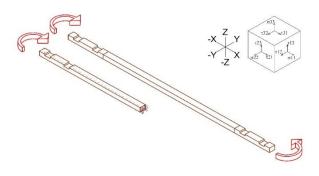
RB0shearY with bending		
V_yd	16480,35	N
K_cr	0,67	
A_(net)	5025,00	mm
τ_(d)	4,92	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	NOT VE	RIFIED
V_yd max	12,28	kN

RB0shearZ		
V_zd	0,00	N
A_(net)	7500,00	mm
τ_(d)	0,00 N/mm^2	
f_(v,d)	3,67	N/mm^2
Verification	VERIFIED	
V_zd max	18,33 kN	

RB0shearY		
V_yd	16480,35	N
A_(net)	7500,00	mm
τ_(d)	3,30	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VERIFIED	
V_yd max	18,33	kN

# RB0shearY with bending is satisfied for a load multiplier $\alpha = 0.7\,$



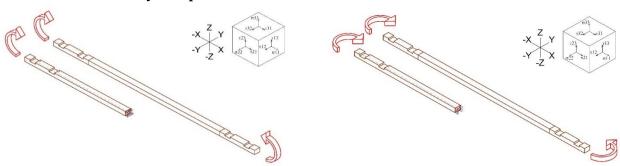


RB0mY		
$M_{(y,d)}$	7635895,48	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(y,d)	93750,00	mm^3
$\sigma_{m,y,d}$	81,45	N/mm^2
kh	1,15	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	58,97	N/mm^2
Verification	NOT VER	IFIED
M_(y,d)max	5528110,83	Nmm
M (v.d)max	5.53	kNm

RB0mZ		
M_(z,d)	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(z,d)	125000,00	mm^3
$\sigma_{}(m,z,d)$	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIFI	ED
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

RB0mY with bending is satisfied for a load multiplier  $\alpha = 0.7$ 

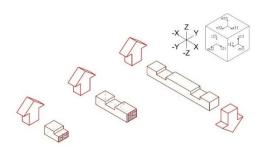
10.5.5.2 Body midspan



RB0mY		
$M_{(y,d)}$	3817947,74	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
$W_{-}(y,d)$	93750,00	mm^3
$\sigma_{m,y,d}$	40,72	N/mm^2
kh	1,15	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	58,97	N/mm^2
Verification	VERIF	IED
M_(y,d)max	5528110,83	Nmm
M_(y,d)max	5,53	kNm

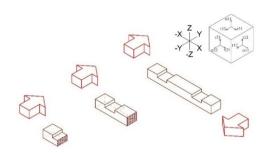
RB0mZ		
$M_{}(z,d)$	0,00	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
$W_{}(z,d)$	125000,00	mm^3
$\sigma_{m,z,d}$	0,00	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	IED
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

### 10.5.5.3 Notch ends



CPNotch0shearZ with bending		
V_zd	0,00	N
K_cr	0,67	
A_(net)	3350,00	mm^2
τ_(d)	0,00	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	VER	IFIED
V_zd max	8,19	kN

CPNotch0shearZ			
V_zd	V_zd 0,00 N		
A_(net)	5000,00	mm	
τ_(d)	0,00	N/mm^2	
f_(v,d)	3,67	N/mm^2	
Verification	VERIFIED		
V_zd max	12,22	kN	

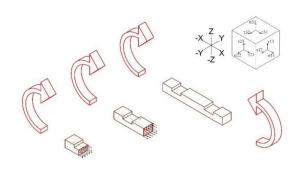


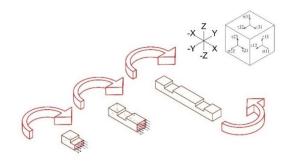
CPNotch0shearY with bending		
V_yd	16480,35	N
K_cr	0,67	
A_(net)	3350,00	mm^2
τ_(d)	7,38	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	NOT VE	RIFIED
V_yd max	8,19	kN

CPNotch0shearY		
V_yd	16480,35	N
A_(net)	5000,00	mm
τ_(d)	4,94	N/mm^2
f_(v,d)	3,67	N/mm^2
Verification	NOT VERIFIED	
V_yd max	12,22	kN

CPNotch0shearY with bending is satisfied for a load multiplier  $\alpha=0.4$ 

CPNotch0shearY is satisfied for a load multiplier  $\alpha = 0.7$ 





CPNotch0mY				
$M_{(y,d)}$	7635895,48	Nmm		
K_m	0,70			
b	100,00	mm		
h	50,00	mm		
$W_{(y,d)}$	41666,67	mm^3		
$\sigma_{}(m,y,d)$	183,26	N/mm^2		
kh	1,25			
f_(m,d)	51,33	N/mm^2		
f_(m,y,d)	63,95	N/mm^2		
Verification	NOT VER	RIFIED		
M_(y,d)max	2664480,07	Nmm		
M_(y,d)max	2,66	kNm		

CPNotch0mZ				
$M_{}(z,d)$	0,00	Nmm		
K_m	0,70			
b	100,00	mm		
h	50,00	mm		
$W_{}(z,d)$	83333,33	mm^3		
$\sigma_{m,z,d}$	0,00	N/mm^2		
kh	1,08			
f_(m,d)	51,33	N/mm^2		
f_(m,z,d)	55,67	N/mm^2		
Verification	VERIF	IED		
M_(z,d)max	4639129,24	Nmm		
M_(z,d)max	4,64	kNm		

CPNotch0mY is satisfied for a load multiplier  $\alpha = 0.3\,$ 

## 10.5.6 Verifications on corner joint, seismic event parallel to Roof Rafter

The verifications on the corner joint have been computed appling the actions resulted by the analysis of the scheme of the rigid frame solved in the sub-chapter 10.2.2.

All the verified sections have been defined in the previous chapters.

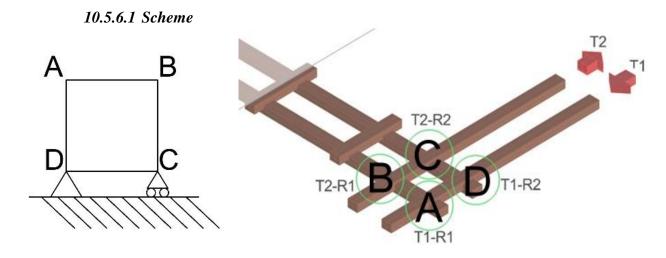


Figure 10-34 Rigid-Jointed Frame - names of the corners- scheme

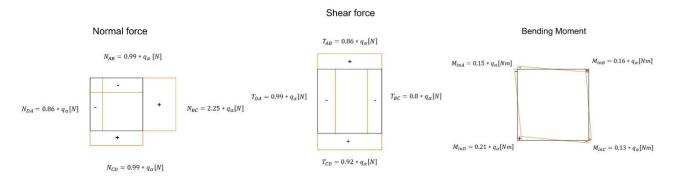
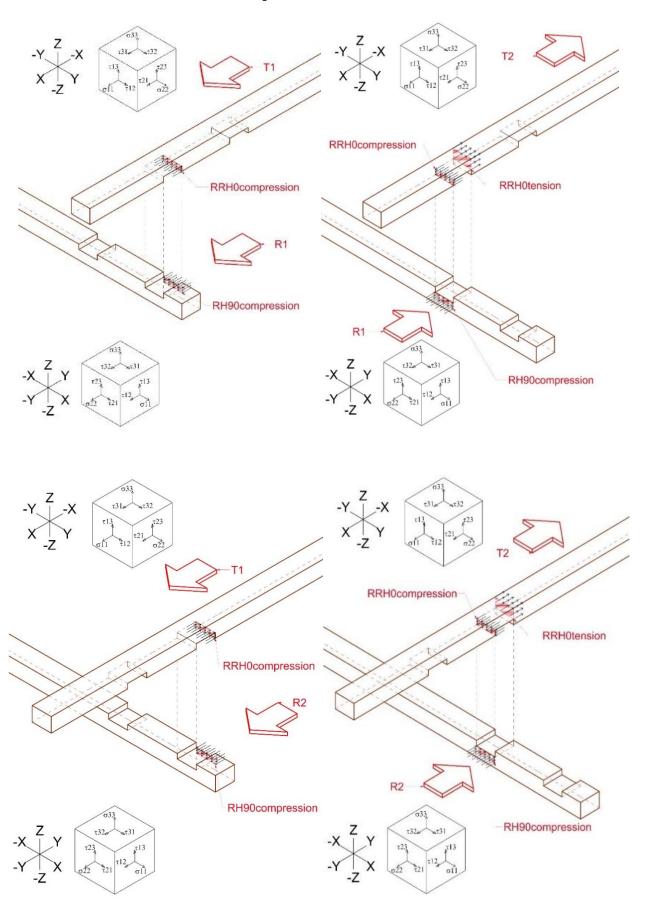


Table 67 Flexible behavior - Rafter body reactions for each beam - Verifications

		Nab/q(α)j	Nbc /q(α)j	Ned /q(α)j	Nda /q(α)j	Tab /q(α)j	Tbc /q(α)j	Tcd /q(α)j	Tda /q(α)j	BM in A /q(a)j	BM in B /q(a)j	BM in C /q(a)j	BM in D /q(α)j
α	1,00	0,99	2,25	0,99	0,86	0,86	0,80	0,92	0,99	0,15	0,16	0,13	0,21
		Normal force Shear force Bending Mor			Shear force			Moment					
Beam	$q(\alpha)j$ with $\alpha=1$ [kN/m]		Nbc [kN]	Ncd [kN]	Nda [kN]	Tab [kN]	Tbc [kN]	Tcd [kN]	Tda [kN]	BM in A [kNm]	BM in B [kNm]	BM in C [kNm]	BM in D [kNm]
1	23,71	23,47	53,46	23,47	20,50	20,50	18,95	21,92	23,47	3,53	3,85	2,97	4,92
2	5,36	5,31	12,09	5,31	4,64	4,64	4,29	4,96	5,31	0,80	0,87	0,67	1,11
3	5,51	5,46	12,43	5,46	4,77	4,77	4,41	5,10	5,46	0,82	0,90	0,69	1,15
4	3,99	3,95	9,00	3,95	3,45	3,45	3,19	3,69	3,95	0,59	0,65	0,50	0,83
5	2,47	2,45	5,57	2,45	2,14	2,14	1,98	2,29	2,45	0,37	0,40	0,31	0,51
6	0,81	0,80	1,82	0,80	0,70	0,70	0,64	0,75	0,80	0,12	0,13	0,10	0,17

The table 56 and table 61 differ only for the sign, this because the verifications are specifically for cases of tension or compression.

10.5.6.2 Axial stresses: Compression and Tension



Nab [kN]	
	23,47

RRH0compression	A3	
N_0d	23475,00	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
$\sigma_{(c,0,d)}$	9,39	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	VERIF	FIED
N_(od)max	62,33	kN

RRH0tension	A4	
N_0d	23475,00	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
$\sigma_{\mathbf{t}}(t,0,d)$	4,69	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERIF	FIED
N_(od)max	154,00	kN

Ncd [kN]	
	23,47

RRH0compression	A3	
N_0d	23475,00	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
$\sigma_{\mathbf{c}}(\mathbf{c},0,\mathbf{d})$	9,39	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	VERI	FIED
N_(od)max	62,33	kN

RRH0tension	A4	
N_0d	23475,00	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
$\sigma_{t}(t,0,d)$	4,69	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERI	FIED
N_(od)max	154,00	kN

Nbc [kN]	
	53,46

RRH0compression	A3	
N_0d	53458,21	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
$\sigma_{(c,0,d)}$	21,38	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	VERII	FIED
N_(od)max	62,33	kN

RRH0tension	A4	
N_0d	53458,21	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
$\sigma_{\mathbf{t}}(t,0,d)$	10,69	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERII	FIED
N_(od)max	154,00	kN

Nda [kN]	
	20,50

RRH0compression	A3	
N_0d	20497,51	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
$\sigma_{(c,0,d)}$	8,20	N/mm^2
f_(c,0,d)	24,93	N/mm^2
Verification	VERI	FIED
N_(od)max	62,33	kN

RRH0tension	A4	
N_0d	20497,51	N
b	100,00	mm
h	50,00	mm
A_(net)	5000,00	mm^2
$\sigma_{-}(t,0,d)$	4,10	N/mm^2
kh	1,08	
f_(t,0,d)	30,80	N/mm^2
Verification	VERI	FIED
N_(od)max	154,00	kN

Tab [kN]	
	20,50

RH90compression	A3	
N_90d	20497,51	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	8,20	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	VERII	FIED
N_(90d)max	16,50	kN

Tcd [kN]	
	21,92

RH90compression	A3	
N_90d	21924,13	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	8,77	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	VERI	FIED
N (90d)max	16,50	kN

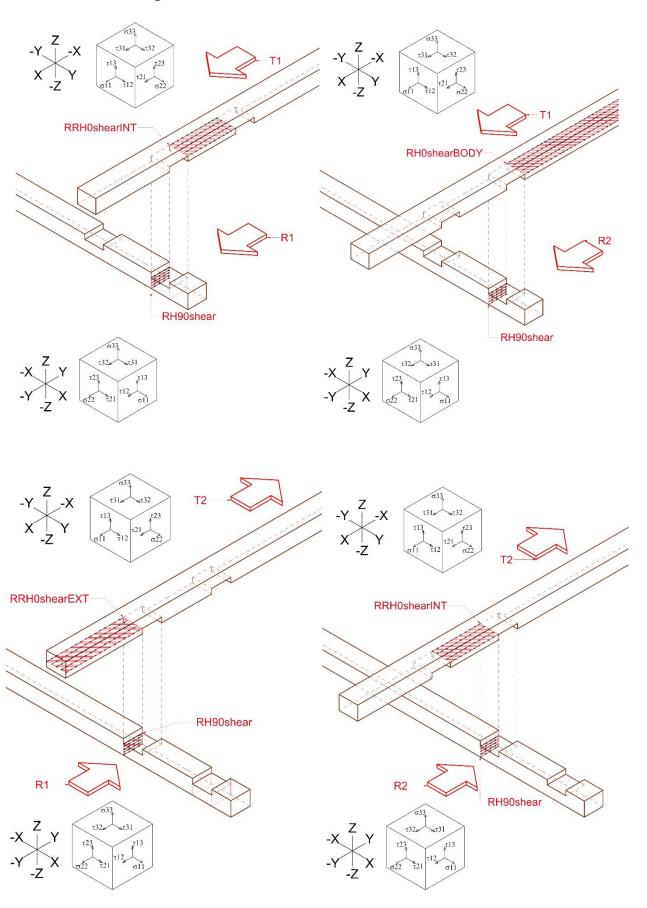
Tbc [kN]	
	18,95

RH90compression	A3	
N_90d	18946,64	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
σ_(c,90,d)	7,58	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	VERI	FIED
N_(90d)max	16,50	kN

Tda [kN]	
	23,47

RH90compression	A3	
N_90d	23475,00	N
b	100,00	mm
h	25,00	mm
A_(net)	2500,00	mm^2
$\sigma_{(c,90,d)}$	9,39	N/mm^2
k_(c,90)	1,50	
f_(c,90,d)	9,90	N/mm^2
Verification	VERI	FIED
N_(90d)max	16,50	kN

10.5.6.3 Tangential stresses : Shear



Nab [kN]			Nbc [kN]			Ncd [kN]			Nda [kN]		
23,47			53,46			23,47			20,50		
23,17			33,10	l		23,17	l		20,30	l	
RRH0shearEXT	A7		RRH0shearEXT	A7		RRH0shearEXT	A7		RRH0shearEXT	A7	
with l	ending	ı	with	bending		with	bending		with	bending	
V_0d	23475,00	N	V_0d	53458,21	N	V_0d	23475,00	N	V_0d	20497,51	N
K_cr	0,67		K_cr	0,67		K_cr	0,67		K_cr	0,67	
A_(net)	26800,00	mm^2	A_(net)	26800,00	mm^2	A_(net)	26800,00	mm^2	A_(net)	26800,00	mm^2
τ_(d)	1,31	N/mm^2	τ_(d)	2,99	N/mm^2	τ_(d)	1,31	N/mm^2	τ_(d)	1,15	N/mm^2
f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2
Verification	VERII	FIED	Verification	VERII	FIED	Verification	VERII	FIED	Verification	VERI	FIED
V_0d max	65,51	kN	V_0d max	65,51	kN	V_0d max	65,51	kN	V_0d max	65,51	kN
										•	
RRH0shearINT	A1		RRH0shearINT	A1		RRH0shearINT	A1		RRH0shearINT	A1	
with l	ending		with	bending		with	bending		with	bending	
V_0d	23475,00	N	V_0d	53458,21	N	V_0d	23475,00	N	V_0d	20497,51	N
K_cr	0,67		K_cr	0,67		K_cr	0,67		K_cr	0,67	
A_(net)	17420,00	mm^2	A_(net)	17420,00	mm^2	A_(net)	17420,00	mm^2	A_(net)	17420,00	mm^2
τ_(d)	2,02	N/mm^2	τ_(d)	4,60	N/mm^2	τ_(d)	2,02	N/mm^2	τ_(d)	1,76	N/mm^2
f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2
Verification	VERII	FIED	Verification	NOT VE	RIFIED	Verification	VERI	FIED	Verification	VERI	FIED
V_0d max	42,58	kN	V_0d max	42,58	kN	V_0d max	42,58	kN	V_0d max	42,58	kN
RH0shearBODY	A6		RH0shearBODY	A6		RH0shearBODY	A6		RH0shearBODY	A6	
with l	ending		with	bending		with bending		with bending			
V_0d	23475,00	N	V_0d	53458,21	N	V_0d	23475,00	N	V_0d	20497,51	N
K_cr	0,67		K_cr	0,67		K_cr	0,67		K_cr	0,67	
A_(net)	179560,00	mm^2	A_(net)	179560,00	mm^2	A_(net)	179560,00	mm^2	A_(net)	179560,00	mm^2
τ_(d)	0,20	N/mm^2	τ_(d)	0,45	N/mm^2	τ_(d)	0,20	N/mm^2	τ_(d)	0,17	N/mm^2
f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2
Verification	VERII	FIED	Verification	VERII	FIED	Verification	VERII	FIED	Verification	VERI	FIED
V_0d max	438,92	kN	V_0d max	438,92	kN	V_0d max	438,92	kN	V_0d max	438,92	kN
RRH0shearINT	A1		RRH0shearINT	A1		RRH0shearINT	A1		RRH0shearINT	A1	
V_0d	23475,00	N	V_0d	53458,21	N	V_0d	23475,00	N	V_0d	20497,51	N
A_(net)	26000,00	mm^2	A_(net)	26000,00	mm^2	A_(net)	26000,00	mm^2	A_(net)	26000,00	mm^2
τ_(d)	1,35	N/mm^2	τ_(d)	3,08	N/mm^2	τ_(d)	1,35	N/mm^2	τ_(d)	1,18	N/mm^2
f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2
1_(v,u)											
Verification	VERII	FIED	Verification	VERII	FIED	Verification	VERII	FIED	Verification	VERI	FIED

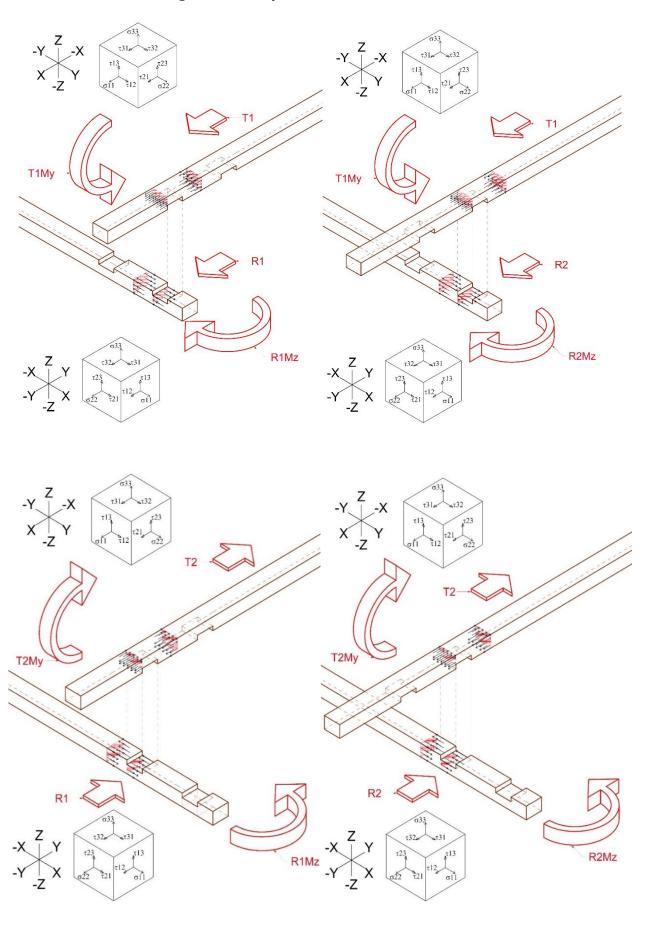
RRH0shearINT with bending is satisfied for a load multiplier  $\alpha = 0.8\,$ 

Tab [kN] 20,50			Tbc [kN]			Tcd [kN] 21,92			Tda [kN] 23,47		
RH90shear	A5		RH90shear	A5		RH90shear	A5		RH90shear	A5	
wit	h bending	7	with	h bending	7	wit	h bending	5	with	h bending	
V_90d	20497,51	N	V_90d	18946,64	N	V_90d	21924,13	N	V_90d	23475,00	N
K_cr	0,67		K_cr	0,67		K_cr	0,67		K_cr	0,67	
A_(net)	5025,00	mm^2	A_(net)	5025,00	mm^2	A_(net)	5025,00	mm^2	A_(net)	5025,00	mm^2
τ_(d)	6,12	N/mm^2	τ_(d)	5,66	N/mm^2	τ_(d)	6,54	N/mm^2	τ_(d)	7,01	N/mm^2
ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2
Verification	NOT VE	ERIFIED	Verification	NOT VE	ERIFIED	Verification	NOT VE	RIFIED	Verification	NOT VE	RIFIED
V_90d max	2,95	kN	V_90d max	2,95	kN	V_90d max	2,95	kN	V_90d max	2,95	kN
RH90shear	A5		RH90shear	A5		RH90shear	A5		RH90shear	A5	
V_90d	20497,51	N	V_90d	18946,64	N	V_90d	21924,13	N	V_90d	23475,00	N
A_(net)	7500,00	mm^2	A_(net)	7500,00	mm^2	A_(net)	7500,00	mm^2	A_(net)	7500,00	mm^2
τ_(d)	4,10	N/mm^2	τ_(d)	3,79	N/mm^2	τ_(d)	4,38	N/mm^2	τ_(d)	4,69	N/mm^2
ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2
Verification	NOT VE	ERIFIED	Verification	NOT VE	ERIFIED	Verification	NOT VE	RIFIED	Verification	NOT VE	RIFIED
V_90d max	4,40	kN	V_90d max	4,40	kN	V_90d max	4,40	kN	V_90d max	4,40	kN

RH90shear for Tab, Tbc and Tcd are satisfied for a load multiplier  $\alpha=0.2$ 

RH90shear for Tda and RH90shear with bending for Tbc are satisfied for a load multiplier  $\alpha=0.15$  RH90shear with bending for Tab, Tcd and Tda are satisfied for a load multiplier  $\alpha=0.125$ 

10.5.6.4 Bending moments My and Mz



BM in A [kNm]

BM in B [kNm] 3,85 BM in C [kNm] 2,97 BM in D [kNm] 4,92

Body0mZ		
M_(z,d)	3526531,56	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(z,d)	125000,00	mm^3
σ_(m,z,d)	28,21	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	TED
M_(z,d)max	6958693,87	Nmm
M_(z,d)max	6,96	kNm

Body0mZ					
$M_{}(z,d)$	3852573,21	Nmm			
K_m	0,70				
b	100,00	mm			
h	75,00	mm			
W_(z,d)	125000,00	mm^3			
$\sigma_{\underline{}}(m,z,d)$	30,82	N/mm^2			
kh	1,08				
f_(m,d)	51,33	N/mm^2			
f_(m,z,d)	55,67	N/mm^2			
Verification	VERIFIED				
M_(z,d)max	6958693,87	Nmm			
M_(z,d)max	6,96	kNm			

Body0mZ					
M_(z,d)	2968218,14	Nmm			
K_m	0,70				
b	100,00	mm			
h	75,00	mm			
W_(z,d)	125000,00	mm^3			
σ_(m,z,d)	23,75	N/mm^2			
kh	1,08				
f_(m,d)	51,33	N/mm^2			
f_(m,z,d)	55,67	N/mm^2			
Verification	VERIFIED				
M_(z,d)max	6958693,87	Nmm			
M_(z,d)max	6,96	kNm			

Body0mZ					
M_(z,d)	4924468,05	Nmm			
K_m	0,70				
b	100,00	mm			
h	75,00	mm			
W_(z,d)	125000,00	mm^3			
σ_(m,z,d)	39,40	N/mm^2			
kh	1,08				
f_(m,d)	51,33	N/mm^2			
f_(m,z,d)	55,67	N/mm^2			
Verification	VERIF	TED			
M_(z,d)max	6958693,87	Nmm			
M_(z,d)max	6,96	kNm			

BM in A [kNm]

BM in B [kNm] 3,85

BM in C [kNm] 2,97

BM in D [kNm] 4,92

Notch0mZ		
$M_{}(z,d)$	3526531,56	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(z,d)	83333,33	mm^3
$\sigma_{-}(m,z,d)$	42,32	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	VERIF	TED
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

Notch0mZ					
$M_{}(z,d)$	3852573,21	Nmm			
K_m	0,70				
b	100,00	mm			
h	50,00	mm			
$W_{-}(z,d)$	83333,33	mm^3			
$\sigma_{\underline{}}(m,z,d)$	46,23	N/mm^2			
kh	1,08				
f_(m,d)	51,33	N/mm^2			
f_(m,z,d)	55,67	N/mm^2			
Verification	VERIFIED				
M_(z,d)max	4639129,24	Nmm			
M_(z,d)max	4,64	kNm			

53458,21

2031412,10

38,00

0,70

100,00

21,67

93750,00

mm

mm

mm

mm^3

N/mm^2

Nmm

Notch0mZ				
2968218,14	Nmm			
0,70				
100,00	mm			
50,00	mm			
83333,33	mm^3			
35,62	N/mm^2			
1,08				
51,33	N/mm^2			
55,67	N/mm^2			
ification VERIFIED				
4639129,24	Nmm			
4,64	kNm			
	2968218,14 0,70 100,00 50,00 83333,33 35,62 1,08 51,33 55,67 VERIF 4639129,24			

23475,00

38,00 mm

0,70

100,00

75,00 mm

93750,00 mm^3

2664480,07 Nmm

2,66 kNm

892049,96 Nmm

mm

9,52 N/mm^2

N	otch0mZ	
$M_{}(z,d)$	4924468,05	Nmm
K_m	0,70	
b	100,00	mm
h	50,00	mm
W_(z,d)	83333,33	mm^3
σ_(m,z,d)	59,09	N/mm^2
kh	1,08	
f_(m,d)	51,33	N/mm^2
f_(m,z,d)	55,67	N/mm^2
Verification	NOT VEF	
M_(z,d)max	4639129,24	Nmm
M_(z,d)max	4,64	kNm

Notch0mZ for BM in D satisfied for a load multiplier  $\alpha = 0.9$ 

They have been verified the "parasitic" bending moment as well.

Nab [kN] 23,47

Nbc [kN] 53,46

Body0mY

N\_0d

 $\delta$  for My

M\_(y,d) K\_m

W\_(y,d)

f\_(m,d)

f\_(m,y,d)

Verification

 $\sigma_{\underline{\phantom{}}}(m,y,d)$ 

b

kh

Ncd [kN] 23,47

Body0mY

N\_0d

 $\delta$  for My

M\_(y,d)

W\_(y,d)

 $\sigma_{m,y,d}$ 

 $K_m$ 

b

Nda [kN] 20,50

25,	L	
Body0mY		
N_0d	23475,00	N
δ for My	38,00	mm
M_(y,d)	892049,96	Nmm
K_m	0,70	
b	100,00	mm
h	75,00	mm
W_(y,d)	93750,00	mm^3
$\sigma_{\underline{}}(m,y,d)$	9,52	N/mm^2
kh	1,15	
f_(m,d)	51,33	N/mm^2
f_(m,y,d)	58,97	N/mm^2
Verification	VERIF	TED
M_(y,d)max	5528110,83	Nmm
M_(y,d)max	5,53	kNm
-	· · · · · ·	

M_(y,d)max	5528110,83	Nmm	M_(y,d)max	5528110,83	Nmm
M_(y,d)max	5,53	kNm	M_(y,d)max	5,53	kNm
Notch0mY			Notch0mY		
N_0d	23475,00	N	N_0d	53458,21	N
δ for My	38,00	mm	δ for My	38,00	mm
$M_{(y,d)}$	892049,96	Nmm	M_(y,d)	2031412,10	Nmm
K_m	0,70		K_m	0,70	
b	100,00	mm	b	100,00	mm
h	50,00	mm	h	50,00	mm
W_(y,d)	41666,67	mm^3	W_(y,d)	41666,67	mm^3
σ_(m,y,d)	21,41	N/mm^2	σ_(m,y,d)	48,75	N/mm^2
kh	1,25		kh	1,25	
f_(m,d)	51,33	N/mm^2	f_(m,d)	51,33	N/mm^2
f_(m,y,d)	63,95	N/mm^2	f_(m,y,d)	63,95	N/mm^2
Verification	VERIF	TED	Verification	VERIFIED	
M_(y,d)max	2664480,07	Nmm	M_(y,d)max	2664480,07	Nmm
M_(y,d)max	2,66	kNm	M_(y,d)max	2,66	kNm

1,15		kh	1,15	
51,33	N/mm^2	f_(m,d)	51,33	N/mm^2
58,97	N/mm^2	f_(m,y,d)	58,97	N/mm^2
VERIF	IED	Verification	VERIF	IED
110,83	Nmm	M_(y,d)max	5528110,83	Nmm
5,53	kNm	M_(y,d)max	5,53	kNm
		Notch0mY		
458,21	N	N_0d	23475,00	N
38,00	mm	δ for My	38,00	mm
412,10	Nmm	M_(y,d)	892049,96	Nmm
0,70		K_m	0,70	
100,00	mm	b	100,00	mm
50,00	mm	h	50,00	mm
666,67	mm^3	W_(y,d)	41666,67	mm^3
48,75	N/mm^2	σ_(m,y,d)	21,41	N/mm^2
1,25		kh	1,25	
51,33	N/mm^2	f_(m,d)	51,33	N/mm^2
63,95	N/mm^2	f_(m,y,d)	63,95	N/mm^2

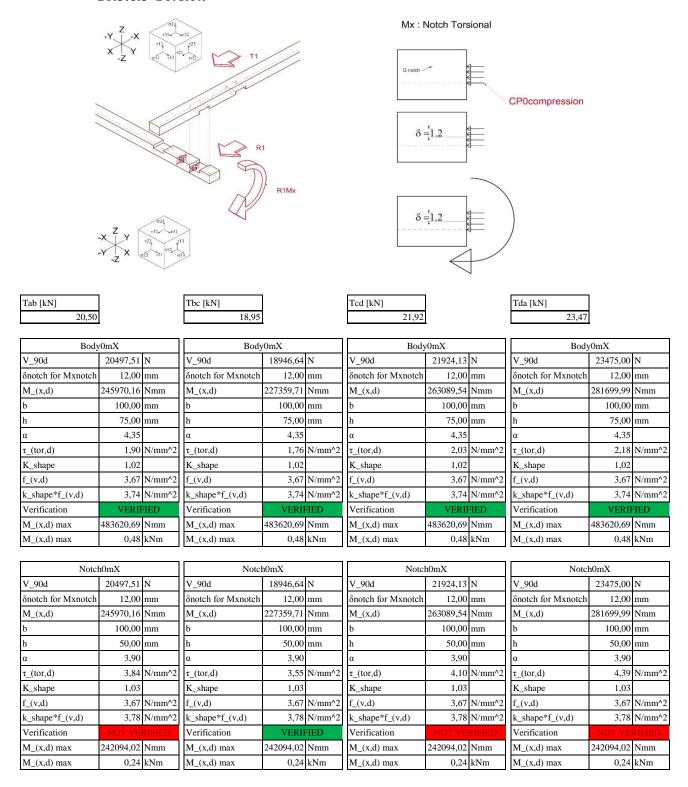
Verification M\_(y,d)max

M\_(y,d)max

20497,51	N		
38,00	mm		
778905,50	Nmm		
0,70			
100,00	mm		
75,00	mm		
93750,00	mm^3		
8,31	N/mm^2		
1,15			
51,33	N/mm^2		
58,97	N/mm^2		
VERIF	IED		
5528110,83	Nmm		
5,53	kNm		
	38,00 778905,50 0,70 100,00 75,00 93750,00 8,31 1,15 51,33 58,97 VERIF 5528110,83		

Notch0mY				
N_0d	20497,51	N		
δ for My	38,00	mm		
$M_{-}(y,d)$	778905,50	Nmm		
K_m	0,70			
b	100,00	mm		
h	50,00	mm		
$W_{-}(y,d)$	41666,67	mm^3		
$\sigma_{-}(m,y,d)$	18,69	N/mm^2		
kh	1,25			
f_(m,d)	51,33	N/mm^2		
f_(m,y,d)	63,95	N/mm^2		
Verification	VERIF	VERIFIED		
M_(y,d)max	2664480,07	Nmm		
M_(y,d)max	2,66	kNm		

10.5.6.5 Torsion



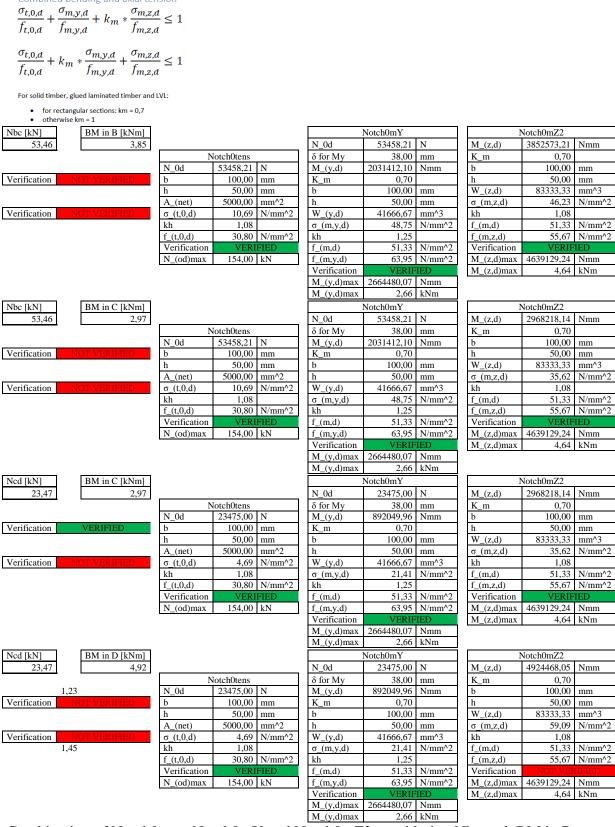
Notch0mX for Tab, Tcd are satisfied for a load multiplier  $\alpha = 0.9$ 

Notch0mX for Tda is satisfied for a load multiplier  $\alpha = 0.8$ 

### 10.5.6.6 Combinations

Combined bending and axial tension

The combinations for bending and tension have been computed for the weakest section, thus the notch.



Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Nbc and BM in B

are satisfied for a load multiplier  $\alpha = 0.5$ 

Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Nbc and  $\,BM$  in C are satisfied for a load multiplier  $\alpha=0.6$ 

Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Ncd and BM in C are satisfied for a load multiplier  $\alpha=0.9$ 

Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Ncd and  $\,BM$  in D are satisfied for a load multiplier  $\alpha=0.6$ 

The combinations for bending and compression have been computed for both sections, thus for the notch and the body section.

### Combined bending and axial compression

$$\left( \frac{\sigma_{c,0,d}}{f_{c,0,d}} \right)^2 + \frac{\sigma_{m,y,d}}{f_{m,y,d}} + k_m * \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$$\left( \frac{\sigma_{c,0,d}}{f_{c,0,d}} \right)^2 + k_m * \frac{\sigma_{m,y,d}}{f_{m,y,d}} + \frac{\sigma_{m,z,d}}{f_{m,z,d}} \le 1$$

$\left(\frac{c,o,a}{f_{c,0,d}}\right) + k_m * \frac{m,y}{f_{m,y,}}$	$\frac{\alpha}{d} + \frac{m_{,z,\alpha}}{f_{m,z,d}}$	≤ 1						
For solid timber, glued laminated tim								
· -								
<ul> <li>for rectangular sections: km</li> <li>otherwise km = 1</li> </ul>	= 0,7							
Nab [kN] BM in A [kNm]				Notch0mY			Notch0mZ2	
23,47 3,53			N 0d	2,35E+04	N	M_(z,d)		Nmm
	N	otch0comp	δ for My	38,00	mm	K_m	0,70	
	N_0d	23475,00 N	$M_{(y,d)}$	892049,96	Nmm	b	100,00	mm
Verification VERIFIED	b	100,00 mm	K_m	0,70		h	50,00	mm
	h	50,00 mm	b	100,00	mm	W_(z,d)	83333,33	mm^3
Verification NOT VERIFIED	$A_{\text{(net)}}$ $\sigma$ (c,0,d)	5000,00 mm^2 4,69 N/mm^2	h W_(y,d)	50,00 41666,67	mm mm^3	σ_(m,z,d) kh	42,32 1,08	N/mm^2
vernication NOT VERTILED	0_(c,0,u)	4,09 14/11111 2	$\sigma$ (m,y,d)	21,41	N/mm^2	f_(m,d)		N/mm^2
	f_(c,0,d)	24,93 N/mm^2	kh	1,25	1011111 2	f_(m,z,d)		N/mm^2
	Verification	VERIFIED	f_(m,d)	51,33	N/mm^2	Verification	VERIF.	IED
	N_(od)max	124,67 kN	f_(m,y,d)	63,95	N/mm^2	M_(z,d)max		Nmm
			Verification	VERIF		M_(z,d)max	4,64	kNm
	non esiste		M_(y,d)max	2664480,07				
Not find			M_(y,d)max	2,66	kNm			
Nab [kN] BM in B [kNm] 23,47 3,85				Body0mY	1		Body0mZ2	
25,47	F	3ody0comp	N 0d	2,35E+04	N	M_(z,d)	3852573.21	Nmm
	N_0d	23475,00 N	δ for My	38,00	mm	K_m	0,70	TVIIIII
Verification VERIFIED	b	100,00 mm	M_(y,d)	892049,96	Nmm	b	100,00	mm
<u> </u>	h	75,00 mm	K_m	0,70		h	75,00	mm
	A_(net)	5000,00 mm^2	b	100,00	mm	$W_{-}(z,d)$		mm^3
Verification VERIFIED	σ_(c,0,d)	4,69 N/mm^2	h	75,00	mm	σ_(m,z,d)	30,82	N/mm^2
	f (a 0 d)	24,93 N/mm^2	$W_{}(y,d)$ $\sigma (m,v,d)$	93750,00 9,52	mm^3 N/mm^2	kh f (m,d)	1,08	N/mm^2
	f_(c,0,d) Verification	VERIFIED	kh	1,15	IN/IIIIII 'Z	f_(m,z,d)		N/mm^2
	N_(od)max	124,67 kN	f_(m,d)		N/mm^2	Verification	VERIF.	
	_(***)	7	f_(m,y,d)	58,97	N/mm^2	M_(z,d)max		Nmm
			Verification	VERIF	IED	M_(z,d)max	6,96	kNm
			M_(y,d)max	5528110,83				
			M_(y,d)max	5,53	kNm			
Nda [kN] BM in C [kNm]				Notch0mY	N		Notch0mZ2	NT.
20,50 2,97		10	N_0d	2,05E+04	N	M_(z,d)	2968218,14	Nmm
	N_0d	otch0comp 20497,51 N	δ for My M_(y,d)	38,00 778905,50	mm Nmm	K_m b	0,70 100,00	mm
Verification VERIFIED	b	100,00 mm	K_m	0,70	INIIIIII	h	50,00	mm
	h	50,00 mm	b	100,00	mm	W_(z,d)	83333,33	mm^3
	A_(net)	5000,00 mm^2	h	50,00	mm	σ_(m,z,d)	35,62	N/mm^2
Verification VERIFIED	σ_(c,0,d)	4,10 N/mm^2	$W_{\underline{}}(y,d)$	41666,67	mm^3	kh	1,08	
	6 ( 0 1)	24.93 N/mm^2	σ_(m,y,d)	18,69	N/mm^2	f_(m,d)		N/mm^2 N/mm^2
				1.05				N/mm^/
	f_(c,0,d)	, · · ·	kh f (m d)	1,25	N/mm^2	f_(m,z,d)	,	
	Verification	VERIFIED	f_(m,d)	51,33	N/mm^2 N/mm^2	Verification	VERIF	IED
		, · · ·		51,33	N/mm^2		,	
	Verification	VERIFIED	f_(m,d) f_(m,y,d)	51,33 63,95	N/mm^2 IED	Verification M_(z,d)max	VERIF. 4639129,24	IED Nmm
	Verification N_(od)max	VERIFIED	f_(m,d) f_(m,y,d) Verification	51,33 63,95 VERIF 2664480,07	N/mm^2 IED	Verification M_(z,d)max	VERIF. 4639129,24	IED Nmm
Nda [kN] BM in D [kNm]	Verification N_(od)max	VERIFIED	f_(m,d) f_(m,y,d) Verification M_(y,d)max	51,33 63,95 VERIF 2664480,07 2,66	N/mm^2 IED Nmm	Verification M_(z,d)max M_(z,d)max	VERIF 4639129,24 4,64 Body0mZ2	IED Nmm kNm
Nda [kN] BM in D [kNm] 4,92	Verification N_(od)max non esiste	VERIFIED 124,67 kN	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max	51,33 63,95 VERIF 2664480,07 2,66 Body0mY	N/mm^2 IED Nmm kNm	Verification M_(z,d)max M_(z,d)max M_(z,d)max	VERIF 4639129,24 4,64 Body0mZ2 4924468,05	Nmm kNm
	Verification N_(od)max non esiste	VERIFIED 124,67 kN  Body0comp	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04	N/mm^2 IED Nmm kNm	Verification M_(z,d)max M_(z,d)max M_(z,d)max	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70	Nmm kNm
20,50 4,92	Verification N_(od)max non esiste  B N_0d	VERIFIED  124,67 kN  60dy0comp  20497,51 N	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max N_0d δ for My	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04 38,00	N/mm^2 IED Nmm kNm N mm	Verification M_(z,d)max M_(z,d)max  M_(z,d)  M_(z,d)  K_m b	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70 100,00	Nmm kNm
	Verification N_(od)max  non esiste  N_0d b	VERIFIED  124,67 kN  30dy0comp  20497,51 N  100,00 mm	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max  N_0d  \[ \delta\ \text{for My} M_(y,d) \]	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04 38,00 778905,50	N/mm^2 IED Nmm kNm	Verification M_(z,d)max M_(z,d)max M_(z,d)max	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70 100,00 75,00	Nmm kNm
20,50 4,92	Verification N_(od)max non esiste  B N_0d	VERIFIED  124,67 kN  60dy0comp  20497,51 N	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max N_0d δ for My	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04 38,00	N/mm^2 IED Nmm kNm N mm	Verification M_(z,d)max M_(z,d)max  M_(z,d)  M_(z,d)  K_m b	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70 100,00 75,00 125000,00	Nmm kNm
20,50 4,92	Verification N_(od)max  non esiste  N_0d b h	VERIFIED  124,67 kN  30dy0comp  20497,51 N  100,00 mm  75,00 mm	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max  N_0d δ for My M_(y,d) K_m	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04 38,00 778905,50 0,70	N/mm^2 IED Nmm kNm  N mm Nmm	Verification M_(z,d)max M_(z,d)max M_(z,d) M_(z,d) K_m b h W_(z,d)	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70 100,00 75,00 125000,00	Nmm kNm  Nmm  mm  mm  mm/3
20,50 4,92  Verification VERIFIED	Verification N_(od)max  non esiste  N_0d b h A_(net)	VERIFIED  124,67 kN  124,67 kN  60dy0comp  20497,51 N  100,00 mm  75,00 mm  500,00 mm^2  4,10 N/mm^2	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max  N_0d δ for My M_(y,d) K_m b	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04 38,00 778905,50 0,70 100,00	N/mm^2 IED Nmm kNm  N mm Nmm nmm	Verification M_(z,d)max M_(z,d)max M_(z,d) M_(z,d) K_m b h W_(z,d) σ_(m,z,d)	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70 100,00 75,00 125000,00 1,08 51,33	Nmm kNm Nmm mm mm/3 N/mm^2
20,50 4,92  Verification VERIFIED	$\begin{tabular}{ c c c c c c c c c c c c c c c c c c c$	VERIFIED  124,67 kN  124,67 kN  100,00 mm  75,00 mm  500,00 mm^2  4,10 N/mm^2  24,93 N/mm^2	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max N_0d δ for My M_(y,d) K_m b h W_(y,d) σ (m,y,d)	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04 38,00 778905,50 0,70 100,00 75,00 93750,00 8,31	N/mm^2 IED Nmm kNm  N mm Nmm nmm mm	Verification   M_(z,d)max   M_(z,d)max   M_(z,d)max   M_(z,d)   K_m   b   h   W_(z,d)   σ_(m,z,d)   kh   f_(m,d)   f_(m,z,d)   f_(m,z,d	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70 100,00 75,00 125000,00 39,40 1,08 51,33 55,67	Nmm kNm mm mm mm^3 N/mm^2 N/mm^2
20,50 4,92  Verification VERIFIED	$\begin{tabular}{ll} Verification \\ N_{-}(od)max \\ & non \ esiste \\ \end{tabular}$ $\begin{tabular}{ll} R_{-}(od) \\ R_{-}(od)$	VERIFIED  124,67 kN  124,67 kN  60dy0comp  20497,51 N  100,00 mm  75,00 mm  500,00 mm^2  4,10 N/mm^2	f_(m,d) f_(m,y,d) Verification M_(y,d)max M_(y,d)max  N_0d δ for My M_(y,d) K_m b h W_(y,d)	51,33 63,95 VERIF 2664480,07 2,66 Body0mY 2,05E+04 38,00 778905,50 0,70 100,00 75,00 93750,00 8,31 1,15	N/mm^2 IED Nmm kNm  N mm Nmm nmm mm mm mm/3	Verification   M_(z,d)max   M_(z,d)max   M_(z,d)max   M_(z,d)   K_m   b   h   W_(z,d)   \( \sigma_c(m,z,d) \) kh   f_(m,d)	VERIF 4639129,24 4,64 Body0mZ2 4924468,05 0,70 100,00 75,00 125000,00 1,08 51,33	Nmm Nmm Nmm mm mm/3 N/mm/2 N/mm/2 N/mm/2

Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Nab and BM in A are satisfied for a load multiplier  $\alpha = 0.9$ 

 $f_{m,y,d}$ 

Verification  $M_{(y,d)}$ max

M\_(y,d)max

N/mm^2.

Nmm

kNm

5528110,83

M\_(z,d)max

6,96 kNm

The combinations for bending and compression have been computed for both sections, thus for the notch and the body section.

Combined Torsion and Shear - CNR-DT 206/2007

$$\frac{\tau_{tor,d}}{k_{shape} * f_{v,d}} + \left(\frac{\tau_d}{f_{v,d}}\right)^2 \le 1$$

Tab [kN] 20,50

Verification

JOT VERIFIED

Notch90shearY		
V_90d	20497,51	N
A_(net)	5000,00	mm^2
τ_(d)	6,15	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification		
V_90d max	2,93	kN

Verification VERIFIED

Body90shearY		
V_90d	20497,51	N
A_(net)	7500,00	mm^2
τ_(d)	4,10	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VER	
V_90d max	4,40	kN

Tbc [kN] 18,95

Verification

NOT VERIFIED

Notch90shearY		
V_90d	18946,64	N
A_(net)	5000,00	mm^2
τ_(d)	5,68	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VER	
V_90d max	2,93	kN

Verification

VERIFIED

Body90shearY		
V_90d	18946,64	N
A_(net)	7500,00	mm^2
τ_(d)	3,79	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VER	
V_90d max	4,40	kN

Notch0mX			
V_90d	2,05E+04	N	
δnotch for Mxnotch	12,00	mm	
$M_{(x,d)}$	245970,16	Nmm	
b	100,00	mm	
h	50,00	mm	
α	3,90		
τ_(tor,d)	3,84	N/mm^2	
K_shape	1,03		
f_(v,d)	3,67	N/mm^2	
k_shape*f_(v,d)	3,78	N/mm^2	
Verification	NOT VE	RIFIED	
M_(x,d) max	242094,02	Nmm	
M_(x,d) max	0,24	kNm	

Body0mX		
V_90d	2,05E+04	N
δnotch for Mxnotch	12,00	mm
$M_{x,d}$	245970,16	Nmm
b	100,00	mm
h	75,00	mm
α	4,35	
τ_(tor,d)	1,90	N/mm^2
K_shape	1,02	
f_(v,d)	3,67	N/mm^2
k_shape*f_(v,d)	3,74	N/mm^2
Verification	VERII	FIED
M_(x,d) max	483620,69	Nmm
M_(x,d) max	0,48	kNm

Notch0mX		
V_90d	1,89E+04	N
δnotch for Mxnotch	12,00	mm
$M_{(x,d)}$	227359,71	Nmm
b	100,00	mm
h	50,00	mm
α	3,90	
τ_(tor,d)	3,55	N/mm^2
K_shape	1,03	
f_(v,d)	3,67	N/mm^2
k_shape*f_(v,d)	3,78	N/mm^2
Verification	VERII	FIED
M_(x,d) max	242094,02	Nmm
M_(x,d) max	0,24	kNm

Body0mX			
V_90d	1,89E+04	N	
δnotch for Mxnotch	12,00	mm	
$M_{}(x,d)$	227359,71	Nmm	
b	100,00	mm	
h	75,00	mm	
α	4,35		
$\tau_{\text{(tor,d)}}$	1,76	N/mm^2	
K_shape	1,02		
f_(v,d)	3,67	N/mm^2	
k_shape*f_(v,d)	3,74	N/mm^2	
Verification	VERII	FIED	
M_(x,d) max	483620,69	Nmm	
M_(x,d) max	0,48	kNm	

Tcd [kN] 21,92

Verification

Notch90shearY		
V_90d	21924,13	N
A_(net)	5000,00	mm^2
τ_(d)	6,58	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification		
V_90d max	2,93	kN

Verification

VERIFIED

Body90shearY		
V_90d	21924,13	N
A_(net)	7500,00	mm^2
τ_(d)	4,38	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VER	
V_90d max	4,40	kN

Tda [kN] 23,47 Verification

Notch90shearY		
V_90d	23475,00	N
A_(net)	5000,00	mm^2
τ_(d)	7,04	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification	NOT VER	
V 90d max	2.93	kN

Verification VERIFIED

Body90shearY		
V_90d	23475,00	N
A_(net)	7500,00	mm^2
τ_(d)	4,69	N/mm^2
ft,90,d	0,44	N/mm^2
f_(v,d)	0,88	N/mm^2
Verification		
V_90d max	4,40	kN

Note	h0mX	
V 90d	2,19E+04	N
δnotch for Mxnotch	12,00	mm
M_(x,d)	263089,54	Nmm
b	100,00	mm
h	50,00	mm
α	3,90	
τ_(tor,d)	4,10	N/mm^2
K_shape	1,03	
f_(v,d)	3,67	N/mm^2
k_shape*f_(v,d)	3,78	N/mm^2
Verification	NOT VE	
M_(x,d) max	242094,02	Nmm
M_(x,d) max	0,24	kNm
Bod	y0mX	
V_90d	2,19E+04	N
δnotch for Mxnotch	12,00	mm
$M_{(x,d)}$	263089,54	Nmm
b	100,00	mm
h	75,00	mm
α	4,35	
τ_(tor,d)	2,03	N/mm^2
K_shape	1,02	
f_(v,d)	3,67	N/mm^2
k_shape*f_(v,d)	3,74	N/mm^2
Verification	VERI	FIED
M_(x,d) max	483620,69	Nmm
M_(x,d) max	0,48	kNm

Notch0mX				
V_90d	2,35E+04	N		
δnotch for Mxnotch	12,00	mm		
$M_{(x,d)}$	281699,99	Nmm		
b	100,00	mm		
h	50,00	mm		
α	3,90			
τ_(tor,d)	4,39	N/mm^2		
K_shape	1,03			
f_(v,d)	3,67	N/mm^2		
k_shape*f_(v,d)	3,78	N/mm^2		
Verification	NOT VE	RIFIED		
M_(x,d) max	242094,02	Nmm		
M_(x,d) max	0,24	kNm		
Bod	y0mX			
V_90d	2,35E+04	N		
δnotch for Mxnotch	12,00	mm		
M_(x,d)	281699,99	Nmm		
b	100,00	mm		
h	75,00	mm		
α	4,35			
τ_(tor,d)	2,18	N/mm^2		
K_shape	1,02			
f_(v,d)	3,67	N/mm^2		
k_shape*f_(v,d)	3,74	N/mm^2		
Verification	VERII	FIED		
M_(x,d) max	483620,69	Nmm		
M_(x,d) max	0,48	kNm		

The verifications about the combination of torsion and shear is strongly affected by the fragile behavior of the timber subjected to a shear force perpendicular to the fibers.

Combination of Notch90shearY and Notch0mX considering Tab and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=0.7$  Combination of Body90shearY and Body0mX considering Tab and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=1$ 

Combination of Notch90shearY and Notch0mX considering Tbc and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=0.7$  Combination of Body90shearY and Body0mX considering Tbc and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=1$ 

Combination of Notch90shearY and Notch0mX considering Tcd and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=0.6$  Combination of Body90shearY and Body0mX considering Tcd and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=1$ 

Combination of Notch90shearY and Notch0mX considering Tda and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=0.6$  Combination of Body90shearY and Body0mX considering Tda and the pertinent torsional parasitic bending moment is satisfied for a load multiplier  $\alpha=1$ 

It is important to underline that the verifications about the combination of torsion and shear result satisfied with the load multiplier shown above but the singular verifications about the shear is not verified. In order to obtain the shear verification they are required the values of the load multipliers listed in the sub-chapter "10.5.6.3 Tangential stresses: Shear"

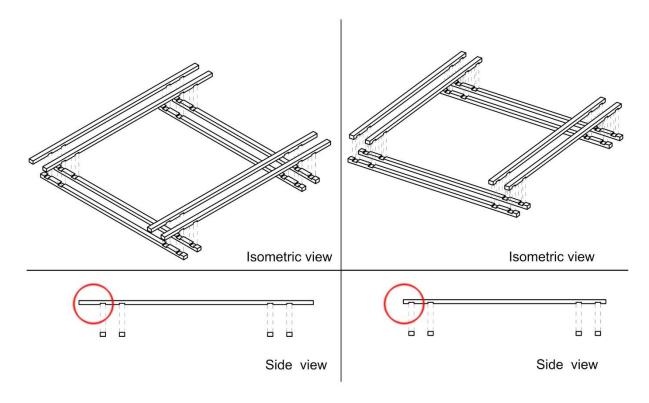
### 10.5.7 Verifications on corner joint, seismic event parallel to Rafter

It has been studied the case of a seismic event parallel to the normal rafter and perpendicular to the roof rafter in the tie-timber beam at the roof level. This has been done because the behavior is similar but normal rafter has few peculiar differences which made it weaker.

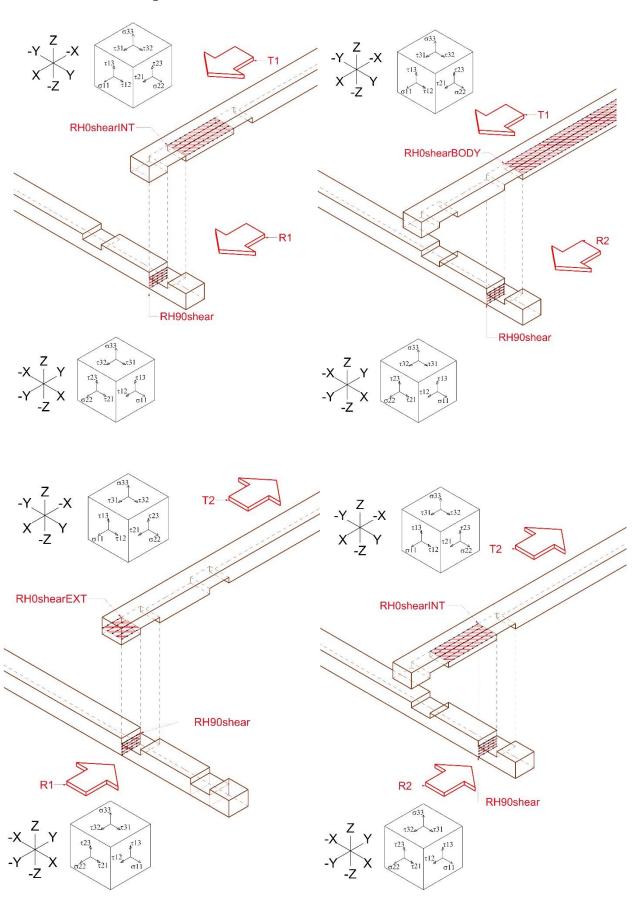
### 10.5.7.1 Scheme

The difference between the roof rafter and the rafter is the length of the head. The length of the normal rafter is shorter and the difference affects the longitudinal shear resistance of the element.

For all the other verifications nothing changes, that is why in the following, they are reported only the verifications about the shear resistance.



10.5.7.2 Tangential stresses : Shear



		[]	1		[ ]	1			4	
23,47		53,46			23,47			20,50	1	
RH0shearEXT	A2	RH0shearEXT	A2		RH0shearEXT	A2	1	RH0shearEXT	A2	
	ending		bending	<u> </u>	with bending			bending		
V_0d	23475 N	V 0d	53458,21	N	V_0d	23475,00	N	V_0d	20497,51	N
K_cr	0,67	K_cr	0,67		K_cr	0,67		K cr	0,67	
A_(net)	6700,0 mm^2	A_(net)	6700,00	mm^2	A_(net)	6700,00	mm^2	A_(net)	6700,00	mm^2
τ_(d)	5,26 N/mm^2	τ_(d)	11,97	N/mm^2	τ_(d)	5,26	N/mm^2	τ_(d)	4,59	N/mm^2
f_(v,d)	3,67 N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2
Verification	NOT VERIFIED	Verification	NOT VE	ERIFIED	Verification	NOT V	ERIFIED	Verification	NOT VI	ERIFIED
V_0d max	16,38 kN	V_0d max	16,38	kN	V_0d max	16,38	kN	V_0d max	16,38	kN
RH0shearBODY	A6	RH0shearBODY	A6		RH0shearBODY	A6		RH0shearBODY	A6	
with b	ending	with	bending		with bending		with bending			
V_0d	23475 N	V_0d	53458,21	N	V_0d	23475,00	N	V_0d	20497,51	N
K_cr	0,67	K_cr	0,67		K_cr	0,67		K_cr	0,67	
	179560		179560,0			179560,0			179560,0	
A_(net)	,00 mm^2	A_(net)		mm^2	A_(net)		mm^2	A_(net)		mm^2
τ_(d)	0,20 N/mm^2	τ_(d)	- , -	N/mm^2	τ_(d)	- , -	N/mm^2	τ_(d)		N/mm^2
f_(v,d)	3,67 N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)		N/mm^2	f_(v,d)		N/mm^2
Verification	VERIFIED	Verification		FIED	Verification		IFIED	Verification		IFIED
V_0d max	438,92 kN	V_0d max	438,92	kN	V_0d max	438,92	kN	V_0d max	438,92	kN
RH0shearINT	A1	RH0shearINT	A1		RRH0shearINT	A1		RRH0shearINT	A1	
V_0d	23475 N	V_0d	53458,21	N	V_0d	23475,00	N	V_0d	20497,51	N
A_(net)	26000 mm^2	A_(net)	26000,00	mm^2	A_(net)	26000,00	mm^2	A_(net)	26000,00	mm^2
τ_(d)	1,35 N/mm^2	τ_(d)		N/mm^2	τ_(d)		N/mm^2	τ_(d)		N/mm^2
f_(v,d)	3,67 N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2	f_(v,d)	3,67	N/mm^2
Verification	VERIFIED	Verification		FIED	Verification		IFIED	Verification		IFIED
V_0d max	63,56 kN	V_0d max	63,56	kN	V_0d max	63,56	kN	V_0d max	63,56	kN

Ncd [kN]

Nda [kN]

RRH0shearEXT with bending for Nab is satisfied for a load multiplier  $\alpha=0.6$  RRH0shearEXT with bending for Nbc is satisfied for a load multiplier  $\alpha=0.3$  RRH0shearEXT with bending for Ncd is satisfied for a load multiplier  $\alpha=0.6$  RRH0shearEXT with bending for Nda is satisfied for a load multiplier  $\alpha=0.7$ 

Nbc [kN]

Nab [kN]

Tab [kN]			Tbc [kN]			Tcd [kN]			Tda [kN]			
20,50			18,95			21,92			23,47			
RH90shear	A5		RH90shear	A5		RH90shear	A5		RH90shear	A5		
wi	th bending		wi	th bending		wi	th bending		W	with bending		
V_90d	20497,51	N	V_90d	18946,64	N	V_90d	21924,13	N	V_90d	23475,00	N	
K_cr	0,67		K_cr	0,67		K_cr	0,67		K_cr	0,67		
A_(net)	5025,00	mm^2	A_(net)	5025,00	mm^2	A_(net)	5025,00	mm^2	A_(net)	5025,00	mm^2	
τ_(d)	6,12	N/mm^2	τ_(d)	5,66	N/mm^2	τ_(d)	6,54	N/mm^2	τ_(d)	7,01	N/mm^2	
ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	
f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	
Verification		RIFIED	Verification	NOT VE	ERIFIED	Verification	NOT VE	ERIFIED	Verification	NOT VE	RIFIED	
V_90d max	2,95	kN	V_90d max	2,95	kN	V_90d max	2,95	kN	V_90d max	2,95	kN	
RH90shear	A5		RH90shear	A5		RH90shear	A5		RH90shear	A5		
V_90d	20497,51	N	V_90d	18946,64	N	V_90d	21924,13	N	V_90d	23475,00	N	
A_(net)	7500,00	mm^2	A_(net)	7500,00	mm^2	A_(net)	7500,00	mm^2	A_(net)	7500,00	mm^2	
τ_(d)	4,10	N/mm^2	τ_(d)	3,79	N/mm^2	τ_(d)	4,38	N/mm^2	τ_(d)	4,69	N/mm^2	
ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	ft,90,d	0,44	N/mm^2	
f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	f_(v,d)	0,88	N/mm^2	
Verification	NOT VE	RIFIED	Verification	NOT VE	ERIFIED	Verification	NOT VE	ERIFIED	Verification	NOT VE	RIFIED	
V_90d max	4,40	kN	V_90d max	4,40	kN	V_90d max	4,40	kN	V_90d max	4,40	kN	

RH90shear for Tab, Tbc and Tcd are satisfied for a load multiplier  $\alpha=0.2$  RH90shear for Tda and RH90shear with bending for Tbc are satisfied for a load multiplier  $\alpha=0.15$  RH90shear with bending for Tab, Tcd and Tda are satisfied for a load multiplier  $\alpha=0.125$ .

# 10.6 Conclusions on seismic analysis out of plane – Flexible

# 10.6.1 Safetybehavior under seismic multiplier $\alpha$ =0,125

All the verifications have been computed in function of the seismic load multiplier  $\alpha$ .

Summing up the results it can be noticed that in this configuration the timber elements with the function of chain is affected by the keyed scarf joint. This due to the fact that in the chain beam there is just un rafter under tension and the action is large compared to the overturning configuration.

The most critical section, again, is in the rafters of the timber beam belonging to failing wall. This section has been named RH90shear but also RRH90shear.

The verification of this section is satisfied for a seismic load multiplier  $\alpha = 0.125$ 

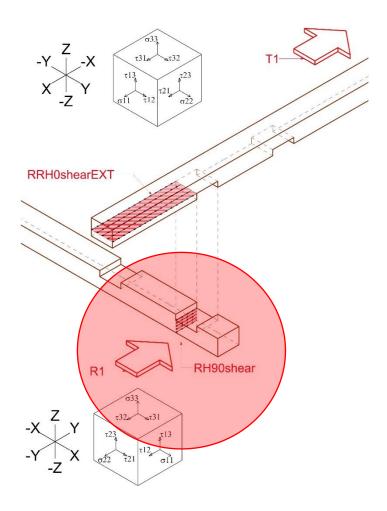


Figure 10-35 Flexible - RH90Shear most critical section

# 11 PRACTICAL RULES OF THUMB FOR CONSTRUCTION OF BHATAR SYSTEM

# 11.1 Arch Tom Schacher's rule of thumb an new specifications

The rules of thumb proposed by Arch Tom Schacher are valid but they do not ensure a perfectly earthquake proof behavior. From the results obtained in the analysis we can assert the structure may hold out against an earthquake with peack ground acceleration about 0.1 g.Some suggestions in reference to Tom shacher's rule of thumb are reported in the following.

### 11.1.1 Specifications on wall joints

With reference to the sub chapter "2.2.4 Wall – joints" it is specified that the keyed scarf joint (or Kashmir joint) must be placed in different position and not along a vertical line on the Z direction.

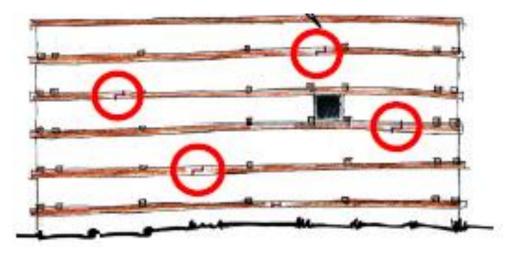


Figure 11-1 Spread the connection points.

The same specifications must be respected on the plane XY of the timber band, as shown in the topof figure 11-2. The joints have to be placed paying attention to do not have opening

A congruent pattern is shown in the figure 11-3, which shows the same wall, on the left the internal surface of the wall and on the right the external surface of the wall.

This kind of joint should be avoided on the rafters at the roof level, for a modul box of a 3.6m square plan If it is not possible it is necessary to respect the pattern described above.

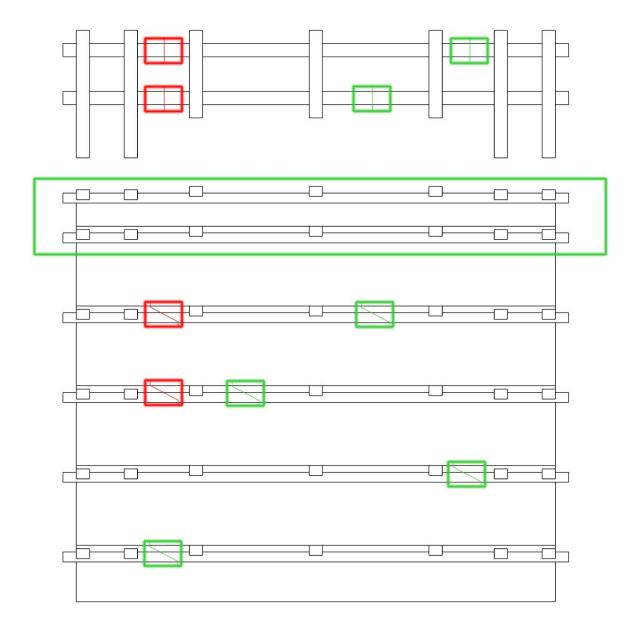


Figure 11-2 Pattern of Keyed scarf joint (or Kashmir joint)

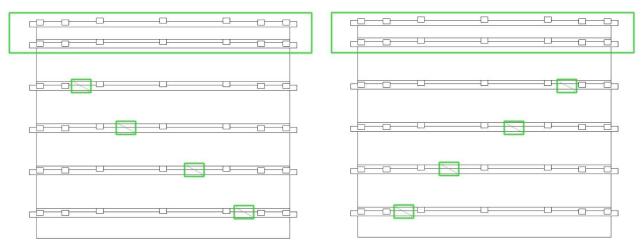


Figure 11-3Pattern for internal and external surface of the same wall

### 11.2New Rules of thumb

### 11.2.1 Consideration about vertical component of the seismic event,

The analysis have been carried out considering the seismic actions applied on an horizontal plane parallel to the ground. Let us consider a spacial reference system with the Z axixs normal to the ground surface, the analysis were focused on the X and Y axis. The seismic action has a vertical component along the Z axis.

The vertical component of seismic action cannot be neglected. The in plane analysis is based on the Barton's model for rockfill which works properly if the surfaces of the rubble stones are in contact. The results of the in plane analysis are actually good even if any safety factors was applied neither to the actions or to the material. In order to ensure the behavior of bhatar analysed previously it is necessary to ensure that the stones composing the rockfill cannot be separated. The idea is to ensure a box behavior for each stone layers between the timber bands.

The connectors may be of different material like rope of vegetable fibersor cords, which are weaker but cheap, or rust preventer steel wire which is more expensive but stronger.

### 11.2.2 Steel wire connectors

In order to be able to sustain eventual vertical component of the seismic force, it is necessary to include some reinforcements where the tension stresses appear.

It can be notice from the picture that the wire is working in pure shear only at the bended part, elsewhere the wire is working in tension.

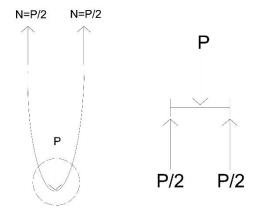


Figure 11-4 Forces acting on the steel wire connectors

"The magnitude of the shear yield stress in pure shear is  $(\sqrt{3})$  times lower than the tensile yield stress in the case of simple tension" [4]. Thus, we have:

$$\tau \leq \frac{f_y}{\sqrt{3}}$$

The general shear stress for the forces acting on the wire is:

$$\tau \leq \frac{P}{2A}$$

Thus,

$$\frac{P}{2A} \le \frac{f_y}{\sqrt{3}}$$

It has been assumed the yield stress of the steel as fy=3000kg/cm<sup>2</sup> and the diameter of the wire as  $\Phi$ =3mm

### 11.2.2.1 Vertical fasten connectors

In order to constrict consecutive timber bands it is possible to take advantage of the cross pieces. The cross pieces stick out to the wall with a length about 10 cm. The cross piece end of the above timber band must be tied to end of the second below timber band, this must be done on the external surface of the wall and on the internal surface of the wall when it is possible.

The vertical connectors (purple line) are placed as shown in the figure below. Each connectors links just two cross pieces with the shown pattern.

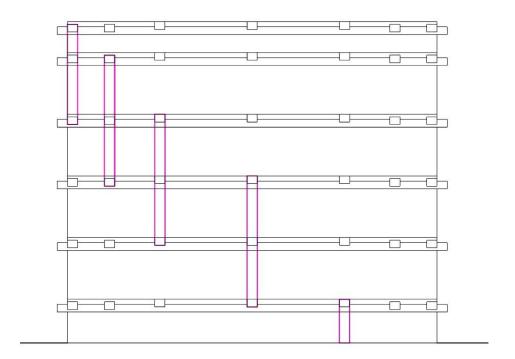


Figure 11-5 Pattern of vertical fasten connector

#### 11.2.2.2 Diagonal fasten connectors

Similarly to the vertical fasten connectors it is usefull to install the diagonal fasten connectors.

The results in the conclusions of the seismic analysis in plane shows that the first four layers from the top may be subjected to sliding. The diagonal connectors guarantee a prevention against this event. The sliding may happen in the direction parallel to the wall and in two sense so the diagonal connectors must be installed with a right sense. In the following figures they are shown diagonal connectors with a positive rotation with respect to the vertical connectors (green line - figure 11-7)

as well diagonal connectors with a negative rotation (red line - figure 11-8). Each connectors links just two cross pieces.

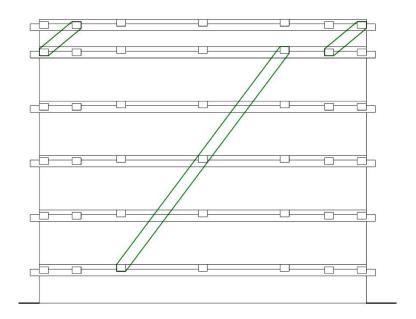


Figure 11-6 Example of single diagonal connector with positive orientation

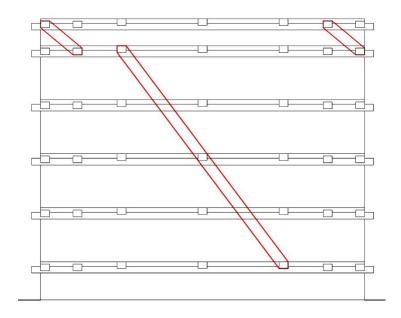


Figure 11-7 Example of single diagonal connector with negative orientation

### 11.2.2.3 Preliminary design of diagonal fasten connectors

The diagonal connectors at roof level have an inclination with respect to the horizontal of  $40^{\circ}$ , the main diagonal connectors in the central position The seismic force distribution have been recalled from the analysis below the timber bands.

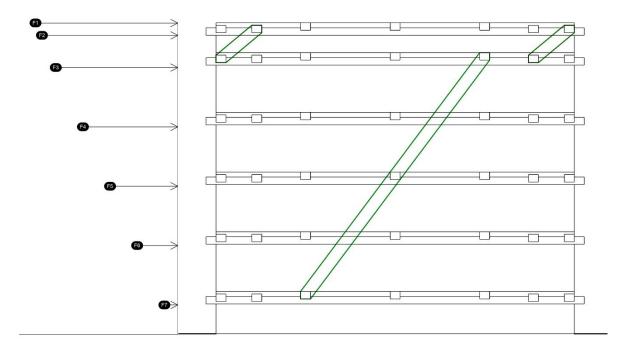


Figure 11-8 Preliminary design of diagonal connectors

The preliminary design of the diagonal connectors is pointed to obtain the numbers of connectors in each position.

The connectors at the roof level : are placed in the corners and they are subjected to a force called Froof, and have an inclination with respect to the horizontal of  $40^{\circ}$ .

$$F_{roof} = F_1 + F_2$$

$$\beta_{roof} = 40^{\circ}$$

The connectors in the central position are subjected to a force called Fwall and have an inclination with respect to the horizontal of  $53^{\circ}$ .

$$F_{wall} = F_3 + F_4 + F_5 + F_6$$
$$\beta_{wall} = 53^{\circ}$$

In order to know the number of connectors for each position it is required the verification of the shear stress  $\tau$  acting on a single connector. The computation of the yeald shear stress has been shown in the sub-chapter "11.2.2 Steel wire connectors".

The component of the seismic force vector acting on the connector at roof level is:

$$P = F_{roof} * \cos(\beta_{roof})$$

$$\tau = \frac{P}{2 * A} = \frac{F_{roof} * \cos(\beta_{roof})}{2 * A}$$

In order to know the numbers of connectors we can write:

$$\frac{\tau}{n} \le \frac{f_y}{\sqrt{3}}$$

Where

n is the number of connectors.

In the case of the roof level it is necessary to consider that the connectors are at the both cornes, thus n must multiplied by 2

$$\frac{\tau}{2*n} \le \frac{f_y}{\sqrt{3}}$$

$$\frac{F_{roof} * \cos(\beta_{roof})}{n * 4 * A} \le \frac{f_y}{\sqrt{3}}$$

Rearranging:

$$\frac{F_{roof} * \cos(\beta_{roof})}{n * 4 * A} \le \frac{f_y}{\sqrt{3}}$$

$$n \ge \frac{F_{roof} * \cos(\beta_{roof}) * \sqrt{3}}{4 * A * f_y}$$

The component of the seismic force vector acting on the connector in the central position is:

$$P = F_{wall} * \cos(\beta_{wall})$$

$$\tau = \frac{P}{2*A} = \frac{F_{wall} * \cos(\beta_{wall})}{2*A}$$

In order to know the numbers of connectors we can write:

$$\frac{\tau}{n} \le \frac{f_y}{\sqrt{3}}$$

Where

n is the number of connectors.

$$\frac{F_{roof} * \cos(\beta_{roof})}{n * 2 * A} \le \frac{f_y}{\sqrt{3}}$$

It has been assumed the yield stress of the steel as fy=3000kg/cm<sup>2</sup> (0.294kN/mm<sup>2</sup>) and the diameter of the wire as  $\Phi$ =3mm

$$A_{\Phi 3} = \frac{\pi * (0.3)^2}{4} = 0.0707 \ cm^2$$

Distribution factors and Forces					
Fj	βj	Fj=Fs*βj			
	/	kN			
F1	0,50	104,63			
F2	0,09	18,38			
F3	0,15	31,26			
F4	0,12	24,32			
F5	0,08	17,37			
F6	0,05	10,42			
F7	0,01	1,80			

	P	β	P/2	$P/2*\cos(\beta)$	$P/(2A)*\cos(\beta)$	n
	kN	deg	kN	kN	kN/mm^2	
Froof=F1+F2=	123,02	40	61,50848	47,11822635	6,665865452	19,622089
Fwall=F3+F4+F5+F6=	83,37	53	41,68481	25,08654208	3,549019713	20,894265

The number of connectors at the roof corners must be at least 20, each corners.

The number of connectors in the center of the wall must be at least 21.

### 11.2.2.4 Foundation

The connectors which guarantee the fastening of the first line of the cross pieces to the ground must be installed in the initial step of the construction of the bhatar structure. The steel wire must be placed under the foundation paying attention to pass it under the first stone layer ,to b more clear the positions of the steel wire is shown in Figure 11-5. The connectors on the corners of the box module plant cannot be placed at the foundation because of the impossibility of installing a straight steel wires without compenetrating the stones.

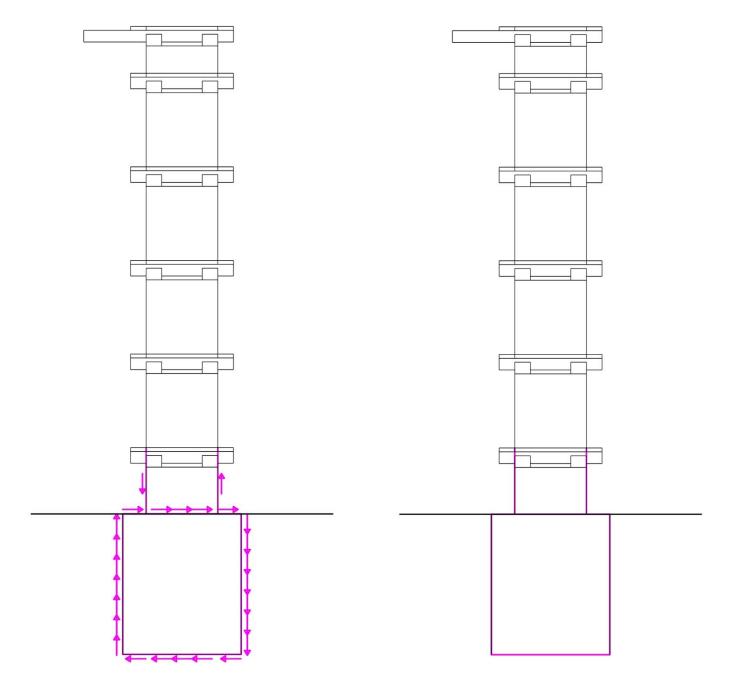


Figure 11-9 Connectors for foundation

### 11.2.2.5 Whole wall distribution of connectors

As written in the previuos sub-chapter the first four layers from the top are subjected to sliding so the priority is to install the vertical (Figure 11-6) and diagonal (Figure 11-8) connectors in order to avoid this event. In the vertical direction the connectors must be installed on all the wall height

The vertical connectors must be installed on the external surface of the wall and on the internal surface of the wall (Figure 11-7).

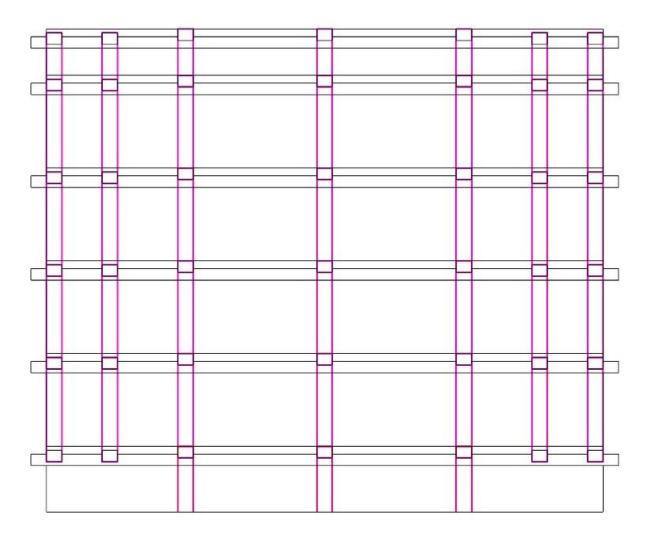


Figure 11-10 Vertical connectors total wall - external

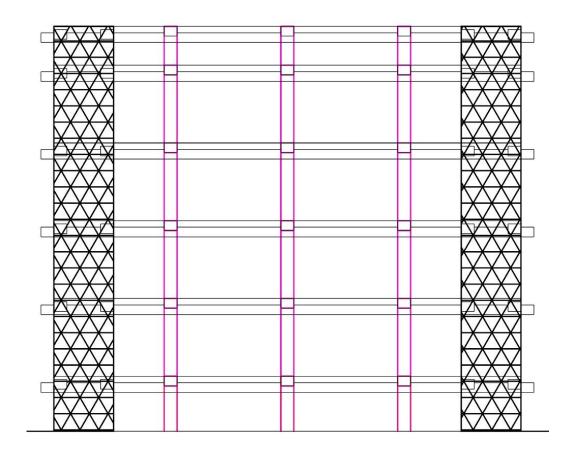
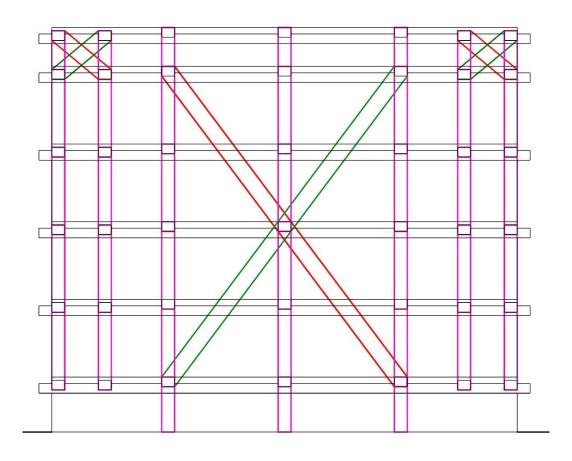


Figure 11-11 Vertical connectors total wall - internal



 $Figure\ 11-12\ Connectors\ on\ \ total\ wall\ -\ external$ 

### 11.2.1 Vertical rafters

The best solution from practical and economical point of view, is to install vertical timber rafters which are available and already known by the Bhatar users.

### 11.2.1.1 Single vertical rafter

The vertical rafters must be placed on the Bhatar structure as the last steps of the bulding process of the load-bearing elements, the walls. The dimension are approximatively

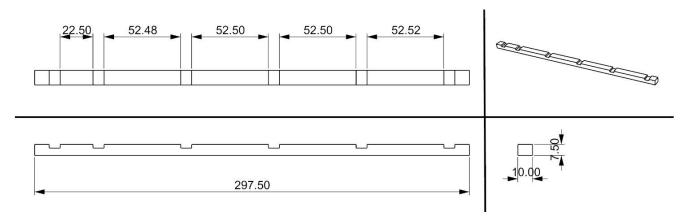


Figure 11-13 Vertical Rafters – gross measuraments in cm

The vertical rafters must be placed in order to embend all the cross pieces along a vertical line. For each line of cross piaces the vertical rafters must be placed at the right side and at the left side. It is also needed to set the vertical rafters externalside and internal side of the box walls. The vertical rafters must be embended to eachothers with connectors which may be of steel wire or rope.

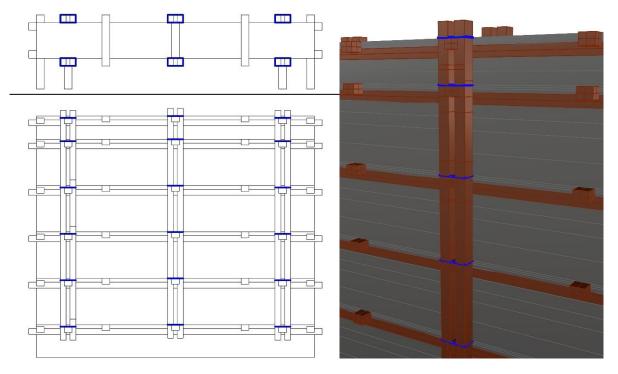


Figure 11-14 Connectors for vertical rafters.

In the next two pages are shown respectively one thrifty solution and one optimal solution for the placement of the vertical rafters.

# 11.2.1.2 Thrifty disposition of vertical rafters

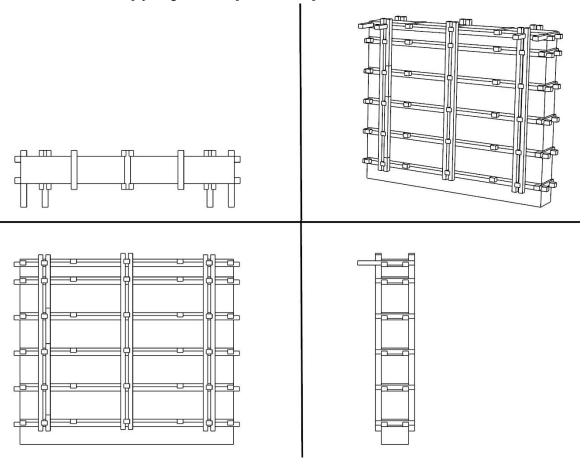


Figure 11-15 Thrifty Solution orthogonal projections

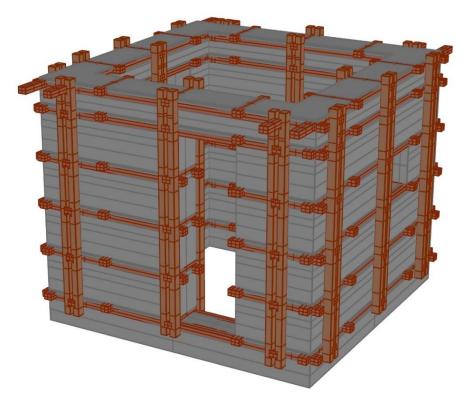


Figure 11-16 Thrifty solution

# 11.2.1.3 Optimal disposition for vertical rafters

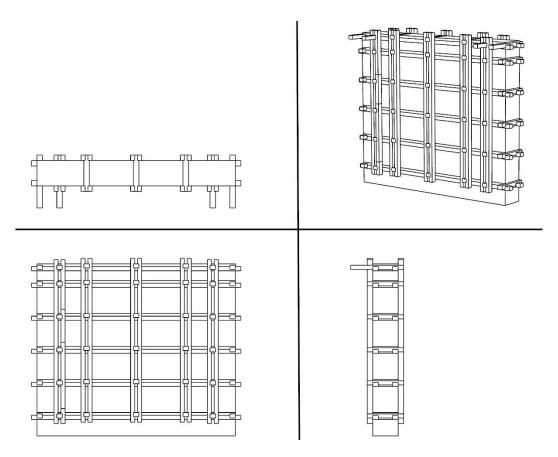


Figure 11-17 Optimal sSolution orthogonal projections

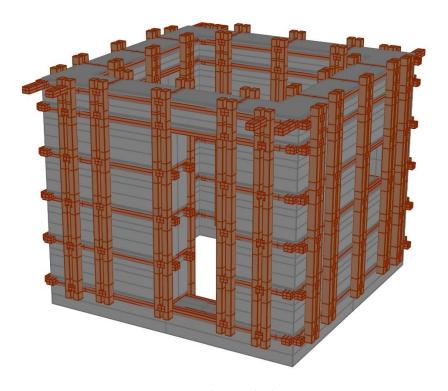


Figure 11-18 Optimal solution

# 11.2.2Roof timber band

From the analysis it is clear that the most stressed timber band is the one where the heavy flat roof is placed. In order to renforce the last timber band on the top it is usefull to install two rafters instead of the two central cross pieces as shown in Figure 11-13. These new kind of rafters are generally equal to the rafter described in the previous chapter axept for the notch in the middle the length. The central joint is a half lap joint with depth of 5 cm as shown in Figure 11-14.

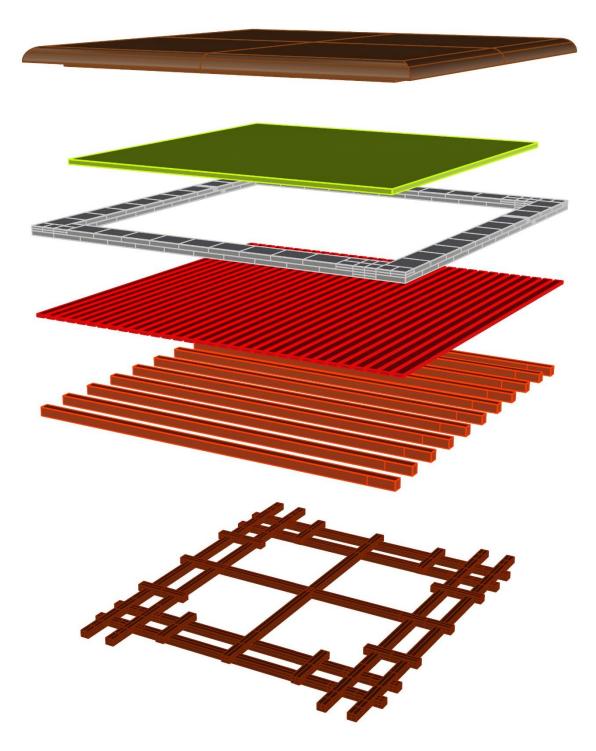


Figure 11-19 Rule of thumb for the roof

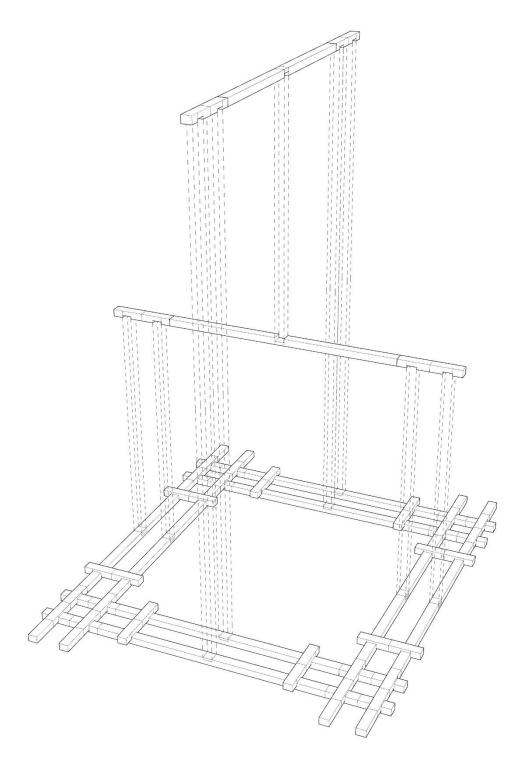


Figure 11-20 Rule of thumb for the roof-Timber band at roof level exploded

# 12 CONCLUSIONS

# 12.1Analysis performed

After the initial observations on the bhatar box module they have been performed the main important seismic analysis used to describe the possible failure mechanisms on a dry-stacked masonry wall like the Bhatar: static, in-plane and out of plane analysis.

The analysis has been carried out starting from a research about materials properties commonly used in Nepal regions like shorea robusta wood and limestone rocks. The habitative unit has been decomposed in the elementar part. The basic geometric elements have been drawn with the Rhinoceros 3D computer graphics and computer-aided design (CAD) application software which allows to get information about volumes and other geometrical properties

The static analysis has been carried out studing the effect of the gravity acceleration on the mass of each layers. The effects of the vertical loads have been studied at different levels, e.g (i)in the middle of the stones layers and (ii)immediately below the timber bands

Generally the failure mechanism, in reference to to the in plane resistance, mayhappen due to shear stresses. The shear stress may produce buckling, sliding or cracks. For the Bhatar system the only possible failure mechanism is the sliding between the stones. The buckling is a of secondary importance because the system is not compact enough, thus the failure happens before. The Bhatar system is characterized by the absence of mortar, the wall is composed by rubble stone masonry and timber beam which is naturally already cracked.

In order to study the sliding failure mechanism, the analysis have been conduct by the use of Barton model. The Barton model is a relationship between the normal stress and the shear stress developing in a gap filled with rocks. This method is used in the field of geotechnical engineering mostly in the studies of the stone dams.

The failure mechanism, in reference to the out of plane resistance, may happen due to overturning with a rigid behavior or with a bending behavior.

The overturning with a rigid behavior has been studied considering the flat heavy earth roof as deformable slab and the absence of the bond-beam (or spreader-beam) at the roof level. The distribution of the reactions on the timber beam at the roof level has been studied as equal ditribuited on the timber rafters.

The timber elements embedded among them may be considered as bond beams, this is why the overturning with a bending behavior has been studied considering the flat heavy earth roof as deformable slab and the presence of the bond-beam (or spreader-beam) at the roof level. In this case, the distribution of the reactions on the timber beam at the roof level has been studied with a thoroughly analysis on the timber rafters connections in the corner joint.

### 12.2 Results

### 12.2.1 Results on seismic analysis in-plane

Recalling the results, they have been identified the critical layers for the in plane seismic analysis. The color red identified the critical load multiplier smaller than the Nepal peak ground acceleration, which is 0,5 g.

Table 68 Summary of results for the in-plane seismic analysis

	Critical Multiplier for inside layer case	Critical Multiplier below the band case	timber	
do	Layer	α <	Layer	α <
l e t	layer1	0,34	layer1	0,18
	layer2	0,41	layer2	0,21
d a	layer3	0,50	layer3	0,26
pplied at of the wall	layer4	0,58	layer4	0,31
Force applied at the top of the wall	layer5	0,67	layer5	0,36
rce	layer6	0,74	layer6	0,40
Foi	Layer_ground/Foundation	0,76	Layer_ground/Foundation	0,76
a)	Layer	α <	Layer	α <
ral all	Layer1	0,64	Layer1	0,37
ate ver	Layer2	0,63	Layer2	0,36
Triangular lateral distribution over the height of the wall	Layer3	0,63	Layer3	0,35
gulg rtio t of	Layer4	0,65	Layer4	0,36
ang ibu	Layer5	0,69	Layer5	0,38
Tri istr hei	Layer6	0,74	Layer6	0,41
ਰ ਹ	Layer_ground/Foundation	0,76	Layer_ground/Foundation	0,76
e	Layer	α <	Layer	α <
all all	Layer1	0,89	Layer1	0,51
ter vel e w	Layer2	0,86	Layer2	0,49
Uniform lateral distribution over the height of the wall	Layer3	0,82	Layer3	0,47
	Layer4	0,80	Layer4	0,46
nife ribr igh	Layer5	0,78	Layer5	0,44
U istr hei	Layer6	0,76	Layer6	0,43
<del>ا</del>	Layer_ground/Foundation	0,76	Layer_ground/Foundation	0,76

### 12.2.1.1 Critical Multiplier for inside stones layer case

Considering the Nepal peak ground acceleration given PGA = 0.5 g the seismic force results smaller than resisting shear force in both the sliding configurations.

The most critical one is the first configuration of «Force applied at the top of the wall» at the roof level, for layer 1 and layer 2 as shown in the following figure.

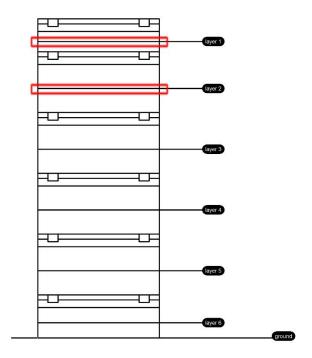


Figure 12-1 Critical layers for the in-plane seismic analysis. Sliding

### 12.2.1.2 Critical Multiplier below the timber band case

Considering the Nepal peak ground acceleration given PGA = 0,5 g the behavior shown is different in the sliding configurations examined considering an amplification of the action due by the safe factor  $\gamma b = 1.5$ . The most critical case is the first configuration of «Force applied at the top of the wall", which shows problems at all the layers. The sliding would occur starting from the roof level with a seismic load multiplier  $\alpha$  =0.18 untill the layer ground/foundation with a seismic load multiplier  $\alpha$  =0.43 The second critical case is the triangular lateral distribution over the height of the wall case, the sliding would occur for a seismic load multiplier  $\alpha$  in a range between 0.37 : 0.43 .

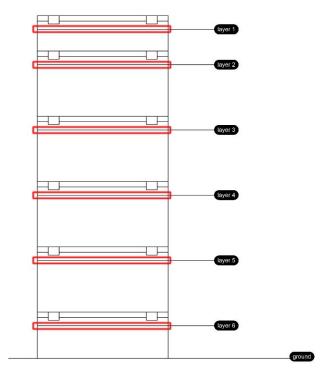


Figure 12-2 Critical layers for the in-plane seismic analysis. Sliding

# 12.2.2 Results on seismic analysis out of plane

### 12.2.2.1 Critical sections for a Nepal seismic event with a PGA=0.5g

The reference country of this thesis is Nepal, as it has been written the peak ground acceleration measured in the last decades in this country is around 0,5 g.

All the seismic load multipliers may be compared with the peak ground acceleration because they have been computed based on the unit measure of the gravity acceleration g. In this sub-chapter they are reported all the sections which do not satisfy the verification for a peak ground acceleration equal or larger to the seismic event expected in Nepal region. Thus, the critical sections are listed specifying the weakness form the most critical to the most safe. Indicators must be read with the following interpretation:

Red: α < 0.5 g</li>
Yellow: α = 0.5 g
Green: α > 0.5 g

RH90shear with bending
RH90shear with bending
Combination of CPNotch0mX and CPNotch0shearY
CPNotch0shearY with bending and CPNotch0shearY
RH90shear
RH90shear
RB0shearY with bending and RB0shearY
Combination of CPNotch0mX and CPNotch0shearZ
CPNotch0mX
RB0tens
Combination of Notch0mX and Notch90shearY
Notch0mZ2
RH0shearEXT with bending

F	LEXIBLE	
	α	
8	0,125	RH90shear with bending for Tab, Tcd and Tda
<b>⊗</b>	0,125	RH90shear with bending for Tab, Tcd and Tda
×	0,15	RH90shear for Tda and RH90shear with bending for Tbc
×	0,15	RH90shear for Tda and RH90shear with bending for Tbc
8	0,2	RH90shear for Tab, Tbc and Tcd
×	0,2	RH90shear for Tab, Tbc and Tcd
8	0,3	CPNotch0mY
8	0,3	RRH0shearEXT with bending for Nbc
×	0,3	The verification about "Influence of keyed scarf joint" is satisfied for a load multiplier $\alpha = 0.3$ .
& & & & & & & & &	0,4	CPNotch0shearY with bending
	0,5	Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Nbc and BM in B
	0,6	Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Nbc and BM in C
<b>⊘</b>	0,6	Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Ncd and BM in D
	0,6	Combination of Notch90shearY and Notch0mX considering Tcd and the pertinent torsional parasitic bending moment
	0,6	Combination of Notch90shearY and Notch0mX considering Tda and the pertinent torsional parasitic bending moment
	0,6	RRH0shearEXT with bending for Nab
	0,6	RRH0shearEXT with bending for Ncd
	0,7	Combination of Notch90shearY and Notch0mX considering Tab and the pertinent torsional parasitic bending moment
	0,7	Combination of Notch90shearY and Notch0mX considering Tbc and the pertinent torsional parasitic bending moment
	0,7	CPNotch0shearY
	0,7	RB0mY with bending
	0,7	RB0shearY with bending
	0,7	RRH0shearEXT with bending for Nda
	0,8	Notch0mX for Tda
<b>⊘</b>	0,8	RRH0shearINT with bending
	0,9	Combination of CPNotch0tens and CPNotch0mY
	0,9	Combination of CPNotch0tens and CPNotch0mZ
	0,9	Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Nab and BM in A
	0,9	Combination of Notch0tens, Notch0mY and Notch0mZ2 considering Ncd and BM in C
$\bigcirc$	0,9	CPNotch0mY
	0,9	Notch0mX for Tab, Tcd
$\bigcirc$	0,9	Notch0mZ for BM in D

The critical sections verified for a load multiplier  $\alpha < 0.5$  are listed in the following figure:

	RIGID	
	α	
8	0,15	RH90shear
×	0,15	RH90shear
8	0,2	Combination of CPNotch0mX and CPNotch0shearY
8	0,2	CPNotch0shearY with bending and CPNotch0shearY
×	0,2	RH90shear
×	0,25	RH90shear
×	0,35	RB0shearY with bending and RB0shearY

FLEXIBLE	
α	
<b>3</b> 0,125	RH90shear with bending for Tab, Tcd and Tda
<b>3</b> 0,125	RH90shear with bending for Tab, Tcd and Tda
<b>8</b> 0,15	RH90shear for Tda and RH90shear with bending for Tbc
<b>3</b> 0,15	RH90shear for Tda and RH90shear with bending for Tbc
<b>3</b> 0,2	RH90shear for Tab, Tbc and Tcd
<b>8</b> 0,2	RH90shear for Tab, Tbc and Tcd
<b>8</b> 0,3	CPNotch0mY
<b>8</b> 0,3	RRH0shearEXT with bending for Nbc
<b>8</b> 0,3	The verification about "Influence of keyed scarf joint" is satisfied for a load multiplier $\alpha = 0.3$ .
<b>8</b> 0,4	CPNotch0shearY with bending

Figure 12-3 Critical sections on the bhatar construction

Basically all the criticalities refer to the notch section of the timber elements with the exeption for the keyed scarf joint.

### 12.2.2.2 Analysis out of plane – Overturning rigidbehavior

The most critical section is in the rafters of the timber beam belonging to overturning wall. This section has been named RH90shear and it is shown in the figure below.

The verification of this section is satisfied for a seismic load multiplier  $\alpha=0.15$ 

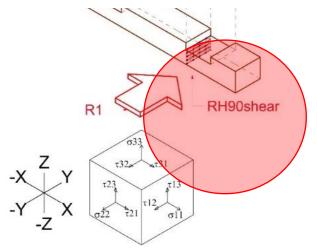


Figure 12-4 RH90Shear most ctitical section

### 12.2.2.3 Analysis out of plane – Flexible response – Bendingbehavior

The most critical section, again, is in the rafters of the timber beam belonging to failing wall. The verification of this section is satisfied for a seismic load multiplier  $\alpha = 0.125$ 

# 12.3 Possible research developemnts



This work reports a full analytical study on the static and seismic behavior, anyhow many subject about this topic need to be examined. In the list below are reported the main important subjects suggested to be thorough:

- Experimental tests on different kind of stones in order to define the specific parameter for the Bartom model for each different kind of stones
- Lab tests on a scale model in order to verify the reliability of the Barton Model for this kind of structure (IN PLANE LAB TESTS)
- Lab tests on Shorea robusta timber, mechanical properties.
- Lab tests on a scale model in order to verify the resistance of the timber elements and the carpentry connections.
- Lab tests on a box module in scale to verify the whole structure behavior.
- Definition of parameters of Barton model for rockfill in order to study the bhatar with a numerical approach.
- Definition of a DEM program in order to verified the hand calculation analysis done.
- Deep study on the horizontal timber bands working as a grounp and the influence on the fragile behavior of the timber.

#### Thumb rules:

- Definition of the dimensioning for design of the vertical and diagonal connectors. This means the diameter of the steel wire and the number of connectors for each cross piece couple. This is because it is needed to ensure a good strength for vertical component of the seismic event and also in order to avoid the sliding of the 4 top timber bands.
- Definition of the dimensioning for design vertical elements at the foundation level for the steel wire or rope case and for vertical rafter case in order to ensure a global scatolar behavior.

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